



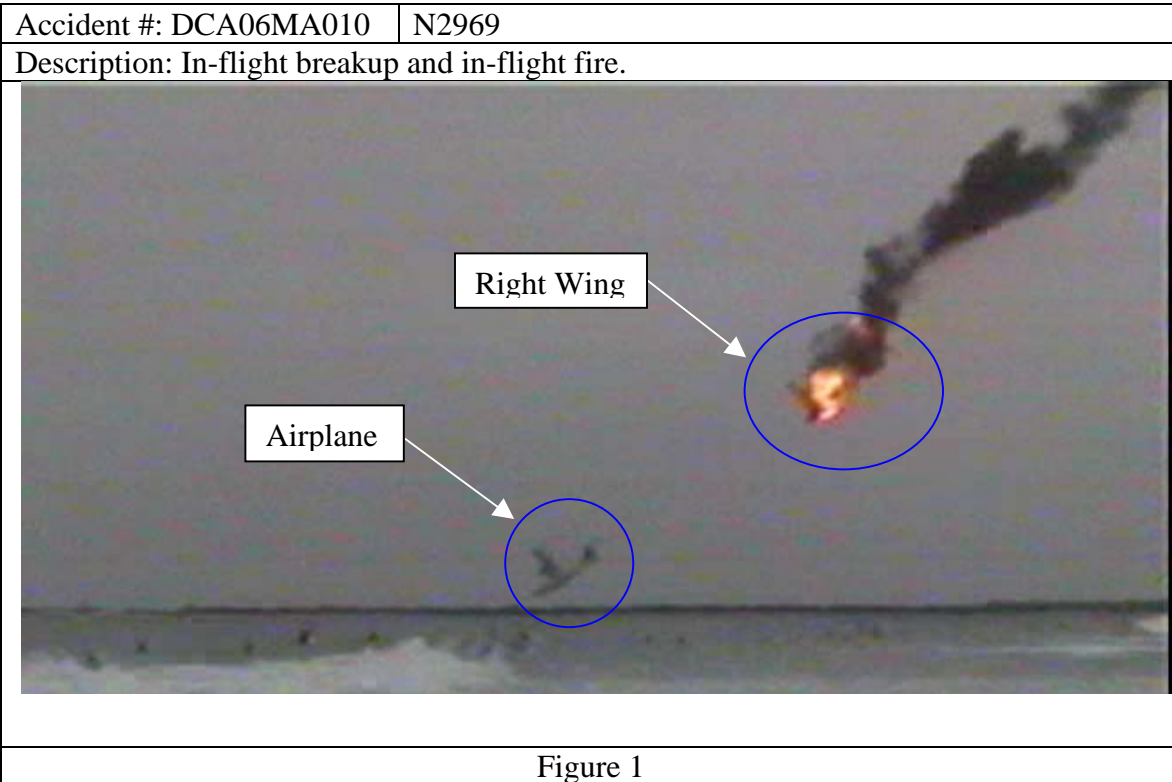
NATIONAL TRANSPORTATION SAFETY BOARD

Office of Aviation Safety
Washington, D.C. 20594

STRUCTURES GROUP CHAIRMANS FACTUAL REPORT

DCA06MA010

Attachment A
Figures



Accident #: DCA06MA010

Description: Accident site.



Figure 3

Accident #: DCA06MA010

Description: Accident site and flight path.



Figure 4

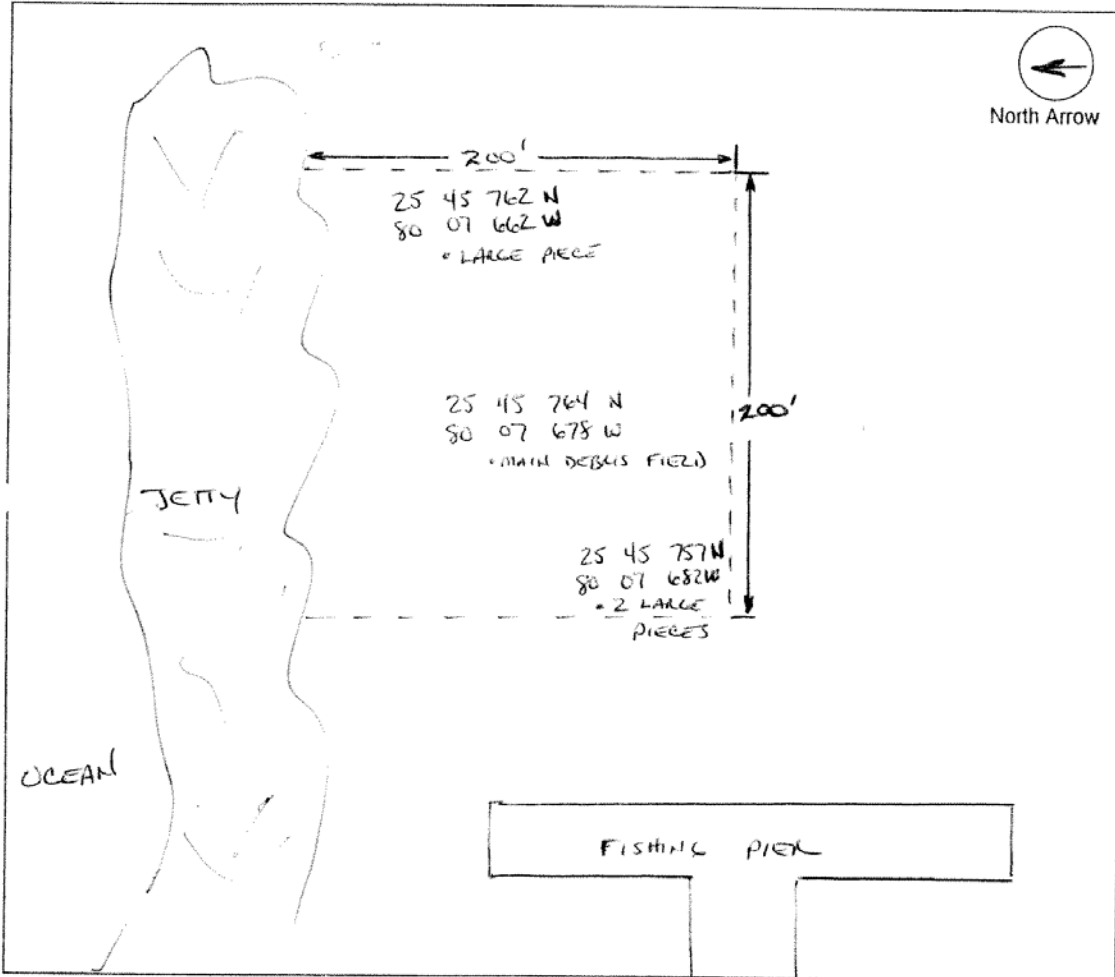
Accident #: DCA06MA010 | N2969

Description: Underwater debris field.

