

DCA21FA174

MAINTENANCE RECORDS

Group Chair's Factual Report - Attachment 6
Airworthiness Directives and Service Bulletins
December 16, 2022



Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20		H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	R1	57-838-01-01	1	TE inboard flap spoilers and rear spar to LG support beam left wing	4000	CYC		09017	3/7/2021	12/17/2023	73,030	3,584
			10	REPEAT WOF TE inboard flap spoilers and rear spar to LG support beam left wing								
			10		540	DAY		09017	3/7/2021	08/29/2022		421
AD	R1	57-874-01-01	1	Wing trailing edge outboard, flap and spoilers right wing	4000	CYC		09017	3/7/2021	12/17/2023	73,030	3,584
			10	REPEAT WOF Wing trailing edge outboard, flap and spoilers right wing								
			10		1095	DAY		09017	3/7/2021	03/06/2024		976
AD	R1	57-882-01-01	1	TE inboard flap, spoilers and rear spar to LH support beam right wing	4000	CYC		09017	3/7/2021	12/17/2023	73,030	3,584
			10	REPEAT WOF TE inboard flap, spoilers and rear spar to LH support beam right wing								
			10		1095	DAY		09017	3/7/2021	03/06/2024		976
AD	R1	57-884-01-01	1	Wing trailing edge inboard flap and spoilers right wing	4000	CYC		09017	3/7/2021	12/17/2023	73,030	3,584
			10	REPEAT WOF Wing trailing edge inboard flap and spoilers right wing								

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			10			1095	DAY	09017	3/7/2021	03/06/2024		976
AD	NA1	74-08-09 R3	10	Lavatory exterior ashtray and smoking placards EQUIP/FURNISH Inspect all lavatory paper and linen waste receptacle enclosure access doors and disposal for proper fit, sealing, and latching for the containment of possible latch fires.		1000	HRS					
AD	R1	85-01-07	10	Fuselage rib upper chord inspection REPEAT Visual Inspection of the BBL 70.85 Rib Upper Chord		5000	CYC	5942	9/26/2018 SC	12/26/2022	71,607	2,161
AD	R1	86-12-04	20	Horizontal stabilizer center section rear spar REPEAT Eddy Current Inspection of the Horizontal Stabilizer center section rear spar upper chord for cracks in the areas of each beam gusset plate		4500	CYC	4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	R1	86-12-05	10	Horizontal stabilizer rear spar attach lug REPEAT Eddy Current inspection of the Horizontal Stabilizer center section rear spar upper chord attach lugs		5700	CYC	4266	6/3/2019 SC	11/16/2023	72,908	3,462
AD	R1	88-22-09	10	Takeoff configuration warning system TAKE OFF CONFIG Perform an operational and functional check of the takeoff configuration warning system in accordance with AMM 31-26-00.		200	HRS	08658-2	6/9/2021	08/31/2021	72,988.50	117.30

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AD	R1	90-06-02	90	Aging Airplane Structural mod. Program INITIAL Wings - Main Frame - Landing Gear Beam Swing Link Attachment Bolt Inspection	75000	CYC		DOM	1/1/2000 SC	04/22/2025	75,000	5,554
			90		18000	CYC		D10014	7/15/2011 SC	06/06/2029	81,023	11,577
AD	R1	90-06-02	110	Aging Airplane Structural mod. Program REPEAT AGING AIRCRAFT INSPECTION [Part 1] Stabilizers-Elevator Rear Spar Inspection and Repair [See also AD 2006-14-07]	1000	HRS		09017	3/7/2021	07/29/2022	73,652.10	780.90
			110		750	CYC		09017	3/7/2021	09/25/2021	69,780	334
AD	R1	90-06-02	120	Aging Airplane Structural mod. Program REPEAT AGING AIRCRAFT INSPECTION Stabilizers-Horizontal Attachment Lug Inspection and Modification [See also CPCP Task Card C53-701-02 and RAI Task Card 55-302-01-01 embedded in 2C]	1460	DAY		4560	5/21/2019 SC	05/20/2023		685
AD	R1	90-06-02	130	Aging Airplane Structural mod. Program REPEAT AGING AIRCRAFT INSPECTION - Part 3 Stabilizers-Elevator Rear Spar Inspection and Repair [See also AD 2006-14-07] NOTE: Part 2 of SB is not applicable to this A/C	8000	HRS		221-001	2/2/2015 SC	06/04/2028	77,926.00	5,054.80
			130		6000	CYC		221-001	2/2/2015 SC	10/01/2021	69,804	358

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AD	R1	92-09-02	20	Wing tanks fuel feed lines REPEAT Applicable to 810TA last done in MASA Malaysia 06/08/99	76381	CYC		DOM	1/1/2000 SC	04/02/2026	76,381	6,935
AD	R1	92-25-09	10	Fuselage corrosion inspections REPEAT <u>WOF</u> Close External Visual Inspection of all Circumferential Skin Butt splices, BS 259.5 to BS 1016	4500	CYC		08911	8/18/2020	11/13/2023	72,895	3,449
			10		456	DAY		08911	8/18/2020	11/17/2021		136
AD	R1	93-05-17	10	Cockpit "E- F" window posts REPEAT X-ray the E-F Window Posts. INITIAL AD para (f) [If cracks are found, see AD para (j)]	5500	CYC		4266	6/3/2019 SC	09/27/2023	72,708	3,262
AD	R1	93-14-10	30	Cargo door lower lobe (aft) frames REPEAT AFT LWR Cargo Doorway Frame, Inspection, Repair and mod. Detailed Visual Inspection of FWD and AFT Body Frames adjacent to AFT LWR Cargo Door	4000	CYC		3746	11/27/2018 SC	07/03/2022	70,902	1,456
AD	R1	96-10-06	10	Fwd galley service door gate hinge REPEAT Detailed Visual Inspection of FWD Galley Service Door Lower Gate Hinge. Replacement of Passenger Door gate hinge would terminate AD.	4500	CYC		4090	1/26/2019 SC	01/09/2023	71,663	2,217
AD	R1	97-14-04	10	Main rudder (PCU) inspections REPEAT Leak Check of Main Rudder PCU	6400	HRS		D10002	1/1/2011 SC	05/09/2022	73,490.00	618.80

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AD	R1	98-11-04 R1	30	Supplemental Structural Inspections (SSID) SSID Bulletin Description: Supplemental Structural Inspection Document (SSID) (Ref. SSI Doc. D6-37089) See AD file.	13924	CYC		5371	3/3/2020 SC	12/31/2029	81,856	12,410
AD	R1	98-11-04 R1	40	Supplemental Structural Inspections (SSID) SSID Bulletin Description: Supplemental Structural Inspection Document (SSID) (Ref. SSI Doc. D6-37089) See AD file.	37924	CYC		5371	3/3/2020 SC	06/05/2046	105,856	36,410
AD	R1	98-14-09	10	Forward engine mount support fitting REPEAT Detailed Visual Inspection of the FWD Engine Mount Support Fittings Lower Flange and verification of engine mounts during inspection	700	CYC		09017	3/7/2021	09/13/2021	69,730	284
AD	S	99-08-23	10	Fuselage and aft pressure bulkhead fastener inspections REPEAT Fuselage BS 1016 AFT Pressure Bulkhead - Inspection for web cracks STR-15L and STR-15R HFEC Inspection AFT side. (superceded by AD 2012-18-13)	12000	CYC						
AD	R1	99-10-12	10	Main landing gear actuator beam arm REPEAT (NDT) Ultrasonic Inspection of Main Gear Trunnion Pin, Actuator Beam Arm and Actuator Bolt Assembly [Repeat every 90 days]	90	DAY		08631-2	5/27/2021	08/25/2021		52

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AD	R1	99-11-05	20	Rudder (PCU) servo valve displacement test REPEAT 810TA last accomplished at MASA Malaysia 07/30/99	80381	CYC		DOM	1/1/2000 SC	12/27/2028	80,381	10,935
AD	R1	99-24-08	10	Airplane battery charger REPEAT Determine the condition of diode CR910 of the GCU	600	HRS		08856-4	6/26/2021	04/21/2022	73,453.20	582.00
AD	R1	FAR 121.1105	10	FAR 121.1105 Administrator audit	2555	DAY		D10154	1/30/2016 SC	01/28/2023		573
AD	R1	FAR 121.1105	20	FAR 121.1105 Notify the Administrator	2495	DAY		D10154	1/30/2016 SC	11/29/2022		513
AD	TER	2000-04-08	10	Fuselage frames / lap joints INITIAL [Initial due at 50,000 cycles, then intervals of 2,500 cycles thereafter.] AD para (a) and (b). Initial: LFEC (sliding probe) in lower skin at stringer 4R lap joint between BS 460 and BS 540 Repeat: Internal Detailed Visual Inspection	50000	CYC		D10202	5/9/2011			
AD	TER	2000-04-08	20	Fuselage frames / lap joints REPEAT Internal Detailed Visual Inspection in frames at BS 480 through BS 520	2500	CYC		D10202	5/9/2011			
AD	R1 A	2000-05-29	10	Fuselage (fwd) pressure bulkhead REPEAT Perform inspections of the the center web, vertical chords, and side chord areas of the forward pressure bulkhead (BS 178) for fatigue cracking AD para (a) [Ref. TASS Report TXE 737-200C, RAI Task No.]	6000	CYC		4266	6/3/2019 SC	01/30/2024	73,208	3,762

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				R51-R002]								
AD	TER	2001-17-18	10	Floor beams & structural integrity REPEAT Detailed Visual Inspection of Floor Beams and Transverse Beams	6000	CYC						
AD	R1	2001-17-18	20	Floor beams & structural integrity REPEAT Bulletin Description: Main Wheel Pressure Floor Transverse Beams and Floor Beams MODIFICATION [See also AD 90-06-02]	75000	CYC		DOM	1/1/2000 SC	04/22/2025	75,000	5,554
AD	P	2002-07-08	20	Cracking of fuselage lap joints HFEC Inspection OD Window Corner Fastener Holes for cracking	50000	CYC						
AD	TER	2003-07-12	10	Landing gear parts Inspect landing gear parts per SB	3650	DAY						
AD	TER	2003-26-01	10	Rudder Actuator Separation REPEAT WOF General Visual Inspection of Flange Bolts in main thrust hinge and actuator assembly and auxiliary actuator support fitting to detect loose bolts. Stage 1	2000	HRS		4095	1/30/2019 SC			
			10		1500	CYC		4095	1/30/2019 SC			

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AD	TER	2003-26-01	30	Rudder Actuator Separation REPEAT Bulletin Description: Rudder Front Spar, Mainm Thrust Hinge, Actuator and Auxilliary Actuator Support Fitting Inspection		1500	CYC	4266	6/3/2019 SC			
AD	NA1	2004-06-18	10	Lavatory Drain System LAV DRAIN SYS. WOF Conduct a leak test of the toilet tank dump valve & the service panel drain valve. Perform a general visual inspection of the outer cap/door and seal mating surface for wear or damage.		4500	HRS					
	(a)(2)		10			547	DAY					
AD	NA1	2004-06-18	20	Lavatory Drain System REPEAT Replace to prevent engine damage, airframe damage, and/or hazard to person or property on the ground result of "Blue Ice" valve seals.		5000	HRS					
	(a)(1)											
AD	NA1	2004-06-18	30	Lavatory Drain System REPEAT Replace to prevent engine damage, airframe damage, and/or hazard to person or property on the ground as a result of "blue ice" seals in the flush/fill ball valve and the toilet tank ant-siphon.		5000	HRS					
	(a)(5)											
AD	R1	2004-19-10	10	Pivot hinge pins of the horizontal stabilizer REPEAT WOF Part 2 Inspection. Inner Hinge Pins are visually inspected for corrosion and then magnetic particle inspected for cracks. Outer Hinge pins are visually inspected in place		6000	HRS	09017	3/7/2021	06/02/2029	78,652.10	5,780.90

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			10			1460	DAY	09017	3/7/2021	03/06/2025		1,341	
AD	R1	2004-19-10	20	Pivot hinge pins of the horizontal stabilizer		12000	HRS	09017	3/7/2021	08/19/2037	84,652.10	11,780.90	
			20	REPEAT SB 737-55A1077, Part 3 Both the inner and outer horizontal stabilizer hinge pins are visually inspected for corrosion and then magenetic particle inspected for cracks.		2920	DAY	09017	3/7/2021	03/05/2029		2,801	
			10	WOFF Overwing Fuel Fill Ports ONCE Do an electrical bonding resistance test between the bulkhead fittings of the engine fuel feed tube and from the spar inside the fuel tank of the wing to determinates the resistance				D10171	3/26/2010				
AD	R1	2005-07-19	40	Cracks in the fuselage skin, doubler, bearstrap, and frames		12000	CYC	D10052	1/1/2012 SC	05/11/2024	73,616	4,170	
			40	REPEAT AD para (f) Detailed v visual and HFEC inspection of skin, doubler, bearstrap at upper corners of Main Cargo door cutout.		8000	CYC	5371	3/3/2020 SC	12/11/2025	75,932	6,486	

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AD	R1	2005-07-19	60	Cracks in the fuselage skin, doubler, bearstrap, and frames REPEAT To find and fix fatigue cracking in the fuselage skin, doubler, bearstrap and frames. AD para (f) [See also AD 93-14-10, SB 737-53-1096, R5]	4000	CYC		4266	6/3/2019 SC	09/17/2022	71,208	1,762
AD	NA1	2005-07-19	70	Cracks in the fuselage skin, doubler, bearstrap, and frames REPEAT S 440 Fuselage Reinforcement Repair Inspection	4000	CYC						
AD	S	2005-13-30	10	Fuselage skin panels SUPERCEDED SUPERCEDED by 2017-02-02	3600	CYC		698-013	5/20/2017 SC			
AD	R1	2005-18-08	10	Main landing gear (MLG) beam REPEAT Detailed Inspection of retaining pin lugs of the support fitting of the MLG beam, or elongation of a bolt hole in a lug AD para (f)	12000	CYC		D10172	5/10/2007 SC	04/29/2024	73,567	4,121
AD	R1	2005-19-25	10	Countersunk rivet holes in the lower lobe REPEAT Perform eddy current inspection for cracks of the surface area around the satellite holes of the radio altimeter cutouts between BS 390 and BS 450 with fasteners installed	4500	CYC		4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	TER	2005-20-01	10	Stiffeners at left buttock line (LBL) and right buttock line (RBL) TERMINATED AD HAS BEEN TERMINATED BY 2017-02-02 AND REPLACED BY AD 2019-16-12	4500	CYC		4266	6/3/2019 SC			

