



Aviation Investigation Final Report

Location: STEVENSVILLE, Maryland Accident Number: NYC94LA129

Date & Time: July 15, 1994, 14:45 Local Registration: N515TN

Aircraft: PITTS S-1S Aircraft Damage: Substantial

Defining Event: 1 Minor

Flight Conducted Under: Part 91: General aviation - Personal

Analysis

THE AIRPLANE'S ENGINE LOST TOTAL POWER JUST AFTER TAKEOFF WHEN THE PILOT REDUCED THE THROTTLE. THE PILOT RE-APPLIED FULL THROTTLE WITH NO RESULT. DURING THE FORCED LANDING IN A FIELD, THE AIRPLANE'S WING STRUCK A TREE, AND THE AIRPLANE NOSED OVER. EXAMINATION OF THE ENGINE REVEALED THAT THE CRANKSHAFT GEAR (13S19646) BECAME DISCONNECTED FROM THE CRANKSHAFT DUE TO A BROKEN BOLT. METALLURGICAL EXAMINATION OF THE ACCESSORY GEAR RETAINING BOLT (P/N STD-2213), AND DOWEL PIN (P/N STD-1065) REVEALED THAT THEY SEPARATED AS A RESULT OF HIGH CYCLE FATIGUE MECHANISM, MOST LIKELY CAUSED BY INSUFFICIENT TORQUE OF THE BOLT.

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: Inadequate maintenance by which undertorquing of the accessory gear retaining bolt resulted in fatigue failure of the bolt and the subsequent loss of engine power.

Findings

Occurrence #1: LOSS OF ENGINE POWER(TOTAL) - MECH FAILURE/MALF

Phase of Operation: CLIMB

Findings

1. ACCESSORY DRIVE ASSY - UNDERTORQUED

2. MAINTENANCE - INADEQUATE - OTHER MAINTENANCE PERSONNEL

3. ACCESSORY DRIVE ASSY - FATIGUE

4. ACCESSORY DRIVE ASSY - DISCONNECTED

Occurrence #2: FORCED LANDING

Phase of Operation: DESCENT - EMERGENCY

Occurrence #3: ON GROUND/WATER COLLISION WITH OBJECT

Phase of Operation: LANDING - ROLL

Findings

5. OBJECT - TREE(S)

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Factual Information

On July 15, 1994, at 1445 eastern daylight time, a Pitts S-15, N515TN, registered to and piloted by Teresa Nagy, sustained substantial damage when it impacted the terrain at the Kentmore Airport, Stevensville, Maryland. The pilot received minor injuries. Visual meteorological conditions prevailed, and no flight plan was filed. The flight was operated under 14 CFR Part 91.

The airplane's engine lost partial power just after takeoff. The pilot was attempting to return to the airport, when the engine lost total power. The pilot made a forced landing in a field, the airplane's wing struck a tree, and the airplane nosed over.

According to the pilot's statement on the NTSB Form 6120.1/2:

I applied power and began climb at 100 IAS...at approximately 500 feet I began to gently reduce power...just prior to the power reduction I thought I heard a "surge" but no vibration or any indication of problem. When I reduced the power the engine stopped, I immediately re applied full throttle-no result....

The engine was removed from the airplane, and retained for further examination. On October 6, 1994, the engine from N515TN, was disassembled under the supervision of NTSB Investigator, Margaret Napolitan, at Lycoming Engine's facilitates, Williamsport, Pennsylvania.

The engine disassembly revealed that engine continuity could not be established as result of the crankshaft gear (13S19646) becoming disconnected from the crankshaft. The accessory gear retaining bolt (STD-2213) was fractured, and a crack was observed in the root of the adjacent thread. According to Lycoming, "...this is consistent with overtorqueing of the bolt." The head of the bolt was worn from coming in contact with the oil pump drive after it was fractured and while it was encapsulated in the crankshaft gear counterbore. The lockplate (LW-18639) was worn, a portion of it was missing, and was not recovered. The dowel (STD-1065), used to position the gear for internal timing, was fractured. The fracture surface of the dowel displayed crack arrest lines (beach marks).

The engine disassembly also revealed that the connecting rods were not torqued to the specified 40 foot pounds. Several bolts used in various locations were not the correct parts, and safety wire was not used were required.

The following parts to include, two pieces of the accessory gear retaining bolt, P/N STS-2213; the separated dowel pin. P/N STD-1065; and lockplate, P/N STD-1065, were sent to the NTSB Materials Laboratory, Washington, DC.

