

# **Aviation Investigation Final Report**

Location: Los Banos, California Accident Number: WPR10LA127

Date & Time: February 2, 2010, 13:20 Local Registration: N36MV

Aircraft: Hiller UH-12E Aircraft Damage: Substantial

**Defining Event:** Powerplant sys/comp malf/fail **Injuries:** 1 None

Flight Conducted Under: Part 137: Agricultural

### **Analysis**

The pilot of the helicopter had just completed an aerial application flight and was returning to land. As he approached the landing area, he began to apply power, but the engine did not respond. He continued to twist the throttle, but it felt loose with no resistance and the engine speed did not change. He aborted the landing approach and performed a run-on landing in a field; during the approach the engine speed began to decrease and the helicopter landed hard, causing substantial damage to the tail boom and lower fuselage. Postaccident examination revealed that the throttle control attach fitting located in the aft section of the collective control had become separated. The nuts and bolts attaching the fitting were not located. The lack of witness marks on the throttle fitting indicates that either the attachment bolt failed or rapidly unscrewed from its associated nut. The disconnection resulted in a loss of engine control and subsequent partial loss of engine power. The last annual inspection occurred 11 months and 250 flight hours prior to the accident.

### **Probable Cause and Findings**

The National Transportation Safety Board determines the probable cause(s) of this accident to be: A partial loss of engine power due to a failure of the throttle interconnect hardware, which resulted in a hard landing.

### **Findings**

Aircraft

Power lever - Failure

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#### **Factual Information**

### **History of Flight**

Approach-VFR pattern final	Powerplant sys/comp malf/fail (Defining event)
Emergency descent	Collision with terr/obj (non-CFIT)

#### HISTORY OF FLIGHT

On February 2, 2010, about 1320 Pacific standard time, a Hiller UH-12E helicopter, N36MV, landed hard near Los Banos, California. S and S Helicopters was operating the helicopter under the provisions of Title 14 Code of Federal Regulations Part 137 as an aerial application flight. The certificated commercial pilot was not injured. The helicopter sustained substantial damage. The local flight departed from a field near Los Banos. Visual meteorological conditions prevailed, and no flight plan had been filed.

The pilot reported that he had just completed an aerial application over a wheat field, and was returning to land. As he approached the landing area at an altitude of about 30 feet above ground level, he began to apply power, but the engine did not respond. He continued to twist the throttle, but it felt loose with no resistance, and the engine speed did not change. He aborted the landing approach, and turned the helicopter towards a larger open field. He circled the field about three times while attempting to diagnose the problem. He then elected to perform a run-on landing; during the landing approach, the engine speed began to decrease, and the helicopter landed hard.

During the landing sequence, the tailboom separated from the aft bulkhead, and the lower fuselage sustained crush damage.

Post accident examination revealed that the throttle control fitting, located at the aft end of the collective stick assembly, had become separated from the cam lever rod bearing.

According to the Hiller service manual applicable to this helicopter, the throttle control fitting is connected to the cam lever rod with an AN23-16A bolt, NAS43DD3-4 spacer, AN960PD10L washer, and a NAS679A3 nut. None of these fittings were located on the engine deck, and examination of the control arm mating surfaces revealed no gouges or fretting signatures.

According to maintenance logbook entries, the helicopter underwent an annual inspection on February 25, 2009, at an airframe total time of 6,815 flight hours. According to the FAA inspector who responded to the accident, the helicopter's total flight time at the time of the accident was 7,064 hours.

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## **Pilot Information**

Certificate:	Commercial	Age:	54,Male
Airplane Rating(s):	Single-engine land; Multi-engine land	Seat Occupied:	Center
Other Aircraft Rating(s):	Helicopter	Restraint Used:	
Instrument Rating(s):	Airplane	Second Pilot Present:	No
Instructor Rating(s):	Airplane single-engine	Toxicology Performed:	No
Medical Certification:	Class 2 With waivers/limitations	Last FAA Medical Exam:	November 5, 2009
Occupational Pilot:	Yes	Last Flight Review or Equivalent:	November 24, 2009
Flight Time:	18500 hours (Total, all aircraft), 700 Command, all aircraft)	0 hours (Total, this make and model),	18500 hours (Pilot In
Instrument Rating(s): Instructor Rating(s): Medical Certification: Occupational Pilot:	Airplane Airplane single-engine Class 2 With waivers/limitations Yes 18500 hours (Total, all aircraft), 700	Second Pilot Present:  Toxicology Performed:  Last FAA Medical Exam:  Last Flight Review or Equivalent:	No November 5, 2009 November 24, 2009

## Aircraft and Owner/Operator Information

Aircraft Make:	Hiller	Registration:	N36MV
Model/Series:	UH-12E	Aircraft Category:	Helicopter
Year of Manufacture:		Amateur Built:	
Airworthiness Certificate:	Restricted (Special)	Serial Number:	5060
Landing Gear Type:	Skid	Seats:	3
Date/Type of Last Inspection:	February 25, 2009 Annual	Certified Max Gross Wt.:	2700 lbs
Time Since Last Inspection:	249 Hrs	Engines:	1 Reciprocating
Airframe Total Time:	7064 Hrs at time of accident	Engine Manufacturer:	LYCOMING
ELT:	Not installed	Engine Model/Series:	VO-540 SERIES
Registered Owner:	JEFF STREETER MOTOR SPORTS INC	Rated Power:	310 Horsepower
Operator:	JEFF STREETER MOTOR SPORTS INC	Operating Certificate(s) Held:	
Operator Does Business As:	S and S Helicopters	Operator Designator Code:	JF5G

## Meteorological Information and Flight Plan

Conditions at Accident Site:	Visual (VMC)	Condition of Light:	Day
Observation Facility, Elevation:	MCE,155 ft msl	Distance from Accident Site:	28 Nautical Miles
Observation Time:	13:01 Local	Direction from Accident Site:	55°
<b>Lowest Cloud Condition:</b>	Few / 700 ft AGL	Visibility	3 miles
Lowest Ceiling:	None	Visibility (RVR):	
Wind Speed/Gusts:	/	Turbulence Type Forecast/Actual:	/
Wind Direction:		Turbulence Severity Forecast/Actual:	/
Altimeter Setting:	29.82 inches Hg	Temperature/Dew Point:	11°C / 9°C
Precipitation and Obscuration:	N/A - None - Mist		
Departure Point:	Los Banos, CA	Type of Flight Plan Filed:	None
Destination:	Los Banos, CA	Type of Clearance:	None
Departure Time:	13:00 Local	Type of Airspace:	

## Wreckage and Impact Information

Crew Injuries:	1 None	Aircraft Damage:	Substantial
Passenger Injuries:		Aircraft Fire:	None
Ground Injuries:	N/A	Aircraft Explosion:	None
Total Injuries:	1 None	Latitude, Longitude:	37.051109,-120.923889(est)

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#### **Administrative Information**

Investigator In Charge (IIC):	Simpson, Eliott
Additional Participating Persons:	Gregg H Schmidt; Federal Aviation Administration FSDO; Fresno, CA
Original Publish Date:	October 21, 2010
Last Revision Date:	
Investigation Class:	<u>Class</u>
Note:	
Investigation Docket:	https://data.ntsb.gov/Docket?ProjectID=75334

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The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, "accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person" (Title 49 Code of Federal Regulations section 831.4). Assignment of fault or legal liability is not relevant to the NTSB's statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 United States Code section 1154(b)). A factual report that may be admissible under 49 United States Code section 1154(b) is available here.

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