<image/>						
ENG	SINE MODEL	O-300-D (6B)				
ENGINE SERI	AL NUMBER	27160-D-2-D				
	KE & MODEL 1958 Cessna 172					
AIRCRAFT SERI	AL NUMBER	36320				
AIRCRAFT REC	GISTRATION	N8620B				
FI	LE NUMBER	18-370				
NAME	SIC	DATE				
Phillip Grice		03-02-2020				
Mike Council	2-18-2020					

18-370

ENGINE EXAMINATION REPORT ENGINE S/N:

27160-D-2-D

PAGE 2 of 39

GENERAL INFORMATION						
EX	AMINA	TION	AC		IT DATA	
DATE	6-17-	2019	NTSB ACCIDE	IDENT # CEN19FA099		A099
FACILITY		nental Aerospace nologies-Analytical	NTSB INVESTIGA	ATOR	Michael Central	Hodges, IIC Region
ADDRESS	2039	South Broad Street	FAA INVESTIGA	ATOR	NA	
	Mobil	e, Alabama		DATE	3-14-20	19
	_		ACCIDENT LOCA		Lakewa	y, TX
	_	ENGINE	INFORMATION			
ENGINE POSI	TION	Single				
TOTAL	TIME	4630.5 calculated				
TIME	SOH	153.3 hours calculat	ted			
TYPE & TIM	E SLI	99.3 hours calculate	ed since last 100-hou	ır inspe	ction date	ed 11-7-2018
BUILD	DATE	3-26-1962 (shipped	to Cessna Aircraft) a	accordir	ng to CMI	Serial Data
IN SERVICE I	DATE	Original undetermin	ed			
Significant logbook information: According to the engine logbooks, the accident engine was overhauled to factory new limits in the field 4-21-2016 using new Superior Air Parts cylinders. See copy of logbook excerpt below: 04/21/2016 Cessna 172 N8620B S/N 17236326 Tach reads 728.8 TTIS: 4477.2 owner : E. Mulloy Continental 0300D S/N 27160-D-2-D TSO: 0.0 Removed engine s/n 3333-D-5-D and installed overhauled engine s/n 27160-D-2-D. Engine overhauled is accordance with Continental Overhaul manual X30013 to new limits. New Superior cylinder assemblies See Forms 8130 and component work orders for details. Installed engine using new engine mount rubbers, new mount to airframe bushings and washers, bolts and lock nuts. Refurbished engine mount and painted. Cleaned firewall and inside of cowlings, primed and painted. Installed carburetor, air-box generator, starter and exhaust system in "as removed" condition. Installed serviceable Slick magnetos new Autolite spark plugs. Fabricated new fuel hose and installed with fire-sleeve. Installed new Airform Baffle Kit. Serviced oil using Phillips XC 20/50 oil. Installed new spin on filter. Pre-oiled engine to 15 PSI Reconditioned McCauley propeller and installed using new lock huts. Engine run up, operational cheor performed, adjustments made as needed. Installed cowling that this engine is airworthy with respect to work performed. Joseph Ø. Sills A&P						
Report Summary: Search Code(s): 15-12-68 The engine was disassembled and inspected under the supervision of the NTSB IIC. This inspection did not reveal any pre-impact anomalies that would have prevented Its ability to produce rated horsepower. Disposition of engine following exam: The engine was released by the NTSB IIC and returned to Air Salvage of Dallas. See shipper # AD01162 dated 6-19-2019.						

18-370

ENGINE EXAMINATION REPORT ENGINE S/N: 2716

27160-D-2-D

PAGE 3 of 39

INSPECTION WITNESSES							
NAME	Phillip Grice	NAME	Lisa Jersild				
ADDRESS	Mobile, Alabama	ADDRESS	Mobile, Alabama				
ORGANIZATION	Continental Aerospace Technologies	ORGANIZATION	Continental Aerospace Technologies				
PHONE		PHONE					
NAME	Randy Bryant	NAME	Michael Hodges, IIC				
ADDRESS	Mobile, Alabama	ADDRESS	Central Region				
ORGANIZATION	Continental Aerospace Technologies	ORGANIZATION	NTSB				
PHONE		PHONE					
NAME	Peter Basile	NAME	NA				
ADDRESS	Wichita, KS	ADDRESS					
ORGANIZATION	Textron Aviation	ORGANIZATION					
PHONE		PHONE					

ASI-Mike Council did not attend the engine tear down examination.

ENGINE EXAMINATION REPORT

ENGINE S/N: 27160-D-2-D

PAGE 4 of 39

EXTERNAL INSPECTION OF ENGINE

The accident engine was received in a Continental Aerospace Technologies engine crate. The engine crate was opened in the presence of the NTSB IIC and the engine was removed and placed on an examination stand. A data plate attached to the crankcase identified the engine as a Continental Motors model O-300-D (6B) engine, serial number 27160-D-2-D.

Engine mounts were broken. The oil sump was impact damaged and the left forward/side separated. Cylinder # 2 intake and exhaust push rods and push rod tubes were impact damaged.

Exhaust mufflers, tailpipes and risers were impact damaged.

18-370

Left induction manifold was impact damaged.

Ignition harness leads were cut and pinched.

Carburetor was impact damaged and shipped loose.

This engine has been visually inspected previously at the accident site. The following parts had been removed at the previous examination and not reinstalled:

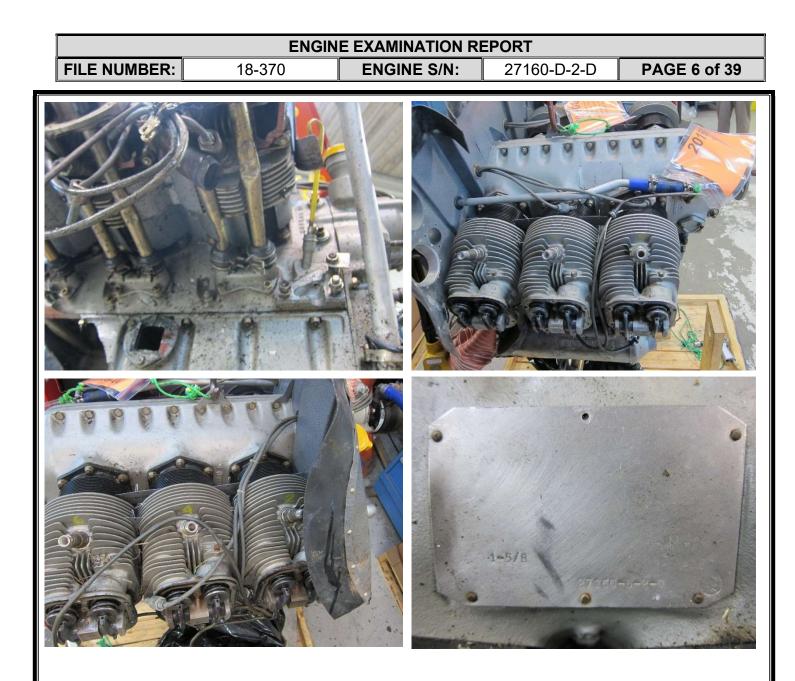
Cylinder rocker arm covers

Left and right magnetos

Spin-on engine oil filter

Engine driven generator





FILE NUMBER:	

18-370

27160-D-2-D

PAGE 7 of 39

AIRFRAME COMPONENTS RETURNED WITH ENGINE

Cooling baffles Engine driven generator Right muffler assembly (Left muffler not received) Instrument air pump Aftermarket oil filter adapters and hoses Oil Cooler – NOTE: The current Continental Aerospace Technologies Illustrated Parts Catalog for this engine does not indicate that an oil cooler was included during the manufacturing of engine model O-300-D (6B) engines.

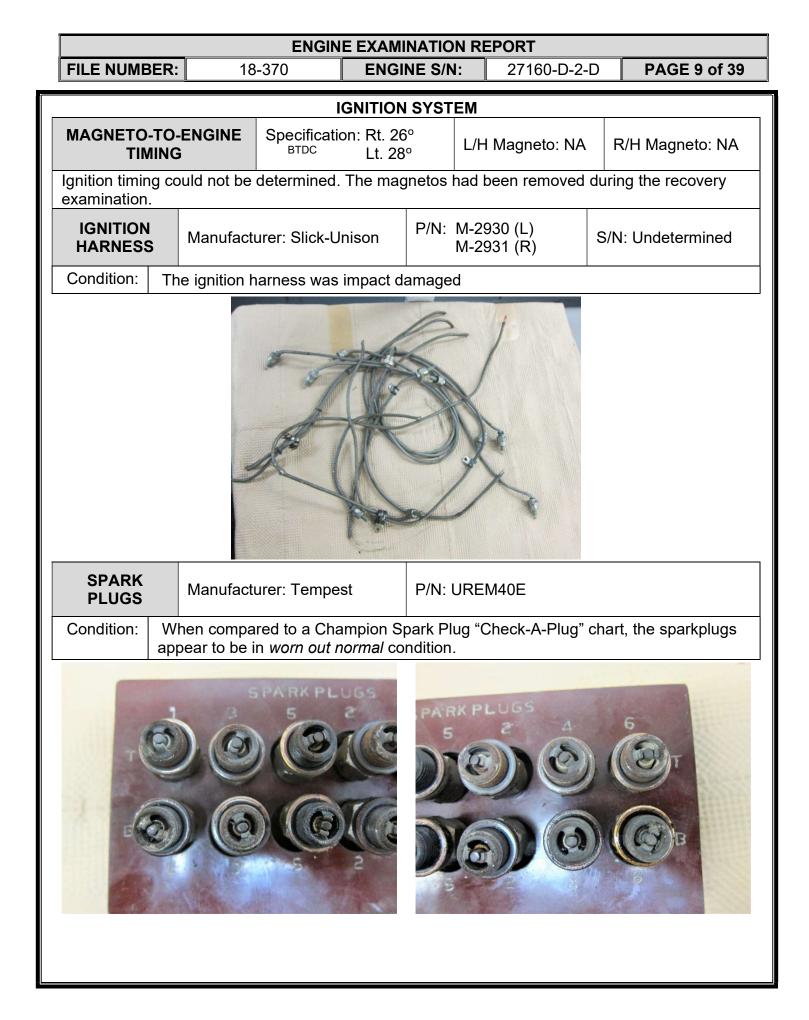
 ENGINE EXAMINATION REPORT

 FILE NUMBER:
 18-370
 ENGINE S/N:
 2716

27160-D-2-D

PAGE 8 of 39

ENGINE TEARDOWN AND COMPONENT EXAMINATION **EXHAUST** SYSTEM The exhaust system components were impact damaged. The muffler/heat Condition: exchanger was impact damaged. Left muffler assembly not shipped with engine. -INDUCTION SYSTEM The left induction manifold was impact separated and not returned with the engine. Condition: The right induction manifold was intact except damage to the mounting flange.