Miami Beach Police
Underwater Recovery Unit
Dive Report/Diagram

Date: 12 23 2005

Case Number 2005-45223



Location EAST END OF GOVERNMENT CUT

Figure 5



Accident #: DCA06MA010 | N2969

Description: Left wing, control cables and fuselage.



Figure 7

Accident #: DCA06MA010 | N2969

Description: Right wing skin, stringer and spar cap fractures overview at WS 34.

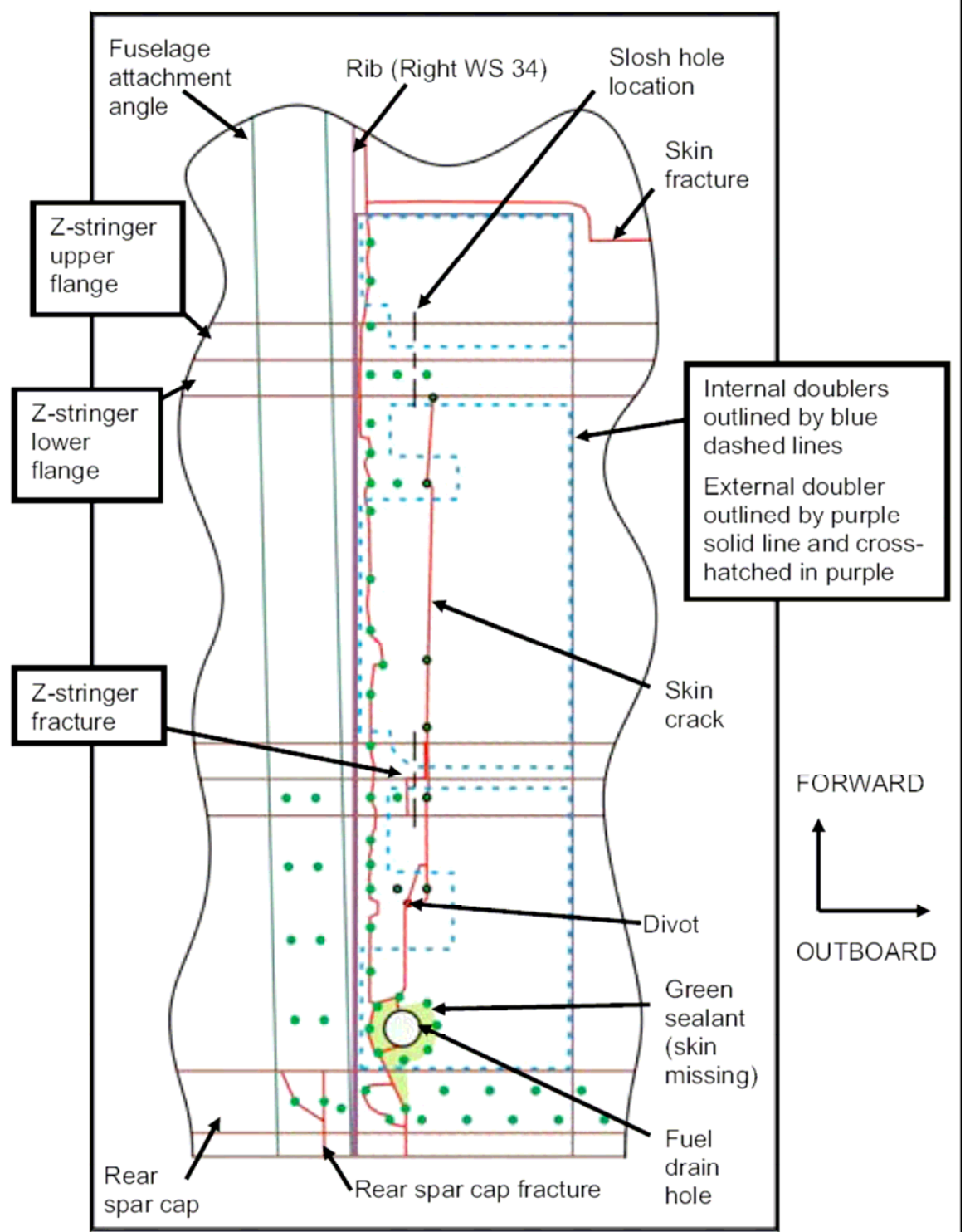

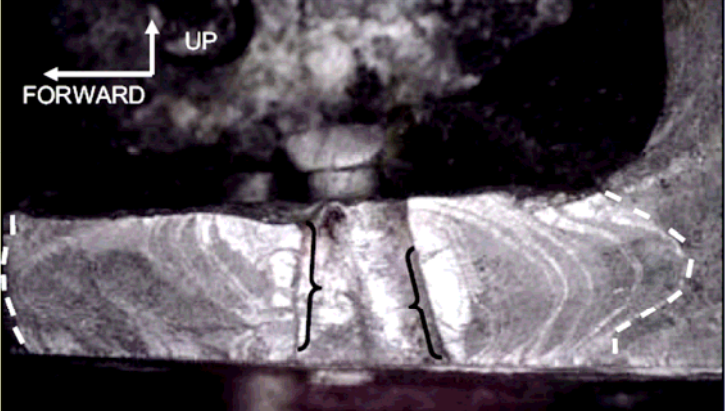


Image No.:0603A00715, Project No.: 2005120013

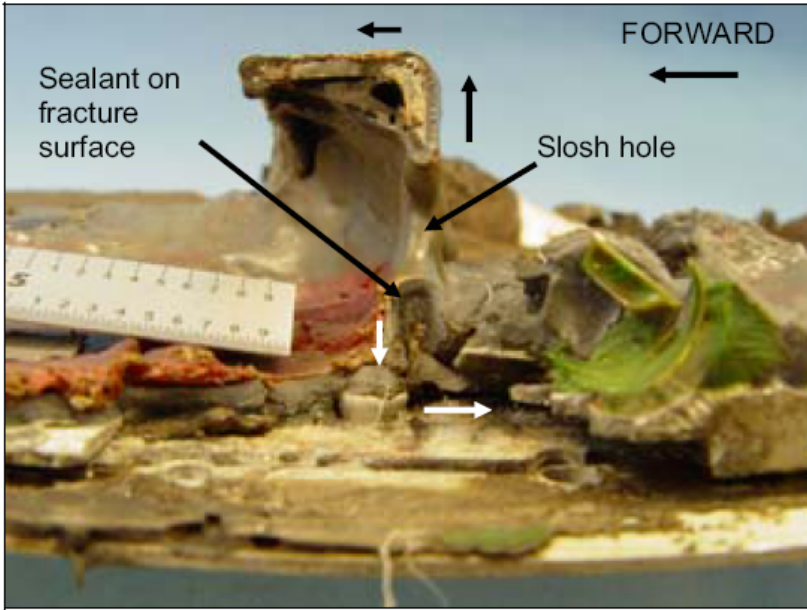
Figure 8

Accident #: DCA06MA010	N2969
Description: Right wing rear spar lower spar cap fracture at WS 34.	
 <p>Image No.:0512A00890, Project No.: 2005120013</p>	<p>Close view of the outboard side of the rear spar lower spar cap transverse fracture just inboard of right WS 34. Fatigue features emanated from a double drilled fastener hole.</p>
Figure 9	

Accident #: DCA06MA010	N2969
Description: Right wing rear spar lower spar cap fatigue region.	
	<p>Closer view of the fatigue region on horizontal flange of rear spar lower spar cap. Dashed lines indicate the fatigue boundaries, and brackets indicate the fatigue origin areas.</p>
Figure 10	

Accident #: DCA06MA010 | N2969

Description: Right wing aft Z-stringer fracture at WS 34 fuel slosh hole.



Close view of the rear Z-stringer fracture near right WS 34. Fatigue features emanated from a slosh hole in the stringer web as indicated with unlabeled arrows.

Image No.:0512A00944, Project No.: 2005120013

Figure 11


Accident #: DCA06MA010 | N2969

Description: Right wing aft Z-stringer fatigue region at WS 34.



Close view of fatigue at upper side of hole in rear Z-stringer fracture near right WS 34. A dashed line indicates the fatigue boundary, and an unlabeled arrow indicates the local fatigue propagation direction.

Figure 12

Accident #: DCA06MA010 N2969	
Description: Right wing aft Z-stringer fatigue region at WS 34.	
	<p>Close view of fatigue at the lower side of the slosh hole in the rear Z-stringer fracture near right WS 34. Unlabeled arrows indicate the local fatigue propagation directions.</p>
io.:0601A01200, Project No.: 2005120013	
Figure 13	

Accident #: DCA06MA010 | N2969

Description: Left wing skin, stringer and spar cap fractures overview at WS 34.

