

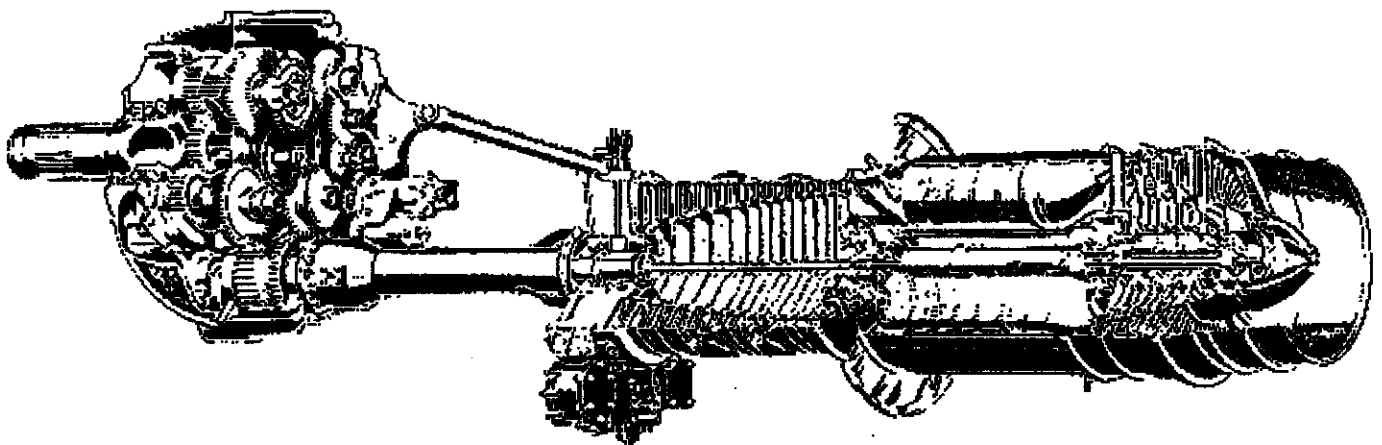


Rolls-Royce

Accident investigation

On-site engine
Investigation report

Rolls-Royce Corporation
Model T56-A-9D
Engines AE100843, AE101504,
AE101267, A100814



Hawkins & Powers
Walker, California

Michael A. Weber

Michael A. Weber
Senior Air Safety Investigator

Accident date: June 17, 2002
Investigation date: June 18-20, 2002
Report date: July 26, 2002

TABLE OF CONTENTS

1. SYNOPSIS.

2. FACTUAL INFORMATION.

- 2.1 History of Flight
- 2.2 Personnel Information
- 2.3 Injuries to Persons
- 2.4 Aircraft Information
- 2.5 Meteorological Conditions
- 2.6 Aids to Navigation
- 2.7 Communications
- 2.8 Aerodrome Information
- 2.9 Flight Recorder
- 2.10 Wreckage and Impact Information
- 2.11 Engine Examination
- 2.12 Engine Maintenance and Records
- 2.13 Additional Information

3. FINDINGS.

4. APPENDICES.

- A. On-Site Photographs at Walker, California
- B. Engine #1 Logbook
- C. Engine #2 Logbook
- D. Engine #3 Logbook
- E. Engine #4 Logbook
- F. Map of the area
- G. Participants
- H. Wreckage Diagram

1. SYNOPSIS

On June 17, 2002, about 15:00 hours local time, a Lockheed C130A aircraft, N130HP, operated by Hawkins & Powers, experienced an in flight structural break up and was destroyed in a post crash fire. The mishap occurred near Walker, California. The three crew members received fatal injuries. The NTSB-SWR office was charged with conducting the investigation. The NTSB IIC requested assistance from the airframe and engine manufacturers for the investigation. Rolls-Royce and Lockheed Martin complied with the request. This report presents the facts and findings of the aircraft and engine examinations conducted on-site at Walker, California on June 19-21, 2002.

2. FACTUAL INFORMATION

2.1 History of Flight

The C130A Aircraft had departed Minden, Nevada, at an undetermined time, to participate in fire fighting efforts near Walker, California. A Company flight plan had been filed and visual meteorological conditions prevailed. The U.S. Forestry Service was operating the aircraft for the public use fire fighting flight under 14 CFR Part 91. The aircraft had just made a pass over a fire and dropped a load of fire retardant when the incident occurred. The aircraft was registered to Hawkins and Powers Aviation, Inc., Greybull, Wyoming.