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AD	TER	2006-03-12	10	Center fuel tank REPEAT Detailed Inspection for chafing of the wire bundles located below the pax comp above the center fuel tank		3650	DAY					
AD	NA4 G	2006-12-17	10	Fuselage skin in the lower lobe cargo compartment REPEAT Internal inspection to detect and correct cracking of the fuselage frames from BS 360 to BS 500B. AD para (g)		9000	CYC					
AD	R1 H	2006-12-17	20	Fuselage skin in the lower lobe cargo compartment INITIAL To detect and correct cracking of the fuselage frames from BS 360 to BS 500B. Prior to accumulation of 75,000 cyc. modification by installation of doublers on the specified frames ...AD para (h) [After modification, see AD para (i)]		75000	CYC	DOM	1/1/2000 SC	04/22/2025	75,000	5,554
AD	R1	2006-12-23	10	Elevator tab assembly REPEAT WOF To prevent excessive in-flight vibrations of the elevator tab. Tab Control Inspection		6000	HRS	06197	7/7/2020	11/13/2028	78,250.90	5,379.70
			10			4500	CYC	06197	7/7/2020	10/07/2023	72,747	3,301
AD	R1	2006-12-23	20	Elevator tab assembly REPEAT WOF To prevent Excessive In-Flight Vibrations of the Elevator Tab. Clamped Hinge Putty Inspection, Hinge Detailed Inspections, Free Play Inspections		2000	HRS	09017	3/7/2021	12/11/2023	74,652.10	1,780.90

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			20			1500	CYC	09017	3/7/2021	04/01/2022	70,530	1,084
AD	R1	2006-14-07	10	Elevator tab inboard hinge support structure		1000	HRS	09017	3/7/2021	07/29/2022	73,652.10	780.90
	K			REPEAT WOF To detect and correct cracking, corrosion, interlaminar corrosion, delamination and disbonding in the elevator rear spar. AD para (k) Part 1 Detailed Inspection - Visual Inspection for cracks at Hinge Bracket Locations and Interlaminar Corrosion of the Spar								
			10			750	CYC	09017	3/7/2021	09/25/2021	69,780	334
AD	R1	2006-14-07	20	Elevator tab inboard hinge support structure		8000	HRS	09017	3/7/2021	02/27/2032	80,652.10	7,780.90
	K			REPEAT WOF To detect and correct cracking, corrosion, interlaminar corrosion, delamination and disbonding in the elevator rear spar. AD para (k) Part 3 Special Detailed Inspection - Ultrasonic Inspection for Interlaminar Corrosion of the Spar 8,000 Hrs. or 6,000 cyc, whichever comes first								
			20			6000	CYC	09017	3/7/2021	04/30/2025	75,030	5,584
AD	R1	2006-21-01	10	Aileron balance tabs		6000	HRS	09017	3/7/2021	06/02/2029	78,652.10	5,780.90
				REPEAT WOF To Prevent Excessive Vibration of the Airframe during flight. Measure freeplay of both aileron								

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Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20		H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
			10	control balance tabs. AD para (g)(1)	730	DAY		09017	3/7/2021	03/07/2023		611
AD	R1	2006-21-01	20	Aileron balance tabs REPEAT WOF To Prevent Excessive Vibration of the Air frame during flight. Lubricate the aileron balance tab components specified in SB. AD para (i)(1)	3000	HRS		09017	3/7/2021	04/24/2025	75,652.10	2,780.90
			20		270	DAY		09017	3/7/2021	12/02/2021		151
AD	R1	2007-03-03	10	Vertical beam webs REPEAT To prevent fatigue cracks in certain vertical beam webs. Perform detailed insp. and HFEC to detect cracks in the BS 178 vertical beam webs. AD para (f)(1)	6000	CYC		698-013	5/20/2017 SC	05/30/2022	70,769	1,323
AD	R1	2007-03-03	30	Vertical beam webs REPEAT To prevent fatigue cracks in certain vertical beam webs. Replace vertical beams at buttock lines (BL) 5.7 and 17.0 of the BS 178 bulkhead. AD para (i)	25000	CYC		D10054	1/1/2010 SC	03/23/2033	86,567	17,121
AD	R1	2007-16-05	10	Elevator tab control rod assemblies REPEAT WOF Perform detailed inspection for discrepancies of the inspection putty of the elevator tab control rod assemblies and/or damage to surrounding structure per SB instructions. AD para (f)	6000	HRS		4266	6/3/2019 SC	03/02/2028	77,737.50	4,866.30

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			10			4500	CYC	4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	R1	2007-24-02	10	Fuel boost pump REPEAT To detect and correct chafing of the fuel boost pump electrical wiring and the surrounding conduit. AD para (f)		15000	HRS	D10002	1/1/2011 SC	02/15/2034	82,090.00	9,218.80
AD	S	2007-26-04	20	Forward fastener row REPEAT To prevent cracking of the STA 259.5 circumferential butt splice. Initial Inspection to be carried out before 50,000 Cyc. [Repeat inspections at 9,000 Cyc.]		9000	CYC	D10051	11/28/2012 SC			
AD	S	2007-26-04	40	Forward fastener row PREVENT MOD To prevent cracking of STA 259.5 circumferential butt splice. Perform preventative modification per SB 737-53A1076, para 3.B.1. Before 75,000 cyc. AD para (h)		75000	CYC	DOM	1/1/2000 SC			
AD	R1	2008-08-10	10	Main landing gear (MLG) REPEAT To detect and correct fatigue cracking of the OTBD stabilizing fitting and stress corrosion cracking of the bolts attaching the fitting to the wing rear spar. AD para (g)		3650	DAY	09017	3/7/2021	03/05/2031		3,531
AD	P	2008-08-23	30	Structural significant item (SSI) INITIAL		66000	CYC					
AD	R1	2008-09-15	10	Flight deck No. 2, 4 and 5 windows REPEAT WOF Internal and External Detailed Inspection for		7500	HRS	4064	1/9/2019 SC	02/11/2030	79,159.40	6,288.20

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
			10	cracking and damage in the vinyl interlayer and glass panes of left and right Flight Deck #2 Windows		1095	DAY	4064	1/9/2019 SC	01/08/2022		188
AD	TER	2008-10-09 R1	10	FAA-approved maintenance program REPEAT To Prevent the Potential for Ignition sources inside Fuel Tanks		3650	DAY	D10213	4/25/2011 SC			
AD	TER	2008-11-03	30	Repetitive inspections to detect cracks of all SSIs SSID Bulletin Description: Paragraph I Before the accumulation of the 66000 total flight cycles, or within 4,000 flight cycles measured from 12 months after the effective date of this AD, whichever occurs later, do initial inspections to detect cracks of all SSIs, in accordance with Appendix A.		63205	CYC	5371	3/3/2020 SC			
AD	S	2008-13-12 R1	20	Upper frame to side frame splice of the fuselage SUPERCEDED AD SUPERCEDED BY AD 2017-04-08		5000	CYC					
AD	R1	2009-04-15	20	Stringer tie clips REPEAT To detect and correct multiple adjacent cracked stringer tie clips and damaged skin and frames. Perform detailed inspection and eddy current. AD para (f)(h)(j) [See also AD 93-08-04]		20000	CYC	D10014	7/15/2011 SC	10/19/2030	83,023	13,577

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AD	R1	2009-04-15	30	Stringer tie clips REPEAT Bulletin Description: To Detect and Correct Multiple adjacent Cracked Stringer Tie-Clips and Damaged skin and frames		15000	CYC	D10014	7/15/2011 SC	05/18/2027	78,023	8,577
AD	NA4	2009-04-15	40	Stringer tie clips INITIAL Initial inspection complied-with by Aero inspections. Ref. SI-009 (B2 Ck) Internal Eddy Current & Detailed Inspection of stringer Tie Clips at Sta 559 to Sta 887 ----- Inspection of stringer Tie Clips at Sta 360 to Sta 540 and Sta 907		74094	CYC					
AD	NA2	2009-09-06	20	Backup intercostals INITIAL Bulletin Description: To Detect and Correct Fatigue Cracking of Backup Intercostals and Upper Sill Web of FWD Airstair Doorway INITIAL INSPECTION [Repeat every 4500 Cyc.]		45000	CYC					
AD	R1	2009-09-07	10	Retract actuator beam REPEAT [LEFT MLG] To detect and correct broken retract actuator beams of the MLG. OVH Retract Actuator Beam [Interval=10 yrs. from most recent OVH date.]		3650	DAY	09017	3/7/2021	03/05/2031		3,531
AD	R1	2009-09-07	20	Retract actuator beam REPEAT [RIGHT MLG] To detect and correct broken retract actuator beams of the MLG. OVH Retract Actuator Beam I/A/W SB 737-32A1355 R2, Part 2. AD para (g) [Interval=10 yrs. from most recent OVH date]		3650	DAY	09017	3/7/2021	03/05/2031		3,531

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	R1	2010-05-13	90	<p>Fuselage</p> <p>REPEAT</p> <p>Bulletin Description: To prevent rapid decompression of the airplane due to fatigue cracks resulting from scribe lines on pressurized fuselage structure. AD para (q) [Initial Insp. I/A/W threshold calculated per SB instructions]</p> <p>ZONE 3Bulletin Description: To prevent rapid decompression of the airplane due to fatigue cracks resulting from scribe lines on pressurized fuselage structure. AD para (r) [Initial Insp. I/A/W threshold calculated per SB instructions.]</p> <p>ZONE 4Bulletin Description: To prevent rapid decompression of the airplane due to fatigue cracks resulting from scribe lines on pressurized fuselage structure. AD para (r) [Initial Insp. I/A/W threshold calculated per SB instructions.</p> <p>ZONE 5Bulletin Description: To prevent rapid decompression of the airplane due to fatigue cracks resulting from scribe lines on pressurized fuselage structure. AD para (s)(4)(i)(ii) (Ref. FAA AMOC No. 120S-10-217]</p>	4500	CYC		4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	P	2010-05-13	150	<p>Fuselage</p> <p>ONE TIME</p> <p>LRTS Zone 1A One time inspection</p>	45000	CYC						
AD	P	2010-05-13	160	<p>Fuselage</p> <p>REPEAT</p> <p>LRTS NDT Zone 1A</p>	250	CYC						
AD	S	2010-25-06	10	<p>Fuselage</p> <p>REPEAT</p> <p>SB 737-53A1254 R1To Detect and Correct Fatigue Cracking of Certian Fuselge Frames & Stub Beams and possible severed Frames [Detailed Insp-4500 Cyc Interval / HFEC Insp-9,000</p>	9000	CYC						

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Interval]												
AD	R1	2011-24-12	20	Fuselage REPEAT Option 1 fuselage skin panels at the chem-milled steps	1800	CYC		09017	3/7/2021	06/15/2022	70,830	1,384
AD	R1	2011-27-03	10	Flight controls REPEAT <u>WOF</u> To Prevent an Undetected failure of the primary load path f for the ballscrew in the drive mechanism of the HSTA and subsequent wear and failure of the secondary load path. INSPECTION	4000	HRS		08231-4	1/12/2021	07/13/2026	76,541.60	3,670.40
			10		730	DAY		08231-4	1/12/2021	01/12/2023		557
AD	R1	2011-27-03	20	Flight controls REPEAT <u>WOF</u> To preventy an undetected failure of the primary load path f for the ballscrew in the drive mechanism of the HSTA and subsequent wear and failure of the secondry load path. LUBRICATION	2000	HRS		08231-4	1/12/2021	10/17/2023	74,541.60	1,670.40
			20		365	DAY		08231-4	1/12/2021	01/12/2022		192
AD	TER	2012-02-09	10	Air conditioning ONE TIME To prevent ignition of the BMS 8-39 or AMS 3570 poly urethane foam insulation on the duct assemblies of the ECS due to potential electrical arc. Rework applicable duct assemblies in the ECS per Appendix A of SB 737-21A1132 R3. AD para (g)	2190	DAY		698-011	5/19/2017			

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AD	S	2012-12-05	10	Fuselage REPEAT Detailed inspection for cracking of the intercostal webs and attachment clips located FWD and AFT of the forward entry door frame, Superseded by 2018-04-06		6000	CYC					
AD	R1	2012-18-13 R1	10	Fuselage REPEAT Detailed and LFEC Inspection [S-5L to S-7L and S-5R to S-9R]		1200	CYC	09017	3/7/2021	01/16/2022	70,230	784
	k		10			3800	CYC					
AD	R1	2012-18-13 R1	20	Fuselage REPEAT Detailed insp and HFEC inspections 15L to 7L, 5L to 5R and 9R to 15R		12000	CYC	221-001	2/2/2015 SC	11/09/2025	75,804	6,358
	k		20			6000	CYC					
AD	P	2012-18-13 R1	30	Fuselage REPEAT Detailed and HFEC inspections of the web from the forward or aft side of the bulkhead for cracks, incorrectly drilled fastener holes, and elongated fastener holes,		12000	CYC					
	l		30			6000	CYC	09017	3/7/2021	04/30/2025	75,030	5,584