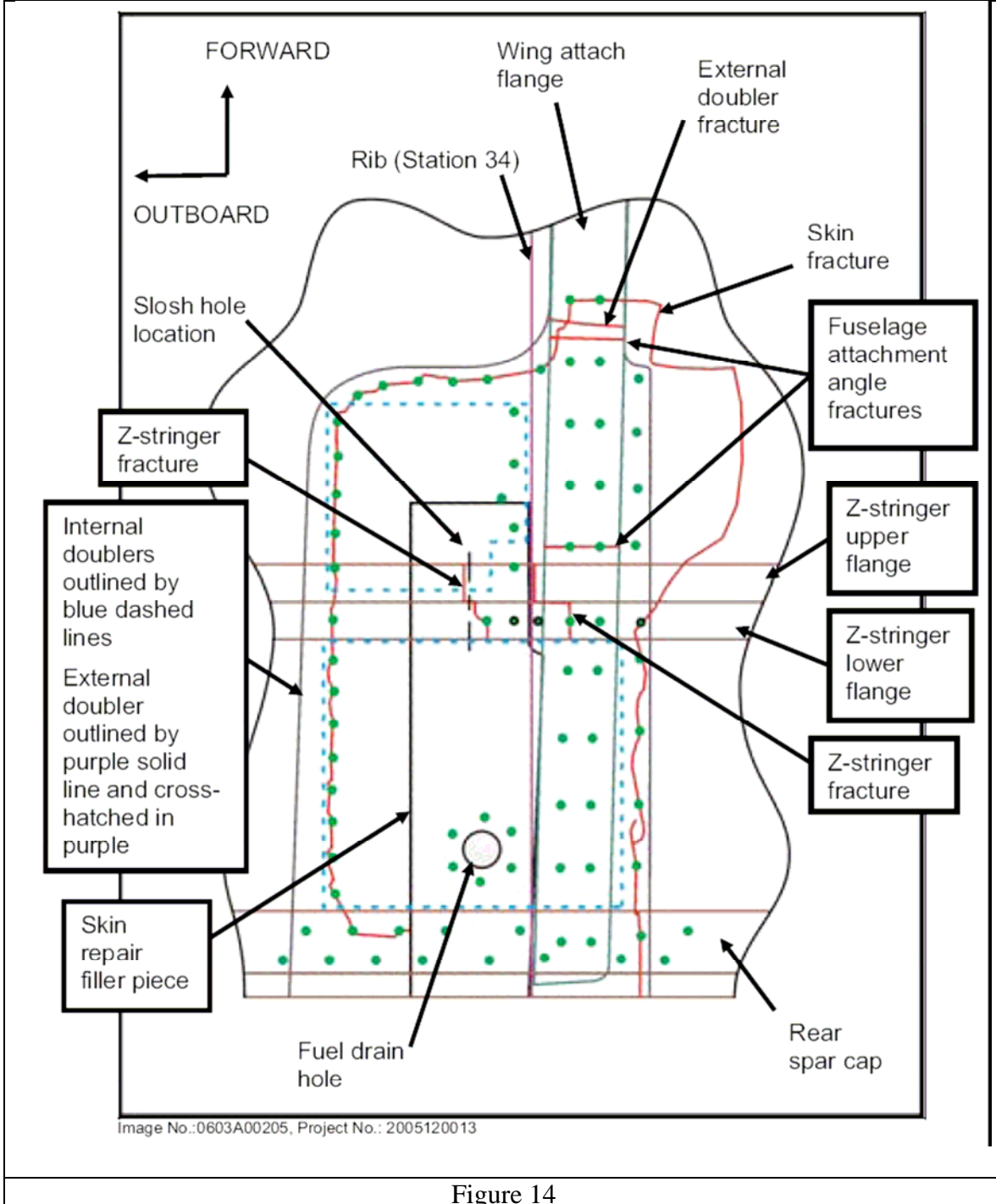
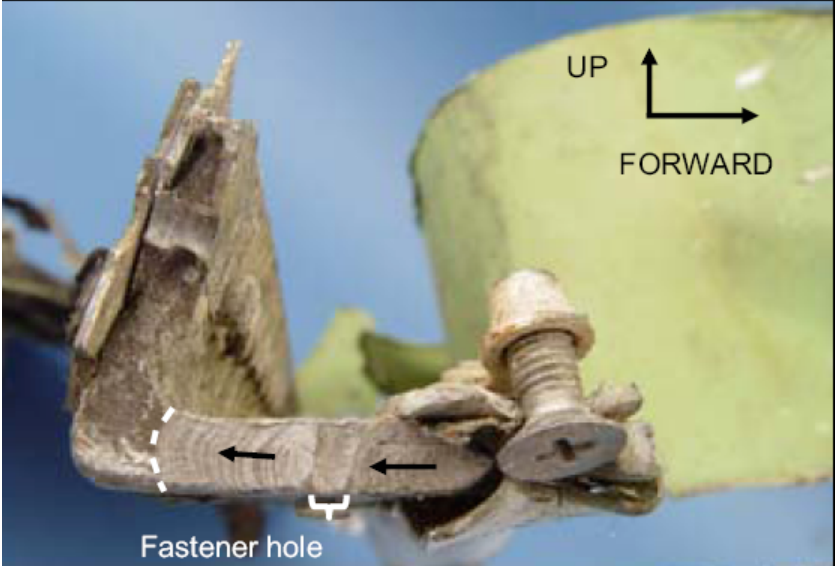
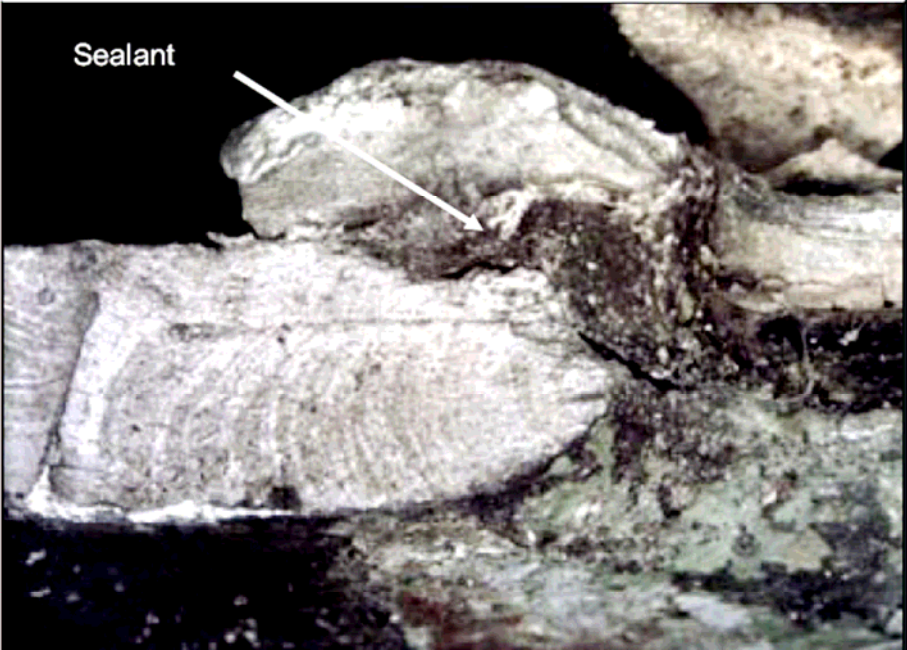
