

Continental Motors

ENGINE EXAMINATION REPORT

LEFT

ENGINE MODEL	IO520C (15B)
ENGINE SERIAL NUMBER	287461-R
AIRCRAFT MAKE & MODEL	Beechcraft D55 Baron
AIRCRAFT SERIAL NUMBER	TE-616
AIRCRAFT REGISTRATION	N533Q
FILE NUMBER	18-366

NAME	SIGNATURE	DATE
Mike Council		4-17-2019

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	GENERAL INFORMATION				
EX	AMINA	TION	ACCIDEN	IT DATA	
DATE	2-23-	2019	NTSB ACCIDENT #	WPR19FA086	
FACILITY	Accid	ent site	NTSB INVESTIGATOR	Jack Vanover	
ADDRESS	Cumr	nings Mountain	FAA INVESTIGATOR	NA	
	Teha	chapi, CA	ACCIDENT DATE	2-21-2019	
			ACCIDENT LOCATION	*Cummings Mountain, Tehachapi, CA	
		ENGINE	INFORMATION		
ENGINE POS	TION	Left			
TOTAL	TIME	Undetermined due to inconsistencies of hobbs times in logbooks. The final log entry for the last 100-hour inspection dated 4-4-2018. Hobbs: 618.9 indicated 653.9 hours TSN. (Unverified)			
TIME	SOH	NA			
TYPE & TIM	E SLI	Undetermined			
BUILD I	DATE	11-28-1995 according to Continental Motors Serial data.			
IN SERVICE I	DATE	12-25-1995 accordi	ng to engine logbook entry	(unspecified airplane)	

^{*}The accident site was near the top of Cummings Mountain, part of Tehachapi Mountains, in a remote area of privately-owned land. The only access to the accident site was by escort with the Kern, Mono and Fresno County Sheriff's Search and Rescue team using snow-cat vehicles. The wreckage was covered in snow in an area that had approximately 4' of snowfall accumulation. The search and rescue IIC required all teams to be back at the SAR base camp before dark for safety reasons. After a two-and-a-half-hour snow-cat ride, the teams arrived at the accident site and had approximately 2.5 hours to complete all tasks before zero hour to depart.

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Significant logbook information:

The Continental Motors Engine logbook which accompanied engine serial number **287461** assembled as a factory rebuilt engine 11-28-1998 acknowledge installation in an unspecified airplane at hobbs 2396 dated 12-25-1995. After an oil change was recorded 9-10-1997, an entry indicates "See AD-Log for year 1998 entries". The Ad Log book marked "LEFT" begins with engine serial number **126306-8-C** and begins on page one, 1-25-1984. The last maintenance entry for this serial number engine is dated 12-16-1986 and indicated that Windham Aircraft Engines overhauled the engine (serial 126306-8-C) on Job Number 1735. The very next entry on an unnumbered page is dated 5-20-2003 at hobbs 496.6. No engine serial number is specified.

The first time that the accident engine (serial number 287461) is mentioned in this same "Left" Ad Log Maintenance logbook is dated April 4, 2018. See the entry below:

Left Engine Log Entry

N533Q

Continental IO-520-C158 S/N 287461-R HOBBS 0618.9 TSN 653.9 April 4th, 2018

- 11. Complied with 100 hour inspection IAW 14 CFR part 43 appendix D.
- Drained oil and removed filter, installed Champion oil filter p/n CH48109-1 and serviced with (12) qts of Aeroshell 100W engine oil. Took oil sample and sent for analysis.
- 13. Engine compression: #1: 68/80 #2: 70/80 #3: 73/80 #4: 66/80 #5: 73/80 #6: 70/80
- 14. Removed injectors and installed after cleaning.
- 15. Replaced spark plug copper gaskets with new p/n AN4027-1
- 16. AD 2014-05-29 does not apply, affected cylinder assemblies not installed.
- 17. AD 2016-16-12 does not apply, affected cylinder assemblies not installed.
- Applicable engine airworthiness directives have been complied with through 2016-16-12. See compliance checklist for details. Ran engine and leak checked, OK.

I certify that this ENGINE has been inspected in accordance with a 100 hour inspection and has been determined to be in airworthy condition.



Report Summary:

Search Code(s):

Examination of the engine was performed by the CMI Investigator under supervision of the NTSB Investigator. The inspection of this engine did not reveal any pre-impact anomalies that would have prevented Its ability to produce rated horsepower.