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Examination of the accessory gear retaining bolt, revealed that the fracture on the head portion of the bolt and the bolt identification markings were completely obliterated by post separation mechanical damage.

The fracture face on the threaded portion of the bolt, revealed features typical of fatigue cracking over approximately 90 percent of the bolt cross section. A secondary crack was found located in the root of the third tread from the fracture surface.

The examination of the bolt revealed no material defects at the fracture origin. Material of the bolt and its hardness were found to be within the specified requirements.

Examination of the dowel pin revealed that most of the fracture contained well defined crack arrest positions consistent with a fatigue fracture mechanism. The fracture features in the final fracture zone were typical of overstress separations. A band of fretting wear was found in the area adjacent to the fracture surface.

Pilot Information

Certificate:	Private	Age:	46,Female
Airplane Rating(s):	Single-engine land	Seat Occupied:	Center
Other Aircraft Rating(s):	None	Restraint Used:	
Instrument Rating(s):	None	Second Pilot Present:	No
Instructor Rating(s):	None	Toxicology Performed:	No
Medical Certification:	Class 3 Valid Medicalw/ waivers/lim	Last FAA Medical Exam:	December 21, 1992
Occupational Pilot:	UNK	Last Flight Review or Equivalent:	
Flight Time:	640 hours (Total, all aircraft), 61 hours (Total, this make and model), 345 hours (Pilot In Command, all aircraft), 17 hours (Last 90 days, all aircraft), 14 hours (Last 30 days, all aircraft), 3 hours (Last 24 hours, all aircraft)		

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Aircraft and Owner/Operator Information

Aircraft Make:	PITTS	Registration:	N515TN
Model/Series:	S-1S S-1S	Aircraft Category:	Airplane
Year of Manufacture:		Amateur Built:	Yes
Airworthiness Certificate:	Experimental (Special)	Serial Number:	7-0255
Landing Gear Type:	Tailwheel	Seats:	1
Date/Type of Last Inspection:	June 10, 1994 Annual	Certified Max Gross Wt.:	1150 lbs
Time Since Last Inspection:	9 Hrs	Engines:	1 Reciprocating
Airframe Total Time:	144 Hrs	Engine Manufacturer:	LYCOMING
ELT:	Not installed	Engine Model/Series:	O-360-A4G
Registered Owner:	TERESA & JOHN NAGY	Rated Power:	180 Horsepower
Operator:		Operating Certificate(s) Held:	None
Operator Does Business As:		Operator Designator Code:	

Meteorological Information and Flight Plan

Conditions at Accident Site:	Visual (VMC)	Condition of Light:	Day
Observation Facility, Elevation:		Distance from Accident Site:	
Observation Time:		Direction from Accident Site:	
Lowest Cloud Condition:	Unknown	Visibility	5 miles
Lowest Ceiling:	Unknown	Visibility (RVR):	
Wind Speed/Gusts:	8 knots /	Turbulence Type Forecast/Actual:	/
Wind Direction:	90°	Turbulence Severity Forecast/Actual:	/
Altimeter Setting:		Temperature/Dew Point:	33°C / 25°C
Precipitation and Obscuration:	N/A - None - Haze		
Departure Point:	(3W3)	Type of Flight Plan Filed:	None
Destination:	EASTON , MD (ESN)	Type of Clearance:	None
Departure Time:	14:52 Local	Type of Airspace:	

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Airport Information

Airport:	KENTMORE AIRPARK 3W3	Runway Surface Type:	Grass/turf
Airport Elevation:	12 ft msl	Runway Surface Condition:	Dry
Runway Used:	10	IFR Approach:	
Runway Length/Width:	2000 ft / 75 ft	VFR Approach/Landing:	Forced landing

Wreckage and Impact Information

Crew Injuries:	1 Minor	Aircraft Damage:	Substantial
Passenger Injuries:		Aircraft Fire:	None
Ground Injuries:	N/A	Aircraft Explosion:	None
Total Injuries:	1 Minor	Latitude, Longitude:	38.989299,-76.299659(est)

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Administrative Information

Investigator In Charge (IIC):	Yurman, Alan	
Additional Participating Persons:	LEO KUNEMAN; BALTIMORE , MD	
Original Publish Date:	March 27, 1995	
Last Revision Date:		
Investigation Class:	<u>Class</u>	
Note:		
Investigation Docket:	https://data.ntsb.gov/Docket?ProjectID=38792	

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