ENGINE EXAMINATION REPORT							
BER:	18-370	ENGI	NE S/N:	27160-D-2-	D PAGE 1	0 of 39	
c	Manufacturer: Slick-U Industries	nison	M/N: 6364		S/N: 07090201		
Condition: The magneto had been removed during the initial recovery examination. The magneto and tested by manually rotating the drive and noting spark at all six ignition leads. The magneto was shipped loose in the shipping container. The magneto was bench tested and functioned normally.							
<complex-block></complex-block>							
c	Manufacturer: Slick-U Industries	nison	M/N: 6364		S/N: 07090204	ŀ	
ma six	gneto and tested by ma ignition leads. The mag	anually ro gneto was	tating the d s shipped lo	ser. NO. MODEL AG. SLICK MAG. FAA-P.M.A.	g spark at four of oping container.	f the	
	D The ma ign ma	SER: 18-370 Manufacturer: Slick-U Industries The magneto had been remagneto and tested by maignition leads. The magneto magneto was bench tested Imagneto magneto had been remagneto was bench tested Imagneto was bench tested Imagneto was bench tested Imagneto had been remagneto was bench tested Imagneto was bench tested Imagneto was bench tested Imagneto had been remagneto was bench tested Imagneto was bench tested Imagneto had been remagneto and tested by magneto and	BER: 18-370 ENGIN Manufacturer: Slick-Unison Industries The magneto had been removed du magneto and tested by manually ro ignition leads. The magneto was sh magneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested and fun Imagneto was bench tested by manually ro Imagneto was Imagneto had been removed du Imagneto had been removed du Imagneto had been removed du Imagneto and tested by manually ro Six ignition leads. The magneto was Imagneto was	BER: 18-370 ENGINE S/N: Manufacturer: Slick-Unison Industries M/N: 6364 The magneto had been removed during the init magneto and tested by manually rotating the d ignition leads. The magneto was shipped loose magneto was bench tested and functioned nor Image: Image	ER: 18-370 ENGINE S/N: 27160-D-2 Manufacturer: Slick-Unison Industries M/N: 6364 The magneto had been removed during the initial recovery ex- magneto and tested by manually rotating the drive and noting ignition leads. The magneto was shipped loose in the shippin magneto was bench tested and functioned normally. Image: the magneto was shipped loose in the shippin magneto was bench tested and functioned normally. Image: the magneto was shipped loose in the shippin magneto was bench tested and functioned normally. Image: the magneto was shipped loose in the shippin magneto was bench tested and functioned normally. Image: the magneto was shipped loose in the shippin magneto was bench tested and functioned normally. Image: the magneto had been removed during the initial recovery ex- magneto and tested by manually rotating the drive and noting six ignition leads. The magneto was shipped loose in the ship magneto was bench tested and functioned normally. Imagneto had been removed during the initial recovery ex- magneto and tested by manually rotating the drive and noting six ignition leads. The magneto was shipped loose in the ship magneto was bench tested and functioned normally. Image: the magneto in the sted and functioned normally. Image: the magneto was bench tested and functioned normally. Image: the magneto was bench tested and functioned normally. Image: the magneto was bench tested and functioned normally. Image: the magneto was bench tested and functioned normally.	ER: 18-370 ENGINE S/N: 27160-D-2-D PAGE 1 Manufacturer: Sick-Unison Industries M/N: 6364 S/N: 07090201 The magneto had been removed during the initial recovery examination. The magneto and tested by manually rotating the drive and noting spark at all six ignition leads. The magneto was shipped loose in the shipping container. The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto was bench tested and functioned normally. Image: The magneto had been removed during the initial recovery examination. The magneto had been removed during the initial recovery examination. The magneto and tested by manually rotating the drive and noting spark at four of six ignition leads. The magneto was shipped loose in the shipping container. magneto was bench tested and functioned normally. Imagneto was bench tested and functioned	

ENGINE EXAMINATION REPORT							
FILE NUME	BER:	18-370	ENGI	NE S/N:	27160-D-	2-D	PAGE 11 of 39
FUEL SYSTEM							
FUEL PUN	IP	Manufacturer: NA		P/N: NA		S/N:N	NA
Condition:	The	engine was not equip	ped with	an engine d	riven fuel pu	ımp in	this application.
CARBURE	TOR	Manufacturer: Pre	cision	P/N: 10-44	139-1	S/N:	FV2192
Condition:	carb	airframe airbox and c uretor was destroyed age consistent with fu	by impac	t forces. Me	tal floats exl	hibited	
		<image/>					

ENGINE EXAMINATION REPORT							
FILE NUMBER:	18-370	ENGINE S/N:	27160-D-2-D	PAGE 12 of 39			





ENGINE EXAMINATION REPORT18-370ENGINE S/N:2716

27160-D-2-D

PAGE 13 of 39

LUBRICATION SYSTEM

OIL SUMP

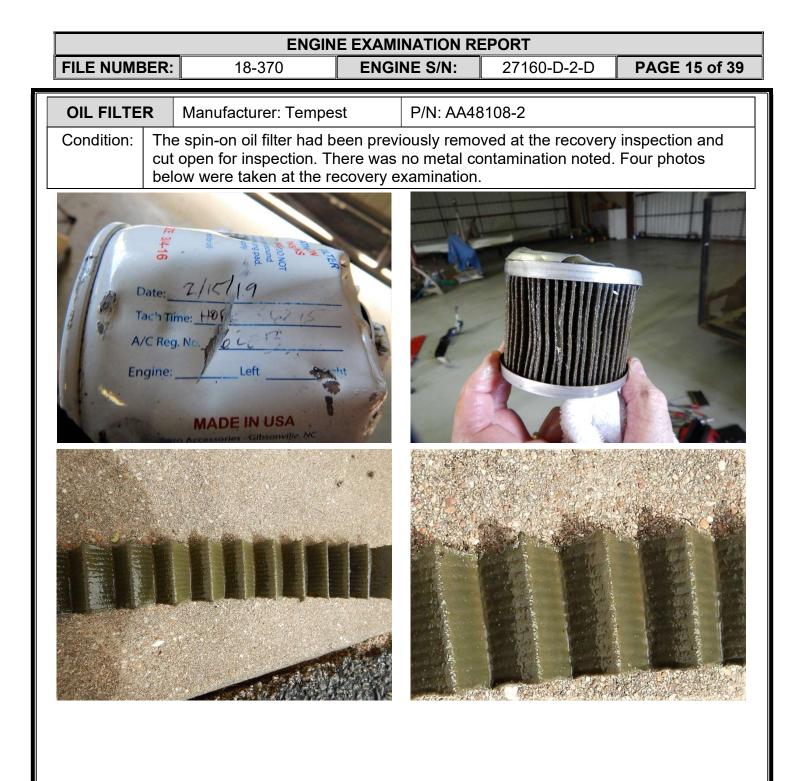
Condition:

tion: The oil sump was impact damaged. The left-forward section impact separated.





	ENGIN	IE EXAMINATION RI	EPORT	
FILE NUMBER:	18-370	ENGINE S/N:	27160-D-2-D	PAGE 14 of 39
OIL PICK-UP SCREEN				
Condition: The	e oil pick-up screen wa	s uncontaminated.		
OIL PUMP			A	
	e oil pump rotated norr	nally with no binding	or internal damage	due to hard
	ticle passage.			