NOTE: According to the engine logbooks, this engine accrued only 13.8 hours between 100-hour inspections dated 8-28-2008 and 4-4-2018. There are no other recorded maintenance entries in the logbook between those dates.

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Disposition of engine following exam: At the time of this writing, there is no plan to recover the wreckage from the accident site. The airplane was not insured, and no recovery business was willing to pick up the wreckage from this isolated site.

*The accident site was near the top of Cummings Mountain, part of Tehachapi Mountains, in a remote area of privately-owned land. The only access to the accident site was by escort with the Kern, Mono and Fresno County Sheriff's Search and Rescue teams using snow-cat vehicles. The Investigation teams traveled as a group which included the Coroner's office Investigators. The wreckage was covered in snow in an area that had approximately 4' of snowfall accumulation. The search and rescue IIC required all teams to be back at the SAR base camp before dark for safety reasons. After a two-and-a-half-hour snow-cat ride, the teams arrived at the accident site and had approximately 2.5 hours to complete all tasks before zero hour to depart.

INSPECTION WITNESSES					
NAME	Mike Council	NAME	Jack Vanover		
ADDRESS		ADDRESS	Western Pacific Region		
	Mobile, Alabama		Seattle, WA		
ORGANIZATION	Continental Motors	ORGANIZATION	NTSB IIC		
PHONE		PHONE			
NAME	Andrew Hall	NAME	NA		
ADDRESS	Wichita, KS	ADDRESS			
ORGANIZATION	Textron Aviation	ORGANIZATION			
PHONE (PHONE			

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EXTERNAL ENGINE CONDITION

The engine separated from its nacelle and was found forward of the main wreckage in the snow. The engine was inverted, and the engine oil sump exhibited impact damage. The rear portion of the oil sump was crushed upward, and a hole was present. The shape of the puncture was consistent with an undetermined object pushing inward from the outside surface of the sump. The right and left oil drain plugs were present and remained secured with safety wire.

The fuel component-throttle body-mixture control remained attached but exhibited impact damage. The throttle and mixture control arms were intact and properly secured. The throttle control cable and mixture control cable were both attached but were separated at approximate mid-length.

Cylinders 1, 3, and 5 remained attached and exhibited damage to cooling fins and rocker box covers. Cylinders 2, 4, and 6 remained attached but all three rocker covers were impact damaged or missing.

The engine was manually rolled over, right side-up. The left forward crankcase half exhibited heavy impact damage. The right upper crankcase was intact and exhibited an engine data plate identifying the engine as a Continental Motors model IO-520C, serial # 287461-R. The fuel manifold remained attached and the assembly screws were secured. No lead seal was attached.

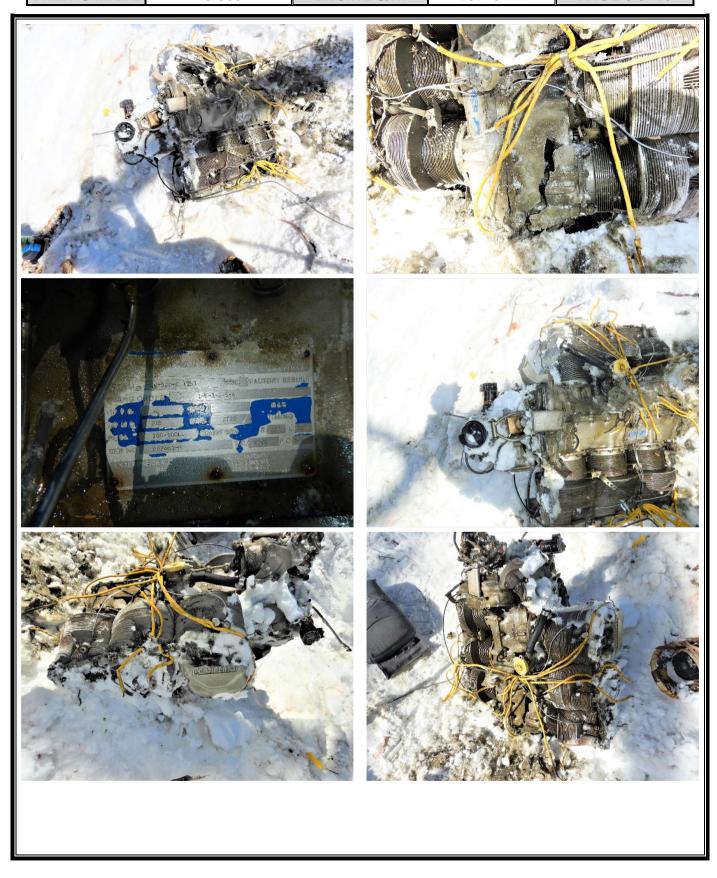
The left and right magnetos separated and were not recovered. The impact damaged ignition leads exhibited cuts and individual leads were broken with missing ends.

