

NATIONAL TRANSPORTATION SAFETY BOARD Office of Aviation Safety Washington, D.C. 20594

STRUCTURES GROUP CHAIRMANS FACTUAL REPORT

DCA06MA010

Attachment C FAA Forms 337 & 8110 Repair Analysis & Drawings for N142PA & N2969

U.5. Department of Transportation	es Group Factual F an Mallard G73	Report MAJOR REI Irframe, Power							· · · · · · · · · · · · · · · · · · ·	ttachment	
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1. Aircraft	Meke (JRUMPAAN				Model G-73						
	Seitat Na Jaz				······	Nationality and N142PA	Registration M	kark.		******	
2. Owner	er Hame (As shown on registration córtificate) SEAPLANE ADVENTURES LLC				Address (As shown on registration certificate) 75 HOLLY LILL LN GREENWICH, CONNECTICUT 06830-6098						
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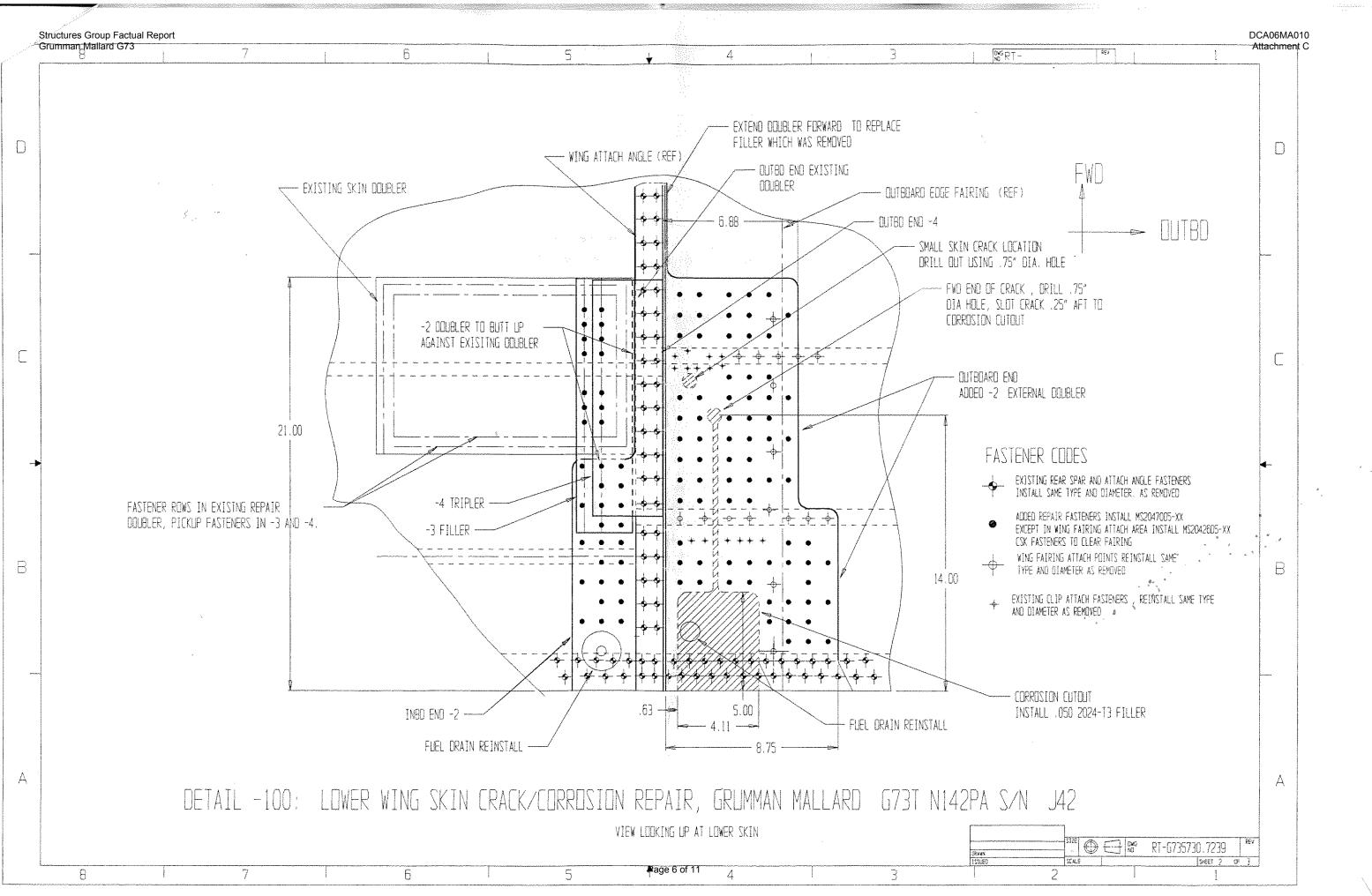
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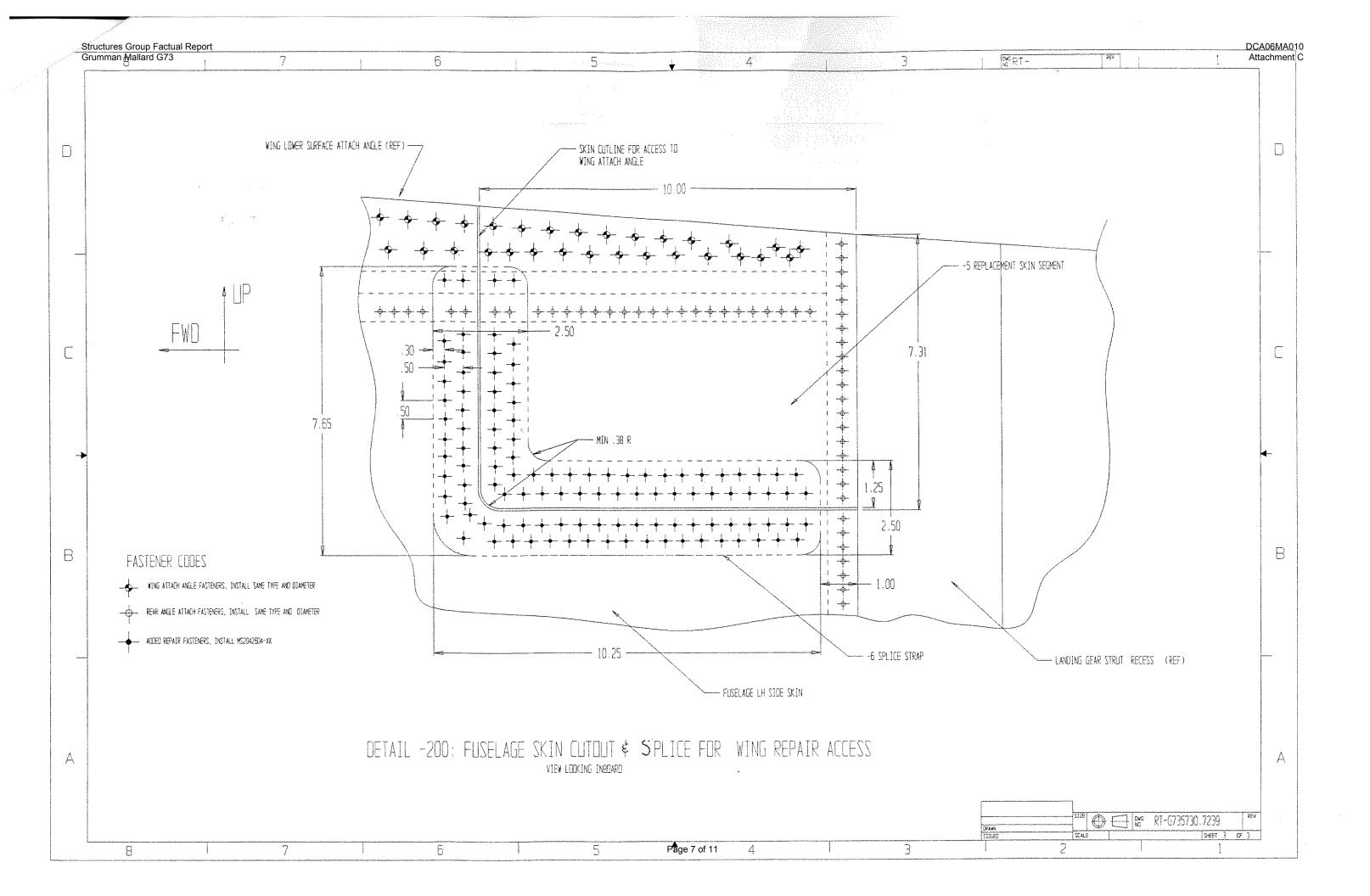
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Dewing no: 050708 rev "A"	"REPAIR SCHE	ME - WING I	Box Lower Skin"		
	This data is app number N142PA	roved for Gru Nonly.	imman model G-73 air	craft seria	l number J42, registration
	j detection at the	outboard WS	er FAB 25.571 require station 68 splice origi ctions at the same time	nal skin na	Iral inspection for crack anel fastener holes at 6020
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o structurally approve a PLICABLE REQUIREMENTS & AR part 48; peragraphs AR Part 25; peragraphs	H specific sactions) 3: 48,302, 48,303,	4a.308	inboard lower skin are	a on the a	above aircraft.