		ENGIN	E EXAMI	INATION RE	EPORT		
FILE NUME	E NUMBER: 18-370 ENGINE S/N: 27160-D-2-D PAGE 16 of 39					PAGE 16 of 39	
OIL COOLERManufacturer: Niagara Thermal Products, IncP/N: 20002AS/N: E15-12882-19						15-12882-19	
Condition:	The	e oil cooler exhibited im	ipact dam	nage			
<image/>							
		ental Aerospace Techr cooler was included du					

AFTERMA OIL FILI ADAPT	TER	
FILTER ADAPTER	M/N: F&M 3?0-1	S/N:10190
OIL COOLER ADAPTER	STEVE'S AIRCRAFT	S/N: 252



 ENGINE EXAMINATION REPORT

 FILE NUMBER:
 18-370
 ENGINE S/N:
 27160-D-2-D
 PAGE 17 of 39

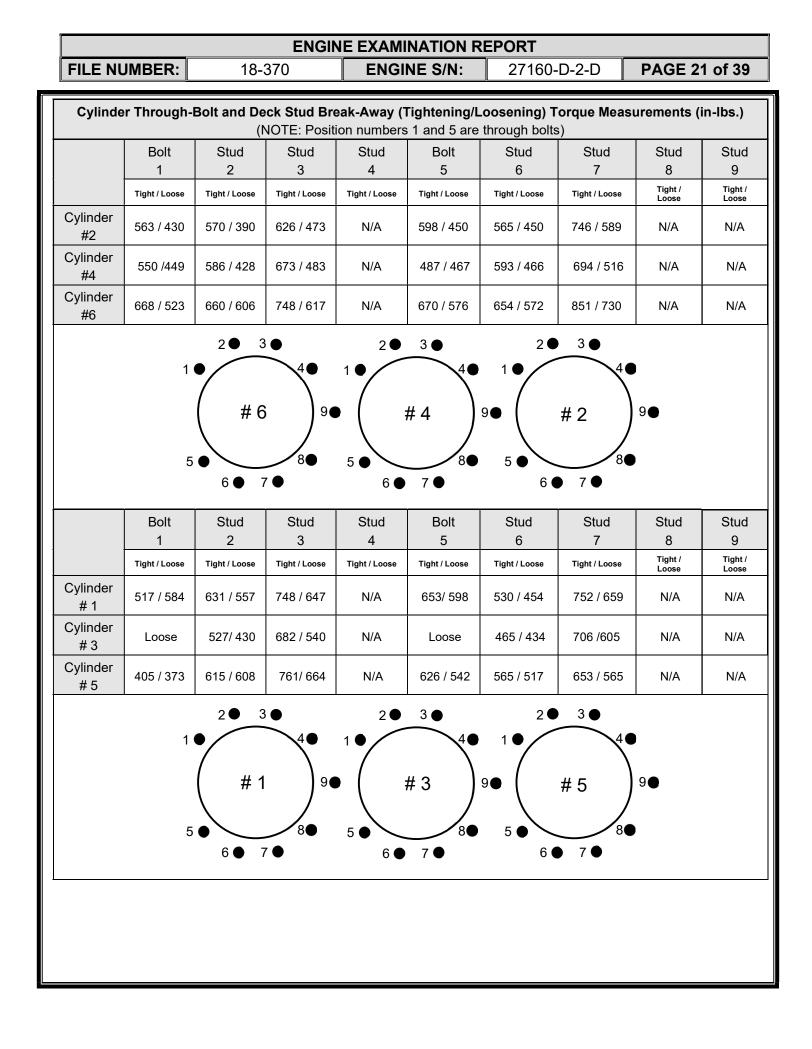


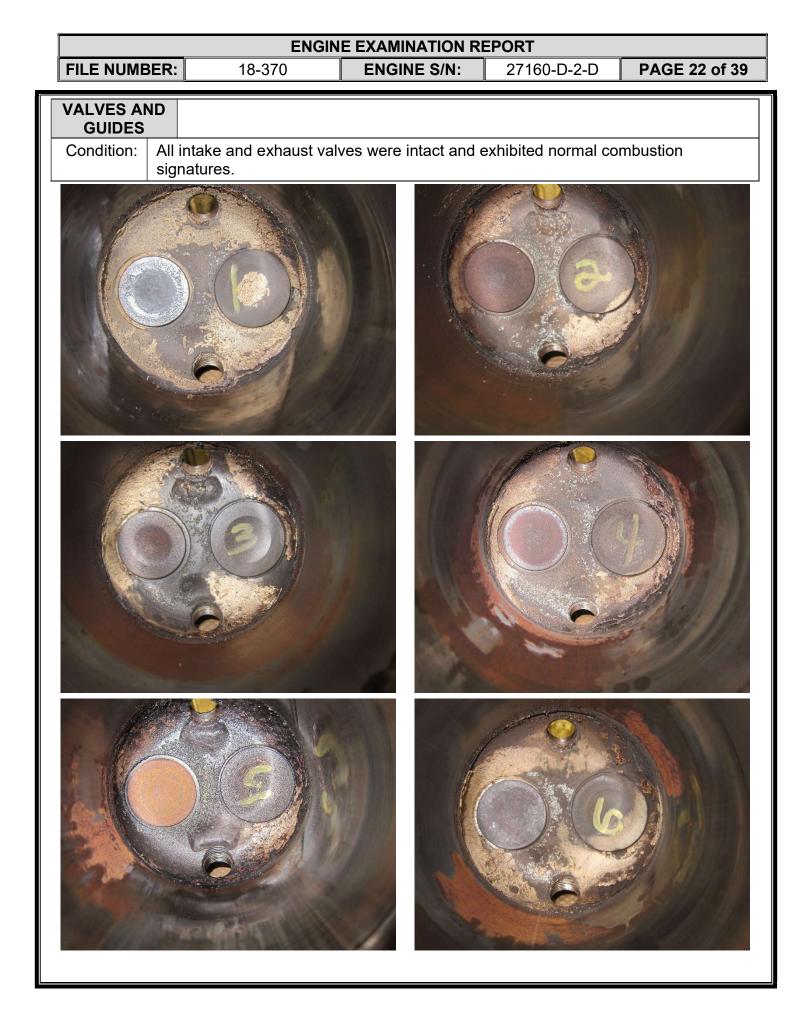
NOTE: The accident engine was equipped with an F&M aftermarket spin-on oil filter adapter model CO-300-1, serial number 10190. In addition, another aftermarket device was attached where the spinon oil filter normally would be attached to the F&M adapter. This device had a placard indicating that it was manufactured by "Steve's Aircraft", serial number 252. This adapter included two flexible oil hoses attached which bypassed to the oil cooler. It was not determined if the two unrelated STC'd appliances were joined within the scope of either STC or with permission of either STC holder.

