

Aviation Investigation Final Report

Location:	Hilo, Hawaii	Accident Number:	LAX06LA236
Date & Time:	July 18, 2006, 15:05 Local	Registration:	N6111A
Aircraft:	Hammack Air Command Elite	Aircraft Damage:	Destroyed
Defining Event:		Injuries:	1 Serious
Flight Conducted Under:	Part 91: General aviation - Personal		

Analysis

The gyrocopter collided with terrain following an in-flight separation of a control system rod. While making a right turn about 500 feet above ground level (agl), the pilot experienced a loss of control and the gyrocopter continued to roll to the right. The pilot attempted to regain control by correcting for the increasing rate of turn and descent of the gyrocopter, but he was not able to. He attempted to perform a forced landing and subsequently collided with terrain adjacent to the destination airport. The gyrocopter was constructed by and regularly maintained by the pilot. A post accident examination of the wreckage revealed that a bolt attaching a control rod in the cyclic system was missing. The bolt and its respective nut were never located. The gyrocopter manual states that during a preflight inspection, the pilot should verify the integrity of the nylon locking nut by twisting it in a counterclockwise direction. If the nut moves during this check, it should be replaced before the gyrocopter is flown.

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: an in-flight separation of a control rod joint assembly, which resulted in the pilot's loss of control. The underlying reason for the separation could not be determined.

Findings

Occurrence #1: AIRFRAME/COMPONENT/SYSTEM FAILURE/MALFUNCTION Phase of Operation: MANEUVERING Findings 1. (C) ROTORCRAFT FLIGHT CONTROL, GYROCOPTER ROTOR CTL ARM - SEPARATION 2. (C) MISCELLANEOUS, BOLT/NUT/FASTENER/CLAMP/SPRING - MISSING

Occurrence #2: LOSS OF CONTROL - IN FLIGHT Phase of Operation: MANEUVERING

Findings
3. (C) AIRCRAFT CONTROL - NOT POSSIBLE - PILOT IN COMMAND

Occurrence #3: IN FLIGHT COLLISION WITH TERRAIN/WATER Phase of Operation: DESCENT - UNCONTROLLED

Findings 4. TERRAIN CONDITION - GROUND

Factual Information

On July 18, 2006, at 1505 Hawaiian standard time, an amateur-built Hammack Air Command Elite gyrocopter, N6111A, collided with terrain after the pilot experienced a loss of control near Hilo International Airport, Hilo, Hawaii. The pilot/owner was operating the gyrocopter under the provisions of 14 CFR Part 91. The commercial pilot, the sole occupant, was seriously injured; the gyrocopter was destroyed. The personal local flight departed from Hilo about 10 minutes prior to the mishap. Visual meteorological conditions prevailed and a flight plan had not been filed.

During a telephone interview with a National Transportation Safety Board investigator, the pilot stated that he departed runway 03. He continued toward the coastline and planned to return back to the airport to perform touch-and-go practice takeoffs and landings. After making an intended shallow right turn (toward the east) inland, while about 500 above ground level (agl), he attempted to maneuver the gyrocopter back to level flight and arrest the turn. The gyrocopter continued to turn and began a shallow descent.

The pilot further stated that the gyrocopter increased the rate of turn and continued to lose altitude. At 100 feet agl he realized that a forced landing was imminent. The aircraft collided with terrain about 15 feet from the main road Kalanianaole, located north of the airport.

The pilot constructed the gyrocopter and performed the maintenance on it regularly. He believed a pushrod had separated in the gyrocopter's controls. Over 2 to 3 years, the pilot had amassed 53 flight hours in the gyrocopter.

A Federal Aviation Administration (FAA) inspector examined the wreckage several weeks after the accident occurred. He stated that a bolt attaching a control rod in the cyclic system was missing. He was unable to locate the bolt or its respective nut.

A representative from Air Command International, Inc., the kit manufacturer, reviewed photographs taken by the FAA of the gyrocopter wreckage. He stated the separated piece appeared to be the heim joint (rod end) of the control tubes for the joystick (akin to a cyclic control). The representative reported that the heim joint was designed to be held in place by a nylon-locking (Nylock) nut and bolt system using a AN4-14A bolt, two AN4 washers, and a AN3625-428A ¼-inch lock nut. He noted that all the aforementioned hardware is included with the kit.

The representative further stated that the Nylock nut ensured a secure joint assembly upon proper fastening. He opined that the Nylock nut was not installed properly at the heim joint of the accident gyrocopter.

