MERIDIAN PA-46-500TP

REFERENCE ONLY

THIS ELECTRONIC VERSION OF THE POH IS NOT APPROVED TO REPLACE ANY OPERATING INFORMATION REQUIRED BY THE REGULATIONS.

PILOT'S OPERATING HANDBOOK

SN 4697174 AND UP

FAA APPROVED AIRPLANE FLIGHT MANUAL

AIRPLANE SERIAL NO. AIRPLANE REGIST. NO.

PA-46-500TP REPORT: VB-1888 FAA APPROVED BY:

Albert J. Mill DOA-510620-CE THE NEW PIPER AIRCRAFT, INC. VERO BEACH, FLORIDA

DATE OF APPROVAL: FEBRUARY 4, 2004

THIS HANDBOOK INCLUDES THE MATERIAL REQUIRED TO BE FURNISHED TO THE PILOT BY THE FEDERAL AVIATION REGULATIONS AND ADDITIONAL INFORMATION PROVIDED BY THE MANUFACTURER AND CONSTITUTES THE FAA APPROVED AIRPLANE FLIGHT MANUAL. THIS HANDBOOK MUST BE CARRIED IN THE AIRPLANE AT ALL TIMES.





PA-46-500TP

SECTION 3 EMERGENCY PROCEDURES

3.9 ENGINE SYSTEM (Continued)

3.9f Feather

Indication: Amber "FEATHER" annunciator illuminated.

On Ground After Engine Start:

Shut down and investigate cause.

In Flight:

Land as soon as practical and investigate cause.

3.9g Beta

Indication: Amber "Beta" annunciator illuminated in flight.

Power Lever	
	POSITION OR FORWARD
	OF FLIGHT IDLE.

3.11 FUEL CONTROL UNIT MALFUNCTION OR POWER LEVER CONTROL LOSS (Manual Override Operation)

WARNING

The manual override system is an emergency system and must only be used in the event of FUEL CONTROL UNIT MALFUNCTION OR POWER LEVER CONTROL LOSS.

CAUTION

The manual override lever is not to be used on the ground for taxiing. During ground opera tions, it may not be possible to recover low Ng with the manual override lever.

The pilot must ensure that the MANUAL OVERRIDE LEVER is in the OFF (full aft) position prior to start otherwise an over-temperature condition may result.

ISSUED: FEBRUARY 4, 2004 REVISED: SEPTEMBER 8, 2015 REPORT: VB-1888 3-13

SECTION 3	
EMERGENCY PROCEDURES	

PA-46-500TP

3.11 FUEL CONTROL UNIT MALFUNCTION OR POWER LEVER CONTROL LOSS (Manual Override Operation) (Cont.)

Utilize slow and smooth movement of the MANUAL OVERRIDE LEVER to avoid engine surges and/or exceeding ITT, Ng, or torque limits. <u>Rapid movement of the MOR lever can cause compressor surges and excessive ITT (over temperature) conditions.</u>

CAUTION

Reverse will not be available for landing.

Land as soon as possible.

After landing:

CONDITION Lever CUT-OFF/FEATHER

If power control using manual override is excessive:

Reduce airspeed to below 168 KIAS by increasing pitch attitude.		
Landing Gear	EXTEND BELOW 168 KIAS	
FLAPS 10°	BELOW 168 KIAS	

Land as soon as possible.

When landing is assured: CONDITION LeverCUT-OFF/FEATHER

REPORT: VB-1888 3-14 ISSUED: FEBRUARY 4, 2004 REVISED: JANUARY 11, 2018

PA-46-500TP	SECTION 7
MERIDIAN	DESCR/OPERATION

7.7 ENGINE CONTROLS (continued)

The MOR is an emergency device that may allow the crew to regain power and continue safe flight and landing following fuel control unit (FCU) malfunction or power lever control loss. The MOR is used to control fuel flow to the engine in the event a pneumatic malfunction occurs in the engine fuel control unit. A malfunction of the pneumatic signal (Py) input to the FCU will result in the fuel flow decreasing to minimum idle (approximately 48% Ng at sea level and increasing with altitude). Additional effects of a Py malfunction are loss of the torque/Ng limiting functions and, Nf governor operation (reverse is not available).

The manual override (MOR) lever is located in the center console to the left of the power lever. To operate the MOR, lift up on the lever and slowly move it forward to take up the dead-band until the engine responds. If possible, allow engine to stabilize before advancing further. Monitor gas generator speed (Ng), ITT, and torque. Rapid movement of the MOR lever can cause compressor surges and excessive ITT (over temperature) conditions.

The friction adjustment lever, located in the middle of the control quadrant, may be adjusted to increase or decrease the friction holding the power lever.

7.8 MEGGITT AVIONICS NEXT GENERATION INTEGRATED COCKPIT (MAGIC)

This section describes the components and operation of the Meggitt Avionics Next Generation Integrated Cockpit (MAGIC).

Refer to Section 7.8a, Meggitt Powerplant and Mechanical System Instrumentation, for the components and operation of the powerplant and mechanical system instrumentation.

Refer to **Section 7.8b**, **Meggitt EFIS Display**, for the components and operation of the Electronic Attitude Director Indicator (EADI) and the Electronic Horizontal Situation Indicator (EHSI).

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