

Aviation Investigation Final Report

Location:	Roosevelt, Utah	Accident Number:	DEN04LA128
Date & Time:	August 20, 2004, 06:30 Local	Registration:	N120F
Aircraft:	Beech 95-B55	Aircraft Damage:	Substantial
Defining Event:		Injuries:	1 Minor
Flight Conducted Under:	Part 91: General aviation - Personal		

Analysis

According to the pilot, the left engine began to run rough shortly after departing runway 07. He noticed a loss in indicated fuel pressure and within a few moments, the left engine lost power and then quit. The pilot made several unsuccessful attempts to restart the engine. The pilot attempted an immediate left turn to return to the airport. The 270-degree left turn resulted in a loss of altitude and improper line-up with runway 25. The pilot completed a 90-degree right turn as the airplane crossed the airport's access road. The airplane continued to descend as it struck several trees and impacted terrain southwest of the airport. The impact separated both engines and the nose and main landing gear assemblies from the fuselage. Approximately 3 feet of the outboard section of each wing was crushed upwards and aft. According to an FAA inspector, the mixture was in the full rich position and the fuel boost pumps were off. Although the airplane's right wing was compromised, there was visible fuel in the left wing.

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: the loss of engine power for undetermined reasons. A contributing factor was the lack of suitable terrain for a forced landing.

Findings

Occurrence #1: LOSS OF ENGINE POWER(TOTAL) - NONMECHANICAL Phase of Operation: CLIMB

Findings
1. (C) REASON FOR OCCURRENCE UNDETERMINED

Occurrence #2: FORCED LANDING Phase of Operation: EMERGENCY DESCENT/LANDING

Occurrence #3: IN FLIGHT COLLISION WITH OBJECT Phase of Operation: EMERGENCY LANDING

Findings 2. OBJECT - TREE(S)

Occurrence #4: IN FLIGHT COLLISION WITH TERRAIN/WATER Phase of Operation: DESCENT - UNCONTROLLED

Findings

- 3. (F) TERRAIN CONDITION NONE SUITABLE
- 4. TERRAIN CONDITION GROUND

Factual Information

On August 20, 2004, at approximately 0630 mountain daylight time, a Beech 95-B55, N120F, piloted by a private pilot, was substantially damaged during a forced landing after takeoff from Roosevelt Municipal Airport (74V), Roosevelt, Utah. The pilot, the sole occupant on board, received minor injuries. Visual meteorological conditions prevailed at the time of the accident. The personal cross-country flight was being conducted under the provisions of Title 14 CFR Part 91 without a flight plan. The flight was originating at the time of the accident.

According to the pilot, the left engine began to run rough shortly after departing runway 07. He noticed a loss in indicated fuel pressure and within a few moments, the left engine lost power and then quit. The pilot said that he adjusted the mixture and made several unsuccessful attempts to restart the engine. The pilot attempted to return to the airport, but he was unable to maintain altitude, and was forced to land in a field southwest of the airport.

According to an FAA inspector, the pilot attempted an immediate left turn to return to the airport. The 270-degree left turn resulted in a loss of altitude and improper line-up with runway 25. The pilot completed a 90-degree right turn as the airplane crossed the airport's access road. The airplane continued to descend as it struck several trees and impacted terrain southwest of the airport. The impact separated the nose and main landing gear assemblies from the fuselage. Approximately 3 feet of the outboard section of each wing was crushed upwards and aft. Both engines were separated from the airplane.

According to the inspector, the pilot stated that he visually checked both tanks for fuel before takeoff. During the initial climb, the pilot kept the mixture in the full rich position and did not turn the fuel boost pumps on after the left engine began to run rough. Although the airplane's right wing was compromised, there was visible fuel in the left wing.

The airplane's last annual inspection was performed on March 3, 2004, at a tachometer reading of 416.8 hours. The tachometer reading at the time of the accident was 417.3 hours.

According to the inspector, the pilot's logbook showed approximately 1,268 hours of flight time, of which 326 hours were in a Beech 95-B55, including 7.2 hours since March 18, 2004, that was logged in N120F.