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AD	TER	2012-18-13 R1	40	Fuselage ONE TIME One time inspection under the Tear Strap		1200	CYC	D10093	10/14/2014			
AD	R1	2012-18-13 R1	50	Fuselage REPEAT No oil canning		6000	CYC	09017	3/7/2021	04/30/2025	75,030	5,584
AD	R1	2012-18-13 R1	60	Fuselage REPEAT Inspection of the Dome Cap at the Center of the Bulkhead. Terminated if AD 2018-04-08 para (h) initial inspection for zone 1 accomplished.		6000	CYC	09017	3/7/2021	04/30/2025	75,030	5,584
AD	R1	2012-18-13 R1	70	Fuselage REPEAT Fuselage - AFT Pressure Bulkhead; Z Stiffeners at FWD Flange AD para (p) Part 5/Table 6 of SB		12000	CYC	221-001	2/2/2015 SC	11/09/2025	75,804	6,358
AD	TER	2012-20-01	10	Fuselage REPEAT To Prevent Cracks at the Adjacent Mating Skins (FWD and AFT) which could initiate just above stringers S-4R and S-4L PART 1 - Inspection		500	CYC	4560	5/21/2019			
AD	TER	2012-20-01	40	Fuselage ONE TIME Bulletin Description: To Prevent Cracks at the Adjacent mating skins (FWD and AFT) which could initiate just above stringers S-4R and S-4L PART 2 - Chemical Test		4500	CYC	4560	5/21/2019			
AD	R1	2012-23-04	10	Fuselage INITIAL If BS 727 Frame Chord has been replaced. Perform Close Visual, Pulse Echo Shear Wave (PESW), and HFEC Inspections to Outboard Chord of the Frame		30000	CYC	D10149	2/15/1996 SC	12/17/2029	81,800	12,354

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at BS727. Retained from AD 95-12-17 para (a)												
AD	P	2012-23-04	20	Fuselage REPEAT Repeat the inspections required by para (g) until the optional terminating action para (r) is accomplished. (1) If, at the time of the most recent inspection required by para (g) or (h) of this AD, airplane has accumulated 27,000 or more flight cycles, but fewer than 50,000 flight cycles since replacement of the outboard chord: Repeat inspection 15,000 flight cycles until airplane has accumulated 50,000 or more flight cycles since the replacement of the outboard chord.	15000	CYC						
AD	P	2012-23-04	21	Fuselage REPEAT Repeat the inspections required by para (g) until the optional terminating action para (r) is accomplished. (2) If, at the time of the most recent inspection required by para (g) or (h) of this AD, the airplane has accumulated 50,000 or more flight cycles, but fewer than 60,000 flight cycles, since replacement of the outboard chord: Repeat inspection 7,500 flight cycles until the airplane has accumulated 60,000 cycles	7500	CYC						
AD	P	2012-23-04	22	Fuselage REPEAT Repeat the inspections required by para (g) until the optional terminating action para (r) is accomplished. (3) If, at the time of the most recent inspection required by para (g) or (h) of this AD, the airplane has accumulated 60,000 or more flight cycles, but fewer than 70,000 flight cycles, since replacement of the outboard chord: Repeat the inspection 5,000 flight cycles until the airplane has accumulated	5000	CYC						

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				70,000								
AD	P	2012-23-04	23	Fuselage REPEAT Repeat the inspections required by para (g) until the optional terminating action para (r) is accomplished (4) If, at the time of the most recent inspection required by para (g) or (h) of this AD, the airplane has accumulated 70,000 or more flight cycles since replacement of the outboard chord: Repeat the inspection thereafter at intervals not to exceed 3,000 flight cycles.		3000	CYC					
AD	TER	2012-23-04	25	Fuselage INITIAL BS 727 frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced Perform close visual, PESW, and HFEC inspections to detect cracks in the outboard chord of the frame at BS 727 (3) For airplanes that have accumulated 60,000 or more total flight cycles, but fewer than 70,000 total flight cycles as of August 18, 1995 (the effective date of AD 95-12-17)): Inspect within 1,500 flight cycles after August 18, 1995		1500	CYC					
AD	TER	2012-23-04	40	Fuselage REPEAT *SUPERSEDED AD95-12-17 para (d) BS 727 frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced. One-Time close visual, PESW & HFEC -To Detect and Correct Fatigue cracking of OTBD and Auxiliary Chords AD para. (j)(1) [After 50,000 cy. see AD para (j)(2)] [See also AD 95-12-17, C/W by Xiamen 11/18/00]		60000	CYC					

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AD	P	2012-23-04	41	Fuselage REPEAT BS 727 f frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced. (1) If, at the time of the most recent inspection required by para (i) or (j) of this AD, the airplane has accumulated 27,000 or more total flight cycles, but fewer than 50,000 total flight cycles. Repeat 15,000 flight cycles until the airplane has accumulated 50,000 or more total flight cycles		15000	CYC					
J (1)												
AD	P	2012-23-04	42	Fuselage REPEAT BS 727 f frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced. (2) If, at the time of the most recent inspection required by para (i) or (j) of this AD, the airplane has accumulated 50,000 or more total flight cycles, but fewer than 60,000 total flight cycles: Repeat 7,500 flight cycles until the airplane has accumulated 60,000 or more total flight cycles.		7500	CYC					
J (2)												
AD	P	2012-23-04	43	Fuselage REPEAT BS 727 f frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced. (3) If, at the time of the most recent inspection required by para (i) or (j) of this AD, the airplane has accumulated 60,000 or more total flight cycles, but fewer than 70,000 total flight cycles: Repeat 5,000 flight cycles until the airplane has accumulated 70,000 cy cles		5000	CYC					
J (3)												
AD	P	2012-23-04	44	Fuselage REPEAT BS 727 f frame outboard chord has not been replaced, or on which only the lower outboard chord has been replaced. (4) If, at the time of the most recent inspection required by para (i) or (j) of this AD, the		3000	CYC					
J (4)												

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airplane has accumulated 70,000 or more total flight cycles: Repeat the inspection thereafter at intervals not to exceed 3,000 flight cycles.												
AD	R1	2012-23-04	60	Fuselage REPEAT Internal Detailed and HFEC Inspections of the Auxiliary Chord Radius of the Frame at BS 727		15000	CYC	4258	2/10/2019 SC	03/26/2030	82,197	12,751
AD	R1	2012-23-04	70	Fuselage REPEAT If BS 727 Frame Chord has not been replaced and has not been modified. Ultrasonic and surface HFEC inspections to detect cracks in the FWD Flange of the Outboard Chord of the Frame at BS 727		5000	CYC	4258	2/10/2019 SC	05/22/2023	72,197	2,751
AD	R1	2012-23-04	80	Fuselage REPEAT If BS 727 Frame Chord has not been replaced and has not been modified. Open Hole Eddy Current Inspection to detect cracks in the FWD Flange of the Outboard Chord of the frame at BS 727.		15000	CYC	4258	2/10/2019 SC	03/26/2030	82,197	12,751
AD	P	2012-23-04	90	Fuselage REPEAT (If BS727 Frame Chord has been replaced and has not been modified) Ultrasonic, LFEC and edge HFEC Inspections to detect cracks in the FWD Flange of the Outboard Chord of the frame at BS 727		5000	CYC					
AD	R1	2012-23-04	95	Fuselage INITIAL 810TA (If BS727 Frame Chord has been replaced and has not been modified) Ultrasonic, LFEC and edge	LH SIDE	27000	CYC	D10149	2/15/1996 SC	11/28/2027	78,800	9,354

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Aircraft Document

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A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20	H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				HFEC Inspections to detect cracks in the FWD Flange of the Outboard Chord of the frame at BS 72. Repeat 5000 cycles. c/w will terminate para (g) and (h)								
AD	TER	2012-23-14	10	Fuselage TERMINATED Bulletin Description: To Detect and Correct Cracking in the Fuselage Skin and Internal Support Structure repeat 4500 cycles after the compliance	124	CYC		5396	4/18/2020			
AD	R1	2013-04-05	10	Fuselage REPEAT Bulletin Description: To Detect and Correct cracks and corrosion of the skin and surrounding structure under the #3 VHF Antenna External Detailed Inspection	1100	CYC		09017	3/7/2021	12/22/2021	70,130	684
AD	R1	2013-04-05	20	Fuselage REPEAT Bulletin Description: To detect and correct cracks and corrosion of the skin and surrounding structure under the #3 VHF Antenna External & Internal HFEC	6000	CYC		09017	3/7/2021	04/30/2025	75,030	5,584
AD	R1	2013-04-05	30	Fuselage REPEAT Bulletin Description: To detect and correct cracks and corrosion of the skin and surrounding structure under the #3 VHF Antenna External DI & HFEC per SB para 3.B.4.a and Internal HFEC per SB para 3.B.5.a	18000	CYC		D10053	1/2/2012 SC	11/10/2029	81,653	12,207
AD	TER	2013-09-02	10	Wings REPEAT WOF To detect and correct damage of the aft end of each	10000	CYC		698-011	5/19/2017 SC			

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Type	Status	Doc	Item	Description / Action Required	Position	Limit	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
			10	INBD flap track of the wing OTBD flap. SUPERSEDES AD 2000-25-07 and AD 2002-05-07		1460	DAY	698-011	5/19/2017 SC			
AD	TER	2013-09-02	50	Wings REPEAT Do a detailes visual inspection to detect damage (corrosion, cracking) of the AFT end of the LH & RH inboard flap track of the wing outboard flap.		1460	DAY					
AD	TER	2013-09-10	10	Doors REPEAT Supersedes AD 2000-07-06 *Cycle intervals adjusted due to cycles previously accumulated on door since last inspection by Aeroinvisiones on 8/12/05. To Prevent fatigue cracking of LWR corners of the door frame and No. 5 cross beam of the FWD cargo door. NDT & Visual		4500	CYC					
AD	PCW	2013-09-10	20	Doors REPEAT Supersedes AD 2000-07-06 *Cycle intervals adjusted due to cycles previously accumulated on door since last inspection by LAM LINHAS Aereas de Mozambique 5/9/02To prevent fatigue cracking of LWR corners of the door frame and No. 5 cross beam of the FWD Cargo Door MODIFICATION		10625	CYC					

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AD	TER	2013-13-12	10	Doors REPEAT To prevent fatigue cracking of the corners of the door frame and the lower cross beams of the aft cargo door. Perform Visual Inspection and HFEC AD para (k) Supersedes A D 2000-06-13.(Terminated due to modification)		4500	CYC					
AD	TER	2013-13-12	20	Doors REPEAT Doors - AFT Cargo Cross Beam E, Adjacent LWR corner frames and reinforcing angles. MODIFICATION (Ref: Transmile Modification Bulletin MB-732-52-012)		9000	CYC					
AD (g)(p)	TER	2013-18-08	10	Fuselage REPEAT To Find and Fix fatigue cracking of certain upper and lower skin panels. Initial External Visual and Eddy Current Inspection I/A/W Part 2 of SB		35000	CYC					
AD (h)	TER	2013-18-08	20	Fuselage REPEAT External Visual Inspection of Lower Lobe Area and Section 41 of Fuselage		40000	CYC					
AD PQRS	TER	2013-18-08	30	Fuselage REPEAT To Find and Fix fatigue cracking of certain upper and lower skin panels [Superseded AD2004-18-06]		4500	CYC					
AD	TER	2013-18-08	40	Fuselage REPEAT Bulletin Description: To Find and Fix fatigue cracking of certain upper and lower skin panels Threshold - 1800 Cyc from Effective Date of AD 11/6/13 [Repeat at 1800 Cyc. Intervals]		1800	CYC	4266	6/3/2019	SC		

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[Superseded AD2004-18-06]												
AD	TER	2013-18-08	(r)	50	Fuselage TERMINATING WOF To detect and correct fatigue cracking of skin panels TERMINATING ACTION for repetitiv e inspections	55000	HRS					
				50		55000	CYC					
AD	TER	2013-19-08	(G) 2	10	Stall Warning System STALL WARN. SYS (ONE TIME) - Do a general visual inspection of the left & right angle of attack (AOA) sensor as applicable, to determine if a certain AOA sensor with a paddle type vane is installed, in accordance with the Accomplishment Instruction of the applicable service informatino specified in paragh (G) (2) of AD.							
AD	TER	2014-01-05		30	Fuselage INITIAL HFEC & LFEC inspections STA 360 IAW SB 737-53-1322; Lap Joint Modification as required IAW SB 737-53A1177	5000	CYC	4255	2/7/2019	SC		
AD	R1	2014-01-05		40	Fuselage REPEAT HFEC & LFEC inspections STA 908 IAW SB 737-53-1322; Lap Joint Modification as required IAW SB 737-53A1177	13000	CYC	4255	2/7/2019	SC	11/11/2028	80,194
AD	S	2014-03-06		10	Wings AD is replaced by 2014-12-13							

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AD	TER	2014-04-05	10	Air conditioning ONE TIME To Prevent damage to wire bundles; General Visual Inspection for any Metal Clamp installed on hoses to Air=Conditioning Temp. Sensor, Gasper Air Outlet, and Diff user, and for damage to adjacent wire bundles at Sta. 259.5 [Within 60 mos. after effective date of AD - 4/1/14		1825	DAY	4560	5/21/2019			
AD	NA4	2014-05-12	10	Wings REPEAT <u>WOL</u> Work package 1: NDT (New Carriage splindles never been overhauled with CSN > 20000 flight cycles or 8 years in service since spindle was new, whichever is less)		150	CYC					
	J		10			15	DAY					
AD	NA4	2014-05-12	20	Wings REPEAT <u>WOL</u> Work package 2: Gap Check (New Carriage splindles never been overhauled with CSN > 20000 flight cycles or 8 years in service since spindle was new, whichever is less)		30	CYC					
	h		20			3	DAY					
AD	NA4	2014-05-12	30	Wings REPEAT <u>WOL</u> Work package 1: NDT (New Carriage splindles never been overhauled with CSN 12000 to 20000 flight cycles)		450	CYC					
	J		30									

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			30			45	DAY					
AD	NA4	2014-05-12	40	Wings REPEAT <u>WOL</u> Work package 2: Gap Check (New Carriage splindles never been overhauled with CSN 12000 to 20000 flight cycles)		150	CYC					
	h		40			15	DAY					
AD	NA4	2014-05-12	50	Wings REPEAT <u>WOL</u> Work package 2: Gap Check (Overhauled Carriage Spindles with CSO of > 20,000 cycles or 8 years in service since spindle was overhauled whichever is less)		30	CYC					
	h		50			3	DAY					
AD	NA4	2014-05-12	60	Wings REPEAT <u>WOL</u> Work package 1: NDT (Overhauled Carriage Spindles with CSO of 12,000 to 20,000 cycles or 8 years in service since spindle was overhauled whichever is less)		150	CYC					
	J		60			15	DAY					
			60									

