



# Aviation Investigation Final Report

<b>Location:</b>	Stillwater, Oklahoma	<b>Accident Number:</b>	CEN09FA230
<b>Date &amp; Time:</b>	March 26, 2009, 20:30 Local	<b>Registration:</b>	N373E
<b>Aircraft:</b>	Cessna 337C	<b>Aircraft Damage:</b>	Substantial
<b>Defining Event:</b>	Fuel exhaustion	<b>Injuries:</b>	1 Fatal
<b>Flight Conducted Under:</b>	Part 91: General aviation - Personal		

## Analysis

Witnesses reported that the airplane departed a private airstrip and the engine or engines were running "rough or misfiring" before the engines "went silent" and the airplane impacted a ravine. The on-site examination of the airplane found that the left auxiliary and main fuel tanks contained about three gallons of fuel each, and that the right auxiliary and main fuel tanks contained less than a gallon total. The position of the fuel valves prior to impact could not be determined. The front fuel strainer/gascolator was found submerged and filled with water and the rear strainer/gascolator was found empty. The airframe and engine examinations did not reveal any abnormalities that would have prevented normal operation. Witnesses reported that the pilot was flying to another airport to refuel the airplane.

## Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: The total loss of dual engine power due to fuel exhaustion as a result of the pilot's inadequate fuel planning.

## Findings

<b>Aircraft</b>	Fuel - Fluid level
<b>Personnel issues</b>	Fuel planning - Pilot

## Factual Information

### History of Flight

<b>Initial climb</b>	Fuel exhaustion (Defining event)
<b>Initial climb</b>	Loss of engine power (total)
<b>Maneuvering</b>	Aerodynamic stall/spin
<b>Maneuvering</b>	Collision with terr/obj (non-CFIT)

### HISTORY OF FLIGHT

On March 26, 2009, about 2030 central daylight time, a twin-engine, Cessna Skymaster 337 airplane, N373E, was substantially damaged upon impact with terrain shortly after departure from a private airstrip located about 7 miles south of Stillwater, Oklahoma. The airline transport pilot, the sole occupant, was fatally injured. The airplane was owned and operated by a private individual. No flight plan was filed and night visual meteorological conditions prevailed for the Title 14 Code of Federal Regulations Part 91 personal flight.

Several witnesses in the vicinity reported seeing or hearing the airplane depart the runway, and circle the area before the airplane impacted the ground. The witnesses added that the airplane was "low" and the engine (or engines were) running "rough or misfiring." They added that the engines went silent, before the airplane nosed sharply down. It was also reported that the pilot had intended to fly to the Stillwater Regional airport to refuel the airplane.

The NTSB Investigator in Charge (IIC), two Federal Aviation Administration (FAA) inspectors, and technical representatives from the airframe and engine manufacturers responded to the accident scene. After a preliminary inspection of the airplane and on-site documentation, the airplane wreckage was recovered and transported to a secure facility for further examination.

### PERSONNEL INFORMATION

The pilot held an Airline Transport Pilot (ATP) license for airplane single and multi-engine land. His third class Federal Aviation Administration (FAA) medical was issued June, 2007. The former military and airline pilot's logbook was located and the pilot had recorded about 23,620 total flight hours.

### AIRCRAFT INFORMATION

The airplane was a 1968 model Cessna 337, Skymaster, which was a twin-engine, push-pull configuration, high-wing, all-metal, retractable tricycle landing gear airplane. The airplane was powered by two Teledyne Continental Motors (TCM), IO-360-C, reciprocating engines, rated at 210 horsepower each. The last annual inspection was performed May 22, 2008, at a time of

3169.3 hours.

## METEOROLOGICAL INFORMATION

At 2053, the automated weather observing system at the Stillwater Regional Airport, Stillwater, Oklahoma, approximately 10 miles north of the accident site, reported winds from 050 degrees at 10 knots, 10 miles visibility, a clear sky, temperature 51-degrees Fahrenheit, dew point 46-degrees Fahrenheit, and an altimeter setting of 29.71 inches of Mercury

## COMMUNICATIONS

The pilot was not in communication with air traffic control at the time of the accident and no distress calls were reported.

