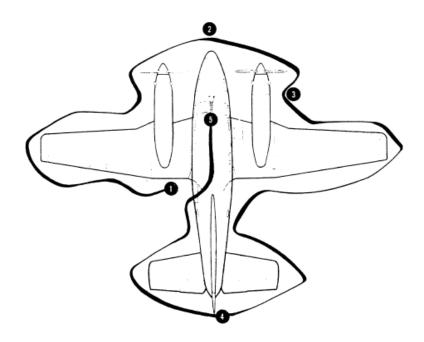
# Factual Report – Attachment 16 Beech A90 Normal Procedures

# OPERATIONAL FACTORS

WPR19MA177

# SECTION V **NORMAL PROCEDURES**



# PREFLIGHT CHECK

# 1 LEFT WING

- General Condition CHECK
- Flaps CHECK
- 3. Aileron and Tab - CHECK
- Lights CHECK 4.
- 5.
- Stall Warning CHECK Deicer Boots CHECK Pitot Cover REMOVE
- Auxiliary Fuel Tank CHECK
- Drain 4 Fuel Sumps:
  - Just outboard of Nacelle
    - Transfer Pump
    - Main Boost Pump
  - Inside Wheel Well
- a. Inside wheel well
  Landing Gear, Strut, Brake and Wheel Well CHECK
  Tie-Down and Chocks REMOVE
  Supercharger Gear Box CHECK OIL LEVEL
  Propeller CHECK
  OIL LEVEL
  Propeller CHECK 10.
- 11.
- 12.
- 13.
- Engine Air and Oil Cooler Intakes CLEAR; Inertial Separator Vane -RETRACTED
- Engine Air Intake Lip Boot CHECK
- Engine Oil CHECK QUANTITY, CAP SECURE 16.
- Firewall Fuel Filter DRAIN

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### LEFT WING (Continued)

#### NOTE

The fuel filter drain cock may be actuated while the top engine cowl door is open for the oil check. Adjacent to the door, a placard (which reads: FUEL STRAINER DRAIN - RELEASE INSIDE DOOR) references the location.

- 18. Cowling CHECK DOORS SECURE AND GENERAL CONDITION
- Main Fuel Tank CHECK
- Heat Exchanger Inlet CLEAR
- Access Panels CHECK
- 2 NOSE SECTION
  - Ram Air Inlet CLEAR
  - Access Panels SECURE
  - 3. Air Conditioner Duct CLEAR
  - 4. Nose Gear, Strut, and Wheel Well CHECK
- 3 RIGHT WING
  - Access Panels CHECK
  - Propeller CHECK
  - Engine Air and Oil Cooler Intakes CLEAR; Inertial Separator Vane -RETRACTED
  - 4. Engine Air Intake Lip Boot CHECK
  - 5. Engine Oil CHECK QUANTITY, CAP SECURE
  - Firewall Fuel Filter DRAIN
  - Cowling CHECK DOORS SECURE AND GENERAL CONDITION
  - Main Fuel Tank CHECK
  - Drain 4 Fuel Sumps:
    - a. Just Outboard of Nacelle
    - b. Transfer Pump
    - c. Main Boost Pump
    - Inside Wheel Well
  - 10. Landing Gear, Strut, Brake and Wheel Well CHECK
  - Tie-Down and Chocks REMOVE
  - 12. Auxiliary Fuel Tank CHECK
  - Pitot Cover REMOVE
  - Deicer Boot CHECK
     Lights CHECK
  - 16. Aileron CHECK
  - 17. Flaps CHECK
  - General Condition CHECK
- 4 TAIL SECTION
  - General Condition CHECK
  - Oxygen Door SECURE
  - Access Panels SECURE
  - 4. Static Ports CLEAR
  - Deicer Boots CHECK
  - Control Surfaces and Tabs CHECK
  - Emergency Locator Transmitter ARMED

# BEFORE STARTING ENGINES

- Exterior Inspection COMPLETED
- Cabin Door LOCKED
- 3. Baggage SECURE; Weight and C.G. CHECKED

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#### BEFORE STARTING ENGINES (Continued)