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AD	h	NA4 2014-05-12	70	Wings REPEAT <u>WOL</u> Work package 2: Gap Check (Overhauled Carriage Spindles with CSO of 12,000 to 20,000 cycles or 8 years in service since spindle was overhauled whichever is less)		30	CYC					
			70			3	DAY					
AD	J	NA4 2014-05-12	80	Wings REPEAT <u>WOL</u> Work package 1: NDT (Overhauled Carriage Spindles with CSO of 3,000 to 12,000 cycles)		500	CYC					
			80			60	DAY					
AD	h	NA4 2014-05-12	90	Wings REPEAT <u>WOL</u> Work package 2: Gap Check (Overhauled Carriage Spindles with CSO of 3,000 to 12,000 cycles)		70	CYC					
			90			7	DAY					
AD	J	R1 2014-05-12	100	Wings REPEAT <u>WOL</u> Work package 1: NDT (Overhauled Carriage Spindles with CSO of < 3,000 cycles) After 2nd overhaul track at items 80 and 90		1500	CYC	09017	3/7/2021	04/01/2022	70,530	1,084
			100			180	DAY	09017	3/7/2021	09/03/2021		61

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AD	NA4	2014-05-12	J	Wings REPEAT Work package 1: NDT (Overhauled Carriage Spindles with more than 40,000 cycles or undocumented lif e) <u>WOL</u>	110	60	CYC					
					110	6	DAY					
AD	NA4	2014-05-12	h	Wings REPEAT Work package 2: Gap Check (Overhauled Carriage Spindles with more than 40,000 cycles or undocumented lif e) <u>WOL</u>	120	20	CYC					
					120	1	DAY					
AD	R1	2014-05-12	o	Wings REPEAT Overhaul of the flap carriages. <u>WOF</u>	130	12000	CYC	4560	5/21/2019	03/07/2028	79,200	9,754
					130	2920	DAY	4560	5/21/2019	05/19/2027		2,145
AD	TER	2014-05-12	p	Wings REPEAT Carriage Spindle Replacement for Model 737-100, -200, and -200C Series Airplanes	140	40000	CYC	D10057	1/1/2013			
AD	NA1	2014-05-12	q	Wings REPEAT Carriage Spindle Replacement for Model 737-300, -400, and -500 Series Airplanes	150	40000	CYC					

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AD	TER	2014-05-12	160	Wings REPEAT Carriage Spindle Replacement for Airplanes with an undocumented carriage	7500	CYC		D10057	1/1/2013			
			160		1095	DAY		D10057	1/1/2013			
AD	R1	2014-05-12	170	Wings REPEAT 40,000 cycles is the life limit of the flap carriage and spindle needs replacement.	40000	CYC		D10057	1/1/2013 SC	12/12/2044	103,697	34,251
AD	NA4	2014-05-12	180	Wings INITIAL <u>WOF</u> For New (never been overhauled) Carriage Spindles: Start clock for NDT and Gap check	12000	CYC						
			180		2920	DAY						
AD	R1	2014-05-12	190	Wings INITIAL For Overhauled Carriage Spindles has CSO of < 3,000 cycles: Start clock for Gap check inspection item 90	3000	CYC		4560	5/21/2019 SC	01/08/2022	70,200	754
AD	R1	2014-05-21	10	Fuselage REPEAT To detect and correct cracks found in the skin, bearstrap, and/or frame outer chord in the hinge cutout areas of the FWD entry and FWD galley service doorways. Repeat Inspection - FWD Entry Doorway [See also AD 2008-11-04]Supersedes AD 2008-11-04	18000	CYC		698-013	5/20/2017 SC	08/16/2030	82,769	13,323

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AD	NA4	2014-05-27	10	Navigation REPEAT To correct possible misalignment issues with TPR-720 and TPR 900 Mode S Transponders; Test and Calibration (within 2yrs. of effective date of AD - 5/2/14)		1460	DAY					
AD	R1	2014-08-08	10	Fuselage REPEAT To detect and correct cracking in the LWR corners of the FWD entry doorway. Perform Detailed Insp-internal & external, and HFEC- internal & external of bear strap and skin assembly		4500	CYC	4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	TER	2014-08-08	20	Fuselage REPEAT Bulletin Description: To detect and correct cracking in the LWR corners of the FWD entry doorway. Perform Detailed Insp-internal & external, and HFEC- internal & external of bear strap and skin assembly		65550	CYC					
AD	R1	2014-12-13	10	Wings REPEAT [Compliance due within 4500 Cyc. after effective date of AD 2014-03-06 - 4/9/14] Supersedes AD 2014-03-06 Option 3 Action 1: do a HFEC surface inspection for any crack in the rear spar upper chord and for any crack in the rear spar web figure 15 and 16.		4500	CYC	4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	R1	2014-12-13	20	Wings INITIAL Option 3 Action 2: HFEC open hole inspection for any crack in the forward support fitting, the rear spar upper chord, and the rear spar web at the 12 fasteners holes (location 1-12). Figure 3,4,15, and 16. Repeat 24000 cycles		18000	CYC	221-001	2/2/2015 SC	12/18/2029	81,804	12,358

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AD	P	2014-12-13	30	Wings REPEAT Option 3 Action 2: HFEC open hole inspection for any crack in the forward support fitting, the rear spar upper chord, and the rear spar web at the 12 fasteners holes (location 1-12). Figure 3,4,15, and 16.		24000	CYC					
AD	TER	2014-16-09	10	Oxygen REPEAT To prevent inadvertent electrical current from passing through an internal, anti-collapse spring of the low-pressure oxygen hose. Replacement of the low-pressure hoses in the flight compartment with non-conductive low-pressure oxygen hoses.				2118	9/22/2017			
AD	R1	2014-18-02	10	Fuselage REPEAT This AD supersedes AD 2014-05-02 To detect and correct corrosion or cracking of the aft pressure bulkhead, Inspection, Modification and Repair. BS1016		730	DAY	09017	3/7/2021	03/07/2023		611
AD	R1	2014-22-06	30	Fuselage REPEAT Detailed and Surface HFEC Inspection of the STA 291.5 Frame web around the stringer 16R intercoastal strap cutout part 2, Step 1 Supersedes AD 2005-07-12		4500	CYC	06309	7/22/2020 SC	10/15/2023	72,779	3,333
AD	R1	2014-22-06	40	Fuselage REPEAT Detailed Inspection of the STA 291.5 frame web from stringer 8R to 15R and a surface HFEC inspection of the STA 291.5 frame outer chord where the frame web is not visible. Part 2, step 2		9000	CYC	06611	9/27/2020	01/14/2027	77,529	8,083

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AD	P	2014-22-06	50	Fuselage REPEAT Detailed Inspection of the intercostals from stringer 8R to 16R Part 2 step 3		9000	CYC					
AD	P	2014-22-06	60	Fuselage REPEAT Surface HFEC Inspection of the inner chord of the door hinge intercostals Part 2 step 4		9000	CYC					
AD	P	2014-22-06	70	Fuselage REPEAT Surface HFEC Inspection of FWD Inner Radius Trimout of the intercoastal at stringer 14R for any crack. Part 2 step 5		9000	CYC					
AD	P	2014-22-06	80	Fuselage REPEAT Detailed Inspection of the STA 277 frame, frame shear ties and stringer clips from stringer 7R to stringer 17R for any crack. Part 2 step 6		9000	CYC					
AD	R1	2014-23-10	10	Fuselage REPEAT SUPERSEDES AD 2008-17-03 To detect and correct cracking which could prevent LH FWD Entry Door from sealing correctly [See also AD 2008-17-03]		4500	CYC	3733	11/20/2018 SC	10/26/2022	71,365	1,919
AD	R1	2014-25-03	10	Fuselage REPEAT FWD Galley Service Doorway Upper AFT Corner External LFEC and Detailed Inspection of skin and bearstrap		4500	CYC	4255	2/7/2019 SC	01/17/2023	71,694	2,248

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AD	R1	2014-25-03	40	Fuselage REPEAT FWD galley service doorway; Internal Detailed inspection of skin and bearstrap	18000	CYC		4255	2/7/2019 SC	04/14/2032	85,194	15,748
AD	NA2	2014-25-09	10	Fuselage INITIAL To Detect and Correct fatigue cracking in the Skin Assembly and Bearstrap of the AFT LWR of FWD airstair stowage doorway. HFEC and Detailed Inspection. [4500 Cyc from effective date of AD] [Repeat interval 6000 Cyc.]	4500	CYC						
AD	TER	2015-02-26	10	Fuselage ONE TIME Supersedes AD 2013-24-13 To Prevent seat detachment in an emergency landing; Replacement of seat track link assembly [60 mos. after effective date of AD 2013-24-13 - 1/7/14 compy with paragraph G and I)	1825	DAY		4061	1/5/2019 SC			
AD	R1	2015-13-05	10	Fuselage REPEAT External Detailed Inspection of the skin assembly and LFEC Inspection of skin assembly and bearstrap at FWD Entry Doorway Upper Corners	4500	CYC		4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	R1	2015-15-05	10	Fuselage REPEAT SUPERSEDES AD 98-22-10To detect and correct fatigue cracking of the AFT frame and frame support structure of the FWD service doorway around the six doorstep fittings. Compliance due 4500 CYC after effective date of AD - 8/28/15	4500	CYC		4090	1/26/2019 SC	01/09/2023	71,663	2,217
AD	NA1	2015-21-06	10	Fuselage INITIAL Lap Joint Modification (Repair) - Crown Areas	50000	CYC						

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Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20	H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	R1	2015-21-06	20	Fuselage INITIAL (LFEC Inspection on Areas outside the Crown Area) Before accumulation of 70,000 cyc. Thereafter repeat every 5000 cyc.	70000	CYC		DOM	1/1/2000 SC	11/19/2021	70,000	554
AD	NA1	2015-21-06	30	Fuselage REPEAT For airplanes that have (NACA modification) of the Crown lap joint stringers. Do External or Internal LFEC inspection specified in fig 8 and 9. Repeat 1500 cycles if External Inspection and 4500 cycles if Internal Inspection performed	12000	CYC						
AD	NA1	2015-21-06	40	Fuselage REPEAT For airplanes with (NACA modification) outside the crown areas done. Do External or Internal LFEC Inspection specified in fig 8 and 9. Repeat 1500 cycles for External or 4500 cycles Internal Inspection performed.	20000	CYC						
AD	NA1	2015-21-06	50	Fuselage ONE TIME After lap joint repair, modify the Tear Strap Splice Straps per fig 10,11, and 12	45000	CYC						
AD	P	2015-21-06	60	Fuselage INITIAL External or Internal LFEC Inspection of Lap Joint Repair	45000	CYC						
AD	P	2015-21-06	65	Fuselage REPEAT External or Internal LFEC Inspection iaw SB 737- 53A1177 Rev 6 or 7 Part 1.E.7.	2800	CYC						

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AD	TER	2015-21-06	80	Fuselage INITIAL (HFEC Inspections of Window Corners) Before accumulations of 50,000 cyc. Thereafter repeat at 4,500 cyc after lap joint repair.		50000	CYC					
AD	TER	2015-21-06	90	Fuselage REPEAT HFEC Inspections of Window Corners		4500	CYC					
AD	TER	2015-21-06	100	Fuselage ONE TIME Surface HFEC inspection at applicable window corner fastener holes and do a preventative modification as applicable IAW Part 4 of SB 737-53A1177 Rev.7. Inspection will terminate para (n). Preventative Mod will terminate repetitive inspection for corner fastener locations		60000	CYC					
AD	NA1	2015-21-06	110	Fuselage TERMINATING Replacement of the skin panel as specified in Part VIII or Part IX. Terminates repetitive inspections at the window corners in para (p) for windows common to the replaced panel only								
AD	NA1	2015-21-06	120	Fuselage TERMINATING Replacement of the skin panel as specified in Part VIII or Part IX. Terminates the lap joint modification required by para (g) for the S-10 and S-14 lap joints common to the replaced panel only								
AD	NA1	2015-21-06	130	Fuselage TERMINATING Replacement of the skin panels as specified in Part VIII or Part IX. Terminates the inspections required by para (g) and (i) of AD 2013-09-01 for the replaced								

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				skin panel only								
AD	TER	2015-21-08	10	Wings REPEAT Eddy Current Inspection of the Inspar Upper Skin common to the Rear Spar at WBL 157 and if installed, the repair parts		500	CYC	5366	2/27/2020			
AD	TER	2015-23-08	10	Fuselage ONE TIME Detailed and Eddy Current inspection to detect and correct missing fuselage frame straps and frame cracking. Inspections at Sta 328, 344, 360, Ref. AD para (j)		2200	CYC					
AD	TER	2016-03-01	20	Wings INITIAL WOF DI for cracks on the LH & RH wing front spar LWR chord and inspar skin between (WBL) 114 and (WBL) 183		5000	CYC	644-001	2/29/2016 SC			
	h		20			730	DAY	644-001	2/29/2016 SC			
AD	R1	2016-03-01	40	Wings REPEAT DI for cracks on the LH & RH wing front spar LWR chord and inspar skin between (WBL) 114 and (WBL) 183		5000	CYC	08231-4	1/12/2021	07/03/2024	73,829	4,383
AD	R1	2016-10-04	10	Fuselage REPEAT Inspection for cracks at the number 2 VHF antenna location. Action 1: repeat Part 1: External detailed inspection for cracks.		680	CYC	09017	3/7/2021	09/08/2021	69,710	264

