NATIONAL TRANSPORTATION SAFETY BOARD

Office of Aviation Safety Washington, D.C. 20594

June 12, 2020

Attachment 9 – United Airlines Airbus 319/320 Flight Manual

[Excerpt]

OPERATIONAL FACTORS

DCA20CA058



UNITED

Weather Radar Control Panels

Note: 48XX/49XX aircraft may either have one of two different weather radar control panels, as shown below.

Location: Control Pedestal



320-BUL-019

1 Radar Switch Position 1 activates the weather radar for display on the ND in any mode except PLAN.

- **Note:** Position 2 is deactivated. Inadvertently selecting this position will not turn on the radar.
- **Note:** The OFF position is the middle position; therefore, exercise care to ensure the system is off after landing since the switch is positioned either full left or full right on UA-configured aircraft.

 1
 GAIN
 MULTISCAN
 GCS
 TILT
 5

 -3
 CAL
 MAN
 AUTO
 OFF
 FAUTO
 5

 -0
 +4
 MODE
 0
 0
 0
 6

 -12
 +12
 TURB
 MAP
 5
 10
 6

 3
 OFF
 2
 0
 0
 6

 3
 OFF
 2
 0
 0
 6

 3
 OFF
 2
 0
 0
 7

 4
 SYS
 PWS
 320-BUL-6150-018

Location: Control Pedestal

The normal settings are:

- (1) MULTISCAN AUTO
- (2) GAIN CAL
- (3) SYS 1
- (4) MODE WX+T
- (5) GCS AUTO

319-320

7 AUG 19



- (6) TILT 0
- (7) PWS AUTO

MULTISCAN In AUTO the radar alternatively scans at two antenna tilt settings. The weather radar image that is displayed is the result of the stored and combined information from each beam. It also removes the weather that is not on the aircraft flight path or is not a threat to the aircraft (post convective cell). In MAN the TILT setting becomes active and must be manually adjusted.

2 GAIN CAL provides automatic gain adjustment based on altitude, geographical area, season, and time of the day.

3 SYS Turns on/off weather radar system 1 or 2. The normal setting is 1 as SYS 2 may be deactived.

MODE WX+T is the normal mode. WX operates like UA 320s, the T feature displays turbulence areas within 40 NM in magenta.

GCS Is ground clutter suppression and removes ground returns from the ND. GCS is not available when using manual modes.

6 TILT Zero indicates the horizon reference, as provided by the ADIRS. If MULTISCAN is set to MAN, utilize the TILT knob per normal SOP as stated in the Flight Manual.

PWS Operates like UA 320s.

319-320

UNITED

24 JAN 20

	Audio control panel (ACP) As required		
	Note:	On VHF 1, in the high frequency band (e.g., 135.97), the an increase in background noise due to electrical interfe ground or in flight; use VHF 2 if necessary. A maintenan is not required.	ere may be rence on the nce report
	Thrust le	evers	Idle
•	WX RADAR Off Verify gain is in AUTO position and the image selector is in WX/TURB.		
	PWS sw	<i>r</i> itch	Off
•	Flight de Confiri	eck door open light ns the flight deck door locking system is powered.	Verify on
	Gravity	gear extension handle	Stowed
	Passen Check	ger address system	Check
	ACARS The A time d indepe Procee Procee	FMGC Initia CARS/FMGC may be initialized by either pilot at any uring the flight deck preparation; the other pilot endently verifies all entries are correct. See Supplement dures>Communications/Navigation>Normal FMGC dures>FMGC Initialization.	ılize/program ary
	Noto	Independent verification means that verification is account	mplished

Note: Independent verification means that verification is accomplished without assistance or influence from the other pilot.

FINAL PREFLIGHT PROCEDURE

CAUTION: The only confirmation the parking brake is set is an indication of brake pressure on the BRAKES and ACCU PRESS indicators.

CAUTION: The parking brake should be set if the ramp is icy or if either wind speed or gusts of 40 knots or higher are anticipated.