The Air Command International, Inc. Commander Elite Single-Place Assembly Manual (provided with the gyrocopter kit purchase) contains a section regarding the installation of hardware. It states that that the Nylock nuts comprise a nylon insert that locks the nut onto the threads of the bolt, preventing the nut from loosening. It adds that the Nylock nuts can be used more than once and still ensure a proper lock. The manual specifies that during a preflight inspection, the pilot should verify the integrity of the nut by twisting it in a counterclockwise direction. If the nut moves during this check, it should be replaced before the gyrocopter is flown. The manual additionally gives the bolt torque specification of 30 to 50 inch-pounds.

Pilot Information

Certificate:	Commercial; Flight instructor	Age:	71,Male
Airplane Rating(s):	Single-engine land	Seat Occupied:	Single
Other Aircraft Rating(s):	None	Restraint Used:	
Instrument Rating(s):	Airplane	Second Pilot Present:	No
Instructor Rating(s):	Airplane single-engine; Instrument airplane	Toxicology Performed:	No
Medical Certification:	Class 2	Last FAA Medical Exam:	November 1, 2004
Occupational Pilot:	No	Last Flight Review or Equivalent:	August 1, 2005
Flight Time:	2413 hours (Total, all aircraft), 53 hours (Total, this make and model), 2092 hours (Pilot In Command, all aircraft), 5 hours (Last 90 days, all aircraft), 2 hours (Last 30 days, all aircraft), 1 hours (Last 24 hours, all aircraft)		

Aircraft and Owner/Operator Information

Aircraft Make:	Hammack	Registration:	N6111A
Model/Series:	Air Command Elite	Aircraft Category:	Gyroplane
Year of Manufacture:		Amateur Built:	Yes
Airworthiness Certificate:	Experimental (Special)	Serial Number:	1
Landing Gear Type:	Tricycle	Seats:	1
Date/Type of Last Inspection:	Annual	Certified Max Gross Wt.:	650 lbs
Time Since Last Inspection:	20 Hrs	Engines:	1 Reciprocating
Airframe Total Time:	65 Hrs at time of accident	Engine Manufacturer:	Rotax
ELT:	Not installed	Engine Model/Series:	532
Registered Owner:	On file	Rated Power:	65 Horsepower
Operator:	On file	Operating Certificate(s) Held:	None

Meteorological Information and Flight Plan

Conditions at Accident Site:	Visual (VMC)	Condition of Light:	Day
Observation Facility, Elevation:	PHTO,38 ft msl	Distance from Accident Site:	1 Nautical Miles
Observation Time:	14:53 Local	Direction from Accident Site:	180°
Lowest Cloud Condition:	Few / 2500 ft AGL	Visibility	10 miles
Lowest Ceiling:	None	Visibility (RVR):	
Wind Speed/Gusts:	9 knots /	Turbulence Type Forecast/Actual:	/
Wind Direction:	40°	Turbulence Severity Forecast/Actual:	/
Altimeter Setting:	30.03 inches Hg	Temperature/Dew Point:	29°C / 21°C
Precipitation and Obscuration:	No Obscuration; No Precipitation		
Departure Point:	Hilo, HI (PHTO)	Type of Flight Plan Filed:	None
Destination:	(PHTO)	Type of Clearance:	VFR
Departure Time:	14:55 Local	Type of Airspace:	

Airport Information

Airport:	Hilo International Airport PHTO	Runway Surface Type:	
Airport Elevation:	38 ft msl	Runway Surface Condition:	
Runway Used:		IFR Approach:	Unknown
Runway Length/Width:		VFR Approach/Landing:	Forced landing;Touch and

Wreckage and Impact Information

Crew Injuries:	1 Serious	Aircraft Damage:	Destroyed
Passenger Injuries:		Aircraft Fire:	None
Ground Injuries:	N/A	Aircraft Explosion:	None
Total Injuries:	1 Serious	Latitude, Longitude:	19.690149,-155.080032(est)

Administrative Information

Investigator In Charge (IIC):	Keliher, Zoe
Additional Participating Persons:	Dave Lusk; Federal Aviation Administration; Honolulu, HI
Original Publish Date:	April 25, 2007
Last Revision Date:	
Investigation Class:	<u>Class</u>
Note:	
Investigation Docket:	https://data.ntsb.gov/Docket?ProjectID=64148

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, "accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person" (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB's statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available <u>here</u>.