



# Aviation Investigation Final Report

<b>Location:</b>	Queen Anne, Maryland	<b>Accident Number:</b>	NYC01FA158
<b>Date &amp; Time:</b>	July 2, 2001, 18:00 Local	<b>Registration:</b>	N526GC
<b>Aircraft:</b>	Moravan Zlin 526F	<b>Aircraft Damage:</b>	Destroyed
<b>Defining Event:</b>		<b>Injuries:</b>	1 Fatal, 1 Serious
<b>Flight Conducted Under:</b>	Part 91: General aviation - Personal		

## Analysis

The pilot and passenger departed on a local aerobatic flight. A few minutes after becoming airborne, the airplane entered a local aerobatic training area. After completing several aerobatic maneuvers, including an immelmann, the pilot attempted a second immelmann. The pilot entered the maneuver, and the airplane climbed about 1,000 feet before completing the course reversal. While the airplane was at the top of the maneuver and inverted, the pilot applied right stick to roll the airplane to the upright position. As the airplane started to roll, the nose began "creeping" right, and then suddenly the airplane entered an inverted spin. The passenger bailed out, but the pilot remained with the airplane until impact. The pilot had approximately 7,000 hours of flight experience, and was actively flying aerobatics. No pre-impact failures or malfunctions were identified with either the engine or the airframe.

## Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: The pilot's failure to maintain airspeed during an aerobatic maneuver, which resulted in an inadvertent inverted spin.

## Findings

Occurrence #1: LOSS OF CONTROL - IN FLIGHT  
Phase of Operation: MANEUVERING

Findings

1. (C) AIRSPEED - NOT MAINTAINED - PILOT IN COMMAND
2. STALL/SPIN - INADVERTENT - PILOT IN COMMAND
3. AEROBATICS - PERFORMED - PILOT IN COMMAND

-----

Occurrence #2: IN FLIGHT COLLISION WITH TERRAIN/WATER  
Phase of Operation: DESCENT - UNCONTROLLED

## Factual Information

### Pilot Information

<b>Certificate:</b>	Airline transport	<b>Age:</b>	61, Male
<b>Airplane Rating(s):</b>	Single-engine land; Multi-engine land	<b>Seat Occupied:</b>	Front
<b>Other Aircraft Rating(s):</b>	None	<b>Restraint Used:</b>	
<b>Instrument Rating(s):</b>	Airplane	<b>Second Pilot Present:</b>	No
<b>Instructor Rating(s):</b>	Airplane multi-engine; Airplane single-engine; Instrument airplane	<b>Toxicology Performed:</b>	Yes
<b>Medical Certification:</b>	Class 2 Valid Medical--no waivers/lim.	<b>Last FAA Medical Exam:</b>	July 9, 1999
<b>Occupational Pilot:</b>		<b>Last Flight Review or Equivalent:</b>	
<b>Flight Time:</b>	7000 hours (Total, all aircraft)		

The pilot held an airline transport pilot certificate with an airplane single-engine-land, and multi-engine-land ratings. In addition, he held a certified flight instructor certificate with ratings for airplane single-engine-land, multi-engine-land, and instrument. He also held an advanced ground instructor rating. The pilot's logbook was not recovered. On the pilot's last Federal Aviation Administration (FAA) second-class medical certificate, which was dated July 9, 1999, he reported a total flight experience of 7,000 hours. In addition, witnesses reported that the pilot was actively fly aerobatics.

The passenger held a private pilot certificate with an airplane single-engine-land rating. His last FAA third-class medical certificate was dated March 3, 2000. According to the passenger, he had 475 hours of total flight experience, which included about 250 hours of aerobatics. His last flight review was conducted in a Pitts S2B, on September 11, 1999.

## Aircraft and Owner/Operator Information

<b>Aircraft Make:</b>	Moravan	<b>Registration:</b>	N526GC
<b>Model/Series:</b>	Zlin 526F	<b>Aircraft Category:</b>	Airplane
<b>Year of Manufacture:</b>		<b>Amateur Built:</b>	
<b>Airworthiness Certificate:</b>	Experimental (Special)	<b>Serial Number:</b>	1185
<b>Landing Gear Type:</b>	Retractable - Tailwheel	<b>Seats:</b>	2
<b>Date/Type of Last Inspection:</b>		<b>Certified Max Gross Wt.:</b>	2070 lbs
<b>Time Since Last Inspection:</b>		<b>Engines:</b>	1 Reciprocating
<b>Airframe Total Time:</b>		<b>Engine Manufacturer:</b>	LOM
<b>ELT:</b>	Installed, not activated	<b>Engine Model/Series:</b>	M137AZ
<b>Registered Owner:</b>	John E. Nagy Jr.	<b>Rated Power:</b>	180 Horsepower
<b>Operator:</b>		<b>Operating Certificate(s) Held:</b>	None

According to the Pilot's Operating Handbook, the airplane was an all metal, two seat, low-wing monoplane manufactured in Czechoslovakia as an aerobatic trainer. It was approximately 35 feet long, had a wingspan of approximately 26 feet, and a maximum gross weight of approximately 2,070 pounds. The airplane was equipped with a six-cylinder engine cable of producing 180 horsepower. The fuselage was constructed of welded steel tubes that were covered with a combination of fabric and metal.