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STATEMENT OF COM	PLIANCE WITH THE FEDERA	L AVIATION RECHLAR		7-22 0
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GRUMMAN	G73T All	RÐLANE	CONSULT	ANPATI
		<u>AF DIX</u>		
IDENTIFICATION		OF DATA		
		INTLE		
	DRAWING			
RT-G735730.7239	LOWER WING SKIN CRACK		RUMMAN	
DTD 7/23/99	MALLARD G73T N42PA S/I	N J42."		
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	ENGINEERING D	ATA ONLY.		
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GENERAL NOTES									
1. THIS DRAWING PROVIDES DETAILS FO	OR REPAIR OF THE LH WING LOWER								
SKIN AT FUSELAGE ATTACHMENT DU ON GRUMMAN MALLARD G73T N142PA	E TO A SKIN CRACK AND CORROSION A S/N J42.			geltin genoeten		i.			
2. CUTOUT FUSELAGE SKIN AS SHOWN S WING ATTACH ANGLE. REMOVE FAST OUT EXISTING SHIM BETWEEN WING L				算错的分子。 算符 持					
CUTOUT CORRODED SKIN AREA AND O STOP DRILL FWD END OF CRACK USING									NUTVO V VIVO A A HARVAN V A A A A
3. BREAK ALL SHARP EDGES .020/.030 R									
 FINISH ALUMINUM PARTS AND REWOR 2 COATS EPOXY PRIMER, REFER TO G7 FINISH REQUIREMENTS. 	KED AREAS USING ALODINE 1201S AND 3 SRM CHAPTER 51 FOR ADDITIONAL								
5. DELETED		N - 1		ALC AND					
 MAINTAIN MINIMUM 2D + 1/16 EDGE I OTHERWISE SPECIFIED. 	DISTANCE ON ALL FASTENERS UNLESS	• 4 • •		ling of the second s					
7. SHIM/TAPER SHIM AS REQUIRED.							, e		
