



Aviation Investigation Final Report

| Location: | WEST MIFFLIN, Pennsylvania | Accident Number: | IAD02LA079 |
|-------------------------|----------------------------|------------------|-------------|
| Date & Time: | July 31, 2002, 20:45 Local | Registration: | N3869Z |
| Aircraft: | Beech A-36 | Aircraft Damage: | Substantial |
| Defining Event: | | Injuries: | 1 None |
| Flight Conducted Under: | Part 91: General aviation | | |

Analysis

The pilot reported that immediately after takeoff, he lost the use of the radios and other electrical equipment. The red landing gear position light remained dimly illuminated, which indicated that the landing gear was neither securely stowed or in the down and locked position. The red alternator-out warning light was also illuminated, and the ammeter registered a discharge. The pilot contacted a 911 dispatcher on his cell phone, reported his landing intentions and proceeded toward the airport. He attempted to manually extend the landing gear with the emergency landing gear extension hand crank located behind his seat, but was unable to engage the hand crank handle. Once at the airport, the pilot flew by the control tower and a controller confirmed that the landing gear were not fully extended. The pilot then flew several traffic pattern circuits before landing on a grassy area next to a runway. Upon touchdown, the airplane continued on a straight path until it collided with a runway marker, veered left, and stopped. Examination of the alternator revealed an open rotor circuit between the F1 and F2 terminals. In addition, alternator disassembly revealed an open rotor between the rotor and slip rings. According to the airplane's pilot operating handbook, an inoperative alternator would have placed the entire electrical system operation of the airplane except engine ignition on the battery, and would have been indicated by the illumination of the ALT OUT warning light and a discharge on the ammeter.

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: The failure of the alternator. A factor was the pilot's inability to manually extend the landing gear.

Findings

Occurrence #1: AIRFRAME/COMPONENT/SYSTEM FAILURE/MALFUNCTION Phase of Operation: TAKEOFF

Findings
1. (C) ELECTRICAL SYSTEM, ALTERNATOR - FAILURE, TOTAL

Occurrence #2: MISCELLANEOUS/OTHER Phase of Operation: CRUISE

Findings 2. (F) GEAR EXTENSION - NOT SUCCESSFUL - PILOT IN COMMAND

Occurrence #3: ON GROUND/WATER COLLISION WITH OBJECT Phase of Operation: EMERGENCY LANDING

Findings 3. OBJECT - AIRPORT SIGN/MARKER

Factual Information

On July 31, 2002, at 2045 eastern daylight time, a Beech A-36, N3869Z, was substantially damaged during an emergency landing at Allegheny County Airport (AGC), West Mifflin, Pennsylvania. The certificated private pilot was not injured. No flight plan was filed for the flight that originated at Arnold Palmer Regional Airport (LBE), Latrobe, Pennsylvania, about 2010. Visual meteorological conditions prevailed for the business flight conducted under 14 CFR Part 91.

The pilot stated that immediately after takeoff, he lost use of the radios and other electrical equipment. He noted that the red landing gear position light remained dimly illuminated, which indicated that the landing gear was neither securely stowed or in the down and locked position. The red alternator-out warning light was also illuminated, and the ammeter registered a discharge.

The pilot contacted a 911 dispatcher on his cell phone, who then contacted the Allegheny County Airport Control Tower. The pilot reported his intentions of landing at Allegheny County Airport and proceeded toward it. While en route, he attempted to manually extend the landing gear with the emergency landing gear extension hand crank located at the rear of both front seats. However, he was unable to engage the hand crank handle.

Once at the airport, the pilot flew by the control tower and a controller confirmed that the landing gear were not fully extended. The pilot then flew several traffic pattern circuits before landing on the grass area next to runway 28. Upon touchdown, the airplane continued on a straight path until it collided with a runway marker, veered left, and stopped.

A witness, who flew for a major airline, was standing on the airport ramp when he saw the airplane fly overhead about 200 feet above ground level (agl). He said all three landing gear were "three-quarters" of the way extended. The witness observed the airplane make three more passes over the airport, each time with the landing gear in the same position. On the fifth pass, the airplane approached the grass area next to runway 28 and landed. As the airplane touched down, the left main landing gear collapsed and the airplane turned left. Then, the right main landing gear and nose gear collapsed. The witness said the airplane slid for a few hundred feet before it came to rest perpendicular to the runway.

Three Federal Aviation Administration (FAA) inspectors performed an examination of the airplane after the accident. According to an inspector, the fuselage was twisted. External examination of the alternator and voltage regulators revealed that all components were intact and securely installed.

According to a mechanic, during the recovery process, the airplane was raised on jacks, and

the landing gear was lowered by the manual extension system hand crank. The airplane was then towed to a hangar.

The alternator was removed and examined under FAA supervision. Examination of the alternator revealed that there was an open rotor circuit between the F1 and F2 terminals. Additionally, when the alternator was disassembled, there was an open rotor between the rotor and slip rings.

A review of aircraft maintenance records revealed that the alternator was last overhauled on April 25, 2002, and had accumulated less than 3 hours since installation.

Two voltage regulators were also removed and examined under FAA supervision. Examination of both regulators revealed no anomalies.

According to the Beechcraft Bonanza A36 Pilot Operating Handbook (POH), page 3-8, "An inoperative alternator will place the entire electrical system operation of the airplane except engine ignition on the battery. An alternator failure will be indicated by the illumination of the ALT OUT warning light, located on the instrument panel below the flight instruments."