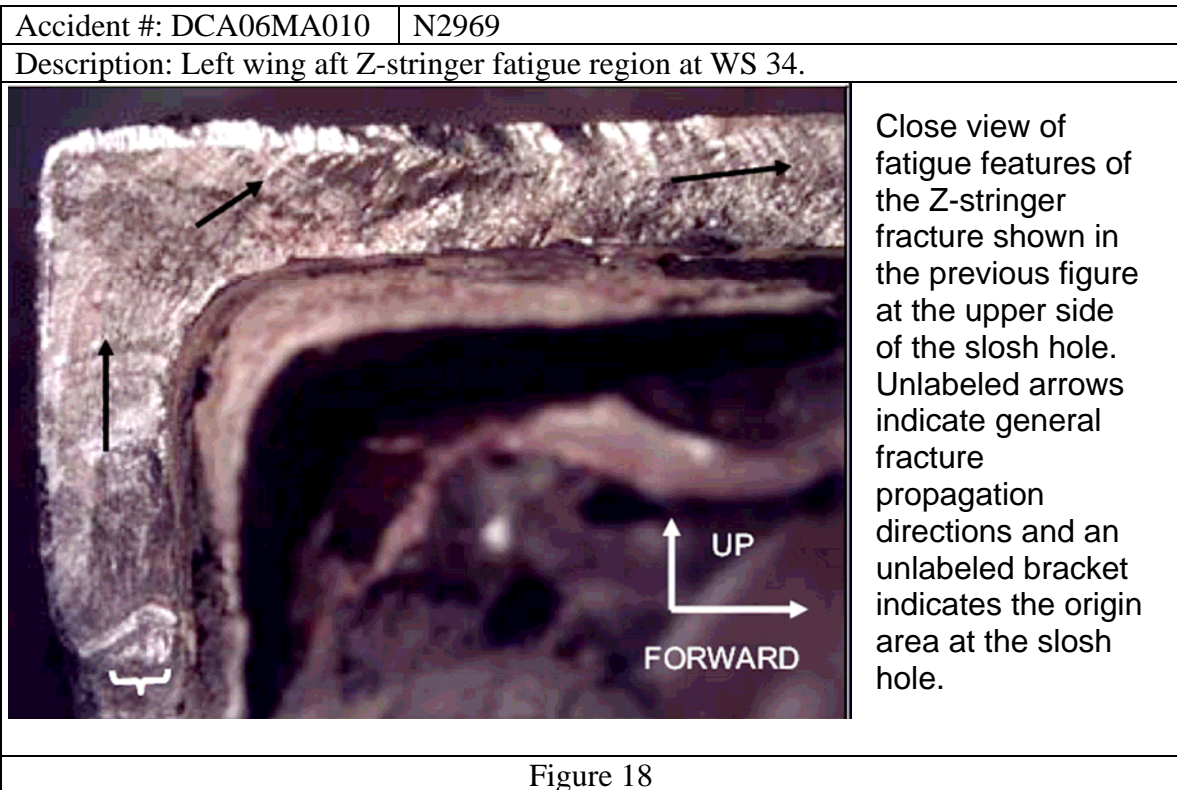
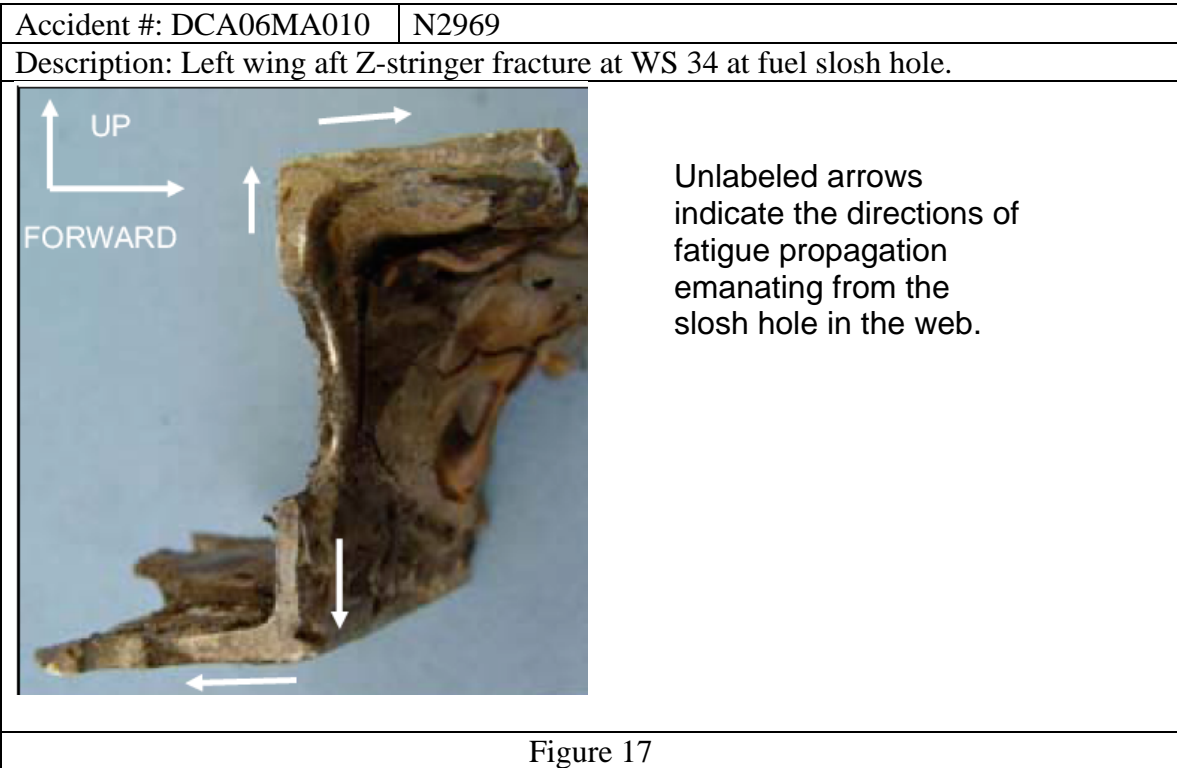


