



# Aviation Investigation Final Report

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<b>Location:</b>	Okmulgee, Oklahoma	<b>Accident Number:</b>	FTW01LA143
<b>Date &amp; Time:</b>	June 14, 2001, 13:15 Local	<b>Registration:</b>	N114BW
<b>Aircraft:</b>	Commander 114-B	<b>Aircraft Damage:</b>	Substantial
<b>Defining Event:</b>		<b>Injuries:</b>	1 Minor
<b>Flight Conducted Under:</b>	Part 91: General aviation - Personal		

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## Analysis

The flight had been airborne for three hours when the pilot heard a "loud bang" and the engine lost total power. He executed a forced landing to a field, during which the airplane contacted dirt mounds that were 4 feet high, before it came to a stop upright. Postaccident examination of the airplane revealed that fresh engine oil was adhering to the left side of the fuselage from the engine cowling extending back to the empennage. The oil dipstick was removed and indicated only 2 quarts (8 quart capacity). The oil was black and smelled as though it had been exposed to heat. The engine crankcase displayed a 2.5-inch diameter hole between the #5 and #6 cylinders. The engine was disassembled and it was noted that the connecting rod for the #6 cylinder had separated from its piston and the crankshaft. The #6 connecting rod journal displayed evidence of thermal damage and metal transfer. Metallurgical examination of the #6 connecting rod end cap and bolt revealed fractures indicative of overstress at extreme temperatures. Additionally, the rod end cap bearing surface displayed fretting and metal transfer. No other anomalies with the engine were noted.

## Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: the failure of the #6 cylinder's connecting rod, which resulted in a complete loss of engine power. A contributing factor to the accident was the lack of suitable terrain for the forced landing.

## Findings

Occurrence #1: LOSS OF ENGINE POWER(TOTAL) - MECH FAILURE/MALF

Phase of Operation: CRUISE

### Findings

1. ENGINE ASSEMBLY,CONNECTING ROD - FAILURE,TOTAL

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Occurrence #2: FORCED LANDING

Phase of Operation: EMERGENCY DESCENT/LANDING

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Occurrence #3: ON GROUND/WATER COLLISION WITH OBJECT

Phase of Operation: LANDING - ROLL

### Findings

2. TERRAIN CONDITION - DIRT BANK/RISING EMBANKMENT

3. (F) TERRAIN CONDITION - NONE SUITABLE

## Factual Information

On June 14, 2001, at 1315 central daylight time, a Commander 114-B airplane, N114BW, was substantially damaged when it impacted the ground during a forced landing following a complete loss of engine power near Okmulgee, Oklahoma. The airplane was registered to a private individual and operated by the pilot. The commercial pilot, sole occupant of the airplane, sustained minor injuries. Visual meteorological conditions prevailed, and an instrument flight rules (IFR) flight plan was filed for the 14 Code of Federal Regulations Part 91 personal flight. The cross-country flight originated from the Kyle-Oakley Field Airport, Murray, Kentucky, at 1000, and was destined for the Wiley Post Airport, Oklahoma City, Oklahoma.

The pilot reported that after three hours of normal flight, he detected a change in the sound of the engine. The oil pressure gauge needle dropped to zero; however, the cylinder head temperature and oil temperature gauge needles remained in their green arcs. The pilot contacted the local flight service station and requested vectors to the nearest airport, which was in Okmulgee. The pilot noticed an increasing vibration and reduced the throttle. Subsequently, he heard a "loud bang" and the engine lost total power. He executed a forced landing to a field, located 2.5 miles northeast of the Okmulgee Airport. During the landing roll, the airplane contacted dirt mounds that were 4 feet high, before it came to a stop upright.

The FAA inspector, who examined the aircraft at the accident site, reported that both wings, the engine firewall, and landing gear were structurally damaged. He observed a film of oil extending from the engine cowling along the fuselage to empennage. He removed the engine oil dipstick and it indicated 2 quarts out of an 8 quart capacity.

The Textron-Lycoming IO-540-T4B5 engine was examined in Oklahoma by an NTSB investigator and a representative from Textron-Lycoming. The oil sump plug was removed and only two drops of oil could be drained. The oil appeared black and smelled as though it had been exposed to heat. The engine's oil system was examined and no evidence of a leak was noted. A hole, that was 2.5 inches in diameter, was observed on the top of the engine crankcase, between the #5 and #6 cylinders. A rod end cap and bolt from the #6 cylinder's connecting rod were extracted from the hole. The connecting rod end cap and bolt were sent to the NTSB Materials Laboratory in Washington, D.C. for further examination. The bolt displayed features typical of bending and torsion overstress while heated. The rod end cap displayed features typical of bending overstress while heated. Additionally, the bearing surface of the rod end cap exhibited fretting and metal transfer, which are signatures consistent with relative movement and looseness in the joint.

The engine was transported to Air Salvage of Dallas, Lancaster, Texas, and examined by the NTSB investigator and the representative from Textron Lycoming. The #1 through #5 cylinders were removed; the #6 cylinder was internally damaged and could not be removed. The

crankcase was separated, and the #6 cylinder's connecting rod was observed to be separated from its piston and the crankshaft. The #1 through #5 connecting rod cap bolts were torque tested using a flexible beam torque wrench. According to the Textron-Lycoming representative, all of the bolts were within acceptable tolerances. The connecting rods were removed from the #1 through the #5 positions on the crankshaft. The #1 through #5 connecting rod journals on the crankshaft were not scored and there was no apparent thermal damage. The #6 connecting rod journal displayed thermal damage and metal transfer. Ferrous debris was found in the oil filter, at the oil suction screen, and in the oil sump. The oil pump was disassembled and no anomalies were noted.

