



Aviation Investigation Final Report

Location:	FREDERICKSBURG, Texas	Accident Number:	FTW95LA018
Date & Time:	October 16, 1994, 11:15 Local	Registration:	N821AC
Aircraft:	PIPER PA-23-250	Aircraft Damage:	Substantial
Defining Event:		Injuries:	2 None
Flight Conducted Under:	Part 91: General aviation - Ferry		

Analysis

DURING THE EN ROUTE CLIMB, THE LEFT ENGINE MADE A LOUD NOISE AND LOST POWER. THE PILOT RETARDED THE THROTTLE AND TURNED AROUND TO RETURN TO THE AIRPORT. HE NOTICED THE ENGINE NACELLE WAS ON FIRE AND HE SHUT OFF THE FUEL SUPPLY. THE FIRE APPEARED TO EXTINGUISH. THE PILOT THEN RETARDED THE MIXTURE AND PROPELLER CONTROLS, WHICH IS NOT THE PROPER SEQUENCE OUTLINED IN THE AIRPLANE OWNER'S MANUAL. THE PROPELLER FAILED TO FEATHER. DESPITE FULL POWER ON THE RIGHT ENGINE AND MAINTAINING BEST SINGLE ENGINE RATE OF CLIMB SPEED, THE AIRPLANE DESCENDED 200 TO 250 FEET PER MINUTE. THE PILOT MADE A FORCED WHEELS DOWN LANDING IN AN OPEN FIELD. THE AIRPLANE STRUCK A TERRACE AND THE LANDING GEAR COLLAPSED. POST ACCIDENT ENGINE INSPECTION DISCLOSED A LOOSE FUEL INJECTOR SUPPLY LINE, AND HEAT DAMAGE AND A 1-1/2 INCH BROWN STAIN ON THE CLAMP THAT ATTACHES THE TAILPIPE TO THE EXHAUST SIDE OF THE TURBOCHARGER. ENGINE MAINTENANCE, REQUIRING THE REMOVAL AND REATTACHMENT OF FUEL LINES, HAD BEEN PERFORMED ON THE AIRPLANE JUST PRIOR TO THE FLIGHT.

Probable Cause and Findings

The National Transportation Safety Board determines the probable cause(s) of this accident to be: IMPROPER MAINTENANCE RESULTING IN AN ENGINE FIRE, AND THE PILOT'S IMPROPER PROPELLER FEATHERING PROCEDURE, CAUSING THE PROPELLER NOT TO FEATHER AND THE AIRPLANE'S ENGINE OUT CAPABILITY TO BE EXCEEDED. A FACTOR WAS THE UNSUITABLE TERRAIN CONDITIONS.

Findings

Occurrence #1: FIRE

Phase of Operation: CLIMB - TO CRUISE

Findings

1. (C) MAINTENANCE, MAJOR REPAIR - IMPROPER - OTHER MAINTENANCE PERSONNEL
 2. FUEL SYSTEM, LINE - LOOSE
 3. EXHAUST SYSTEM, TURBOCHARGER - LEAK
 4. (C) PROPELLER FEATHERING - IMPROPER - PILOT IN COMMAND
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Occurrence #2: FORCED LANDING

Phase of Operation: DESCENT - EMERGENCY

Occurrence #3: IN FLIGHT COLLISION WITH TERRAIN/WATER

Phase of Operation: LANDING - ROLL

Findings

5. (F) TERRAIN CONDITION - NONE SUITABLE
6. TERRAIN CONDITION - ROUGH/UNEVEN

Factual Information

On October 16, 1994, at 1115 central daylight time, N821AC, a Piper PA-23-250, was substantially damaged during a forced landing near Fredericksburg, Texas. The pilot and passenger were not injured. Visual meteorological conditions prevailed.

The following is based on the pilot's report. During the en route climb, the pilot heard a loud noise and noted the left engine losing power. He retarded the throttle, and turned the airplane around in an attempt to return to the airport. He then noticed the left engine nacelle was on fire and he shut off the fuel supply. The fire appeared to extinguish. The pilot then retarded the mixture control to idle cutoff and retarded the propeller control to the feather position.

The airplane owner's manual states (in part): "...the propeller on the dead engine should be feathered by pulling the throttle to idling position and the prop pitch control back fully; then the mixture should be set at idle cut-off and the ignition off." The handbook further states: "The Hartzell feathering propellers can only be feathered while the failed engine is rotating, and not if the engine drops below 1,000 RPM, because the centrifugal force due to rotation is necessary to hold out a stop-pin which keeps the propeller from feathering each time the engine is stopped on the ground. . .single engine flight can be maintained with the dead engine propeller unfeathered, although a noticeable decrease in single engine performance will take place." Retarding the mixture before retarding the propeller will cause the engine to stop with a resultant loss of oil pressure.