Figure 14

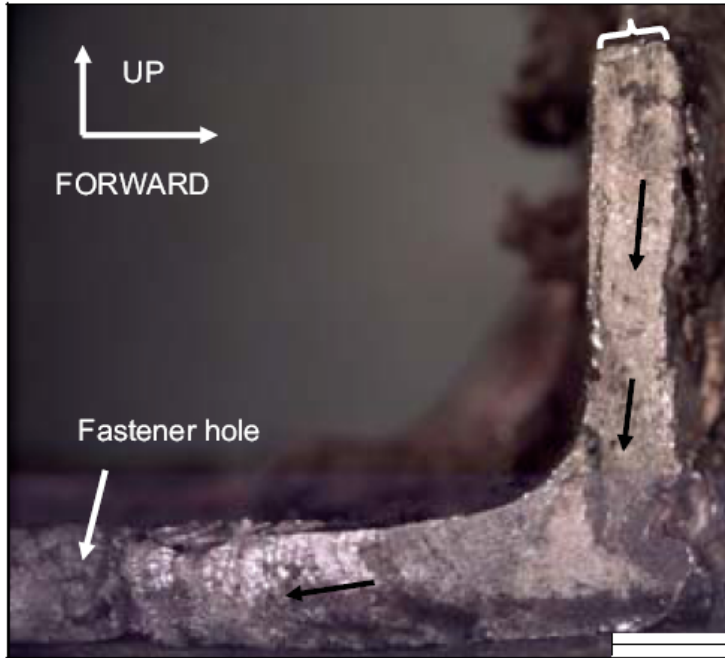
Accident #: DCA06MA010 N2969	
Description: Left wing front lower spar cap fracture at WS 34.	
	View of the outboard side of the fracture in the front spar lower spar cap 5 inches outboard of WS 34. A dashed line indicates the fatigue boundary, and unlabeled arrows indicate the direction of fatigue propagation.
Figure 15	

Accident #: DCA06MA010 N2969	
Description: Left wing front spar lower cap fatigue region.	
	Closer view of the fatigue origin area in the front spar cap fracture.
Figure 16	



Accident #: DCA06MA010 | N2969

Description: Left wing aft Z-stringer fatigue region at WS 34.



