



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

I. Aircraft	Make Cessna	Model U206E
	Serial No. U20601603	Nationality and Registration Mark N9403G
II. Owner	Name (As shown on registration certificate) Payette River Industries	Address (As shown on registration certificate) PO Box 1046 McCall, Id. 83638

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				XXX
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

D. Agency's Name and Address Chris Binford [Redacted]	E. Kind of Agency <input checked="" type="checkbox"/> U.S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	F. Certificate No. [Redacted]
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[Redacted]	[Redacted]
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BY	FAA Fit Standards Inspector	Manufacturer	XX	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection 12-6-94	Certificate or Designation No. [Redacted]	Signature of Authorized Individual [Redacted]
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed TCM IO-550-F engine and McCauley model D3A34C401 propeller with associated spinner assembly IAW Bonaire Aviation Company "Installation Instructions for Continental IO-550-F Engine in Cessna 206 Series Aircraft" Rev.D dated April 1, **1994** and Bonaire Aviation Co. drawing numbers 1 through 4 and 4a through 4g, all Rev.D. and all dated April **1,1994**, or later FAA Approved Revision. This work is covered under STC SA 283080.

Weight and Balance and Equipment List was updated to reflect the changes made.

-----End-----

☐ Additional Sheets Are Attached



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Office Identification

1. Aircraft	Make Cessna	Model U206E
	Serial No. U20601603	Nationality and Registration Mark N9403G
2. Owner	Name (As shown on registration certificate) Payette River Industries	Address (As shown on registration certificate) PO Box 1046 McCall, Id. 83638

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~			XX	
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address Chris Binford [Redacted]	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate NO. [Redacted]
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Date 12-6-94	Signature of Authorized Individual [Signature]
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BY	FAA Flt. Standards Inspector	Manufacturer	XX	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 12-6-94		Certificate or Designation No. [Redacted]	Signature of Authorized Individual [Signature]		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

I. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Repaired the nose cowling area by replacing the following skin pieces;

- 1) Lower nose cap PN 1252410-38
- 2) Lower LH cowl PN 1213660-7


All work was done IAW Cessna Repair Manual and AC 43:13 1A, Chapter 2 Section 3.

Weight and Balance not affected because of direct skin replacement.

End

☐ Additional Sheets Are Attached

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 U.S. Department of Transportation Federal Aviation Administration				MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, & Appliance)		Form Approved OMB No. 2120-0020	
						For FAA Use Only	
						Office Identification	
I. Aircraft	Make Cessna			Model U206E			
	Serial No. U20601603			Nationality and Registration Mark N9403G			
II. Owner	Name (As shown on registration certificate) Payette River Industries			Address (As shown on registration certificate) PO Box 1046 McCall, Id. 83638			
	3. For FAA Use Only						
4. Unit Identification						5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration		
AIRFRAME	***** (As described in Item 1 above) *****			X			
POWERPLANT							
PROPELLER							
APPLIANCE	Type						
	Manufacturer						
A. Agency's Name and Address Anton Gordon Bakke [Redacted]				B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer		C. Certificate No. [Redacted] A & P	
Date Oct 15, 1994				Signature of Authorized Individual [Redacted Signature]			
BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)			
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group				
Date of Approval or Rejection Oct 15, 1994		Certificate or Designation No. [Redacted]		Signature of Authorized Individual [Redacted Signature]			

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

The following structural component was repaired in accordance with Cessna Repair Manual and 43:13 - 1A - Change Three, Chapter 2, Section 3.

R.H. Cessna 206 Wing

Skins replaced: Outboard-most leading edge skin, top and bottom tip skins. Part numbers 0720601-16, 1220059-6, 1220059-5.

Outboard leading edge cuff replaced: Horton part number 2050-3.

Ribs replaced: Nose ribs sta. 190 and sta. 208, main rib sta. 208. Part numbers 0720607-4, 0720608-26, 0720608-7.

Tip fairing replaced: Part number 0721107-2.

Leading edge stringer spliced at sta. 195 with stock stringer reinforcement material.

Above item not installed on aircraft.