On November 17, 2004, an examination of the left engine revealed that the engine's oil pan and exhaust manifold were crushed upward, the intake manifold was crushed aft and the left magneto was separated from the engine. Due to impact damage, a test-run of the engine could not be accomplished. The propeller spinner was crushed aft and twisted in a counter-clockwise direction. Both propeller blades showed torsional bending and chordwise scratches. Engine continuity was confirmed.

The pilot did not provide a Pilot Operator Aircraft Accident Report (NTSB form 6120.1/2) for this accident.

Pilot Information

Certificate:	Private	Age:	61,Male
Airplane Rating(s):	Multi-engine land	Seat Occupied:	Left
Other Aircraft Rating(s):	None	Restraint Used:	
Instrument Rating(s):	None	Second Pilot Present:	No
Instructor Rating(s):	None	Toxicology Performed:	No
Medical Certification:	Class 3 Valid Medical–w/ waivers/lim	Last FAA Medical Exam:	October 7, 2003
Occupational Pilot:	Yes	Last Flight Review or Equivalent:	
Flight Time:	1268 hours (Total, all aircraft), 326 hours (Total, this make and model), 4 hours (Last 90 days, all aircraft), 3 hours (Last 30 days, all aircraft), 1 hours (Last 24 hours, all aircraft)		

Aircraft and Owner/Operator Information

Aircraft Make:	Beech	Registration:	N120F
Model/Series:	95-B55	Aircraft Category:	Airplane
Year of Manufacture:		Amateur Built:	
Airworthiness Certificate:	Normal	Serial Number:	TC-684
Landing Gear Type:	Retractable - Tricycle	Seats:	б
Date/Type of Last Inspection:	March 3, 2004 Annual	Certified Max Gross Wt.:	5400 lbs
Time Since Last Inspection:	0.5 Hrs	Engines:	2 Reciprocating
Airframe Total Time:	4491 Hrs as of last inspection	Engine Manufacturer:	Continental
ELT:	Installed, not activated	Engine Model/Series:	IO-470
Registered Owner:	Leon Ross Drilling and Construction Inc.	Rated Power:	
Operator:		Operating Certificate(s) Held:	None

Meteorological Information and Flight Plan

Conditions at Accident Site:	Visual (VMC)	Condition of Light:	Day
Observation Facility, Elevation:	KVEL,3295 ft msl	Distance from Accident Site:	31 Nautical Miles
Observation Time:	06:53 Local	Direction from Accident Site:	65°
Lowest Cloud Condition:	Clear	Visibility	10 miles
Lowest Ceiling:	None	Visibility (RVR):	
Wind Speed/Gusts:	4 knots / 0 knots	Turbulence Type Forecast/Actual:	/
Wind Direction:	280°	Turbulence Severity Forecast/Actual:	/
Altimeter Setting:	30.15 inches Hg	Temperature/Dew Point:	11°C / 9°C
Precipitation and Obscuration:	No Obscuration; No Precipita	ation	
Departure Point:	Roosevelt, UT (74V)	Type of Flight Plan Filed:	None
Destination:	Roosevelt, UT (GJT)	Type of Clearance:	VFR
Departure Time:	06:20 Local	Type of Airspace:	Class E

Airport Information

Airport:	Roosevelt Municipal 74V	Runway Surface Type:	
Airport Elevation:	5172 ft msl	Runway Surface Condition:	Unknown
Runway Used:		IFR Approach:	None
Runway Length/Width:		VFR Approach/Landing:	None

Wreckage and Impact Information

Crew Injuries:	1 Minor	Aircraft Damage:	Substantial
Passenger Injuries:		Aircraft Fire:	None
Ground Injuries:	N/A	Aircraft Explosion:	None
Total Injuries:	1 Minor	Latitude, Longitude:	40.279907,-109.999298(est)

Administrative Information

Investigator In Charge (IIC):	Scott, Arnold
Additional Participating Persons:	Brent Robinson; Federal Aviation Administration ; Denver, CO
Original Publish Date:	March 30, 2005
Last Revision Date:	
Investigation Class:	<u>Class</u>
Note:	
Investigation Docket:	https://data.ntsb.gov/Docket?ProjectID=59960

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The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, "accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person" (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB's statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available here.