

Aviation Investigation Final Report

| Location: | DARRINGTON, Washi | ngton | Accident Number: | SEA99LA020 |
|-------------------------|--------------------------------|------------|------------------|-------------|
| Date & Time: | December 14, 1998, 1 | 1:00 Local | Registration: | N94NW |
| Aircraft: | GARLICK | UH-1H | Aircraft Damage: | Substantial |
| Defining Event: | | | Injuries: | 1 Serious |
| Flight Conducted Under: | Part 133: Rotorcraft ext. load | | | |

Analysis

While in a hover during heavy-lift aero-logging operations, the vertical fin separated from the tail boom, resulting in an uncontrolled descent into the terrain. During the investigation, the NTSB Materials Laboratory determined that the separation was due to the propagation of a number of fatigue cracks which had initiated within the fourth rivet hole above where the left side of the spar fin cap protruded from the tail boom. Although the FAA had previously issued Airworthiness Directive (AD) 97-20-09 to address the separation of vertical fins from UH-1 helicopters used in repeated heavy-lift operations, the Rotorcraft Directorate determined that the AD was not adequate. As a result, the Directorate initiated the formulation of a Priority Letter Airworthiness Directive requiring the replacement of the original fin spar with one that '...has been shown to account for the effects of repeated heavy-lift operations.'

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: The total separation of the vertical fin (stabilizer), as the result of fatigue cracks which initiated in the fourth rivet hole above where the fin extends from the tailboom.

Findings

Occurrence #1: AIRFRAME/COMPONENT/SYSTEM FAILURE/MALFUNCTION Phase of Operation: HOVER

Findings

1. (C) VERTICAL STABILIZER - FATIGUE 2. (C) VERTICAL STABILIZER - FRACTURED 3. (C) VERTICAL STABILIZER - SEPARATION

Occurrence #2: IN FLIGHT COLLISION WITH TERRAIN/WATER Phase of Operation: DESCENT - UNCONTROLLED

Factual Information

On December 14, 1998, approximately 1100 Pacific standard time, a Garlick UH-1H helicopter, N94NW, owned by Northwest Helicopters, and operated as a 14 CFR Part 133 aero-logging operation, impacted the terrain after experiencing a separation of the vertical fin in flight. Visual meteorological conditions prevailed, and no flight plan was filed. The aircraft sustained substantial damage, and the commercial pilot received serious injuries. The flight originated at a nearby helicopter service pad.

According to the pilot, he was in a hover at the end of a 150 foot long-line that had just been attached to a set of logs. He reported that he had not yet began to lift the logs, but had taken the slack out of the line when he heard a loud bang. The aircraft immediately began spinning to the right and settled into the terrain.

The aft portion of the tail boom was sent to the NTSB materials laboratory for examination. That examination revealed the presence of fatigue cracking in all five layers of the left side of the vertical fin spar cap. According to the laboratory, the cracks initiated from the hole for the fourth rivet above where the fin protrudes from the tail boom, and propagated through most of the left side of the spar cap prior to final separation.

During the investigation, it was determined that the vertical fins of at least two other UH-1 helicopters had previously separated in flight, although not through the fourth hole. The FAA had previously issued Airworthiness Directive (AD) number 97-20-09 to address the problem of fin separation from UH-1 helicopters used in repeated heavy-lift operation, but the FAA Rotorcraft Directorate subsequently determined that the AD was not adequate. As a result, the Directorate initiated the formulation of a Priority Letter Airworthiness Directive calling for the replacement of the original equipment manufacturer's (OEM) fin spar with one that "...has been proven to account for the effects of repeated heavy-lift operations."