## WRECKAGE AND IMPACT INFORMATION

The airplane came to rest in a steep nose down, wings level attitude. The front engine and compartment was mostly submerged in water. All major components of the airplane were accounted for at the accident site. The impact site was a small ravine, which contained several large trees. The ground and impact signatures were consistent with the airplane being in a nose low attitude during the collision.

The forward engine and cabin areas were largely destroyed by the impact. The left and right wings and tail booms remained attached to the fuselage. The main landing gear were extended and the flaps were retracted. The tail surfaces were attached in their respective positions, and exhibited only minor damage. Control continuity to the flight controls was established up to the front cabin area.

The overhead fuel selector panel and control cables were broken/stretched and the position of the fuel valves prior to impact could not be determined. The left auxiliary and main wing fuel tanks contained approximately 3 gallons of fuel each. The right wing and auxiliary fuel tanks contained a total of about a gallon of fuel. The front fuel strainer/gascolator was found submerged and filled with water. The rear strainer/gascolator was found empty. The on-site examination of the engines failed to identify any pre-impact malfunctions.

The airplane's rear engine was removed from the airframe and shipped to the TCM engine test facility, and on April 27-28, 2009, the engine was examined under the supervision of the NTSB IIC, and technical representatives from Teledyne Continental Motors (TCM) and Cessna Aircraft Company.

In order to run the engine, several minor components were either replaced or removed. Those items mainly consisted of the engine mounts, oil cooler, and magneto assembly cover. The rear engine was placed in an engine test cell. The engine was then fitted with a test propeller. The engine started, and then run for several minutes at various (idle to full) power settings.

During the tests, the engine was able to produce rated horsepower, without hesitation or interruption.

The front engine was removed from the airframe to facilitate an inspection. On May 7, 2009, the NTSB IIC along with technical representatives from the engine and airframe manufacturers examined the front engine at the salvage facility. The engine displayed accident impact forces with the crankshaft propeller flange separated from the crankshaft. The sparkplugs were removed from the engine and appeared "normal." The engine was rotated by hand; continuity through the engine was established. A thumb compression and borescope inspection was conducted on each cylinder. The left and right magnetos, which had been submerged in water, were dried, and then rotated by hand and produced spark at all the magneto posts. The inspection did not reveal any abnormalities that would have prevented normal engine operation.

The front propeller remained attached to the engine crankshaft flange. One propeller blade had a "forward" curve starting about mid-span. The other blade had a slight gentle curve towards the non-cambered side, starting near the root of the blade.

The rear propeller remained attached to the engine. During the accident sequence the rear engine moved forward, and the propeller contacted the engine cowling. The propeller rotated less than one complete revolution, once in contact with the cowling. The rear propeller blades had no apparent damage.

#### MEDICAL AND PATHOLOGICAL INFORMATION

The Oklahoma Office of the Chief Medical Examiner, Oklahoma City, Oklahoma, conducted autopsy the pilot on March 27, 2009. The cause of death was determined to be chest trauma, crushing blunt force.

The FAA Toxicology Accident Research Library, Oklahoma City, Oklahoma, conducted toxicological Testing. The pilot tested negative for carbon monoxide, cyanide, ethanol, and drugs.

## Pilot Information

<b>Certificate:</b>	Airline transport	<b>Age:</b>	58, Male
<b>Airplane Rating(s):</b>	Single-engine land; Multi-engine land	<b>Seat Occupied:</b>	Left
<b>Other Aircraft Rating(s):</b>	None	<b>Restraint Used:</b>	
<b>Instrument Rating(s):</b>	Airplane	<b>Second Pilot Present:</b>	No
<b>Instructor Rating(s):</b>	None	<b>Toxicology Performed:</b>	Yes
<b>Medical Certification:</b>	Class 3 Without waivers/limitations	<b>Last FAA Medical Exam:</b>	June 1, 2007
<b>Occupational Pilot:</b>	No	<b>Last Flight Review or Equivalent:</b>	September 19, 2008
<b>Flight Time:</b>	23620 hours (Total, all aircraft)		