	ENGINE EXAMINATION REPORT						
FILE NUMBER:	18-370	ENGI	NE S/N:	27160-[D-2-D	PAGE 18 of 39	
ALL CYLINDERS	WERE MANUFACTURED I		IDERS IOR AIR PART	'S AS FAA-P	MA APP	ROVED CYLINDERS	
CYLINDER #1	P/N: SA 10200-A1		S/N:20-C1	5-41895	Head [Undete	Date: ermined	
Work Orders:	B-03309		·				
Condition: No	damage was noted.						
CYLINDER #3	P/N: SA 10200-A1		S/N:20-C1	5-41893	Head [Date: Undetermined	
Work Orders:	B-03291						
Condition: Inta	ake and exhaust push r	rods and	push rod tub	es were ir	npact da	amaged.	

FILE NUMBER:		E EXAMINATION RI ENGINE S/N:	EPORT 27160-D-2-	D PAGE 19 of 39
CYLINDER #5	P/N: SA 10200-A1	S/N: Unde	etermined He	ad Date: Undetermined
Work Orders:	B03929		·	
Condition: No	damage was noted			
CYLINDER #2	P/N: SA 10200-A1	S/N:20-C	15-41889 He	ad Date: Undetermined
Work Orders:	B-03365			
Condition: No	damage was noted exc	cept impact damage	to pushrod tub	es and pushrods

FILE NUMBER:	ENGINE EXAM 18-370 ENG	INATION REP	ORT 27160-D-2-D	PAGE 20 of 39
CYLINDER #4	P/N: SA 10200-A1	S/N:20-G15	-42231 Hea	d Date: Undetermined
Work Orders:	B-03970			
Condition: No	damage was noted			
CYLINDER #6	P/N: SA 10200-A1	S/N:20-C15-	-41886 Hea	d Date: Undetermined
Work Orders:	B-03478			
Condition: No	damage was noted			





18-370

ENGINE EXAMINATION REPORT ENGINE S/N:

27160-D-2-D

PAGE 23 of 39

ROCKER ARMS AND SHAFTS

Condition:

All rocker arms and shafts were intact and moved normally when the engine was rotated manually except cylinder 2 which sustained impact damage to the intake and exhaust pushrods.