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
Intital: 1250 cycles and repeat: 680 cycles												
AD	R1	2016-10-04	20	Fuselage REPEAT Action 2: Repeat Part 2: External detailed inspection, external HFEC surface inspection, and HFEC open hole inspection for cracks.	6000	CYC		673-003	6/20/2016 SC	11/29/2021	70,039	593
AD	R1	2016-10-04	30	Fuselage REPEAT Preventive modification due at 12000 flight cycles after the effective date of AD Part 4 of service bulletin constitutes as terminating action	12000	CYC		673-003	6/20/2016 SC	01/07/2026	76,039	6,593
AD	NA2	2016-13-11	10	Wings REPEAT To Detect and correct fatigue cracking of upper and lower chords and web of longitudinal floor beams supersedes 2008-05-06. HFEC and Detailed Inspection.	6000	CYC						
AD	g	2016-18-16	10	Fuel ONE TIME Install the Automatic Shutoff System for the Center and Auxiliary Fuel Tank Boost Pumps	1095	DAY		7270	10/31/2019 SC			
AD	l	2016-18-16	20	Fuel REPEAT For Model 737-100, -200, and -200C series airplanes without Boeing auxiliary fuel tanks installed: Incorporate AWL No. 28-AWL-21 of Section C. a year after accomplishment of the installation of Automatic Shutoff System.	365	DAY		06646-2	10/31/2020	10/31/2021		119

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Type	Status	Doc	Item	Description / Action Required	Position	Lim iter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	TER	2016-18-16	30	Fuel ONE TIME Installation of Un-commanded on (UCO) protection system	1825	DAY		09017	3/7/2021			
AD	R1	2016-18-16	40	Fuel REPEAT For Model 737-100, -200, and -200C series airplanes without Boeing auxiliary fuel tanks. Incorporate 28-AWL-24 of Section C 1 year after accomplishment of the UCO Protection System installation.	365	DAY		09017	3/7/2021 SC	03/07/2022		246
AD	R1	2017-02-02	30	Fuselage REPEAT This AD was prompted by an evaluation by the design approval holder indicating that th fuselage skin is subject to widespread fatigue damage, and reports of cracks at the chem milled steps in the fuselage skin. We are issuing this AD to detec and correct fatigue cracking of the fuselage skin panels, which could cause rapid decompression of the airplane (Zone 1)	3700	CYC		4560	5/21/2019 SC	07/02/2022	70,900	1,454
AD	PCW	2017-02-02	50	Fuselage REPEAT Bulletin Description: This AD was prompted by an evaluation by the design approval holder indicating that th fuselage skin is subject to widespread fatigue damage, and reports of cracks at the chem milled steps in the fuselage skin. We are issuing this AD to detec and correct fatigue cracking of the fuselage skin panels, which could cause rapid decompression of the airplane	68469	CYC		DOM	1/1/2000 SC			
AD	R1	2017-02-02	70	Fuselage REPEAT Bulletin Description: This AD was prompted by an	6000	CYC		698-013	5/20/2017 SC	05/30/2022	70,769	1,323

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
<p>evaluation by the design approval holder indicating that the fuselage skin is subject to widespread fatigue damage, and reports of cracks at the chem milled steps in the fuselage skin. We are issuing this AD to detect and correct fatigue cracking of the fuselage skin panels, which could cause rapid decompression of the airplane</p>												
AD	R1	2017-02-02	100	Fuselage REPEAT PART 1: Zone 1 Do an internal or external inspection for cracks. Replaces AD 2005-13-30	3700	CYC		4560	5/21/2019	07/02/2022	70,900	1,454
AD	R1	2017-02-02	110	Fuselage REPEAT Part 2: Zone 2 do a detailed inspection	4500	CYC		09017	3/7/2021	04/20/2024	73,530	4,084
AD	R1	2017-02-02	120	Fuselage ONE TIME Skin Panel Replacement terminates para (h)	6000	CYC		698-011	3/23/2017 SC	05/30/2022	70,768	1,322
AD	P	2017-02-02	130	Fuselage REPEAT Detailed Inspection of Zone 2	7500	CYC						
AD	R1	2017-02-05	10	Fuselage REPEAT Part 2 and 6: LFEC and Detailed Inspection of AFT Entry and Galley Doorway Corners	4500	CYC		09017	3/7/2021	04/20/2024	73,530	4,084
AD	TER	2017-04-08	20	Fuselage INITIAL Part 2: MFEC Inspection of Upper Frame to Side Frame Splice above Stringer S-14	5000	CYC		4560	5/21/2019			

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AD	R1	2017-04-08	30	Fuselage REPEAT Part 2: MFEC Inspection of Upper Frame to Side Frame Splice above Stringer S-14 (Repeat until Preventative mod is accomplished)	6000	CYC		4560	5/21/2019 SC	01/28/2024	73,200	3,754
AD	K	2017-04-08	50	Fuselage ONE TIME Part 4: Preventative Mod (terminates para (g))	4000	CYC		08226-4	1/4/2021			
AD	J	2017-04-08	70	Fuselage INITIAL Part 6: Post-Preventative Mod Inspection of Upper Frame to Side Frame Splice above Stringer S-14 (repeat 4500 cycles)	4500	CYC		08226-4	1/4/2021			
AD	J	2017-04-08	80	Fuselage REPEAT Part 6: Post-Preventative Mod Inspection of Upper Frame to Side Frame Splice above Stringer S-14	4500	CYC		08226-4	1/4/2021 SC	02/26/2024	73,315	3,869
AD	R1	2017-04-08	100	Fuselage SSID SSI's F-36, F-33C and F-34C	4500	CYC		09017	3/7/2021	04/20/2024	73,530	4,084
AD	(h)	2017-06-14	20	Horizontal stabilizer REPEAT 400 ONLY Cracks in horizontal stabilizer lower skin (configuration 2)	365	DAY						
AD	g	2017-11-01	10	Stabilizers REPEAT Detailed, HFEC, and Ultrasonic Inspection of Center section rear spar upper chord clevis lugs. Repeat 3000 cycles until para (m) complied with then repeat 4500 cycles. Terminates AD 84-23-05	3000	CYC		09017	3/7/2021			

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AD	TER	2017-11-01	20	Stabilizers INITIAL WOF Do HFEC and fluorescent dye penetrant inspections for cracking in the front and rear spar upper clevis lugs of the center section and front and rear spar upper lugs of the horizontal stabilizer. Repeat 18000 Horizontal Stabilizer Center Section flight cycles or 7 years WOF. Terminates AD 86-12-05		9000	CYC	09017	3/7/2021			
	m		20			1825	DAY	09017	3/7/2021			
AD	P	2017-11-01	30	Stabilizers REPEAT Part 2/2A: Detailed, HFEC, and Ultrasonic Inspection of Center Section Rear Spar Upper Chord Clevis Lugs and Horizontal Stabilizer Rear Upper Lugs		4500	CYC					
AD	R1	2017-11-01	40	Stabilizers REPEAT WOF Part 7/7A: HFEC and Fluorescent Dye Penetrant Inspection in the Front and Rear Spar Upper Clevis Lugs of the Center Section and Front and Rear Spar Upper Lugs of the Horizontal Stabilizer		18000	CYC	09017	3/7/2021	SC	07/17/2033	87,030
	m		40			2555	DAY	09017	3/7/2021	SC	03/05/2028	2,436
AD	P	2017-12-05	10	Fuselage INITIAL Part 1: Left side External Skin Detailed and HFEC Surface Inspection of STA 259.5 Circumferential Butt Splice		4500	CYC					

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AD	P	2017-12-05	20	Fuselage INITIAL Part 2: Right side External Skin Detailed and HFEC Surface Inspection of STA 259.5 Circumferential Butt Splice (repeat 6000 cycles)		66000	CYC	DOM	1/1/2000 SC			
AD	TER	2017-12-05	40	Fuselage INITIAL Left side External Skin Detailed and HFEC Surface Inspection of STA 259.5 Circumferential Butt Splice (Supersedes AD 2007-26-04) part 1 (repeat 6000 cycles)		68298	CYC	4560	5/21/2019			
AD	TER	2017-12-05	50	Fuselage INITIAL Bulletin Description: This AD was prompted by an evaluation by the design approval holder (DAH) indicating that certain skin panels are subject to widespread fatigue damage (WFD) (Supersedes AD 2007-26-04)		68354	CYC	4560	5/21/2019			
AD	R1	2017-12-05	90	Fuselage REPEAT Both Left and Right side External Skin Detailed and HFEC Surface Inspection of STA 259.5 Circumferential Butt Splice		6000	CYC	4560	5/21/2019 SC	01/28/2024	73,200	3,754
AD	TER	2017-12-05	100	Fuselage ONE TIME External Countersunk Rivet Inspection of Left and Right side STA 259.5 Circumferential Butt Splice		6000	CYC					
AD	P	2017-12-05	110	Fuselage INITIAL 400 ONL LH STA 259.5 Circumferential butt splice -inspection and repair, Ext countersunk rivet inspection LH and RH STA 259.5 Circumferential butt splice		46000	CYC					

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AD	TER	2017-13-13	30	Fuselage INITIAL Bulletin Description: To detect and correct frame web cracking at station (STA) 344 system penetration holes between stringer S-22L and stringer S-24.	69058	CYC		08493	2/18/2021			
AD	R1	2017-13-13	60	Fuselage REPEAT Part 1 and 2: HFEC, Detailed and General Visual for cracking of Section 41 Lower Frames between STA 268.25 and STA 360 for both left and right side	12000	CYC		09017	3/7/2021	06/08/2029	81,030	11,584
AD	R1	2017-13-13	70	Fuselage REPEAT Part 3: HFEC Inspection for cracking of the right side STA 312, STA 328, STA 344, section 41 lower lobe frames	4500	CYC		09017	3/7/2021	04/20/2024	73,530	4,084
AD	R1	2017-17-04	30	Wings INITIAL Bulletin Description: To detect and correct cracks in the upper AFT skin of the wings at fastener holes along the rear spar upper chord	69697	CYC		DOM	1/1/2000 SC	09/04/2021	69,697	251
AD	P	2017-17-04	60	Wings REPEAT Detailed Inspection of the Upper AFT Skin on the Left and Right Wing from Wing Buttock Line (WBL) 80 to WBL 155	4500	CYC						
AD	TER	2017-20-12	20	Fuselage, Nacelles/Py lons, Stabilizers, and Wings INITIAL Bulletin Description: To detect cracks in each structural significant item (SSI) and repair of any cracked structure.	69708	CYC		698-011	5/19/2017			

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AD	TER	2017-20-12	F-4 30	Fuselage, Nacelles/Pylons, Stabilizers, and Wings REPEAT F-4 SSID		24000	CYC	698-011	5/19/2017	SC			
AD	TER	2017-20-12	F-12 40	Fuselage, Nacelles/Pylons, Stabilizers, and Wings REPEAT F-12 SSID		24000	CYC	698-011	5/19/2017	SC			
AD	TER	2017-20-12	F-39 A 50	Fuselage, Nacelles/Pylons, Stabilizers, and Wings REPEAT F-39A SSID		24000	CYC	698-011	5/19/2017	SC			
AD	TER	2017-20-12	F-39 F 60	Fuselage, Nacelles/Pylons, Stabilizers, and Wings REPEAT F-39 F SSID		3000	CYC	698-011	5/19/2017				
AD	TER	2017-22-04	20	Fuselage INITIAL Bulletin Description: To detect and correct disbanded skin panels, which could result in fuselage skin cracking, rapid decompression and loss of structural integrity of the airplane		70019	CYC	09017	3/7/2021				
AD	R1	2017-22-04	40	Fuselage REPEAT External GV and Detailed Inspection of Fuselage Skin Panels		4500	CYC	09017	3/7/2021	SC	04/20/2024	73,530	4,084
AD	R1	2017-22-04	50	Fuselage REPEAT External High Freq Bond Test Inspection at S-4L and S-4R lap splice between BS 259.50 and BS 1016, S-10L and S-10R lap splice between BS 259.50 and BS 1016, S-25L and S-25R lap splice between BS 908		5000	CYC	09017	3/7/2021	SC	08/23/2024	74,030	4,584

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				and BS 1016								
AD	R1	2017-22-04	60	Fuselage REPEAT External High Freq Bond Test Inspection at S-14L and S-14R lap splice between BS 259.50 and BS 540, S-14L and S-14R lap splice between BS 727 and BS 1016, S-19L and S-19R lap splice between BS 259.50 and BS 540, S-20L and S-20R lap splice between BS 727 and BS 1016, S-24L and S-24R lap splice between BS 259.50 and BS 360, S-25L and S-25R lap splice between BS 727 and BS 908, S-26L and S-26R lap splice between BS 360 and BS 540	11000	CYC		09017	3/7/2021 SC	10/01/2028	80,030	10,584
AD	R1	2017-22-04	70	Fuselage REPEAT External High Freq Bond Test Inspection at BS360 butt joint between S-10L and S-10R, BS 908 butt joint between S-10L and S-10R	47000	CYC		09017	3/7/2021 SC	05/23/2053	116,030	46,584
AD	R1	2017-22-04	80	Fuselage REPEAT External High Freq Bond Test Inspection at BS 540 butt joint between S-10L and S-10R, BS 727 butt joint between S-10L and S-10R	22000	CYC		09017	3/7/2021 SC	04/12/2036	91,030	21,584
AD	R1	2017-23-02	20	Fuselage REPEAT Part 1: Crown Skin Chem-mill Step Internal/External NDI for any crack on the production skin panel	1800	CYC		5371	3/3/2020 SC	09/13/2021	69,732	286
AD	R1	2017-23-02	30	Fuselage ONE TIME Crown Skin Panel Replacement from S-10L to S-10R - STA 360 to STA 540 and STA 727 to STA 908	66000	CYC						NO START