 [C] Parking brake Set, pressure normal Parking brake must be set prior to beginning the Preflight Checklist.

The brake accumulator is normally charged in the green range (approximately 3000 PSI). If the pressure drop is greater than 300 PSI, recharge the accumulator.

WARNING: If a departure procedure requires holding after takeoff (e.g., to climb to MEA/MCA), the holding must be manually loaded, even on line-selectable departures.

FLIGHT MANUAL

3.20.24 Normals Preflight

319-320

UNITED

■ [C/F] ATC clearance Obta	in
Obtain clearance and compare routing with Flight Plan routing. Make any necessary FMGC corrections.	
■ [C, F] FCU Veri	fy
Speed window Manage Indicated by dashes.	эd
 Heading window	וg וו
Altitude window	et
Vertical speed window Dashe	эd
[C, F] PFD Check After the FCU is set, verify autoflight modes for departure:	ck
Climb mode Arme	ed
■ Lateral mode	ed
■ FCU altitude Displaye	ed
[C, F] EGPWS, radar displays As required If use of weather radar is not required, both pilots should select the TERR ON ND or TERR switches on. If weather radar is required, either or both pilots may deselect the TERR ON ND or TERR switch. If both pilots are in the weather radar display mode, the terrain display pop-up feature is still available.) I
 [C] Pilot and flight attendant briefings Complete For review of applicable briefing items, see FOM>Operating Information> Crew Briefings and Departure Briefing. 	te
Preflight Checklist Accomplis	sh
[C] APU Star Start the APU as close to pushback as possible.	art
CAUTION: Select the EXT PWR switch off prior to disconnecting ground power to avoid power surges.	d
■ [C] External power)ff
Ensure the GVR GND GTE Switch ON light remains indfiniated.	

3.80.4 Normals Before Takeoff	319-320 19 OCT 18	
Thrust reg	duction/acceleration altitude	220
 Flap setti if required 	ng. Advise the Captain and	l change the flap handle setting,
FLEX TO	temperature, if appropriate	3
 EO accel 	eration altitude and procedu	ure
■ [F] FCU Ensure the FCU is s	et as follows:	Check
 SPD window Verify the sp PFD speed 	beed window displays dash scale.	es and V2 is displayed on the
 HDG/NAV If departure heading. If I NAV automa 	is to be flown using heading NAV is to be used, the head atically engages at 30 feet o	g, set assigned heading or runway Jing window displays dashes and on departure.
Verify the N heading.	D runway display is closely	aligned to the actual runway
Note: If a l autor to de and the e	neading is preset on the F0 matically engages at 30 fee parture and an engine failu wind adjusted, to track the engine failure procedure.	CU, RWY TRK guidance et. If a heading is not preset prior pre occurs, HDG must be engaged extended centerline or to follow
 ALT Verify the construction FCU altitude 	prrect altitude for the depart window and on the PFD a	ure procedure is displayed in the altitude scale.
■ [F] FMGC (F-PL	۸)	Verify departure programming
■ [F] PWS		AUTO
■ [C, F] Weather ra	ıdar	As required
 [C, F] Radar/terra Select weather feature is still a 	ain radar or terrain as appropri vailable if both pilots have v	As required iate. The terrain display pop-up weather radar displayed.
[C] Before Takeo The Captain ma the line as appr	ff Checklist ay call for the Before Takeo ropriate.	Announce off Checklist in its entirety or to
When the Captain	calls for the Before Takeo	off Checklist:
[F] Final weights Respond with the Example: First	ne Final Weight Manifest G Officer: "133.1, minus 501	(GW), (TOW CHG) W and TOW change. "

UNITED 💹

 If thrust variations become excessive, disengage the autothrust system and adjust thrust levers as necessary.

Note: If any overspeed occurs, a maintenance report is required.