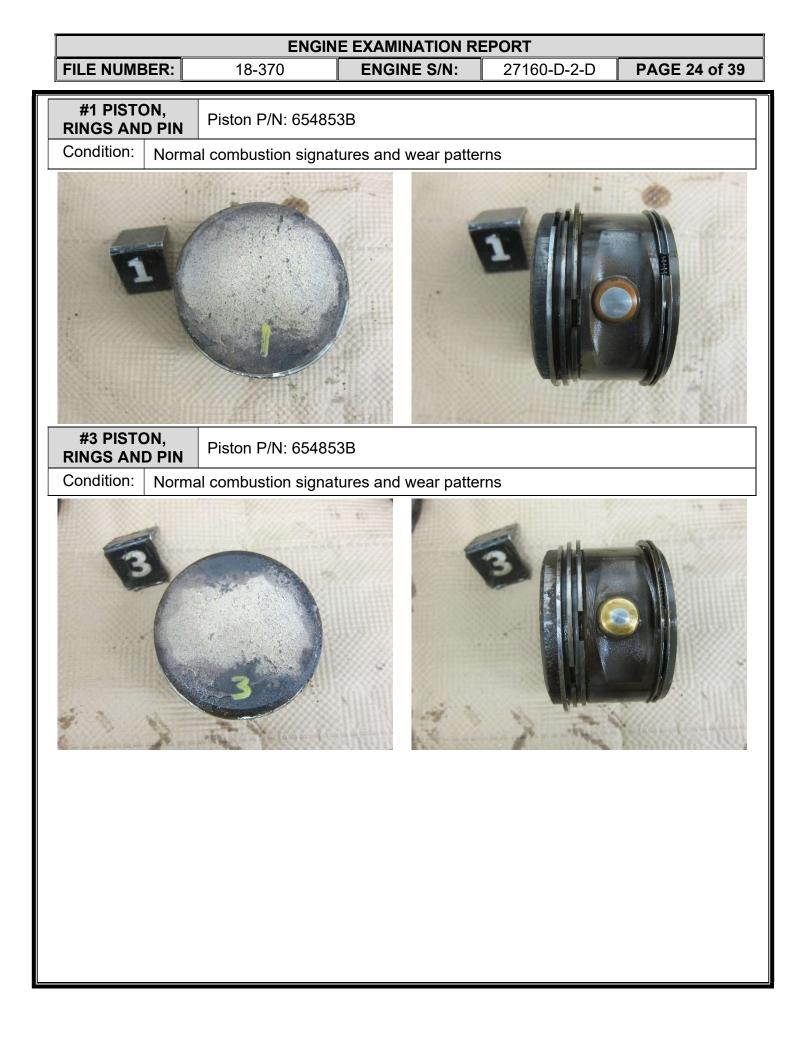


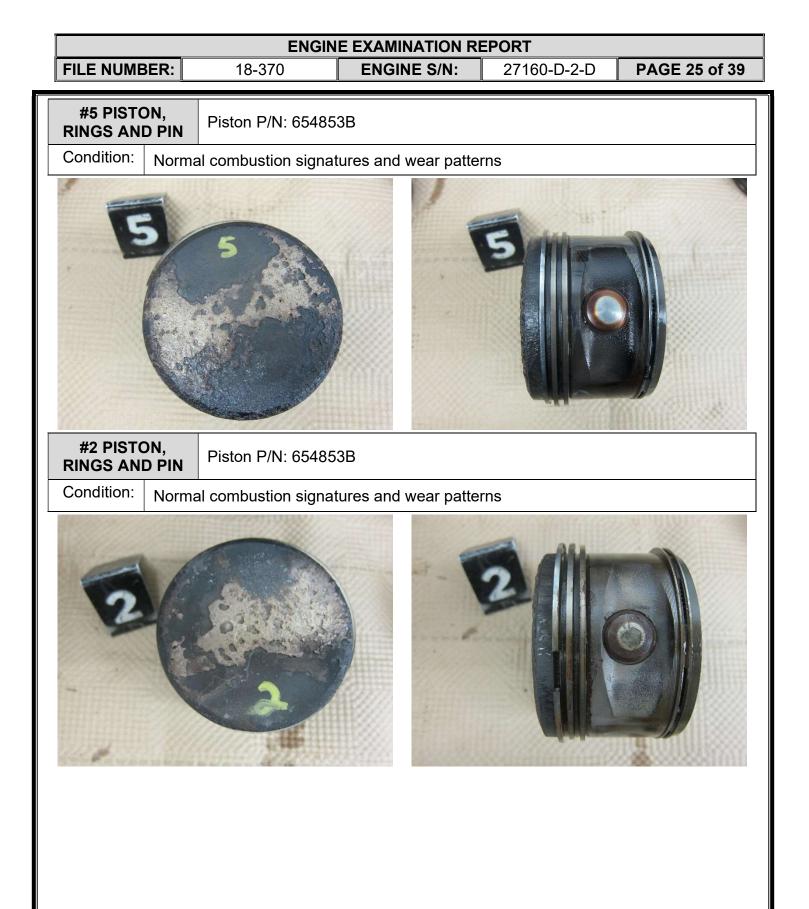




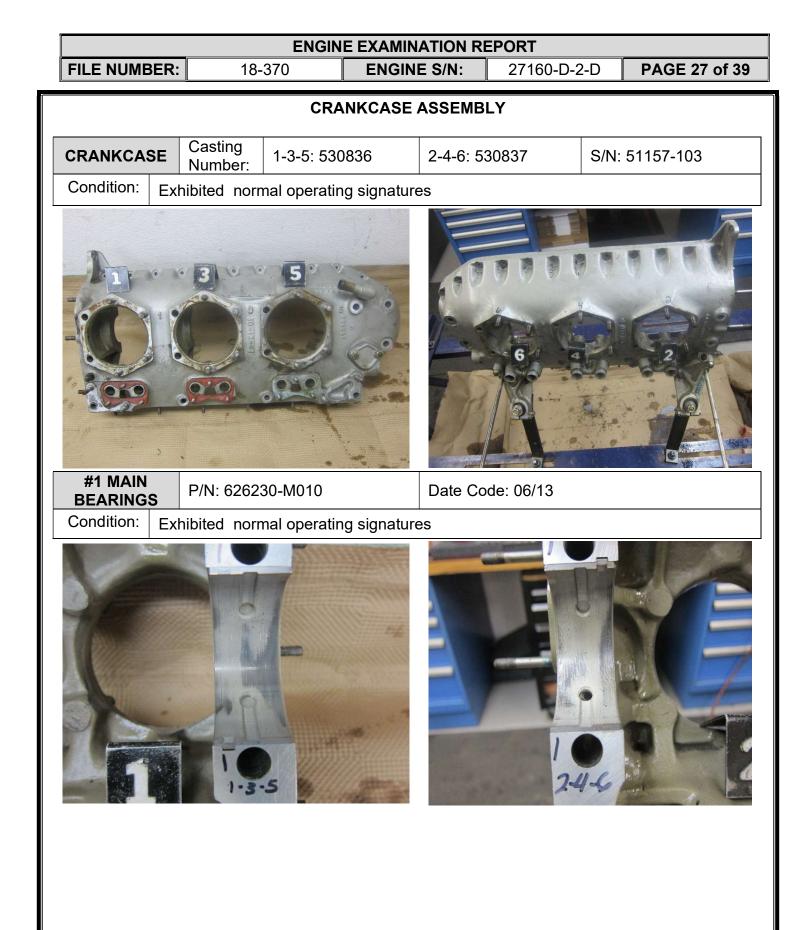


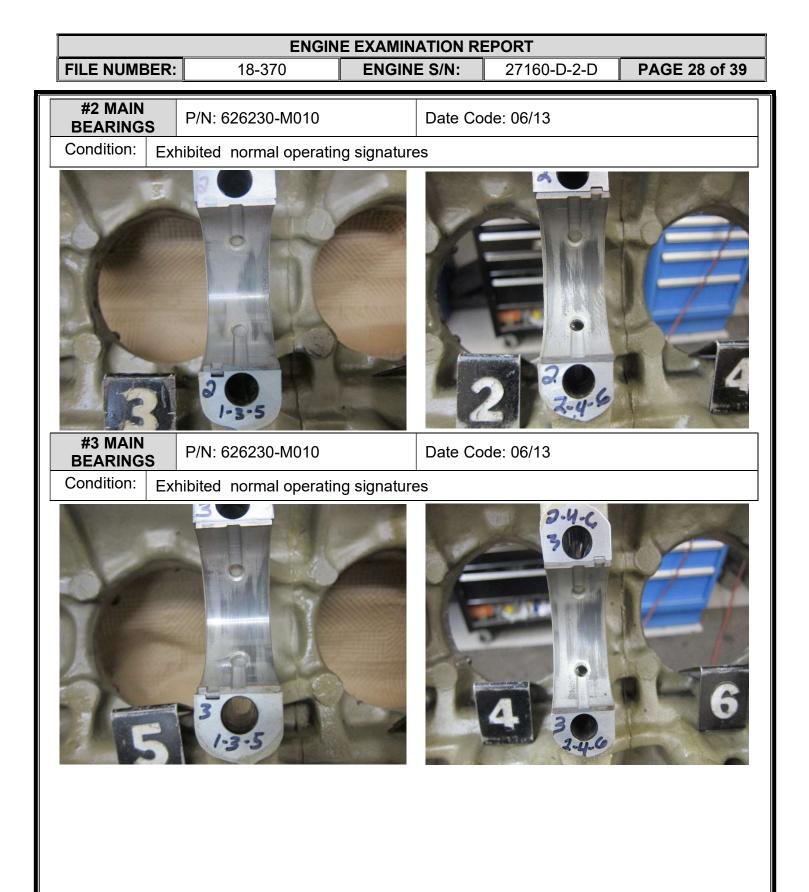


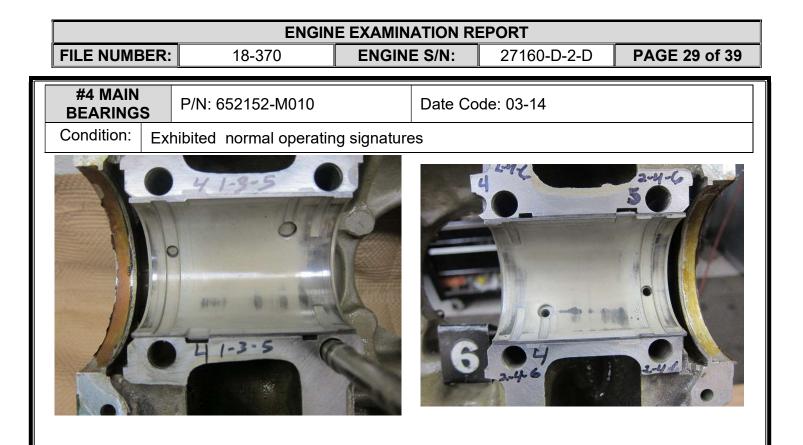


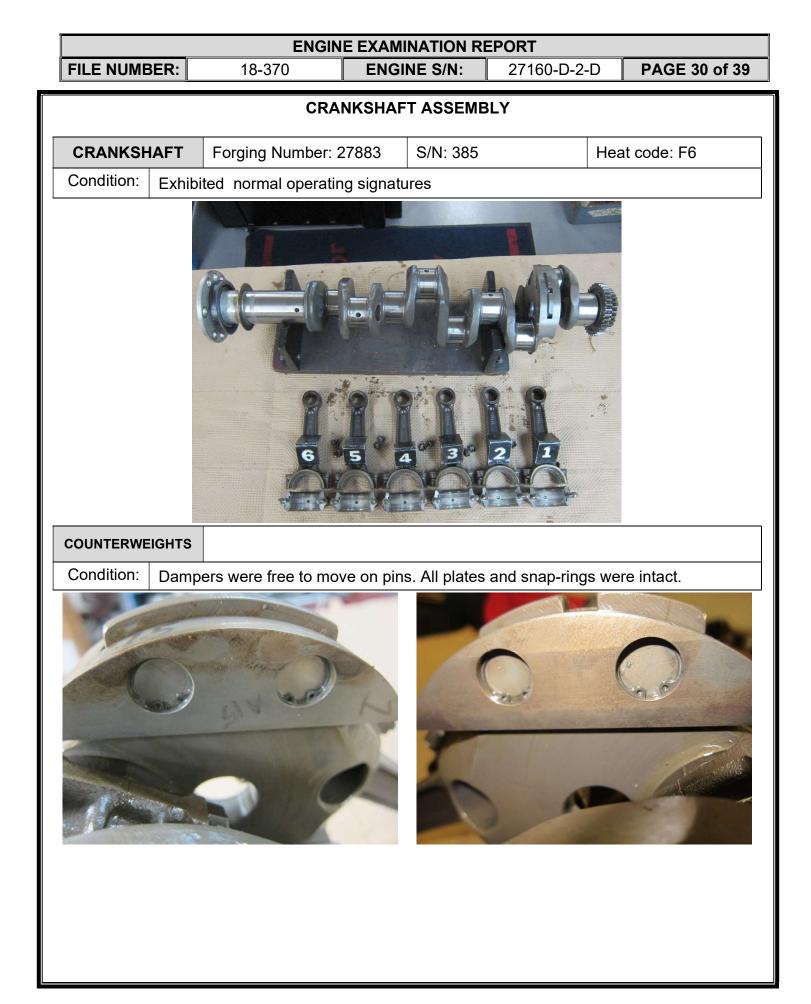












ENGINE EXAMINATION REPORT						
FILE NUME	BER:	18-370	ENGINE S/N:	27160-D-2-D	PAGE 31 of 39	
INTERN TIMIN						
Condition:	Correc gear.	t internal timing w	vas confirmed. Red arr	ow points to timing	mark on camshaft	

ENGINE EXAMINATION REPORT 18-370

ENGINE S/N: 27160-D-2-D PAGE 32 of 39

#1 CONNECTING ROD		P/N: 654795A1	Forging Number: 530186			
Condition:	Condition: Exhibited normal operating signatures					
#1 CONNEC ROD BEAF		P/N: SA639640-M010				
Condition:	Exhibi	ited normal operating signatu	ıres			
#3 CONNEC ROD	TING	P/N: 654795A1	Forging Number: 530186			
Condition:	Exhibi	ited normal operating signatu	ures			
#3 CONNEC ROD BEAF		P/N: SA639640-M010				