8. ALL HI-LOK FASTENERS ARE TO BE IN REAMED HOLES. HOLE TOLERANCE PE ARE TO BE INSTALLED IN NET FIT HOL	9 MARCIO ATT CONTENTS TO A COMPANY	÷							
9. ALL FASTENER LENGTHS TO BE DETER		. · ·							
10. INSTALL ALL REPAIR PARTS WET USIN									
OCTO (OK EQUIVALENT).									
11. RELOCATE FASTENERS AS REQUIRED T BETWEEN ADDED FASTENERS AND EX	TO MAINTAIN MINIMUM 3D SPACING ISTING HOLES IN FRAME WEB.	· · · ·							
12. REINSTALL FUEL DRAINS AND VENT A TYPE AND DIAMETER FASTENERS AS R	T AFT EDGE OF SKIN USING SAME								
 INSTALL FUSELAGE SKIN REPAIR PER S SKIN REPAIR 	SHEET 3 UPON COMP[LETION OF WING		1 -6	SPLICE STRAP, FUSELAGE SKIN	REPAIR		DQ-A-250/5	.063 2024-13	
		alle . Light	1 -5	REPLACEMENT SKIN, FUSELAGE S	SKIN REPAIR		QQ-A-250/5	.063 2024-13	
14. RE-SEAL FUEL TANK AREAS PER MAIN			-4	TRIPLER, WING SKIN REPAIR			00-A-250/(3	.063 7075-16	
15 REVISE AIRCRAFT WEIGHT AND BALA	NCE AS APPLICABLE.		-2	FILLER, WING SKIN REPAIR "DOUBLER, EXTERNAL; WING SKIN	N REPAIR		00-4-250/13	A\$ REQUIRED 2024-13 DR 70 .063 7075-16	
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			1 -200	FUSELAGE SKIN CUTOUT AND SPL					
			I -100 DTY DASH ND.	LOWER WING SKIN CRACK/CORRI DESCRIP		0 G73T N142PA S/N 42, [PART_ND .	SPECIFICATION	MATERIAL	
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			FRACTIONS ANGLES	DECIMALS DRAWN			№ RT-67357.	30.7239 ^{RE}	ĒV
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Structures Group Factual Report Grumman Mallard G73