WEATHER RADAR OPERATION

GENERAL

While penetration of widespread areas of light stratiform rain is possible, caution should be used when flying through extended areas of moderate rain due to the possibility of embedded thunderstorms. Radar equipment is to be used as an aid *to avoid severe weather, not as a penetration tool.*

RADAR ATTENUATION CHARACTERISTICS

In widespread *light* rain, the radar displays *green* to a distance of 70 NM with the remainder of the light rain not displayed due to lower reflectivity. In widespread *moderate* rain, the radar displays *yellow* to a distance of 45 NM, then turns green for an additional 10 NM with the remainder of the moderate rain not displayed. Even though the reflectivity increases in *moderate* rain, the distance precipitation is displayed decreases due to radar attenuation.

A storm cell 3 NM wide and 3 NM tall with heavy rain surrounded by clear air displays on radar when within 230 NM of the aircraft. The same storm cell embedded in widespread light rain displays on radar when within 160 NM of the aircraft. Embedded in widespread moderate rain, the same storm cell displays when within 80 NM of the aircraft. These distances relate to a very intense storm; storms of less intensity are not detectable until closer to the aircraft. As the cell intrudes the colored area of the radar display, it casts some shadow. Whenever flying through widespread rain, using primary tilt position and searching for shadows are very important.

- **Note:** The weather radar may occasionally display red and black wedges or sharp sectors similar to a radar test pattern. This can be disregarded if these disappear in two to three radar sweeps. If the images persist, report the defect to Maintenance.
- **Note:** All grids for spacial comparisons only, not on actual displays.



SYSTEM TEST

WARNING: Do not operate the radar when personnel are within 15 feet of the radar antenna or while fueling or defueling.

On power-up and during final approach, the aircraft runs a system self-test. The CFDS test emits a very brief radar pulse to verify transmitter operation.

To check the radar when clear of the ramp area:

	System switch On
	Weather radar mode selector Weather/turbulence
•	Tilt knob Adjust Tilt the antenna down until ground returns display on the NDs. Verify no fault messages. There is no test pattern display on the ND during this functional check.
To	check the receiver, antenna, and displays, turn the radar on during taxi when

Io check the receiver, antenna, and displays, turn the radar on during taxi when clear of the ramp area. Tilt the antenna down until ground returns display on the NDs. Verify no fault messages.

UNITED 💹

SYSTEM OPERATION

■ Weather radar mode selector Weather/turbulence Provides normal weather radar operation plus Doppler detection of turbulence by measuring the velocity of rain droplets moving parallel to the aircraft flight path. Turbulence may be displayed in light rain or entirely outside the precipitation area. Although Doppler cannot detect clear air turbulence, it can detect turbulence in the presence of rain lighter than that displayed in the weather-only mode. Turbulence returns and Level 4 and higher storms are displayed in magenta. These areas can be very dangerous. Avoid echoes in accordance with the FOM. ND brightness control knobs Adjust Balance the radar returns and ND. ■ ND range selector As required As weather echoes move closer to the aircraft, selecting shorter ranges yields best results. Select progressively shorter ranges during a descent. The longer ranges should be monitored periodically to detect distant storm cells. When different ranges are selected, tilt adjustment is necessary. See Weather Radar - Tilt Management, this section. Inflight radar check..... Accomplish For enroute verification of radar operation, adjust tilt for maximum distance ground mapping. Consider radar accurate if able to display ground returns to the following distances: Above 10,000 feet AGL to at least 100 NM or Above 20,000 feet AGL to at least 150 NM SHUTDOWN



25 JAN 19

UNITED

WEATHER RADAR - TILT MANAGEMENT

Range and azimuth depict the organization of the weather system, the relative position of echoes in a line, and an avoidance path. Measuring the height, gradient, reflectivity, Doppler return, and shape of cells depicts the degree of hazard. Note that shapes are often fleeting; they may appear only at certain tilt settings or disappear after a single sweep. Failure to spot a particular shape or feature does not necessarily mean the weather is nonhazardous. Also, size, intensity, and echo tops are not always clear-cut indicators of hazard. See FOM, Adverse Weather chapter for additional information.