## Meteorological Information and Flight Plan

<b>Conditions at Accident Site:</b>	Visual (VMC)	<b>Condition of Light:</b>	Night
<b>Observation Facility, Elevation:</b>	BWI,146 ft msl	<b>Distance from Accident Site:</b>	42 Nautical Miles
<b>Observation Time:</b>	17:54 Local	<b>Direction from Accident Site:</b>	294°
<b>Lowest Cloud Condition:</b>	Few / 5500 ft AGL	<b>Visibility</b>	10 miles
<b>Lowest Ceiling:</b>	None	<b>Visibility (RVR):</b>	
<b>Wind Speed/Gusts:</b>	/	<b>Turbulence Type Forecast/Actual:</b>	/
<b>Wind Direction:</b>		<b>Turbulence Severity Forecast/Actual:</b>	/
<b>Altimeter Setting:</b>	30.2 inches Hg	<b>Temperature/Dew Point:</b>	23°C / 5°C
<b>Precipitation and Obscuration:</b>	No Obscuration; No Precipitation		
<b>Departure Point:</b>	EASTON, MD (ESN )	<b>Type of Flight Plan Filed:</b>	None
<b>Destination:</b>		<b>Type of Clearance:</b>	None
<b>Departure Time:</b>	17:30 Local	<b>Type of Airspace:</b>	Class G

The weather at the Baltimore/Washington International Airport (BWI), Baltimore, Maryland, about 6 minutes before the accident was wind calm, visibility 10 miles, few clouds at 5,500 feet, scattered clouds at 25,000 feet, temperature 73 degrees Fahrenheit, dew point 41 degrees Fahrenheit, and an altimeter setting of 30.20 Hg.

## Wreckage and Impact Information

<b>Crew Injuries:</b>	1 Fatal	<b>Aircraft Damage:</b>	Destroyed
<b>Passenger Injuries:</b>	1 Serious	<b>Aircraft Fire:</b>	On-ground
<b>Ground Injuries:</b>	N/A	<b>Aircraft Explosion:</b>	None
<b>Total Injuries:</b>	1 Fatal, 1 Serious	<b>Latitude, Longitude:</b>	38.928333,-75.999725

The debris path was approximately 430 feet long, and orientated along a magnetic heading of 320 degrees. The first item in the debris path was the passenger's parachute. The next item was the cockpit canopy, which was located 240 feet past the parachute, and about 60 feet left of the center of the debris path. The next and last item was the main wreckage. The majority of the wreckage came to rest upright, and was confined to the dimensions of the airplane. Both wings, the vertical and horizontal stabilizers, along with all of the flight control surfaces were identified.

The left wing displayed impact damage to the wingtip, and the inboard 4 feet of the leading edge. The top inboard section of the left wing was consumed in the post-crash fire. The majority of the cockpit was

consumed in the post-crash fire. The right wing displayed impact damage the entire length of the leading edge. The top inboard section of the right wing was also consumed in the post-crash fire. The structure of the empennage was intact, but the fire had consumed the majority of the skin. Also consumed in the post-crash fire was the skin of the left and right elevators, along with the rudder. The vertical and horizontal stabilizers were intact and displayed minor impact damage. Both elevator trim tabs were intact, and approximately neutral. Flight control continuity was verified from each of the control surfaces to both of the cockpits. Flight control continuity could not be traced to each of the pilot controls because of impact and fire damage.

The engine was located on top of an impact crater that measured approximately 4 feet wide and 2-1/2 feet deep. The propeller had separated from the engine, and was located in the crater. The propeller blades displayed leading edge gouging and "S" bending. The fracture surfaces for the propeller flange and bolts were grayish in color, and consistent with torsional overload.

### **Medical and Pathological Information**

---

An autopsy was performed on the pilot at the Medical Examiners office in Baltimore, Maryland, on July 3, 2001. The Federal Aviation Administration Toxicology and Accident Research Laboratory in Oklahoma City, Oklahoma, performed toxicological testing on July 20, 2001.

### **Additional Information**

---

According to the passenger, he had flown once before in the accident airplane. It was from the front seat and about 1-1/2 years prior to the accident. On the day of the accident, the passenger did not fly the airplane. However, he did follow along on the controls during a couple of the maneuvers with the pilot's permission, but not during the accident maneuver. His reason for not flying on the day of the accident was a "hamstring injury," which prevented him from applying any force to the rudder pedals. The passenger added that he saw a medical doctor earlier in the day because of the injury.

The wreckage was released to the owner's representative on July 3, 2001.

## Administrative Information

<b>Investigator In Charge (IIC):</b>	Muzio, David
<b>Additional Participating Persons:</b>	Cooper Towers; FAA/FSDO; Baltimore, MD
<b>Original Publish Date:</b>	December 4, 2001
<b>Last Revision Date:</b>	
<b>Investigation Class:</b>	<a href="#">Class</a>
<b>Note:</b>	
<b>Investigation Docket:</b>	<a href="https://data.ntsb.gov/Docket?ProjectID=52604">https://data.ntsb.gov/Docket?ProjectID=52604</a>

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, “accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person” (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB’s statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available [here](#).