- 4. Emergency Exit LATCHED
- 5. Seat Belts and Shoulder Harness SECURE
- 6. Control Locks REMOVED: Flight Controls FREE
- 7. Brakes SET
- 8. Landing Gear Handle DOWN
- 9. Oxygen Pressure 1800 psi at 21°C
- 10. Emergency Static Air Valve CLOSED
- 11. Power Levers IDLE
- 12. Propeller Controls FULL FORWARD
- 13. Condition Levers CUT-OFF
- 14. All Switches OFF
- 15. Circuit Breakers IN
- 16. Crossfeed OPEN (check FUEL CROSSFEED light ON), then CLOSED
- 17. Fuel Firewall Valves CHECKED AND OPEN (listen for operation)
- 18. Boost Pumps ON (listen for operation)
- 19. Battery Switch ON
- 20. Voltmeters CHECK (no voltage indicates current limiter out)

#### CA UTION

If battery voltage indicates less than 20 volts, battery must be recharged, or replaced with a battery indicating a charge of at least 20 volts, before using auxiliary power.

When an auxiliary power source is used, it must be of the negatively grounded type. If the polarity of the source is unknown, turn all switches OFF. Connect the auxiliary power source and observe the airplane voltmeter for proper indication (28.5 volts).

The battery switch must be ON when starting engines with auxiliary power and generators should be OFF until auxiliary power has been disconnected.

After second engine has been started, disconnect the auxiliary power source and secure the access door.

- 21. Annunciator Panel and Warning Lights CHECK
- Transfer Pumps ON (listen for operation), then OFF. If either or both pumps fail to operate, press the Transfer Test Switch and monitor NO FUEL TRANSFER light,
- 23. Fuel Quantity CHECKED

#### ENGINE START

- 1. Right Ignition and Engine start switch ON
- Right Condition Lever LOW IDLE (after N<sub>1</sub> rpm stabilizes for 5 seconds: 12% minimum)
- 3. ITT and N<sub>I</sub> MONITOR (1090°C maximum)
- 4. Right Ignition and Engine Start Switch STOP (at 50% N1, or above)
- Right Engine ADJUST N<sub>1</sub> 10% ABOVE IDLE, MINIMUM (15% above IDLE or to the HI IDLE position on airplanes with 325 ampere current limiters)
- Right Generator ON (BATTERY CONDITION CHECK Refer to NICKEL-CAD-MIUM BATTERY CONDITION CHECK)
- Right Engine Oil Pressure CHECK (Right Propeller unfeathered indicates oil pressure)
- 8. Left Ignition and Engine Start Switch ON
- Left Condition Lever LOW IDLE (after N<sub>1</sub> rpm stabilizes for 5 seconds 12% minimum)
- 10. ITT and N<sub>1</sub> MONITOR (1090°C maximum)
- 11. Left Ignition and Engine Start Switch STOP (at 50% N1 or above)

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# **ENGINE START (Continued)**

- 12. Left Generator ON
- 13. Inverter ON
- 14. Right and left oil pressure and fuel pressure CHECK by gage pressure.
- 15. Right N1 REDUCE TO IDLE

#### CA UTION

If no ITT rise is observed within 10 seconds after moving the Condition Lever to LOW IDLE, move the Condition Lever to cutoff and starter switch to STOP. Allow a 30 second delay to drain fuel, then motor the engine by placing the starter switch in the STARTER ONLY position.

If, for any reason, a starting attempt is discontinued, the entire starting sequence must be repeated after allowing the engine to come to a complete stop.

#### **ENGINE CLEARING PROCEDURE**

- 1. Condition Lever IDLE CUT-OFF
- 2. Ignition and Engine Start Switch STOP
- 3. Battery Switch ON
- 4. Fuel Boost Pump Switch ON
- 5. Ignition and Engine Start Switch STARTER ONLY (for a minimum of 15 seconds)

#### CAUTION

Do not exceed starter time limits. See Section IV.