TAKEOFF

Scan for storms in the departure path at the 10 NM range and move the tilt slowly from +15° to +4°. If the 10 NM range is clear, repeat for the 20 NM and 40 NM ranges. Look for irregular shapes and crescent-shaped storms. Normal antenna tilt position during and after takeoff is +4°.

PRIMARY TILT POSITION

While in flight, move the tilt downward until ground return is displayed.



Raise the beam and observe the ground return moving farther out in range. A radar sweep takes about 4 seconds, so make the adjustments slowly. Stop adjusting the tilt when the ground return stripe moves to the outer edge of the indicator but can still be seen. At the higher ranges, the ground return appears as a sprinkle of ground return spots.



The primary tilt position creates a zone where no storm can go undetected as the radar beam is toward the most reflective section of a storm.



Any significant change of altitude requires a new primary tilt position; on descent, more upward tilt is needed; on climb, more downward tilt. Tilt control is independent of radar stabilization. The stabilization function maintains the antenna position parallel to the earth during aircraft maneuvering. However, tilt must be adjusted for significant altitude changes and when a different range is selected.



OVERSCANNING

Above 25,000 feet, as the aircraft approaches storms, the storms may become dangerously understated or disappear completely from the display. This is due to overscanning the most reflective part of the storm.



If an echo disappears from the scope or becomes understated, due to overscanning, reduce the range and repeat the primary tilt position procedure.

Special Convective Conditions

Storms in regions of unstable air (i.e., Tropics and Subtropics) may also be overscanned. To aid in detection of dry top cells:

- Position mode selector in WX/TURB.
- Position GAIN selector in AUTO.
- Keep flight deck lights low to optimize visual detection.
- Use 40 NM range; keep at least one pilot on this range.
- Use the primary tilt position for the 40 NM range.

LOOKING FOR SHADOWS

Tilt the antenna down to a position that paints ground returns on the outer one-third to one-half of the display:

• If weather can be seen behind a cell, assume it is at least one level more hazardous than depicted, due to attenuation.



If no ground return or other echoes can be seen behind a cell, a radar shadow is indicated. This shadow is an important indicator of hazardous weather; avoid all shadowed areas. If it is an extreme storm, with only its leading edge depicted, it appears as a thin line or band of precipitation in the shape of a crescent, with the horns pointed away from the aircraft. A black area behind it (the radar shadow) is the most dangerous part of the storm. An isolated mountain echo appears as a crescent-shaped cell that casts a shadow, a city will not. Widespread areas of water (e.g., lakes, rivers) may also appear as shadows.



319-320

25 JAN 19



PARALLEL BEAM POSITION

This tilt position results in the bottom of the beam being parallel to the earth, and echoes depicted (weather and/or high terrain) are objects that intrude through the current altitude. It should be selected frequently, particularly when operating below 15,000 feet AGL, but for only a few sweeps. To find the parallel beam position:

Adjust the tilt to paint the ground stripe at a range equal to current AGL altitude (e.g., flying at FL 370 over sea-level terrain, paint it at 37 NM; at FL 370 over Denver, paint it at approximately 32 NM):







UNITED

4.10.25 Supplementary Procedures Adverse Weather Operations

EVALUATING CELL HEIGHT

When used for a few sweeps, this formula gives a quick estimate of radar height of the echo and may be used at any range. This is useful **only** for determining the relative severity of the storm and should not be used to attempt overflight in IMC. Radar height is **not** the storm height, due to low reflectivity in the cell tops. Assume the actual storm top is at least 20% above the measured radar top. **Severe turbulence can exist well above the actual top.**

 Cell height Evaluate Start from the parallel beam position.



. Increase

Increase until echo disappears.



 Cell height evaluation..... Continue The radar top is distance to echo x 100 x degrees of tilt change from parallel beam (e.g., an echo at 40 NM disappears with a 4° tilt increase: 40 x 100 x 4° = *radar top*, which is at least 16,000 feet above current flight level; *actual top* is 120% of 16,000 feet or at least 19,200 feet above current flight level).