Pilot Information

| Certificate: | Commercial | Age: | 50,Male |
|---------------------------|---|-----------------------------------|----------------|
| Airplane Rating(s): | None | Seat Occupied: | Left |
| Other Aircraft Rating(s): | Helicopter | Restraint Used: | |
| Instrument Rating(s): | None | Second Pilot Present: | No |
| Instructor Rating(s): | None | Toxicology Performed: | No |
| Medical Certification: | Class 2 Valid Medicalw/ waivers/lim | Last FAA Medical Exam: | April 29, 1998 |
| Occupational Pilot: | Yes | Last Flight Review or Equivalent: | |
| Flight Time: | 6336 hours (Total, all aircraft), 800 hours (Total, this make and model), 6336 hours (Pilot In Command, all aircraft), 210 hours (Last 90 days, all aircraft), 66 hours (Last 30 days, all aircraft), 7 hours (Last 24 hours, all aircraft) | | |

Aircraft and Owner/Operator Information

| Aircraft Make: | GARLICK | Registration: | N94NW |
|----------------------------------|---|-----------------------------------|-----------------|
| Model/Series: | UH-1H UH-1H | Aircraft Category: | Helicopter |
| Year of Manufacture: | | Amateur Built: | |
| Airworthiness Certificate: | Restricted (Special) | Serial Number: | 66-1101 |
| Landing Gear Type: | Skid | Seats: | 2 |
| Date/Type of Last Inspection: | December 10, 1998 Continuous airworthiness | Certified Max Gross Wt.: | 9500 lbs |
| Time Since Last Inspection: | 10 Hrs | Engines: | 1 Turbo shaft |
| Airframe Total Time: | 7240 Hrs | Engine Manufacturer: | Lycoming |
| ELT: | Installed, not activated | Engine Model/Series: | T53-L13B |
| Registered Owner: | NORTHWEST HELICOPTERS INC. | Rated Power: | 1400 Horsepower |
| Operator: | | Operating Certificate(s) Held: | |
| Operator Does Business As: | | Operator Designator Code: | |

Meteorological Information and Flight Plan

| Conditions at Accident Site: | Visual (VMC) | Condition of Light: | Day |
|---|----------------------------------|---|----------|
| Observation Facility, Elevation: | | Distance from Accident Site: | |
| Observation Time: | | Direction from Accident Site: | |
| Lowest Cloud Condition: | Unknown | Visibility | 10 miles |
| Lowest Ceiling: | None | Visibility (RVR): | |
| Wind Speed/Gusts: | / | Turbulence Type Forecast/Actual: | / |
| Wind Direction: | 0° | Turbulence Severity Forecast/Actual: | / |
| Altimeter Setting: | | Temperature/Dew Point: | 4°C |
| Precipitation and Obscuration: | No Obscuration; No Precipitation | | |
| Departure Point: | | Type of Flight Plan Filed: | None |
| Destination: | | Type of Clearance: | None |
| Departure Time: | 10:30 Local | Type of Airspace: | Class G |

Airport Information

| Airport: | | Runway Surface Type: | |
|----------------------|---|---------------------------|------|
| Airport Elevation: | | Runway Surface Condition: | |
| Runway Used: | 0 | IFR Approach: | None |
| Runway Length/Width: | | VFR Approach/Landing: | None |

Wreckage and Impact Information

| Crew Injuries: | 1 Serious | Aircraft Damage: | Substantial |
|------------------------|-----------|-------------------------|----------------------------|
| Passenger Injuries: | | Aircraft Fire: | On-ground |
| Ground Injuries: | N/A | Aircraft Explosion: | None |
| Total Injuries: | 1 Serious | Latitude, Longitude: | 48.189357,-121.210533(est) |

Administrative Information

| Investigator In Charge (IIC): | Anderson, Orrin | |
|--------------------------------------|--|--|
| Additional Participating Persons: | BRENT MARROW; SEATTLE , WA JACK SUTTLE; FORT WORTH , TX | |
| Original Publish Date: | January 11, 2000 | |
| Last Revision Date: | | |
| Investigation Class: | <u>Class</u> | |
| Note: | | |
| Investigation Docket: | https://data.ntsb.gov/Docket?ProjectID=45496 | |

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The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, "accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person" (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB's statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available <u>here</u>.