- 6. Ignition and Engine Start Switch STOP
- 7. Fuel Boost Pump Switch OFF

# AFTER STARTING AND TAXI

- 1. Radios ON
- 2. Cabin Temperature SET
- 3. Fuel Control Heat ON
- 4. Cabin Sign ON
- 5. Lights AS REQUIRED
- Transfer Pumps ON
   Crossfeed Switch AUTO
- 8. Brakes CHECKED

# NOTE

For taxi speed control, reversing propellers may be used in the Beta Range. Beta range is defined as the control range between the idle stop and the point at which N<sub>1</sub> rpm begins to increase for reverse power. In this range, only blade angle is changed. This mode of control may be used without incurring propeller blade erosion

# BEFORE TAKE-OFF

- 1. Boost Pumps and Auto Crossfeed TEST
  - a. Crossfeed Valve AUTO
  - b. Left Boost Pump OFF. Annunciator Panel should read:
    - (1) L.H. BOOST FAIL
  - (2) FUEL CROSSFEED
  - c. Left Boost Pump ON
  - d. Crossfeed CLOSED, THEN AUTO

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#### **BEFORE TAKE-OFF (Continued)**

- e. Right Boost Pump OFF. Annunciator Panel should read:
  - (1) R.H. BOOST FAIL
  - (2) FUEL CROSSFEED
- f. Right Boost Pump ON g. Crossfeed CLOSED, then AUTO
- 2. Loadmeters and Generators CHECK
- 3. Propeller Synchronization OFF
- 4. Electric Elevator Trim ON (CHECK)
- 5. Trim Tabs SET
- 6. Flaps CHECKED AND SET
- 7. Pressurization TEST:
  - a. Cabin Press. Dump Valve Switch SET TO CABIN PRESSURE
  - b. Cabin Altitude SET BELOW FIELD ELEVATION (Approx. 1000 feet)
  - c. Pressurization Test Switch ACTUATE (Monitor cabin climb indicator for descent)
  - d. Cabin Altitude SET FIELD ELEVATION PLUS 500 FEET
- e. Rate Control AS REQUIRED
- 8. Autopilot CHECK, then OFF
- 9. Overspeed Governors TEST:
  - a. Propeller Control FULL INCREASE RPM
  - b. Power Levers BELOW 1900 RPM
  - c. Overspeed Governor Test Switch ON (Hold)
  - d. Power Lever INCREASE (CHECK GOVERNING 1900 2100 RPM)

#### NOTE

Observe maximum ITT and torque limits.

- e. Power Lever REDUCE TO 1900 RPM
- f. Overspeed Governor Test Switches RELEASE
- \*10. Primary Governors (all propellers) CHECK (exercise at 1900 rpm)
- 11. Engine Inertial Separator Vanes CHECK
- \*12. Secondary Low Pitch (Flight Idle) Stops TEST:
  - a. Condition Levers HIGH IDLE
  - b. Power Levers IDLE (Read propellers rpm)
  - c. Prop Test Switches HOLD TO SECONDARY IDLE STOP TEST
  - d. Power Levers ALIGN AFT EDGE WITH TOP OF BETA RANGE MARKS
  - e. Secondary Low Pitch (Flight Idle) Lights CHECK ON
  - f. RPM CHECK STABILIZED AT 210 ± 40 ABOVE RPM IN STEP "b"
  - g. Prop Test Switch RELEASE
  - h. RPM CHECK (MUST INCREASE ABOVE STEP "f")
  - i. Power Levers IDLE

### CAUTION

Do not force the Power Levers into the FULL REVERSE position with the Secondary Idle Stop Test switches ON.

- 13. Instrument Vacuum and Deice Pressure CHECK (HI-IDLE)
- 14. Autofeather CHECK then ARM
- 15. Propeller Feathering (manual) CHECK (LOW IDLE)
- 16. Flight and Engine Instruments CHECK (Oil temperature must be above the minimum shown in chart (page 4-2) to preclude ice formation in the fuel control)
- 17. Ice Protection AS REQUIRED (refer to Normal Procedure Section for Icing Flight)

#### WARNING

Both boost pumps must be operative for take-off.

\*May be omitted for quick turn around at pilot's discretion.

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# NORMAL TAKE-OFF

Check ITT and engine torque readings while applying power. Remember that increased airspeed will cause torque and ITT to increase. Accelerate to 91 knots before rotating, and then to 100 knots before climb. Retract the landing gear before reaching 130 knots.

