

LOWER SKIN PANEL - WING CENTER TANK

CW-206
C-2160
GREEN

This section extends from the midspar to S-13 and from approx the left SOB to LBL 93. Only the fwd skin flange portion of the midspar lower chord remains attached.

The remaining rivets in the skin panel have a shear failure and two holes have elongation in the inbd/outbd direction. The outbd fracture line passes thru the inbd bolt row common to the side of body splice. The panel is s-shaped with the outbd end bowed up and the inbd end bowed down. There is a partial fracture through the panel at S-11 at approx LBL 103. Portions of the angle mounted on the exterior surface of the panel at LBL 104 remain attached.

The interior and exterior surface of the skin panel is sooted. See the Fire and Explosion Group notes for full documentation.

Stephen F. Klupnick sr
FBI - 12-06-96

W. Myrtle TWA 12-6-96
R. Brachett IAM 12-6-96

SD GREEN 12-6-96 ALPA
CA Reimer BOEING 12/7/96

CW-206

GREEN TAG # C2160

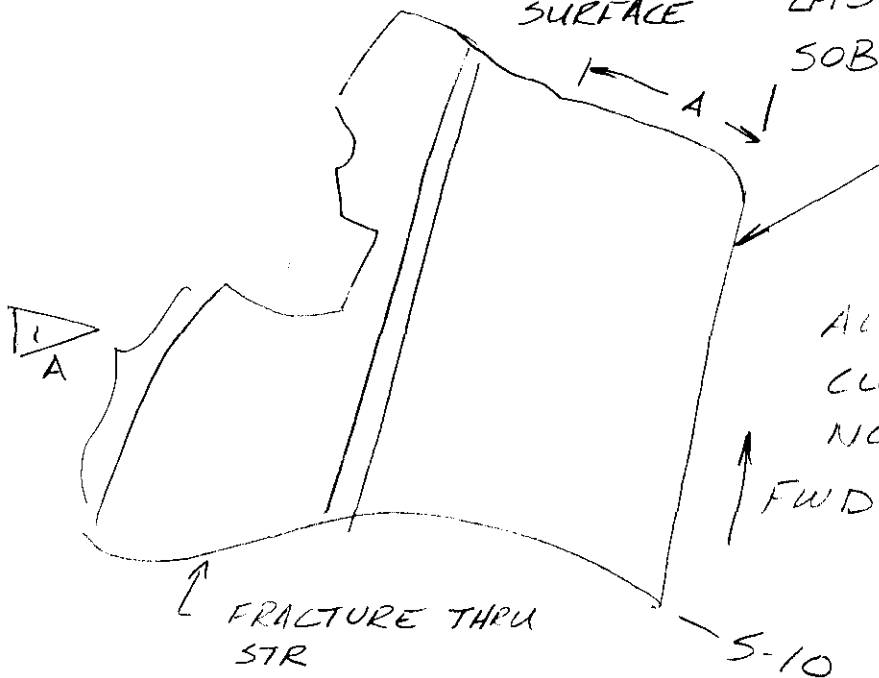
PART E

VIEW LKG @

BOTTOM
SURFACE

LHS

SOB



FAILURE THRU FASTNR
LINE - COMBINATION ANGLED &
V SHAPED FRACTURE FACE

ALL FRACTURE SURFACES
CLEAN EXCEPT AS
NOTED

WHITE ENAMEL STILL INTACT W/ SMEARED HEAVY
SOOT { PART LOOKS LIKE IT'S BEEN IN BACK OF CAR
GARAGE FOR YEARS)

THIS PART LOOKS DISTINCTLY DIFFERENT THAN PART D
IN TERMS OF SOOTING

UPPER SURFACE OF LWR WING SKIN HAS FILM OF
BLACKENING (LIGHT) THRU OUT INCLUDING AREAS
WHERE STR FOOT PRINTS WERE

▷ SLIGHT BLACKENING ON FRACTURE SURFACE

A = ANGLED FRACTURE FACE - SEE PART G