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Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20	H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Lim iter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	R1	2017-23-02	40	Fuselage ONE TIME Crown Skin Panel Replacement f rom S-10L to S-10R - STA 360 to STA 540 and STA 727 to STA 908	6000	CYC		2443	12/5/2017	12/07/2022	71,532	2,086
AD	R1	2017-23-02	60	Fuselage 810 REPEAT ONLY Castle 8100-9 reference ID 201903200017-0008 Fuselage upper lobe skin and stringers, fuselage circumferential splices, fuselage lap splices	70769	CYC		DOM	1/1/2000 SC	05/30/2022	70,769	1,323
AD	TER	2017-23-05	20	Wings INITIAL Detailed Inspection of the Upper Wing AFT Skin from WBL 159 to 220	70098	CYC		09017	3/7/2021			
AD	R1	2017-23-05	40	Wings REPEAT Detailed Inspection of the Upper Wing AFT Skin from WBL 159 to 220	4500	CYC		09017	3/7/2021 SC	04/20/2024	73,530	4,084
AD	TER	2017-23-07	30	Wings INITIAL Bulletin Description: To detect and correct cracking of the lower chord of the rear spar and the lower AFT skin at WBL 157	70098	CYC		09017	3/7/2021			
AD	R1	2017-23-07	40	Wings REPEAT LFEC Inspection of the Lower Chord of the Rear Spar and DTL Inspection of the Lower AFT Skin at WBL 157	4500	CYC		09017	3/7/2021 SC	04/20/2024	73,530	4,084
AD	TER	2017-24-05	30	Wings INITIAL Surface HFEC Inspection of the Upper AFT Skin from WBL 159 to WBL 220 in areas without an External Repair	70135	CYC		09017	3/7/2021			

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Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

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AD	R1	2017-24-05	40	Wings REPEAT Surface HFEC Inspection of the Upper AFT Skin from WBL 159 to WBL 220 in areas without an External Repair	4500	CYC		09017	3/7/2021 SC	04/20/2024	73,530	4,084
AD	TER	2017-25-10	10	Stabilizers INITIAL Inspections and checks of the elevator and horizontal stabilizer at elevator hinge #4 and 6.	821	DAY		4560	5/21/2019 SC			
AD	R1	2017-25-10	20	Stabilizers REPEAT Part 2 and 3: HFEC Inspection of Elevator Hinge #4 and 6 and loose bolt check.	4500	CYC		4560	5/21/2019	01/18/2023	71,700	2,254
AD	R1	2017-25-10	25	Stabilizers REPEAT Part 4 and 5: HFEC and LFEC Inspection of Horizontal Stabilizer at Elevator Hinge #4 and 6 for any cracks and loose bolt check	4500	CYC		4560	5/21/2019 SC	01/18/2023	71,700	2,254
AD	R1	2017-25-10	30	Stabilizers REPEAT TABLE 5 WORK PACKAGE 3. DVI OF THE HORIZONTAL STABILIZER REAR SPAR, GUSSET PLATE, AND INSPAR SKIN FWD OF ELEVATOR HINGE #4 AND 6	730	DAY		08629-2	5/26/2021	05/26/2023		691
AD	R1	2017-25-10	40	Stabilizers REPEAT WOF TABLE 6: GVI of the Elevator Front Spar around Hinge #4 and 6 for any ply damage	4500	CYC		08232-4	1/13/2021	03/01/2024	73,333	3,887
			40		1095	DAY		08232-4	1/13/2021	01/13/2024		923

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AD	R1	2017-25-10	50	Stabilizers REPEAT WOF Table 7: Tap Test inspection of Elevator Skin for any delamination	4500	CYC		4560	5/21/2019 SC	01/18/2023	71,700	2,254
			50		1095	DAY		4560	5/21/2019 SC	05/20/2022		320
AD	R1	2017-25-10	60	Stabilizers REPEAT HFEC & LFEC and loose bolt check at horizontal stabilizer (FWD).	4500	CYC		4560	5/21/2019 SC	01/18/2023	71,700	2,254
AD	R1	2017-25-10	85	Stabilizers REPEAT Table 8: Elevator Hinge #4 and 6 Bearing Plate Replacement	18000	CYC		2764	1/18/2018 SC	03/29/2031	83,669	14,223
AD	P	2017-25-10	105	Stabilizers REPEAT Paragragh H8: Disconnection of elevator hinge No. 4 or No. 6	18000	CYC						
AD	R1	2017-25-12	10	Fuselage REPEAT detect and correct cracking in the webs of the stub beams at certain fuselage stations, which, if not corrected, could result in the loss of structural integrity of the air frame during flight, collapse of the main landing gear, and failure of the pressure deck.	9000	CYC		2776	1/23/2018 SC	02/03/2025	74,689	5,243
AD	NA3	2018-01-08	10	Fuselage REPEAT Table 1 Action 1: Detailed inspection for cracks in S-18A outboard with a thickness of 0.33 inches or less 306AL 326TR and 810TA N/A due to paragraph H3 of AD 2018-01-08 preventative modification was not	5000	CYC						

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				installed on Aircraft. 737CS and 809TA N/A due to variable number.								
AD	NA3	2018-01-08	737-53A1166	20 Fuselage REPEAT Table 2 Action 1: ultrasonic and surface HFEC inspection for cracks in STA 727 f frame outboard chord		5000	CYC					
AD	NA3	2018-01-08	737-53A1166	30 Fuselage INITIAL Table 1 action 2: Eddy current inspection for cracks in S-18A. Repeat is 15000 cycles		20000	CYC					
AD	NA3	2018-01-08	737-53A1166	40 Fuselage INITIAL Table 2 action 2: Open hole Eddy current inspection for cracks in STA 727 f frame outboard. Repeat 15000 cycles		10000	CYC					
AD	NA3	2018-01-08	737-53A1166	50 Fuselage INITIAL Table 5 action 1 and 2: DVI and Eddy Current for cracks in STA 727 aux chord. Repeat 5000 cycles		5000	CYC					
AD	R1	2018-04-06	(j)	40 Fuselage REPEAT (GRP 2) Part 3: Detailed inspection of S-7L to S-16L intercostals from BS 294.5 to BS 303.9 and S-8L, S-10L, S-12L, S-14L, S-15L and S-16L intercostals from BS 348.2 to BS 360 (FWD of the FWD Entry Door)		6000	CYC	221-001	2/2/2015 SC	10/01/2021	69,804	358
AD	R1	2018-04-06	(k)	50 Fuselage REPEAT (GRP 2) Part 3: Detailed inspection of S-7L to S-16L		6000	CYC	221-001	2/2/2015 SC	10/01/2021	69,804	358

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				intercostals from BS 294.5 to BS 303.9 and S-8L, S-10L, S-12L, S-14L, S-15L and S-16L intercostals from BS 348.2 to BS 360 (AFT of the FWD Entry Door)								
AD	R1	2018-04-06	60	Fuselage REPEAT (GRP 2) Detailed and HFEC inspections for cracking of the S-15L aft intercostal between BS 348.2 and BS 360, and do a detailed inspection of the cargo barrier net fitting at the intercostal IAW Fig. 3	6000	CYC		221-001	2/2/2015 SC	10/01/2021	69,804	358
AD	NA4	2018-04-06	70	Fuselage INITIAL (GRP 1) STA 348.2 Frame Inspection IAW Paragraph 3.B., Work Instructions, PART 1.	2250	CYC						
AD	NA4	2018-04-06	80	Fuselage REPEAT (GRP 1) STA 348.2 Frame Inspection IAW Paragraph 3.B., Work Instructions, PART 1.	6000	CYC						
AD	R1	2018-04-06	90	Fuselage INITIAL (GRP 3) Eddy current inspections for cracking of the frame, reinforcement angle, and shear web and DVI for missing fasteners and any crack in the STA 348.2 frame outer chord, inner chord and reinforcement angle IAW Paragraph 3.B., Work Instructions, Step 3.B.4.a and b. Table 3	4500	CYC		3163	3/30/2018 SC	02/22/2022	70,379	933
AD	P	2018-04-06	100	Fuselage REPEAT (GRP 3) Eddy current inspections for cracking of the frame,	6000	CYC						

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				reinforcement angle, and shear web and DVI for missing fasteners and any crack in the STA 348.2 frame outer chord, inner chord and reinforcement angle IAW Paragraph 3.B., Work Instructions, Step 3.B.4.a and b. Table 3								
AD	P	2018-04-06	(p)	110	Fuselage REPEAT (GRP 3) Part 3: Inspection of the STA 351.2 Frame Inner Chord Common to the Shear Web at S-17L for any crack (Alternate SSI Inspection for F-13B or F-14B)	6000	CYC					
AD	P	2018-04-08	H (1)	10	Fuselage REPEAT PART 1 DETAILED HFEC AND LFEC INSPECTIONS OF THE GORE WEBS IN ZONE 1 FOR ANY CRACKS 6000 CYCLES AFTER THE COMPLIANCE OF LAST INSPECTION OR 4500 AFTER THE EFFECTIVE PART OF AD. WOF. Terminates AD 2012-18-13 para (o)	6000	CYC					
AD	R1	2018-04-08	H (2)	30	Fuselage INITIAL PART 2: DETAILED HFEC AND LFEC INSPECTIONS OF THE GORE WEB LAP SPICES IN ZONE 2 FOR ANY CRACK. Initial - 4500 CYCLES AFTER THE EFFECTIVE PART OF THE AD OR 54000 CYCLES. WOL. Repeat 6000 CYCLES	4500	CYC	3178	4/11/2018 SC	02/28/2022	70,405	959
AD	P	2018-04-08	H (2)	40	Fuselage REPEAT PART 2: DETAILED HFEC AND LFEC INSPECTIONS OF THE GORE WEB LAP SPLICES IN ZONE 2 FOR ANY CRACK	6000	CYC					

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AD	R1	2018-04-12	10	Fuel prevent electrical arcing between the fuel boost pump power cable wiring and the surrounding conduit, which could lead to arc-through of the conduit, consequent fire or explosion of the fuel tank, and subsequent loss of the airplane. para G	REPEAT	1825	DAY	3177	4/8/2018 SC	04/07/2023		642
AD	R1	2018-09-01	10	Wings INITIAL Open Hole HFEC Inspection of MLG Beam FWD Support Fitting at Fastener Holes without a Freeze Plug installed		4500	CYC	3265	6/8/2018 SC	04/21/2022	70,611	1,165
AD	P	2018-09-01	20	Wings REPEAT Open Hole HFEC Inspection of MLG Beam FWD Support Fitting at Fastener Holes without a Freeze Plug installed		15000	CYC					
AD	R1	2018-09-08	10	Fuselage INITIAL Detect and correct cracks in the frame web common to the stinger ties adjacent to the air conditioning support brackets		9000	CYC	3269	6/13/2018 SC	05/22/2025	75,121	5,675
AD	R1	2018-09-13	10	Wings INITIAL Do high frequency eddy current (HFEC) open hole inspections for any cracking in the forward support fitting, the aft support fitting, the rear spar upper. TERMINATES AD 2015-21-08 and 2014-12-13		4500	CYC	3276	6/19/2018 SC	04/28/2022	70,639	1,193
AD	P	2018-09-13	20	Wings REPEAT Do high frequency eddy current (HFEC) open hole inspections for any cracking in the forward support fitting, the aft support fitting, the rear spar upper		15000	CYC					

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
Please see Damage tolerance evaluation 12103-R060												
AD	R1	2018-10-12	10	Wings, Fuselage REPEAT Surface HFEC and detailed inspections of the vertical stiffeners of the LBL 6.15 and RBL 6.15	500	CYC		08693-4	6/22/2021	10/18/2021	69,873	427
AD	R1	2018-12-05	10	Wings REPEAT Option 1 per DVI figure 1, 2, 5, 6	3000	CYC		4266	6/3/2019 SC	01/10/2022	70,208	762
AD	P	2018-12-05	20	Wings REPEAT Option 2 HFEC inspection of area 1 and 2. Must be done 3000 cycles after the accomplishment of option 1. After option is accomplished, repeat interval not to exceed 4500 cycles.	3000	CYC						
AD	P	2018-12-05	30	Wings REPEAT Option 2 HFEC inspection of area 1 and 2. Must be done 3000 cycles after the accomplishment of option 1. After option is accomplished, repeat interval not to exceed 4500 cycles.	4500	CYC						
AD	R1	2018-16-09	10	Fuselage REPEAT DO HFEC INSPECTOIN OF THE FUSELAGE FRAME WEBS AT STA 616-639 BETWEEN S-7 AND S-8 LEFT AND RIGHT	6000	CYC		5919	9/12/2018 SC	08/15/2023	72,537	3,091
AD	R1	2018-17-23	10	Fueslage REPEAT Do a HIGH FREQUENCY EDDY CURRENT (HFEC) INSPECTION OF THE WINDSHIELD NO 3,	6000	CYC		3606	10/8/2018 SC	09/12/2023	72,648	3,202

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				AFT SILL WEB AT STA 254.6, BETWEEN S-9 AND S-11 ON THE LEFT AND RIGHT SIDE for any crack.								
AD	NA2	2018-18-03	10	Fuselage REPEAT Forward Airstair Door Cutout - Station (STA) 312 Floor Beam Lower Chord, Upper Sill Structure, and Door Stop Fittings No. 2, No. 5 and No. 8 - Initial High Frequency Eddy Current (HFEC) and Detailed Inspection	4500	CYC						
AD	R1	2018-18-13	10	Wings REPEAT Detailed Inspections and NDT Inspections of the Outboard Flap Tracks at Positions 2 and 7	1460	DAY		4288	6/26/2019 SC	06/25/2023		721
AD	R1	2018-18-13	20	Wings REPEAT Bulletin Description: Detailed Inspections and NDT Inspections of the Outboard Flap Tracks, 1, 8, Overhaul	1460	DAY		7262	8/31/2019 SC	08/30/2023		787
AD	R1	2018-18-13	30	Wings REPEAT Overhaul of Outboard Flap Tracks, 1, 8	2920	DAY		7130	8/7/2019 SC	08/05/2027		2,223
AD	R1	2018-18-13	40	Wings REPEAT Outboard Flap Tracks 2, 7 Overhaul	2920	DAY		713-007	11/29/2016 SC	11/27/2024		1,242
AD	NA2	2018-19-20	10	Fuselage INITIAL Do a detailed or HFEC inspection for cracking of the body station (BS) 616 and 639 frames and stub beams	4500	CYC						