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U.S Departme Transportation Federal Aviat Administratio	lion					Propeller,			Foi	FAA Use Only			
INSTRU and disp	CTIONS: Print or type position of this form. The such violation (Section	his repor	t is required	by law (49	U.S.C	ppendix B, and A . 1421). Failure	C 43.9-1 (to report c	or subsequent rev an result in a civil	ision thereof) for ins penalty not to excee	tructions ed \$1,000			
	Make Grumman						Model G73T	<u></u>					
I. Aircraft	aft Serial No. J27 Nationality and Registration Mark N2969												
2. Owner		Name (As shown on registration certificate) SeaPlane Leasing III, LLC					Address (As shown on registration certificate) 3240 Bird Ave. FL Lauderdale, Florida 33131						
					3	. For FAA Use	Only						
	1			1		. Unit Identific	ation			5. Type	1		
Unit	٨	/lake		<u> </u>		Model		Ser	ial No.	Repair	Alteration		
AIRFRAME		(As described in item 1 above)						\times					
POWERPL.	WT								4				
PROPELLE	R												
APPLIANC													
	Manufacturer												
				·····	6	. Conformity St	atement						
	s Name and Address					ind of Agency U.S. Certified N	la ala ala		C. Certificate No).			
Juan Hered	12					Foreign Certifie		C					
Miami, Flori	da 33147					Certified Repair Station Manufacturer			-				
have	fy that the repair and/or been made in accorda in is true and correct to	nce with	the requirem	nents of Pa	lentifi	ed in item 4 abo	ve and desc ral Aviation	cribed on the reve Regulations and	rse or attachments in the information	hereto fumished			
Date July 6, 2000	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,					Signature of Au	thorized Inc	fividual //					
				7	. Ар	proval (or Retu	m to Servi	ce/					
	to the authority given p rator of the Federal Avia			ow, the unit and is		ified in item 4 wa	as inspecte REJECTED		rescribed by the				
BY	FAA Fit Standards Inspector		Manufacture	r		Inspection Auth	orization		Other (Specify)				
ы. Г	FAA Designee		Repair Static	on		Person Approve Canada Airword							
1						Signature of Au		L					