The following is based on the continuation of the pilot's report. Although the pilot maintained full power on the right engine and 104 MPH (90 knots) indicated airspeed (best single engine rate of climb airspeed), the airplane descended at 200 to 250 feet per minute. He then noticed the propeller had failed to feather, and

it became apparent he was not going to make it to the airport. The pilot made a forced wheels down landing in a field near the airport. During the landing roll, the airplane struck a terrace in the field and the landing gear collapsed.

The left engine was later examined by a Federal Aviation Administration (FAA) airworthiness inspector. According to his report, fire damage was confined to the left side of the engine. Some highlights of his report:

- > Oil seepage around a broken turbocharger oil inlet supply line;
- > Fire damage to the insulated fuel and oil lines in the middle and left side of the accessory section;
- > Heat damage and a 1-1/2 inch brown stain on the clamp that attaches the tailpipe to the exhaust side of the turbocharger;

> Heat damage to the turbocharger heat shield; > Loose fuel injector supply lines (a similar situation was found on the right engine).

According to the FAA, the airplane had been at the Fredericksburg Airport to be painted and some engine maintenance. The latter required the removal and reattachment of fuel lines.

Pilot Information

Certificate:	Airline transport; Commercial	Age:	30, Male
Airplane Rating(s):	Single-engine land; Multi-engine land	Seat Occupied:	Left
Other Aircraft Rating(s):	None	Restraint Used:	
Instrument Rating(s):	Airplane	Second Pilot Present:	No
Instructor Rating(s):	None	Toxicology Performed:	No
Medical Certification:	Class 1 Valid Medical--no waivers/lim.	Last FAA Medical Exam:	March 18, 1994
Occupational Pilot:	No	Last Flight Review or Equivalent:	
Flight Time:	2020 hours (Total, all aircraft), 10 hours (Total, this make and model), 1915 hours (Pilot In Command, all aircraft), 61 hours (Last 90 days, all aircraft), 19 hours (Last 30 days, all aircraft)		

Aircraft and Owner/Operator Information

Aircraft Make:	PIPER	Registration:	N821AC
Model/Series:	PA-23-250 PA-23-250	Aircraft Category:	Airplane
Year of Manufacture:		Amateur Built:	
Airworthiness Certificate:	Normal	Serial Number:	27-7554038
Landing Gear Type:	Retractable - Tricycle	Seats:	2
Date/Type of Last Inspection:	January 11, 1994 Annual	Certified Max Gross Wt.:	5200 lbs
Time Since Last Inspection:	52 Hrs	Engines:	2 Reciprocating
Airframe Total Time:	2880 Hrs	Engine Manufacturer:	LYCOMING
ELT:	Installed	Engine Model/Series:	TIO-540-C1A
Registered Owner:	CHURCH OF BIBLE UNDERSTANDING	Rated Power:	250 Horsepower
Operator:	MITCHELL, DANIEL L.	Operating Certificate(s) Held:	None
Operator Does Business As:		Operator Designator Code:	

Meteorological Information and Flight Plan

Conditions at Accident Site:	Visual (VMC)	Condition of Light:	Day
Observation Facility, Elevation:		Distance from Accident Site:	
Observation Time:		Direction from Accident Site:	
Lowest Cloud Condition:	Scattered / 1100 ft AGL	Visibility	10 miles
Lowest Ceiling:	Broken / 1300 ft AGL	Visibility (RVR):	
Wind Speed/Gusts:	10 knots /	Turbulence Type Forecast/Actual:	/
Wind Direction:	200°	Turbulence Severity Forecast/Actual:	/
Altimeter Setting:	29 inches Hg	Temperature/Dew Point:	26°C / -2°C
Precipitation and Obscuration:	N/A - None - Fog		
Departure Point:	, TX (T82)	Type of Flight Plan Filed:	IFR
Destination:	PENSACOLA , FL (PNS)	Type of Clearance:	IFR
Departure Time:	10:55 Local	Type of Airspace:	Class E

Airport Information

Airport:		Runway Surface Type:	
Airport Elevation:		Runway Surface Condition:	
Runway Used:	0	IFR Approach:	
Runway Length/Width:		VFR Approach/Landing:	Forced landing

Wreckage and Impact Information

Crew Injuries:	1 None	Aircraft Damage:	Substantial
Passenger Injuries:	1 None	Aircraft Fire:	In-flight
Ground Injuries:	N/A	Aircraft Explosion:	None
Total Injuries:	2 None	Latitude, Longitude:	30.270528,-98.869453(est)

Administrative Information

Investigator In Charge (IIC):	Scott, Arnold
Additional Participating Persons:	FRED J CECH; SAN ANTONIO , TX
Original Publish Date:	March 27, 1995
Last Revision Date:	
Investigation Class:	Class
Note:	
Investigation Docket:	https://data.nts.gov/Docket?ProjectID=19285

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