## Aircraft and Owner/Operator Information

<b>Aircraft Make:</b>	Cessna	<b>Registration:</b>	N373E
<b>Model/Series:</b>	337C	<b>Aircraft Category:</b>	Airplane
<b>Year of Manufacture:</b>		<b>Amateur Built:</b>	
<b>Airworthiness Certificate:</b>	Normal	<b>Serial Number:</b>	337-0937
<b>Landing Gear Type:</b>	Retractable - Tricycle	<b>Seats:</b>	6
<b>Date/Type of Last Inspection:</b>	May 22, 2008 Annual	<b>Certified Max Gross Wt.:</b>	
<b>Time Since Last Inspection:</b>		<b>Engines:</b>	2 Reciprocating
<b>Airframe Total Time:</b>	3169 Hrs as of last inspection	<b>Engine Manufacturer:</b>	CONT MOTOR
<b>ELT:</b>	C91 installed, not activated	<b>Engine Model/Series:</b>	IO-360 SER
<b>Registered Owner:</b>	On file	<b>Rated Power:</b>	210 Horsepower
<b>Operator:</b>	On file	<b>Operating Certificate(s) Held:</b>	None

## Meteorological Information and Flight Plan

<b>Conditions at Accident Site:</b>	Visual (VMC)	<b>Condition of Light:</b>	Night/dark
<b>Observation Facility, Elevation:</b>	KSWO	<b>Distance from Accident Site:</b>	10 Nautical Miles
<b>Observation Time:</b>		<b>Direction from Accident Site:</b>	
<b>Lowest Cloud Condition:</b>	Clear	<b>Visibility</b>	10 miles
<b>Lowest Ceiling:</b>	None	<b>Visibility (RVR):</b>	
<b>Wind Speed/Gusts:</b>	10 knots /	<b>Turbulence Type Forecast/Actual:</b>	/
<b>Wind Direction:</b>	50°	<b>Turbulence Severity Forecast/Actual:</b>	/
<b>Altimeter Setting:</b>	29.7 inches Hg	<b>Temperature/Dew Point:</b>	11°C / 8°C
<b>Precipitation and Obscuration:</b>	No Obscuration; No Precipitation		
<b>Departure Point:</b>	Stillwater, OK	<b>Type of Flight Plan Filed:</b>	Unknown
<b>Destination:</b>		<b>Type of Clearance:</b>	None
<b>Departure Time:</b>		<b>Type of Airspace:</b>	

## Wreckage and Impact Information

<b>Crew Injuries:</b>	1 Fatal	<b>Aircraft Damage:</b>	Substantial
<b>Passenger Injuries:</b>		<b>Aircraft Fire:</b>	None
<b>Ground Injuries:</b>	N/A	<b>Aircraft Explosion:</b>	None
<b>Total Injuries:</b>	1 Fatal	<b>Latitude, Longitude:</b>	36.079765,-97.069717(est)

## Administrative Information

<b>Investigator In Charge (IIC):</b>	Hatch, Craig
<b>Additional Participating Persons:</b>	Earnest Holdsclaw; FAA FSDO; Oklahoma City, OK Rocky Patel; FAA FSDO; Oklahoma City, OK Steve Miller; Cessna Aircraft Company; Wichita, KS Jason Lukasik; Teledyne Continental Motors; Mobile, AL
<b>Original Publish Date:</b>	March 3, 2010
<b>Last Revision Date:</b>	
<b>Investigation Class:</b>	<a href="#">Class</a>
<b>Note:</b>	The NTSB traveled to the scene of this accident.
<b>Investigation Docket:</b>	<a href="https://data.nts.gov/Docket?ProjectID=73562">https://data.nts.gov/Docket?ProjectID=73562</a>

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, “accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person” (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB’s statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available [here](#).