The forward mounted engine driven alternator separated and was not recovered. The instrument air pump separated at its mount and was not recovered. Only the mounting ring and plastic shear-drive coupling remained attached. The plastic drive gear was intact.





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ENGINE COMPONENT EXAMINATION

EXHAUST SYSTEM

Condition:

The left side exhaust components were impact damaged and separated from the engine due to impact forces. The right exhaust collector and risers remained attached to cylinder ports but was impact damaged.



INDUCTION SYSTEM

Condition:

The left induction intake tube elbows and hoses separated, and the forward balance tube assembly was crushed.



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IGNITION SYSTEM

LEFT MAGNETO		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition:	Se	parated from engine and not rec	overed.	
RIGHT MAGNETO Manufacturer: Undetermined		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition:	Se	parated from engine and not rec	overed.	
IGNITION HARNESS Manufacturer: TCM P/N: Undetermined S/N: Undetermined				
Condition: Individual ignition leads were cut and pinched due to impact forces.				



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FUEL SYSTEM

				,
FUEL PUMP M		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition:	Unde	etermined		
THROTTLE BODY METERING UNIT		Manufacturer: TCM	P/N: 6299?4-2	S/N: J179524AR
Condition: The fuel component-throttle body-mixture control remained attached but exhibited impact damage. The throttle and mixture control arms were intact and properly secured. The throttle control cable and mixture control cable were both attached but were separated at approximate mid-length.				



FUEL MANIFOLD VALVE Manufacturer: TCM P/N: 63?427-1 S/N: A188121R

Condition: The manifold valve was intact and attached at it mount. The assembly screws were



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LUBRICATION SYSTEM

OIL SUMP

Condition: The rear portion of the oil sump was crushed upward, and a hole was present.



OIL FILTE	TER Manufacturer: Undetermined P/N: Undetermined			
Condition:	Condition: Undetermined. The oil filter separated and was not found in the wreckage.			
OIL COOLER		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition: One oil cooler which had separated was found in the wreckage. Orientation could not be determined.				

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CYLINDERS		P/N: Undetermined	S/N: Undetermined	Head Date: Undetermined	
Condition:	rec	All cylinders remained attached to their respective mount. Cylinders 1, 3, and 5 received minor impact damage to lower cooling fins but rocker covers on cylinders 2, 4, and 6 were impact damaged.			

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CRANKCASE ASSEMBLY

CRANKCASE

Condition: The left forward crankcase half was impact damaged.



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ACCESSORIES

STARTER		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition: The starter separated and was not located in the			located in the wreckage	
STARTER ADAPTER P/N: Undetermined				
Condition: The starter adapter remained attached but was impact damaged.			naged.	
ALTERNATOR		Manufacturer: Undetermined	P/N: Undetermined	S/N: Undetermined
Condition:	The	forward mounted alternator se	parated and was not rec	overed.
VACUUN PUMP	1	Manufacturer:	P/N:	S/N:
Condition:	Condition: The vacuum pump separated and was not recovered. An intact vacuum pump drive spline was found still engaged in the accessory gear.			





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PROPELLER

PROPELLER		Manufacturer: Hartzell	P/N: See note below	S/N: See note below
Condition: Both propellers exhibited impact damage with rotational scoring on the cambered				

Both propellers exhibited impact damage with rotational scoring on the cambered surfaces and torsional bending of the propeller blades. Damaged trees in the area

of initial impact with propeller cuts exhibited 45-degree cuts.

NOTE: Two propellers were found in the wreckage. Orientation was not possible. The letters A & B are arbitrary and do not indicate orientation.

A-Hartzell three blade full feathering propeller model: PHC-C3YF-2UF, hub serial number EB4535A





B- Hartzell three blade full feathering propeller model: PHC-C3YF-2UF, hub serial number EB4536A





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