At the time of the accident, the engine and airframe had accumulated a total of 1743.9 hours, and the airplane had flown 150.0 hours since the last annual inspection on January 5, 2001.

### Pilot Information

<b>Certificate:</b>	Commercial; Flight instructor	<b>Age:</b>	64, Male
<b>Airplane Rating(s):</b>	Single-engine land	<b>Seat Occupied:</b>	Left
<b>Other Aircraft Rating(s):</b>	Glider	<b>Restraint Used:</b>	
<b>Instrument Rating(s):</b>	Airplane	<b>Second Pilot Present:</b>	No
<b>Instructor Rating(s):</b>	Airplane single-engine; Instrument airplane	<b>Toxicology Performed:</b>	No
<b>Medical Certification:</b>	Class 2 Valid Medical-w/ waivers/lim	<b>Last FAA Medical Exam:</b>	May 9, 2000
<b>Occupational Pilot:</b>	No	<b>Last Flight Review or Equivalent:</b>	February 17, 2001
<b>Flight Time:</b>	2110 hours (Total, all aircraft), 69 hours (Total, this make and model), 1833 hours (Pilot In Command, all aircraft), 57 hours (Last 90 days, all aircraft), 23 hours (Last 30 days, all aircraft), 3 hours (Last 24 hours, all aircraft)		

## Aircraft and Owner/Operator Information

<b>Aircraft Make:</b>	Commander	<b>Registration:</b>	N114BW
<b>Model/Series:</b>	114-B	<b>Aircraft Category:</b>	Airplane
<b>Year of Manufacture:</b>		<b>Amateur Built:</b>	
<b>Airworthiness Certificate:</b>	Normal	<b>Serial Number:</b>	14541
<b>Landing Gear Type:</b>	Retractable - Tricycle	<b>Seats:</b>	4
<b>Date/Type of Last Inspection:</b>	May 5, 2001 Annual	<b>Certified Max Gross Wt.:</b>	
<b>Time Since Last Inspection:</b>	105 Hrs	<b>Engines:</b>	1 Reciprocating
<b>Airframe Total Time:</b>	1743.4 Hrs at time of accident	<b>Engine Manufacturer:</b>	Lycoming
<b>ELT:</b>	Installed, activated, did not aid in locating accident	<b>Engine Model/Series:</b>	IO-540-T4B5
<b>Registered Owner:</b>	Robert T. Williams	<b>Rated Power:</b>	260 Horsepower
<b>Operator:</b>	Gary A. Gallo	<b>Operating Certificate(s) Held:</b>	None
<b>Operator Does Business As:</b>	N/A	<b>Operator Designator Code:</b>	

## Meteorological Information and Flight Plan

<b>Conditions at Accident Site:</b>	Visual (VMC)	<b>Condition of Light:</b>	Day
<b>Observation Facility, Elevation:</b>	TUL,677 ft msl	<b>Distance from Accident Site:</b>	30 Nautical Miles
<b>Observation Time:</b>	12:53 Local	<b>Direction from Accident Site:</b>	360°
<b>Lowest Cloud Condition:</b>	3400 ft AGL	<b>Visibility</b>	9 miles
<b>Lowest Ceiling:</b>	Broken / 3400 ft AGL	<b>Visibility (RVR):</b>	
<b>Wind Speed/Gusts:</b>	16 knots /	<b>Turbulence Type Forecast/Actual:</b>	/
<b>Wind Direction:</b>	190°	<b>Turbulence Severity Forecast/Actual:</b>	/
<b>Altimeter Setting:</b>	29.73 inches Hg	<b>Temperature/Dew Point:</b>	29°C / 23°C
<b>Precipitation and Obscuration:</b>	No Obscuration; No Precipitation		
<b>Departure Point:</b>	Murray, KY (CEY)	<b>Type of Flight Plan Filed:</b>	IFR
<b>Destination:</b>	Oklahoma City, OK (PWA)	<b>Type of Clearance:</b>	IFR
<b>Departure Time:</b>	10:00 Local	<b>Type of Airspace:</b>	Class G

## Airport Information

<b>Airport:</b>	Okmulgee Municipal Airport OKM	<b>Runway Surface Type:</b>	
<b>Airport Elevation:</b>	720 ft msl	<b>Runway Surface Condition:</b>	Unknown
<b>Runway Used:</b>		<b>IFR Approach:</b>	Unknown
<b>Runway Length/Width:</b>		<b>VFR Approach/Landing:</b>	Forced landing

## Wreckage and Impact Information

<b>Crew Injuries:</b>	1 Minor	<b>Aircraft Damage:</b>	Substantial
<b>Passenger Injuries:</b>		<b>Aircraft Fire:</b>	None
<b>Ground Injuries:</b>	N/A	<b>Aircraft Explosion:</b>	None
<b>Total Injuries:</b>	1 Minor	<b>Latitude, Longitude:</b>	35.630348,-95.950531(est)

## Administrative Information

<b>Investigator In Charge (IIC):</b>	Ragogna, Jason
<b>Additional Participating Persons:</b>	Cary E Wilcox; Federal Aviation Administration; Oklahoma City, OK John B Butler; Textron Lycoming; Arlington, TX
<b>Original Publish Date:</b>	July 1, 2002
<b>Last Revision Date:</b>	
<b>Investigation Class:</b>	<a href="#">Class</a>
<b>Note:</b>	The NTSB traveled to the scene of this accident.
<b>Investigation Docket:</b>	<a href="https://data.ntsb.gov/Docket?ProjectID=52485">https://data.ntsb.gov/Docket?ProjectID=52485</a>

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, “accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person” (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB’s statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available [here](#).