#### CLIMB

- 1. Climb Power SET (Observe maximum ITT, torque, and N<sub>1</sub> rpm limits.)
- 2. Propeller 2000 RPM
- 3. Propeller Synchronizer ON
- 4. Pressurization Schedule CHECK
- 5. Engine Instruments CHECKED
- 6. Cabin Sign AS REQUIRED

## CRUISE

#### WARNING

Do not lift Power Levers in flight.

- 1. Cruise Power SET
- 2. Engine Instruments CHECKED
- 3. Fuel System CHECK (For fuel leveling, use the following procedure):
  - a. Both Boost Pumps ON
  - b. Crossfeed OPEN
  - c. Boost Pump (low tank) OFF (Check crossfeed light and fuel pressure)

#### NOTE

Turbulent air penetration speed: 161 kts.

#### DESCENT

1. Pressurization - SET

#### NOTE

Set cabin altitude to show landing field elevation plus 500 feet.

2. Altimeter - SET

## CAUTION

- If either of the SECONDARY FLIGHT IDLE STOP (SECONDARY LOW PITCH STOP) warning lights become illuminated in flight, DO NOT attempt propeller reversal upon landing.
- 3. Cabin Sign AS REQUIRED
- Windshield Anti-Ice AS REQUIRED (Turn ON well before descent into warm, moist air, to aid in defogging.)

# LANDING

5-6

Maximum Demonstrated Crosswind . . . . . 25 knots

- 1. Cabin Sign ON
- 2. Propeller Synchronization OFF
- 3. Flaps AS REQUIRED
- 4. Landing Gear DOWN
- 5. Landing and Taxi Lights AS REQUIRED
- 6. Pressurization CHECK
- 7. Power Levers BETA RANGE (AS REQUIRED AFTER TOUCHDOWN)

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#### MAXIMUM REVERSE POWER LANDING

#### CAUTION

To insure constant reversing characteristics, the Propeller Controls must be in FULL INCREASE RPM position.

- 1. Condition Levers HIGH IDLE.
- Propeller Controls FULL INCREASE RPM
- 3. Power Levers LIFT AND REVERSE AFTER TOUCHDOWN
- Condition Levers LOW IDLE

# CAUTION

If possible, propellers should be moved out of reverse at approximately 40 knots, to minimize propeller blade erosion. Care must be exercised when reversing on runways with loose sand or dust on the surface. Flying gravel will damage propeller blades and dust may impair the pilot's forward visibility at low aircraft speeds.

#### AFTER LANDING

- Flaps UP
- Landing and Taxi Lights AS REQUIRED
- Auto-ignition OFF
- 4. Ice Protection OFF
- 5. Electrical Load REDUCE
- Trim SET

# **ENGINE SHUTDOWN AND SECURING**

- Parking Brake SET
- Transfer Pumps OFF
- Crossfeed CLOSED
- 4. Subpanel and Avionics Switches OFF
- Overhead Panel Switches OFF
- Battery CHARGED (BATTERY CONDITION CHECK Refer to page 5-12)
- ITT BELOW 610°C FOR ONE MINUTE
- Propellers FEATHERED
- 9. Condition Levers CUT-OFF

#### CAUTION

Monitor ITT during shutdown. If sustained combustion is observed, proceed immediately to the ENGINE CLEARING procedure on Page 5-4. During shutdown, ensure that the compressor decelerates freely. Do not close the Fuel Firewall Shutoff Valve for normal engine shutdown.

- Boost Pumps OFF
- 11. Battery and Generator Switches OFF
- 12. Control Locks, Pitot Covers, Tie-Down, and Chocks AS REQUIRED

# CAUTION

The boost pumps and crossfeed are connected to the battery bus. Failure to turn these switches OFF will discharge the battery.

# NIGHT OR INSTRUMENT FLIGHT (BEFORE TAKE-OFF)

- 1. Internal Lights CHECK
- External Lights CHECK
- Flight Instruments CHECK
   Instrument Air Pressure CHECK
- Voltage and Loadmeters CHECK
- Auto-ignition CHECK

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