Page 8 of 11

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NOTICE
We first: Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.
8. Description of Work Accomplished
 Description of Work Accompliance (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
N2050
July 6, 2000 N2969
Stringer repaired in Right Hand wing at Sta. 50.5, third stringer from aft. See attached Drawing and 8110 #RT-G7355720.6290. Wing fuel tank resealed and leak check carried out in accordance with Frakes Service Manual. Aircraft was reweighed prior to return to service.
END
Additional Sheets Are Atlached U.S.GPO:1989-0-663-171
U.S.GP0:1989-0-003-171

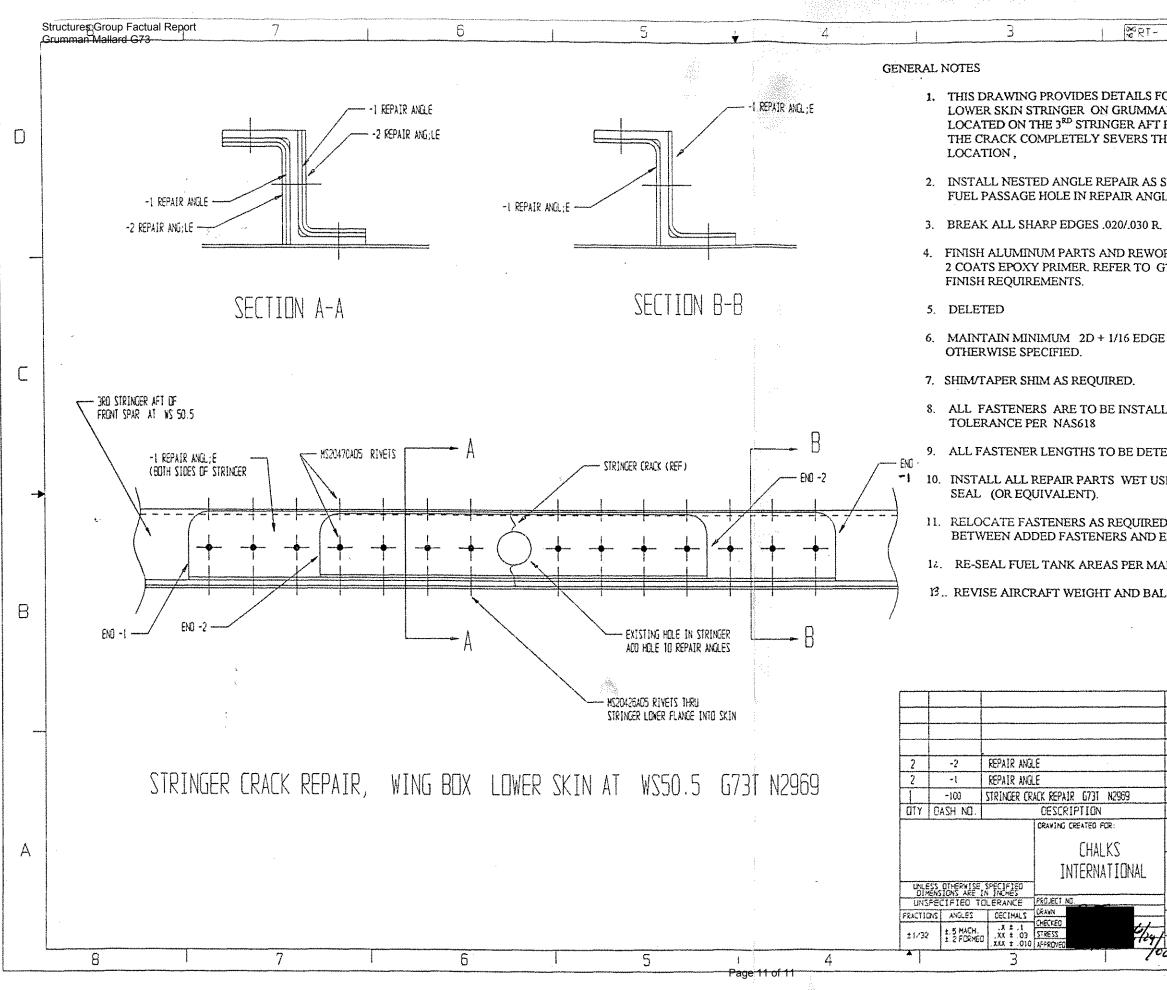
Grumman Mallard G73			·····	<u> </u>	DCA06MA010
	FEDERAL AVIATION	ADMINISTRAT	108		
STATEMENT OF COMP	PLIANCE WITH TH	E FEDERAL	AYIATION REGUL	ATIONS	6-30-00
			COMPONENT IDENTI	1	
IAKE	MODEL NO.		plana, Radio, Helicopter,		APPLICANT
GRUMMAN	G73T		IRPLANE	1	SULTANT
	N2969	-			
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IDENTIFICATION			TITLE		*****
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	PART NO.	SPECIF	ICATION	MATERIAL		
	P. S. BCA 17753 MIA	5/RO	SP/AC			
	TITLE : STRING	er Crack 30x Lover			- A	
h.		R NG R	1-67357	20.6290	REV	
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