"The warning light will not illuminate until the alternator output is almost zero. A verification of alternator malfunction would be a discharge on the ammeter."

The pilot reported a total of 1,846 flight hours, of which, 1,134 hours were in make and model.

Weather at the time of the accident included winds from 340 degrees at 5 knots, visibility 10 statute miles, scattered clouds at 5,500 feet, and broken clouds at 25,000 feet.

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|---------------------------|---|-----------------------------------|----------------|
| Certificate: | Private | Age: | 70,Male |
| Airplane Rating(s): | Single-engine land | Seat Occupied: | Left |
| Other Aircraft Rating(s): | None | Restraint Used: | |
| Instrument Rating(s): | Airplane | Second Pilot Present: | No |
| Instructor Rating(s): | None | Toxicology Performed: | No |
| Medical Certification: | Class 3 Valid Medicalw/ waivers/lim | Last FAA Medical Exam: | March 19, 2001 |
| Occupational Pilot: | No | Last Flight Review or Equivalent: | April 14, 2002 |
| Flight Time: | 1846 hours (Total, all aircraft), 1134 hours (Total, this make and model), 1337 hours (Pilot In Command, all aircraft), 26 hours (Last 90 days, all aircraft), 17 hours (Last 30 days, all aircraft) | | |

Pilot Information

Aircraft and Owner/Operator Information

| Beech | Registration: | N3869Z |
|---|--|--|
| A-36 | Aircraft Category: | Airplane |
| | Amateur Built: | |
| Normal | Serial Number: | E-1788 |
| Retractable - Tricycle | Seats: | 6 |
| July 27, 2002 Annual | Certified Max Gross Wt.: | 3600 lbs |
| 3.6 Hrs | Engines: | 2 Reciprocating |
| 2398 Hrs at time of accident | Engine Manufacturer: | Teledyne Continental |
| Installed, activated, did not aid in locating accident | Engine Model/Series: | IO-520-BB |
| DANIEL M. ROONEY | Rated Power: | 285 Horsepower |
| | Operating Certificate(s) Held: | None |
| DMR Airlines | Operator Designator Code: | |
| | A-36 Normal Retractable - Tricycle July 27, 2002 Annual 3.6 Hrs 2398 Hrs at time of accident Installed, activated, did not aid in locating accident DANIEL M. ROONEY | A-36Aircraft Category:A-36Amateur Built:NormalSerial Number:NormalSeats:July 27, 2002 AnnualCertified Max Gross Wt.:3.6 HrsEngines:2398 Hrs at time of accidentEngine Manufacturer:Installed, activated, did not aid in locating accidentRated Power:DANIEL M. ROONEYRated Power:Operating Certificate(s) Held:Serial Certificate(s) |

Meteorological Information and Flight Plan

| Conditions at Accident Site: | Visual (VMC) | Condition of Light: | Dusk |
|----------------------------------|----------------------------------|---|-------------|
| Observation Facility, Elevation: | AGC,1251 ft msl | Distance from Accident Site: | |
| Observation Time: | 20:53 Local | Direction from Accident Site: | |
| Lowest Cloud Condition: | Scattered / 5500 ft AGL | Visibility | 10 miles |
| Lowest Ceiling: | Broken / 25000 ft AGL | Visibility (RVR): | |
| Wind Speed/Gusts: | 5 knots / | Turbulence Type Forecast/Actual: | / |
| Wind Direction: | 340° | Turbulence Severity Forecast/Actual: | / |
| Altimeter Setting: | 29.97 inches Hg | Temperature/Dew Point: | 29°C / 21°C |
| Precipitation and Obscuration: | No Obscuration; No Precipitation | | |
| Departure Point: | LATROBE, PA (LBE) | Type of Flight Plan Filed: | None |
| Destination: | WEST MIFFLIN, PA (AGC) | Type of Clearance: | VFR |
| Departure Time: | 20:10 Local | Type of Airspace: | Class B |

Airport Information

| Airport: | Allegheny County AGC | Runway Surface Type: | Concrete |
|----------------------|----------------------|---------------------------|-----------------------|
| Airport Elevation: | 1251 ft msl | Runway Surface Condition: | Dry |
| Runway Used: | 28 | IFR Approach: | None |
| Runway Length/Width: | 1500 ft / 100 ft | VFR Approach/Landing: | Precautionary landing |

Wreckage and Impact Information

| Crew Injuries: | 1 None | Aircraft Damage: | Substantial |
|------------------------|--------|-------------------------|------------------|
| Passenger Injuries: | | Aircraft Fire: | None |
| Ground Injuries: | N/A | Aircraft Explosion: | None |
| Total Injuries: | 1 None | Latitude, Longitude: | 40.354167,-79.93 |

Administrative Information

| Investigator In Charge (IIC): | Yeager, Leah |
|--------------------------------------|--|
| Additional Participating Persons: | WILLIAM KOSHAR; FAA/FSDO; ALLEGHENY, PA |
| Original Publish Date: | August 26, 2003 |
| Last Revision Date: | |
| Investigation Class: | <u>Class</u> |
| Note: | |
| Investigation Docket: | https://data.ntsb.gov/Docket?ProjectID=55351 |

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, "accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person" (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB's statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available <u>here</u>.