

ENGINE EXAMINATION REPORT N257AR

ENGINE MODEL	Titan IOX-370-CLD1T4
ENGINE SERIAL NUMBER	E9E010
AIRCRAFT MAKE & MODEL	Commuter Craft - Innovator
AIRCRAFT SERIAL NUMBER	002 (Prototype)
AIRCRAFT REGISTRATION	N257AR
FILE NUMBER	18-372

NAME	SIGNATURE	DATE
Mike Council		4-2-2020

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GENERAL INFORMATION									
EX	AMINA	TION				ACCIDE		Α	
DATE		2019 Or 2019 Re		/	NTSB ACCIDENT # ERA		ERA1	19FA134	
FACILITY	Atlant	a Air Re	ecovery	/	NTSB INVESTIGATOR Bria		Brian	an Rayner	
ADDRESS	Unifo	rm Drive	;		FAA INVESTIGATOR Juli		Juli O'	Gorman	
	Griffir	i, GA			ACCIDENT DATE 3-23		3-23-2	2019	
					ACCIDENT	LOCATION	Calho	un, GA	
			EN	IGINE	INFORMATIO	ON			
ENGINE POSI	TION	Single	– Rea	r-facin	g pusher conf	iguration			
TOTAL	TIME	Undete	ermine	b					
TIME	SOH	NA							
TYPE & TIM	E SLI	Undete	ermine	b					
BUILD	DATE	9-20-2	016						
IN SERVICE I	DATE	Undete	ermine	b					
documented and oil and filter change dated 1-8-2019.									
Report Summary: Search Code(s):									
An examination of NTSB Investigato would have preve	r. The	inspectio	on of th	nis eng	ine did not re	veal any pre-i			
Disposition of en further examination request of the NT	on but v	was retu	-		•				

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INSPECTION WITNESSES						
NAME	Mike Council	NAME	Brian Rayner			
ADDRESS	Mobile, Alabama	ADDRESS	Eastern Region			
ORGANIZATION	Continental Aerospace Technologies	ORGANIZATION	NTSB			
PHONE		PHONE				

EXTERNAL INSPECTION OF ENGINE

An engine data plate attached to the accident engine identified the engine as a Titan Aircraft Engine, model IOX-370-CLD1T4, serial number E9E010, rated at 200 horsepower at 2700 RPM.

The engine mount tubes failed due to impact forces and the engine separated from the fuselage. The engine driven fuel pump separated but was recovered from the wreckage. The electronic "e-mags" received impact damage and the ignition harnesses separated. The right E-mag electronic ignition module was impact damaged.

Cylinder # 1 received impact damage with wood fibers imbedded in the cooling fins. The # 1 and # 4 exhaust tubes were impact damaged and the # 3 induction tube separated. The starter ring gear separated from the pulley. The oil filter adapter separated but was recovered from the wreckage path. The oil filter remained intact and attached to the broken adapter with proper safety wire still attached. The outer oil pump cover was impact damaged and separated from the pump. The oil cooler separated.

The two-blade constant-speed propeller hub remained attached to the engine crankshaft flange. Both composite propeller blades separated approximately 10-12 inches outboard of the propeller root consistent with engine power at impact.











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INTERNAL INSPECTION OF ENGINE

The top sparkplugs were removed and inspected. All four top sparkplugs (automotive type) exhibited normal combustion signatures and wear patterns. Number 4 top sparkplug was impact damaged. Cylinders 2 and 4 produced thumb compression when the engine was manually rotated. Cylinders 1 and 3 did not produce thumb compression. The cylinders were inspected using a lighted electronic borescope. All piston domes and valves faces exhibited normal combustion signatures. Cylinder #1 exhaust valve was not fully closing. Cylinder # 1 rocker cover and cylinder head exhibited impact damage.

Cylinder # 1 rocker cover was removed. The intake and exhaust rocker arms, pushrods and valve springs were intact and moved correctly when the engine was manually rotated. However, the exhaust valve stem did not move fully in the valve guide which kept the exhaust valve from fully closing. Each cylinder rocker cover exhibited impact damage.

The right E-mag position was marked with a yellow paint pen then the E-mag was removed. The accessory gears moved correctly when the engine was manually rotated. The E-mag was then reinstalled.





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