Condition:	Exhibi	ited normal operating signatu	Jres			
#5 CONNEC ROD	TING	P/N: 654795A1	Forging Number: 530186			
Condition:		ited normal operating signatu	Jres			
#5 CONNEC ROD BEAR		P/N: SA639640-M010				
Condition:	Exhibited normal operating signatures					
		π	Т			
#2 CONNEC ROD	TING	P/N: 654795A1	Forging Number: 530186			
Condition:		ited normal operating signatu	Jres			
#2 CONNEC ROD BEAR	_	P/N: SA639640-M010				
Condition:	Exhibi	ited normal operating signatu	Jres			
#4 CONNEC ROD	TING	P/N: 654795A1	Forging Number: 530186			
Condition:	Exhibi	ited normal operating signatu	ures			
#4 CONNEC ROD BEAF	CTING	P/N: SA639640-M010				
Condition:	Exhibi	ited normal operating signatu	Jres			

ENGINE EXAMINATION REPORT						
FILE NUMBER:	18-370	ENGINE S/N:	27160-D-2-D	PAGE 33 of 39		
#6 CONNECTING ROD	P/N: 654795A1	Forging N	lumber: 530186			
	ted normal operatin	g signatures				
#6 CONNECTING ROD BEARING	P/N: SA639640-M	010				
Condition: Exhib	ted normal operatin	g signatures				