Close view of fatigue features of the Z-stringer fracture shown in the previous figure at the lower side of the slosh hole. Unlabeled arrows indicate general fracture propagation directions and an unlabeled bracket indicates the origin area at the slosh hole.

Figure 19


Accident #: DCA06MA010	N2969
Description: Left wing middle Z-stringer fracture at WS 34 at fuel slosh hole.	
	View of the fracture surface for the middle left Z-stringer. An unlabeled bracket indicates the extent of a fatigue region emanating from the forward side of a fastener hole, and an unlabeled arrow indicates a small fatigue region at the aft side of the fastener hole.

Figure 20

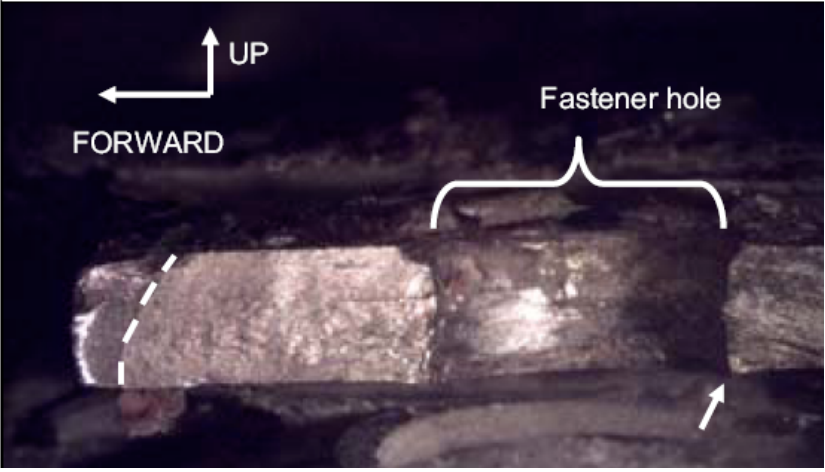
Accident #: DCA06MA010	N2969
Description: Left wing middle Z-stringer fatigue region at WS 34.	
	Close view of the fracture in upper flange of the left wing middle Z-stringer shown in the previous figure. A dashed line indicates the fatigue boundary for the region emanating from the forward side of the fastener hole, and an arrow indicates the smaller fatigue region at the aft side of the fastener hole.

Figure 21

Accident #: DCA06MA010 | N2969

Description: Empennage.



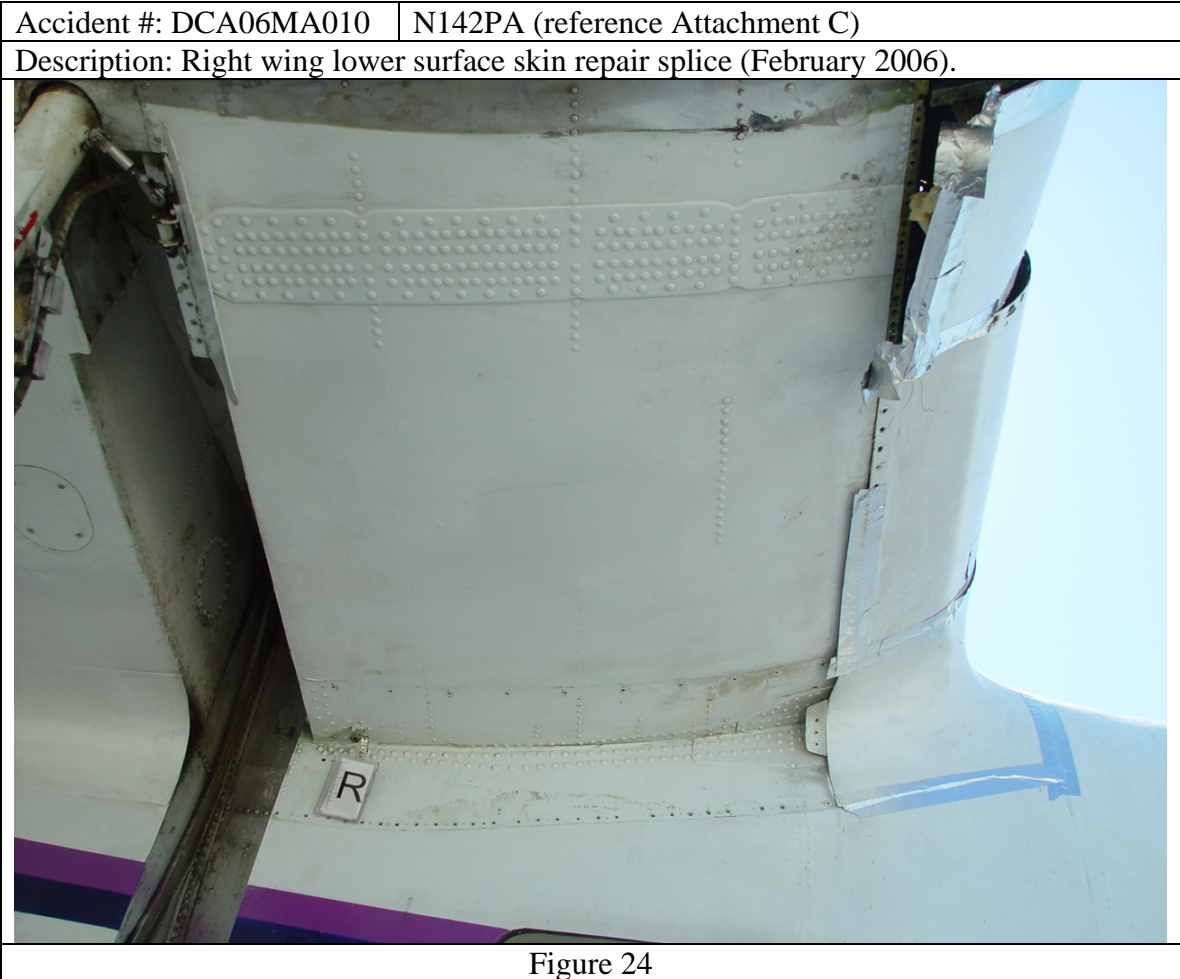
Figure 22


Accident #: DCA06MA010 | N2969


Description: Typical dope and fabric control surface fire damage.