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AD	NA2	2018-19-20	20	Fuselage REPEAT Option 1: Detailed inspection of BS 616 and BS 639 frames and the stub beam, for any crack	4500	CYC						
AD	NA2	2018-19-20	30	Fuselage REPEAT Option 2: HFEC inspection of the BS 616 and BS 639 frames and the stub beam, for any crack	9000	CYC						
AD	NA2	2018-19-20	40	Fuselage INITIAL Open hole HFEC and HFEC Surface inspection for any crack in the BL 45.5 longitudinal floor beam web	9000	CYC						
AD	NA2	2018-19-20	50	Fuselage REPEAT Open hole HFEC inspection for any crack in the BL 45.5 longitudinal floor beam web at each fastener hole common to the stub beam attachment angle	29000	CYC						
AD	NA2	2018-19-20	60	Fuselage REPEAT HFEC Surface inspection for any crack in the BL 45.5 longitudinal floor beam web, around the fastener head/tail at each fastener location common to the back-up strap	9000	CYC						
AD	R1	2018-21-08	10	Fuselage REPEAT Bulletin Description: Do High Frequency Eddy Current (HFEC) and Low Frequency Eddy Current (LFEC) inspections of the STA 259.5 bulkhead frame web from the first stiffener above S-10 to S-13 left and right side for any crack. (a)	6000	CYC		3735	11/21/2018 SC	11/07/2023	72,871	3,425

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AD	NA4	2018-23-12	1	Bulletin Description: Within 42 months after the effective date of this AD, do an inspection to determine if any of the following conditions exist: Per SB 737-103-25-50 provided by air cruisers, our fleet is N/A		1277	DAY					
AD	R1	2019-01-06	10	Fuselage INITIAL Surface HFEC Inspection of the STA 303.9 Frame Web and Doubler at the stop fitting at S-16L for any crack.		4500	CYC	4266	6/3/2019 SC	01/20/2023	71,708	2,262
AD	P	2019-01-06	30	Fuselage REPEAT Surface HFEC Inspection of the STA 303.9 Frame Web and Doubler at the stop fitting at S-16L for any crack. Repeat interval after the initial set at 4500 cycles after the effective date of the AD.		12000	CYC					
AD	R1	2019-08-02	10	Fuselage INITIAL Part 2: Detailed and General Visual Inspection of the Fuselage Upper Frame, Side Frame and Fail-Safe Chord between Stringer S-10 and S-17, from STA 277 to STA 294.5, STA 360 to STA 520, STA 727A to STA 1006, Left and Right side		4500	CYC	4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	20	Fuselage REPEAT Part 2: Detailed and General Visual Inspection of the Fuselage Upper Frame, Side Frame and Fail-Safe Chord between Stringer S-10 and S-17, from STA 277 to STA 294.5, STA 360 to STA 520, STA 727A to STA 1006, Left and Right side		9000	CYC					
AD	P	2019-08-02	40	Fuselage REPEAT Part 3: HFEC Surface Inspection of the Fuselage		9000	CYC					

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Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20		H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				Frame Web Stringer Cutout at STA 277 and from STA 360 to STA 500, Left and Right side								
AD	R1	2019-08-02	50	Fuselage INITIAL Part 4: Eddy Current Inspection of the Fuselage Frame Web at Stringer Clip Attachments, Reinforcement Doubler and Fastener Attachments, Fail-Safe Chord at Stringer Flange Attachments and Detailed Inspection of Stringer Flange at Fail-Safe Chord Attachments from STA 360 to STA 500, Left and Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	60	Fuselage REPEAT Part 4: Eddy Current Inspection of the Fuselage Frame Web at Stringer Clip Attachments, Reinforcement Doubler and Fastener Attachments, Fail-Safe Chord at Stringer Flange Attachments and Detailed Inspection of Stringer Flange at Fail-Safe Chord Attachments from STA 360 to STA 500, Left and Right side	9000	CYC						
AD	R1	2019-08-02	70	Fuselage INITIAL Part 5: General Visual Inspection for Open Tooling Hole in the Fuselage Frame Web between Stringer S-14 and S-15, from STA 360 to STA 520, Left and Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	80	Fuselage REPEAT Part 6: Open Hole HFEC Inspection of the Fuselage Frame Web at the affected Open Tooling Hole between Stringer S-14 and S-15, Left and Right side. If tooling is open and inspection is accomplished, do	6000	CYC						

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Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				initial part 6 and terminate part 5								
AD	R1	2019-08-02	90	Fuselage INITIAL Part 10: HFEC Surface Inspection of the Fuselage Frame Web Stringer Cutout from STA 727A to STA 747, Left and Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	100	Fuselage REPEAT Part 10: HFEC Surface Inspection of the Fuselage Frame Web Stringer Cutout from STA 727A to STA 747, Left and Right side	9000	CYC						
AD	R1	2019-08-02	110	Fuselage INITIAL (Grp 3) Part 11: Detailed Inspection of the Fuselage Frame Integral Inboard Chord and Web from STA 360 to STA 400, Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	120	Fuselage REPEAT (Grp 3) Part 11: Detailed Inspection of the Fuselage Frame Integral Inboard Chord and Web from STA 360 to STA 400, Right side	9000	CYC						
AD	R1	2019-08-02	130	Fuselage INITIAL Part 12: HFEC Surface Inspection of the Fuselage Frame Integral Inboard Chord at the Mop-Sill Attachment from STA 727A to 727B, Left and Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	140	Fuselage REPEAT Part 12: HFEC Surface Inspection of the Fuselage	9000	CYC						

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Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				Frame Integral Inboard Chord at the Mop-Sill Attachment from STA 727A to 727B, Left and Right side								
AD	R1	2019-08-02	150	INITIAL Part 13: Eddy Current Inspection of the Fuselage Frame Web at the Crease Beam Clip from STA 867 to STA 907, Left and Right side	4500	CYC		4272	6/13/2019 SC	01/24/2023	71,722	2,276
AD	P	2019-08-02	160	REPEAT Part 13: Eddy Current Inspection of the Fuselage Frame Web at the Crease Beam Clip from STA 867 to STA 907, Left and Right side	6000	CYC						
AD	R1	2019-08-03	10	INITIAL Table 2 PART 2: Detailed and General Visual Inspection of the Section 43 lower lobe frames from STA 380 to STA 520, left and right side for any crack.	4500	CYC		4280	6/20/2019 SC	01/27/2023	71,734	2,288
AD	P	2019-08-03	20	REPEAT Part 2: Detailed and General Visual Inspection of the Section 43 Lower Lobe Frames from STA 380 to STA 520, Left and Right side	9000	CYC						
AD	P	2019-08-03	30	REPEAT Part 3: HFEC Inspection of Section 43 Lower Lobe Frames from STA 380 to STA 520, Left and Right side (not required if repaired IAW 737-100/200 SRM 53-10-4, Fig 7, 8, 9, or 11)	9000	CYC						

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Type	Status	Doc	Item	Description / Action Required	Position	Lim iter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	NA2	2019-08-03	40	Fuselage REPEAT Part 4: HFEC Inspection of Lower Lobe Frame Integral Inboard Chord and Web at STA 380, Left and Right side	4500	CYC						
AD	R1	2019-08-03	50	Fuselage INITIAL Table 3: HFEC of the sectoin 43 lower lobe frames from STA 380 t0 STA 520 left and right side. Do a general visual inspection to determine if there are any repairs in the part 3 eddy current inspection zones	4500	CYC		D10244	5/18/2019	01/17/2023	71,697	2,251
AD	TER	2019-09-01	10	Ice and rain protection ONE TIME Perform modification	730	DAY		09017	3/7/2021			
AD	R1	2019-09-01	20	Ice and rain protection REPEAT Pitot static and anti icing inspection. 11000 hours after the installation of the J18 junction box.	11000	HRS		09017	3/7/2021 SC	04/06/2036	83,652.10	10,780.90
AD	TER	2019-10-03	10	Fuselage INITIAL Detailed Inspection of the Left and Right Side-Strut Support Fitting Lugs at BS 685 with Bushings installed for any corrosion, crack or severed lug	365	DAY		06189	7/2/2020			
AD	R1	2019-10-03	20	Fuselage REPEAT WOF Part 17 and 18: Detailed and HFEC Inspections of Left and Right Side-Strut Support Fitting Lugs at BS 685 for any corrosion, crack or severed lug	4500	CYC		09017	3/7/2021	04/20/2024	73,530	4,084
			20		730	DAY		09017	3/7/2021	03/07/2023		611

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Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	R1	2019-10-03	40	Fuselage REPEAT GROUP 1 WOF 326 and 810 only: Part 3D and 11D: Detailed and HFEC Inspection of the Left and Right Side-Strut Support Fitting Lugs at BS 685 with bushings removed for any corrosion or cracks (terminates Part 17 and 18 once accomplished)	15000	CYC		7101	7/10/2019 SC	04/07/2030	82,244	12,798
			40			3650	DAY	7101	7/10/2019 SC	07/07/2029		2,925
AD	R1	2019-13-02	10	Fuselage ONE TIME Detailed inspection for any scribe lines. Install new lower love skin panel assembly in area from STA 360 to STA 540, S-14 to S-19 on left and right side	6000	CYC		7252	8/23/2019 SC	03/04/2024	73,345	3,899
AD	R1	2019-14-02	10	Hydraulic power ONE TIME Do an on-airplane inspection for the part numbers. Replace any Parker P/N H61C0552M1 pressure module check valves with Crissair P/N 1C4196 and do a hydraulic system A check valve installation test. Replace with a Crissair P/N 1C4196 check valve if Parker Pressure Module Check Valves P/N H61C0552M1 installed. In 5,360 flight hours after the initial inspection.	5360	HRS		7116	7/25/2019 SC	05/03/2027	77,129.30	4,258.10
AD	R1	2019-14-15	10	Fuselage INITIAL HFEC and Detailed inspection for any crack in the STA 685 and STA 706 stub beams, STA 695 Stub Beam Webs	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-14-15	20	Fuselage REPEAT HFEC and Detailed inspection for any crack in the	4500	CYC						

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Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
				STA 685 and STA 706 stub beams, STA 695 Stub Beam Webs								
AD	P	2019-14-15	30	Fuselage REPEAT GV Inspection for any existing repair and HFEC Inspection for any crack in the STA 685 and STA 706 Stub Beam Webs in Repaired Areas	9000	CYC						
AD	R1	2019-15-07	10	Fuselage INITIAL PART 2: Detailed and General Visual inspections of the Lower Lobe Frames from STA 727A to STA 1006, from S-17L to S-17R for any crack.	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	20	Fuselage REPEAT PART 2: Detailed and General Visual inspections of the Lower Lobe Frames from STA 727A to STA 1006, from S-17L to S-17R for any crack.	8000	CYC						
AD	R1	2019-15-07	30	Fuselage INITIAL Part 3: General Visual and Eddy Current Inspection of the Section 46 Lower Lobe Frames from STA 727A to STA 807 Left side and from STA 727A to STA 887 Right side	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	40	Fuselage REPEAT Part 3: General Visual and Eddy Current Inspection of the Section 46 Lower Lobe Frames from STA 727A to STA 807 Left side and from STA 727A to STA 887 Right side	8000	CYC						

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AD	R1	2019-15-07	50	Fuselage INITIAL Part 4: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 827 to STA 907 Left side	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	60	Fuselage REPEAT Part 4: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 827 to STA 907 Left side	8000	CYC						
AD	R1	2019-15-07	70	Fuselage INITIAL Part 5: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 907 to STA 1006 Left and Right side	4500	CYC		D10184	9/19/2019	02/27/2023	71,861	2,415
AD	P	2019-15-07	80	Fuselage REPEAT Part 5: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 907 to STA 1006 Left and Right side	8000	CYC						
AD	NA2	2019-15-07	90	Fuselage INITIAL Part 6: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 727B to STA 747 Left and Right side (Grp 10-18)	4500	CYC						
AD	NA2	2019-15-07	100	Fuselage REPEAT Part 6: General Visual and HFEC Inspection of the Section 46 Lower Lobe Frames from STA 727B to STA 747 Left and Right side	8000	CYC						

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AD	R1	2019-15-07	110	Fuselage INITIAL Part 7: General Visual and Open Hole HFEC Inspection of the Section 46 Lower Lobe Frames Integral Inboard Chord Right side	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	120	Fuselage REPEAT Part 7: General Visual and Open Hole HFEC Inspection of the Section 46 Lower Lobe Frames Integral Inboard Chord Right side	8000	CYC						
AD	R1	2019-15-07	130	Fuselage INITIAL Part 8: General Visual and HFEC Inspection of the Frame Web at Section 46 Water Tank Bracket Lower Lobe Frames at STA 947.5, S-24R thru S-26R	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	140	Fuselage REPEAT Part 8: General Visual and HFEC Inspection of the Frame Web at Section 46 Water Tank Bracket Lower Lobe Frames at STA 947.5, S-24R thru S-26R	8000	CYC						
AD	R1	2019-15-07	160	Fuselage INITIAL Part 9: General Visual and HFEC Inspection of the Frame Web at Section 46 Lower Lobe Frames from STA 767 to STA 907, from S-18 to S-21 Left and Right side	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	170	Fuselage REPEAT Part 9: General Visual and HFEC Inspection of the Frame Web at Section 46 Lower Lobe Frames from STA 767 to STA 907, from S-18 to S-21 Left and Right side	7600	CYC						

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AD	R1	2019-15-07	180	Fuselage INITIAL Part 10/11: General Visual and Eddy Current Inspections of the Frame at STA 947.5, S-21 Left and Right side	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	190	Fuselage REPEAT (Option 1) Part 10: Open Hole HFEC and HFEC Surface Inspection of the Frame at STA 947.5, S-21 Left and Right Side	14000	CYC						
AD	P	2019-15-07	200	Fuselage REPEAT (Option 2) Part 11: LFEC and HFEC Surface Inspection of the Frame at STA 947.5, S-21 Left and Right Side	3000	CYC						
AD	R1	2019-15-07	210	Fuselage INITIAL Part 12: Open Hole HFEC Inspection of the Frame Web Tooling Insulation Attachment and Coordination Holes	4500	CYC		D10184	9/19/2019 SC	02/27/2023	71,861	2,415
AD	P	2019-15-07	220	Fuselage REPEAT Part 12: Open Hole HFEC Inspection of the Frame Web Tooling Insulation Attachment and Coordination Holes	8000	CYC						
AD	P	2019-15-07	230	Fuselage TERMINATING Part 13: Preventative Modification of the Frame Web Tooling Hole and Insulation Attachment Hole in the Section 46 Lower Lobe Frame (terminates repetitive Open Hole HFEC inspection of para (h) for modified tooling hole or insulation attachment hole location only)								

