



# WEIGHT AND BALANCE

FOR RIGGING EVALUATION PURPOSES

DATE: X/X/XXXX  
 SERIAL: AC-XXX  
 REGISTRATION: NXXXXXX

ITEM	SCALE WEIGHT (lbs)	TARE (lbs)
FRONT LEFT WHEEL	38.0	
FRONT RIGHT WHEEL	34.0	
REAR LEFT WHEEL	656.0	
REAR RIGHT WHEEL	651.0	

NET WEIGHT (lbs)		ARM (in)	=	MOMENT (in-lbs)
38.0	X	22.5	=	855
34.0	X	22.5	=	765
656.0	X	140.3	=	92037
651.0	X	140.3	=	91335
<b>TOTAL</b>		1379.0 lbs		184992 in-lbs

EMPTY C.G. LOCATION = TOTAL MOMENT / TOTAL WEIGHT =

134.1 in.

OR

53%

CHORD

Wing Leading Edge Station: 96.128 in.  
 Chord of Wing: 72.139 in.

ITEM	ARM (in)		WEIGHT (lbs)	=	MOMENT (in-lbs)
EMPTY AIRCRAFT	134.1	X	1379.0	=	184992
FRONT PILOT	45.0	X	156	=	7020
REAR PILOT	92.0	X	150	=	13800
FUEL	114.0	X	168	=	19152
AIRCRAFT CARGO COMPARTMENT	125.0	X	0	=	0
FLOAT CARGO COMPARTMENT	111.0	X		=	
<b>TOTAL</b>			<b>1853</b> lbs		<b>224964</b> in-lbs

LOADED C.G. LOCATION = TOTAL MOMENT / TOTAL WEIGHT =

121.4 in.

C.G. LOCATION MUST BE BETWEEN 110.8 IN. AND 125.4 IN.

GROSS WEIGHT = 2000 LBS.

Cargo Location: 15.0 in.  
 Plus + 100 in.  
 Equals: 115.0 in.