Figure 23




Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: Left wing lower surface wing skin repair (February 2006).	
	
Figure 25	


Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: Left wing interior lower surface wing repair (February 2006).	
	
Figure 26	

Accident #: DCA06MA010	N130FB
Description: N130FB left wing to fuselage attach (February 2006).	
	
Figure 27	

Accident #: DCA06MA010	N130FB
Description: Left wing lower surface wing repair (February 2006).	
	
Figure 28	

Accident #: DCA06MA010	Spare located in Chalk's hangar.
Description: Left wing rear lower spar cap double drilled holes (February 2006).	
 A close-up photograph of a metal spar cap. The spar cap is a long, horizontal metal bar with two rows of circular holes drilled through it. The holes are arranged in two parallel lines, one above the other. The metal surface appears slightly worn and has some discoloration. In the background, a green object is partially visible.	
Figure 29	

Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: N142PA right wing lower surface skin crack (red line, June 2005).	
	
Figure 30 (Source John S. Patterson DER)	

Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: Right wing lower surface skin crack close-up (June 2005).	
	
Figure 31 (Source John S. Patterson DER)	

Accident #: DCA06MA010 | N142PA (reference Attachment C)
Description: Right wing lower surface with previous repair doubler removed (June 2005).



Figure 32 (Source John S. Patterson DER)

Accident #: DCA06MA010 | N142PA (reference Attachment C)
Description: Right wing lower surface with previous repair doubler removed (June 2005).