ENGINE EXAMINATION REPORT ENGINE S/N: 2716

27160-D-2-D

PAGE 34 of 39



18-370











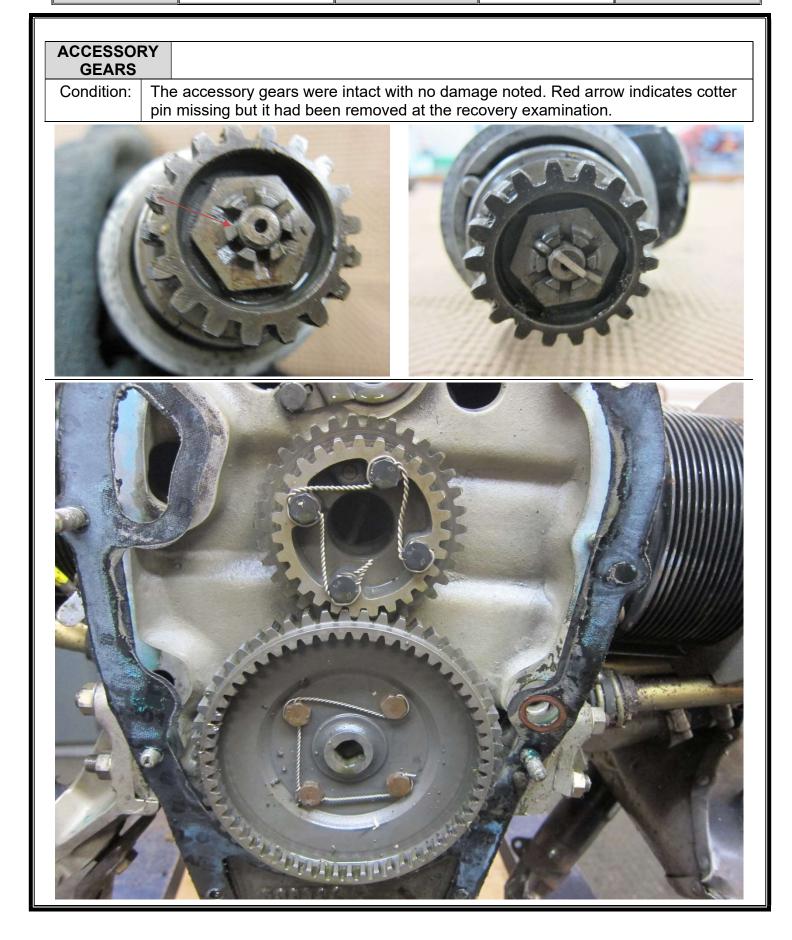
ENGINE EXAMINATION REPORT									
FILE NUMBER: 18-370 ENGINE S/N: 27160-D-2-D PAGE 35 of 39									
CAMSHAFT									
CAMSHAFT	CAMSHAFT P/N: 643068 S/N: 211706								
Condition: Ex									
LIFTERS	#1	#3	#5	#2	#4	#6			
INTAKE	530851	530851	530851	530851	530851	530851			
EXHAUST	530851	530851	530851	530851	530851	530851			
		xhibited norn	hal operating :	signatures		J			

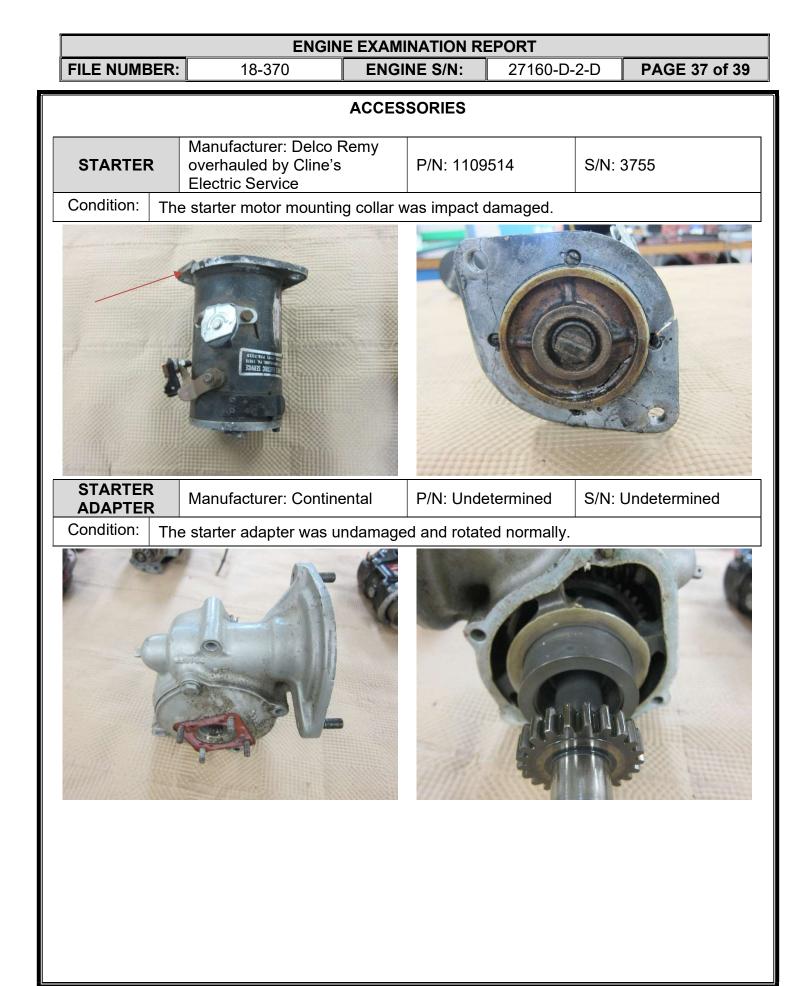
18-370

ENGINE EXAMINATION REPORT ENGINE S/N: 2716

27160-D-2-D

PAGE 36 of 39





ENGINE EXAMINATION REPORT							
FILE NUMBER:18-370ENGINE S/N:2		27160-D-	2-D	PAGE 38 of 39			
GENERATOR Manufacturer: Delco- Overhauled by Aerotech				P/N: 1101898 S/N: A798030			
Condition:							
<text></text>							

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