-----End-----

☐ Additional Sheets Are Attached

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

Form Approved
Budget Bureau No. 04-R050.1

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

FOR FAA USE ONLY

OFFICE IDENTIFICATION

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE Cessna	MODEL U206B
	SERIAL NO. U20601603	NATIONALITY AND REGISTRATION MARK N9403G
2. OWNER	NAME (As shown on registration certificate) Kurth Robert O.	ADDRESS (As shown on registration certificate) [REDACTED]

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****			XX	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS Wade P. Paquette [REDACTED]	B. KIND OF AGENCY U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC " " " CERTIFICATED REPAIR STATION MANUFACTURER	C. CERTIFICATE NO. A&P [REDACTED]
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE February 28, 1984	SIGNATURE OF AUTHORIZED INDIVIDUAL [REDACTED]
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7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persona specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNER	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 2/28/84		CERTIFICATE OR DESIGNATION NO. [REDACTED]		SIGNATURE OF AUTHORIZED INDIVIDUAL [REDACTED]	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed the following new parts in fuselage; 1-1212400-4 Skin-side R.H. tailcone, 1-1212416-10 Stringer-lower R.H. STA. 138.00 to STA. 209.00, 1-1212416-19 Stringer-lower R.H. STA. 166.40 to STA. 209.00, 1-1212417-2 Stringer-side R.H. STA. 152.20 to STA. 230.197, 1-0710601 Bulkhead Assg. upper cabin L.H., 1-1211487-1 Retainer-upholstery L.H., 1-1211443-1 Stringer-aft cabin L.H., 1-1211445-7 Channel-side window Fwd. L.H., 1-1211445-14 Channel-side window Fwd. L.H., 1-1210416-23 Skin Assg. cabin top Fwd., 1-1200461-1 Fairing Assy. wing to fuselage L.H. Aft., 1-1200461-2 Fairing Assy. wing to fuselage R.H. aft, 1-1211442-9 Fillet-wing to fuselage upper rear L.H., 1-1211442-18 Fillet- wing to fuselage upper rear R.H.

Installed the following new parts in left wing Assg. ; 1-0523400-20 Angle-rear spar Inbd., 2-0422229-11 Fitting-rear wing attachment, 1-0721100-27 Rib Assy. wing root, 1-1220083-1 Rib-trailing edge root, 1-0760607-3 Bracket-aileron pulley, 1-0720608-21 Rib Assy. wingtip STA. 208.00, 1-0720607-8 Rib-center STA. 190.00, 1-0720607-4 Rib Assg. nose STA. 190.00, 1-0720607-11 Rib-center STA. 172.00, 1-0720607-3 Rib Assy. nose STA. 172.00, 1-0720601-15 Skin-leading edge Outbd., 1-1220059-6 Skin-upper Outbd., 1-1220059-5 Skin-lower Outbd., 1-2050-2 Horton Skin- leading edge Outbd., 1-0523541-61 Stringer.

Repaired left wing as follows;

Installed front wing spar splice kit SK180-30P at STA. 178.75 as per Drawing No. 10172-304 Rev. B.

Spliced outboard section of PN 1220060-206 skin from STA. 154.00 to STA. 172.00.

Spliced outboard section of PN 1220060-29 skin from STA. 154.00 to STA. 172.00.

Spliced inboard section of skin PN 1220023-5 from wing root to STA 39.0

Spliced inboard section of skin PN 1220059-14 from wing root to Sta. 39.00

Spliced inboard section of skin PN 1220059-13 from wing root to Sta. 39.00.

Spliced inboard rear 8 5/8" X 6 1/2" section of Skin PN 1220006-21.

Spliced rear spar from STA. 33.00 to inboard end.

Cessna 206 series service manual section 18, paragraph 18-1, 18-2, 18-3, 18-5, 18-7, 18-10, 18-11, 18-12, 18-13, 18-14, 18-15, 18-16, 18-18, 18-19, 18-23, 18-24, 18-27, 18-28, 18-30, 18-32, 18-33, 18-40, 18-42, 18-52, 18-53, 18-55, 18-56, figure 18-1, 18-2, 18-6, 18-11, AC43.13-1A Chapter 2 Section 1 Paragraph 54, a, b, c, Section 3 Paragraph 95, 96, c, d, 97, a, 99, b, c, d, e, f, g, h, 100, d, e, were used as an acceptable guide. END.

☐ ADDITIONAL SHEETS ARE ATTACHED