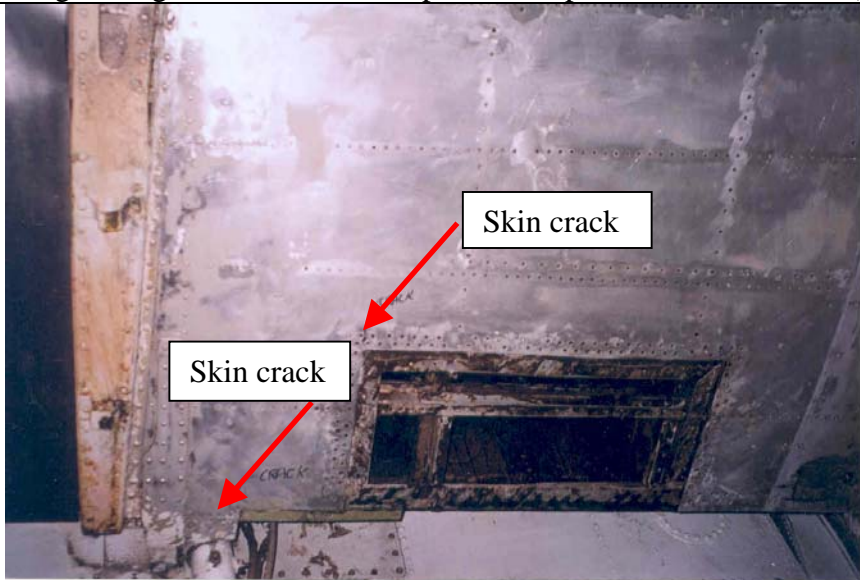
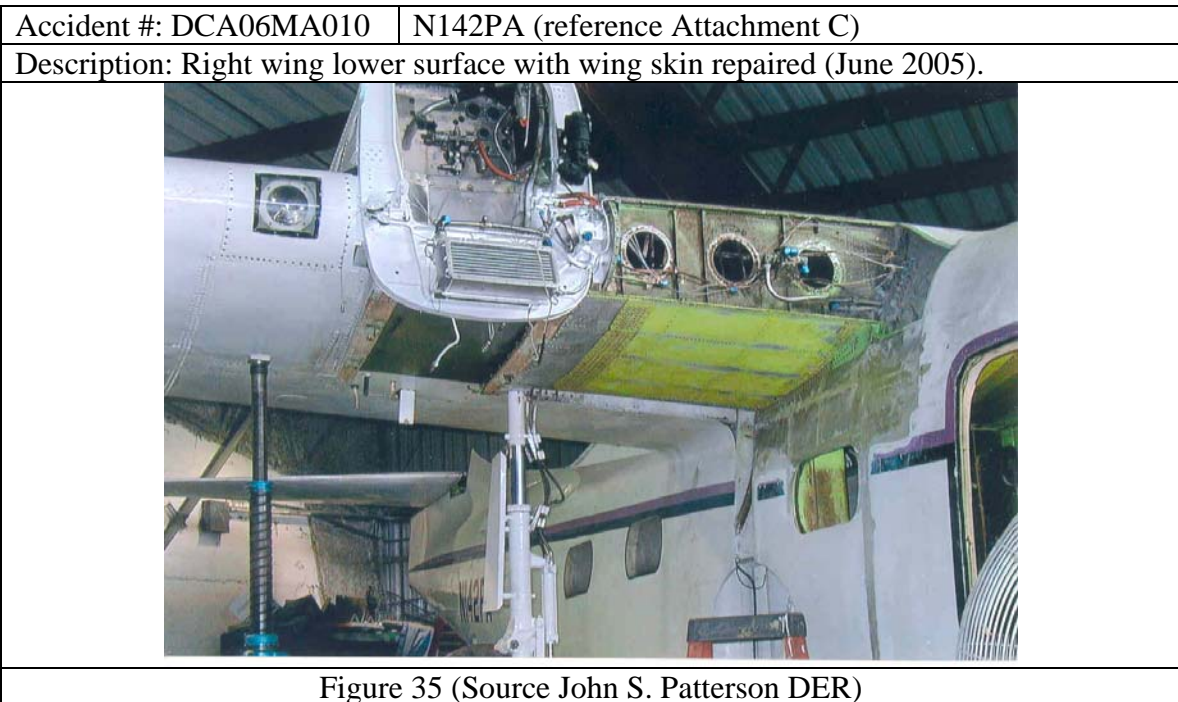
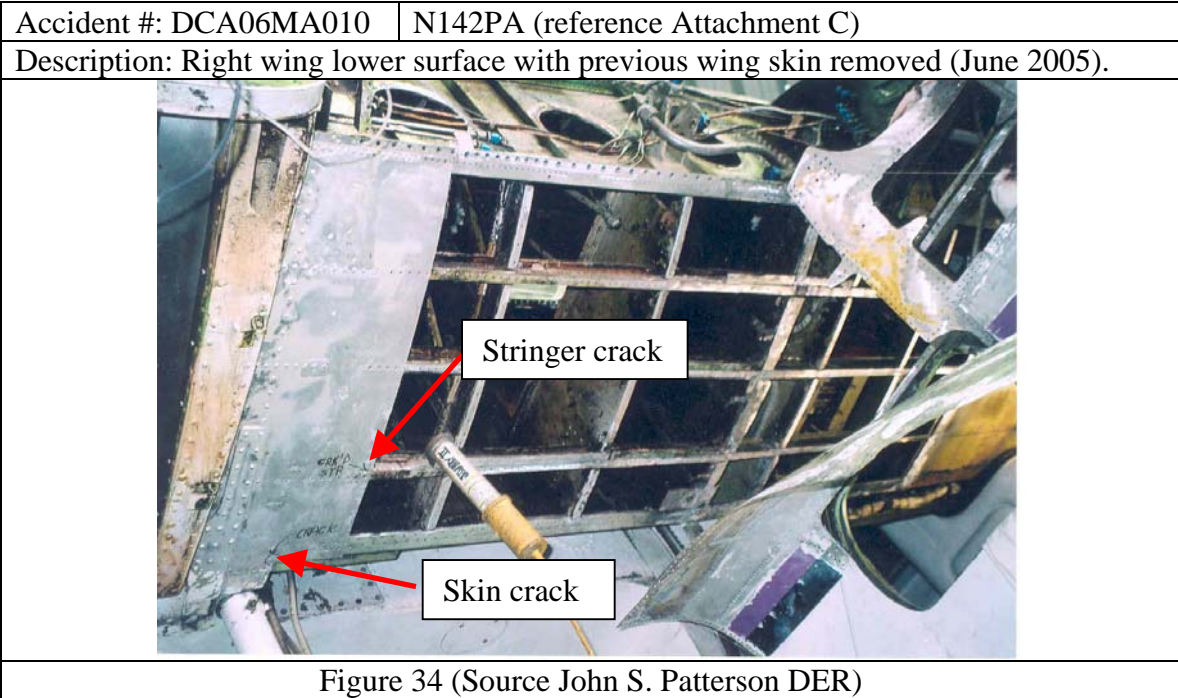
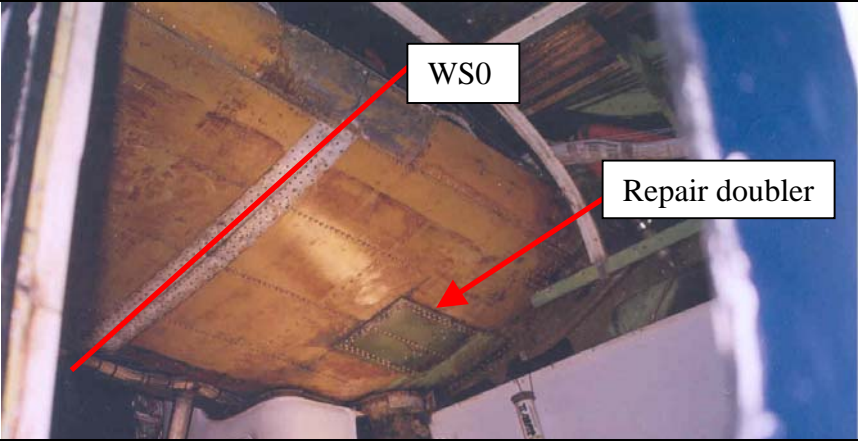
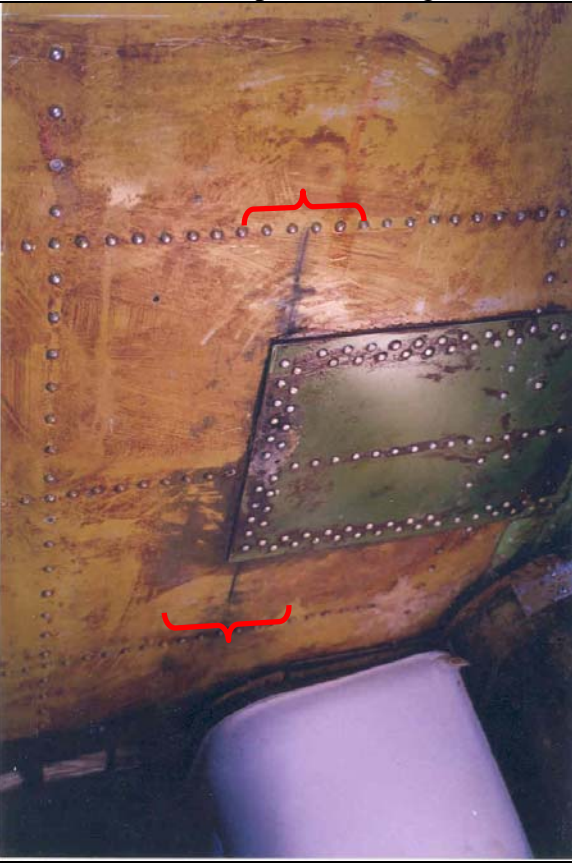


Figure 33 (Source John S. Patterson DER)



Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: Left wing lower surface with previous wing skin repair & crack (June 2005).	
 A photograph showing the underside of an aircraft wing. A diagonal red arrow points to a section of the wing skin labeled 'WSO'. Another red arrow points to a rectangular patch of green-painted metal labeled 'Repair doubler'. The wing structure is made of metal with visible rivets and bolts.	
Figure 36 (Source John S. Patterson DER)	

Accident #: DCA06MA010	N142PA (reference Attachment C)
Description: Left wing lower surface with previous wing skin & crack (June 2005).	
 A close-up photograph of the wing skin repair area. Two red brackets are drawn on the image, one above and one below the green repair doubler, highlighting the riveted joints. The surrounding wing skin is yellowish-brown and shows signs of wear and cracking.	
Figure 37 (Source John S. Patterson DER)	