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AD	P	2019-15-07	(h)	240 Fuselage INITIAL 400 ONL Table 10: Cracks in the frames below the passenger floor, Table 11: Cracks in the frames below the passenger floor, Table 2: Cracks in the frames below the passenger floor, Table 3: Cracks in the frame below the passenger floor, Table 4: Cracks in the frames below passenger floor, Table 5: Cracks in the frame below the passenger floor, Table 6: Cracks in the frame below the passenger floor, Table 7: Cracks in the frames below the passenger floor, Table 8: Cracks in the frames below the passenger floor	30000	CYC						
AD	R1	2019-16-12	g	10 Wings ONE TIME Stiff eners replacements at LBL 6.15 RBL 6.15. Installation of angle and bonding jumpers given in SB. Stiff ener replacement terminates inspection requirements IAW 737-57A1339 AD 2018-10-12	4500	CYC		7274	11/3/2019 SC	03/07/2023	71,890	2,444
AD	TER	2019-23-14	28-AWL-01	10 Nitrogen Generation System INITIAL Detailed Inspection of External Wires over Center Fuel Tank (Done at every 5C Check)	365	DAY		D10218	12/16/2014			
AD	R1	2019-23-14	28-AWL-01	20 Nitrogen Generation System 28-AWL-01 Detailed Inspection of External Wires over Center Fuel Tank (Done at every 5C Check)	3650	DAY		D10218	12/16/2014 SC	12/13/2024		1,258
AD	TER	2019-23-14	28-AWL-29	30 Nitrogen Generation System INITIAL AC Fuel Boost Pump Installation	365	DAY		08245-4	1/21/2021			
AD	R1	2019-23-14	28-AWL-29	40 Nitrogen Generation System 28-AWL-29 AC Fuel Boost Pump Installation - Perform an	2190	DAY		08245-4	1/21/2021 SC	01/20/2027		2,026

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				electrical bonding resistance check								
AD	R1	2019-23-16	10	Wings REPEAT Surface HFEC inspection of the left and right upper wing skin at WBL 157, between stringer 4 and stringer 5. General Visual Inspection of the left and right upper wing skin in the adjacent rib bay areas for any crack		2000	CYC	09017	3/7/2021	08/04/2022	71,030	1,584
AD	g	NA2 2020-02-16	10	Fuselage INITIAL External Detailed Inspection, Dual Frequency Eddy Current and Open Hole High Frequency Eddy Current Inspections of S-14L Lap Splice Skin (Left) and S-14R Lap Splice Skin (Right)		53999	CYC					
AD	g	NA2 2020-02-16	20	Fuselage INITIAL External Detailed Inspection, Dual Frequency Eddy Current and Open Hole High Frequency Eddy Current Inspections of S-14L Lap Splice Skin (Left) and S-14R Lap Splice Skin (Right)		42000	CYC					
AD	g	NA2 2020-02-16	30	Fuselage REPEAT External Detailed Inspection, Dual Frequency Eddy Current and Open Hole High Frequency Eddy Current Inspections of S-14L Lap Splice Skin (Left) and S-14R Lap Splice Skin (Right)		2000	CYC					
AD	P	2020-10-11 1	10	Left and right side HFEC in the radius of STA 540 bulkhead splice angle INITIAL Left and right side HFEC in the radius of STA 540 bulkhead splice angle. Initial is 5000 cycles after the		5000	CYC					

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A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20	H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining	
				effective date of the AD. Repeat inspection is 12000 cycles.									
AD	P	2020-10-11	1	Left and right side HFEC in the radius of STA 540 bulkhead splice angle REPEAT Left and right side HFEC in the radius of STA 540 bulkhead splice angle.	20	12000	CYC						
AD	NA4	2020-18-51		Attitude and Direction Data System ONE TIME Revise AFM for airplane to prohibit operation under IFR or night VFR and prohibit coupling the autopilot with an affected attitude indicator	10								
AD	R1	2020-26-04		Fuselage REPEAT Part 1, 3, 4, 5, 6 External Detailed and NDT Inspection	20	1800	CYC	09017	3/7/2021	06/15/2022	70,830	1,384	
AD	R1	2020-26-09		Wings INITIAL Inspection of the Center Wing Box, STA 663.75 Rear Spar, Lower Skin and Lower Chord 810TA will not be accomplished due to skin replacement	20	3000	CYC	08460-2	1/27/2021	SC	03/10/2023	71,902	2,456
AD	P	2020-26-09		Wings REPEAT (OPT 1) Option 1: External Surface HFEC and Ultrasonic Inspection of the Center Wing Box, STA 663.75 Rear Spar, Lower Skin and Lower Chord	30	9000	CYC						
AD	NA1	2021-01-04		Doors	10								

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Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20		H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446		C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	NA2	2021-07-02	10	Ice and Rain Protection AD 2019-09-01		11000	HRS					
AD	NA4	2021-08-04	10	Oxygen General visual inspection to determine if the serial number of the crew oxygen mask stowage box unit is identified in the Appendix of Intertechnique Service Bulletin MXP1/4-35-175, Revision 2, dated May 10, 2011.		730	DAY					
AD (g)	R1	2021-08-12	10	Wings INITIAL HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #1, 2, 3, 6, 7, and 8 attachment locations		4500	CYC	08633-2	5/28/2021 SC	06/02/2024	73,705	4,259
AD (g)	P	2021-08-12	20	Wings REPEAT HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #1, 2, 7, and 8 attachment locations		23000	CYC					
AD (g)	P	2021-08-12	30	Wings REPEAT HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #3 and 6 attachment locations		9000	CYC					
AD (g)	P	2021-08-12	40	Wings REPEAT (GRP 3) HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #1 and 8 attachment locations		19000	CYC					

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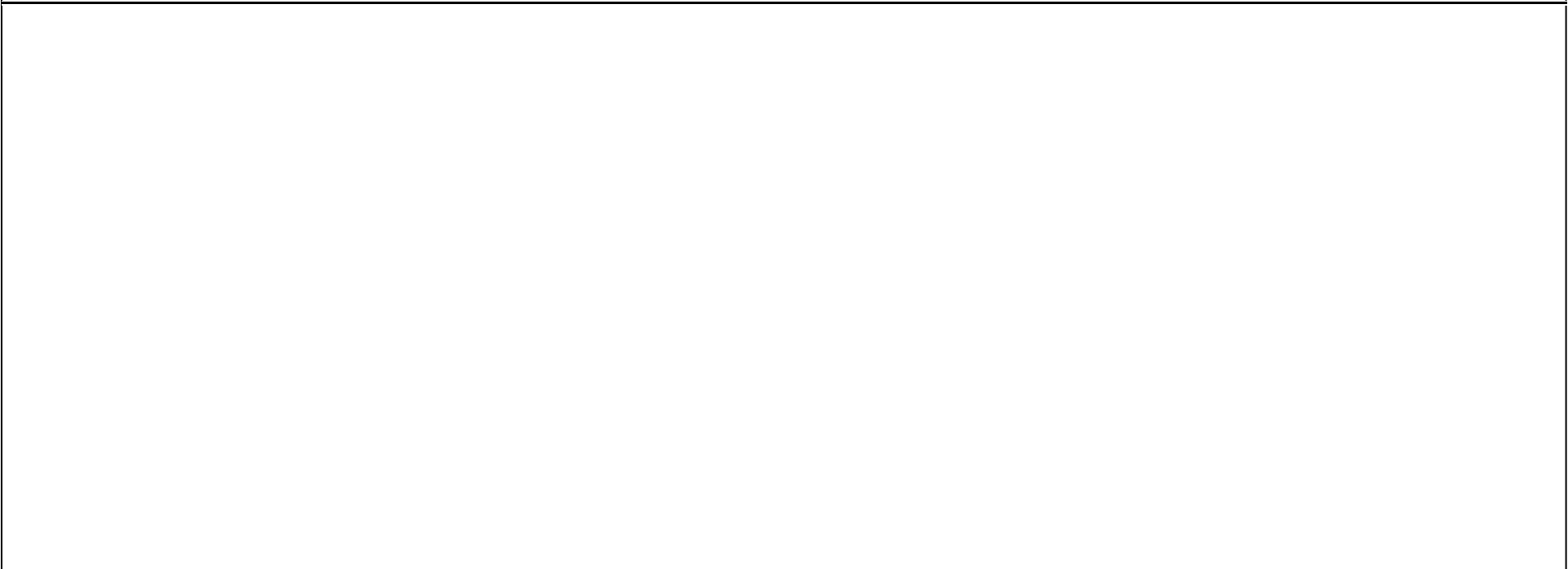


Aircraft Document

A/C Reg: N810TA Doc Type: Airworthiness Directive

A/C Reg: N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20	H/day: 2.00
Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69,446	C/day: 4.00

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied LogPage	Date	Due Date	Next Due	Remaining
AD	P	2021-08-12	(g) 50	Wings REPEAT (GRP 3) HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #2 and 7 attachment locations		16000	CYC					
AD	P	2021-08-12	(g) 60	Wings REPEAT (GRP 3) HFEC Inspection of Left and Right Wing, Lower AFT Wing Skin AFT Edge at Flap Track #3 and 6 attachment locations		7500	CYC					



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Engine Document

A/C Reg: N810TA, Doc Type: Airworthiness Directive

TransAir Hawaii

Hrs/day: 2.00

Cyc/day: 4.00

Engine 647548	Installed 10/28/2019	Current Hours: 32,304.90	A/C Reg N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20
Position: ENG 2	Current Cycles: 33670	Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69446	

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied			Due Date	Next Due	Remaining	
							A/C Reg	LogPage	Date					
AD	P	86-09-02 R2	10	Combustion chamber liner inspection <u>WOF</u> Inspect combustion chambers at 2000 cycles or 3000 hours	ENG 2	3000	HRS							
			10		ENG 2	8500	HRS							
			10		ENG 2	2000	CYC							
			10		ENG 2	6000	CYC							
AD	R1	86-09-02 R2	20	Combustion chamber liner inspection <u>WOF</u>	ENG 2	3097	CYC	N810TA	7263	10/28/2019	SC	03/14/2022	34,682	1,012
			20		ENG 2	6810	HRS	N810TA	7263	10/28/2019	SC	05/20/2029	38,059.70	5,754.80
AD	P	86-09-02 R2	30	Combustion chamber liner inspection <u>WOF</u>	ENG 2	77144	HRS							
			30		ENG 2	103614	CYC							
AD	R1	2003-12-07	10	12th Stage high pressure compressor HPC disc corrosion inspection stages C-7 through C-	ENG 2	1460	DAYS	N810TA	7141	8/15/2019	SC	08/14/2023		771

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Engine Document

A/C Reg: N810TA, Doc Type: Airworthiness Directive

TransAir Hawaii

Hrs/day: 2.00

Cyc/day: 4.00

Engine 647548	Installed 10/28/2019	Current Hours: 32,304.90	A/C Reg N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20
Position: ENG 2	Current Cycles: 33670	Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69446	

Type	Status	Doc	Item	Description / Action Required	Position	Limiters	Adj.	Last Complied			Due Date	Next Due	Remaining
								A/C Reg	LogPage	Date			
			12										
AD	P	2003-12-07	20	12th Stage high pressure compressor	ENG 2	2920	DAYS						
AD	R1	2010-21-17	10	Fan blade Overhaul of first stage fan blades every 4000 cycles in service. Initial compliance due at next shop visit. 306AL #2 engine 700559 N/A	ENG 2	4000	CYC	N810TA	7263	10/28/2019	SC	10/25/2022	35,585 1,915

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Engine Document

A/C Reg: N810TA, Doc Type: Airworthiness Directive

TransAir Hawaii

Hrs/day: 2.00

Cyc/day: 4.00

Engine 657227	Installed 10/28/2019	Current Hours: 70,827.30	A/C Reg N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20
Position: ENG 1	Current Cycles: 101908	Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69446	

Type	Status	Doc	Item	Description / Action Required	Position	Limiters	Adj.	Last Complied		Due Date	Next Due	Remaining
							A/C Reg	LogPage	Date			
AD	P	86-09-02 R2	10	Combustion chamber liner inspection REPEAT Inspect all Combustion Chambers	ENG 1	6000	CYC					
			10		ENG 1	2000	CYC					
			10		ENG 1	8500	HRS					
			10		ENG 1	3000	HRS					
AD	P	86-09-02 R2	20	Combustion chamber liner inspection	ENG 1	6810	HRS					
			20		ENG 1	3097	CYC					
AD	R1	86-09-02 R2	30	Combustion chamber liner inspection	ENG 1	103614	CYC		1/1/2000 SC	11/22/2044	136,076	34,166
			30		ENG 1	77144	HRS		1/1/2000 SC	05/10/2027	75,100.10	4,272.80
AD	R1	2003-12-07	10	12th Stage high pressure compressor HPC disc corrosion inspection stages C-7 through C-12	ENG 1	1460	DAYS	N810TA	D10266	4/1/2021 SC	03/31/2025	1,366

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Engine Document

A/C Reg: N810TA, Doc Type: Airworthiness Directive

TransAir Hawaii

Hrs/day: 2.00

Cyc/day: 4.00

Engine 657227	Installed 10/28/2019	Current Hours: 70,827.30	A/C Reg N810TA	As Of Date: 7/1/2021	Aircraft Time: 72,871.20
Position: ENG 1	Current Cycles: 101908	Serial#: 21116	Log Page: 08868-2	Aircraft Cycles: 69446	

Type	Status	Doc	Item	Description / Action Required	Position	Limiter	Adj.	Last Complied			Due Date	Next Due	Remaining	
							A/C Reg	LogPage	Date					
AD	P	2003-12-07	20	12th Stage high pressure compressor	ENG 1	2920	DAYS							
AD	R1	2010-21-17	10	Fan blade Overhaul of first stage fan blades every 4000 cycles in service. Initial compliance due at next shop visit. 306AL #2 engine 700559 N/A	ENG 1	4000	CYC	N810TA	D10184	9/19/2019 SC	10/25/2022	103,823	1,915	

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