2.2 Personnel Information

The three crewmembers were Steve Wass, Greg Labaire, and Mike Davis. No additional information was provided

2.3 Injuries to Persons

Injuries to all three crewmembers were fatal.

2.4 Aircraft Information

Model	Lockheed C130A
Serial Number	3146
Registration Number	N130HP
Airframe Total Time	21863.0
Last "A" check Inspection	21863.0
Last "B" check Inspection	21695.5
Last "C" check inspection	21401.5

Engine Model	Allison T56-A-9D
Rating:	3750 Shaft Horsepower
Engine Position Number 1	AE100843
Engine Total Hours	10100.4
Last "A" check Inspection	10100.4
Last "B" check inspection	1580.2
Last "C" check inspection	1253.5

Engine Model	Allison T56-A-9D
Rating:	3750 Shaft Horsepower
Engine Position Number 2	AE101504
Engine Total Hours	9943.5
Last "A" check Inspection	9943.5
Last "B" check inspection	9893.1
Last "C" check inspection	9598.7

Engine Model	Allison T56-A-9D
Rating:	3750 Shaft Horsepower
Engine Position Number 3	AE101267
Engine Total Hours	10875.1
Last "A" check Inspection	10875.1
Last "B" check inspection	10679.9
Last "C" check inspection	10385.5

Engine Model	Allison T56-A-9D
Rating:	3750 Shaft Horsepower
Engine Position Number 4	A100814
Engine Total Hours	Unknown/TSO 2612.5
Last "A" check Inspection	2612.5
Last "B" check inspection	N/A
Last "C" check inspection	N/A

2.5 Meteorological Conditions

As reported by the NTSB IIC:

Visibility = 10 miles

Clouds = Few scattered

Altimeter = 30.21

Temperature = 23° C

Dew Point = 1° C

Winds = Southwest to Northwest at 6 knots gusting to 10 knots

2.6 Aids to Navigation

There were no known aids to navigation. The flight was operating under visual flight rules (VFR) conditions and the pilot was said to be familiar with the area.

2.7 Communications

None

2.8 Aerodrome Information

None

2.9 Flight Recorder

The aircraft was not equipped with either a Cockpit Voice Recorder or a Flight Data Recorder.

2.10 Wreckage and Impact Information

General Wreckage Site Observations:

The entire most southerly impact area exhibited fire/heat damage to all of the aircraft, engine, and propeller parts/components as well as scorched earth. No fire damage was observed in the most northerly impact area. The area and the aircraft parts around the initial aircraft impact area were covered with fire retardant. *No large pieces of the fuselage or cockpit area were recovered* except for a section of cargo flooring identified to have been located just aft of the cockpit. Several items were found either on highway 395 or in the brush located west of highway 395, which were not part of the main impact site. A piece of the top of the center wing measuring 276.5-inches long x 44-inches wide (maximum of six stringers) was found on highway 395.

From the seam lines this piece was comprised of 2½ panels with one complete panel. Four panels make up a complete configuration width. A single center wing stringer plus a piece of the upper center wing, measuring 93-inches long x 12.5 inches wide with a lap joint visual, was found in the brush on the west side of highway 395. A wreckage diagram is included as an appendix to this report.

Visual Observations of Individual Components were as noted:

Right wing outer flap and flap track:

The jackscrew was still attached to flap and the jackscrew nut was visible and appeared to be at the 50% position (mid length of the jackscrew). Another jackscrew was found separate from the flap structure but still in this area. The jackscrew nut on this jackscrew was also at the 50% position.

Left outer wing:

The left outer wing was found with a broken outboard rainbow fitting. The wing was found top facing up and the entire wing was consumed by fire.

Bottom of the center wing with the exception of the right outboard 10-inches:

The center wing within the fuselage measured about 110-inches radially (lengthwise) and the entire length should have been 122-inches long. The center wing was found with the bottom facing up and the left inboard engine (No. 2) engine pylon was still attached. The left center wing leading edge was still attached and exhibited heat damage, melting, blistering of the paint and fire consumption of some of the material. Fire damage and blistering of the paint was observed on the inside of the bottom on the center wing section. The most intense fire damage on the inside of the bottom of the center wing section was noted to be in the vicinity of the cross feed manifold. A section of the outer left wing measuring 39-inches long was still attached to the center wing.

From the seam lines this piece was comprised of 2½ panels with one complete panel. Four panels make up a complete configuration width. A single center wing stringer plus a piece of the upper center wing, measuring 93-inches long x 12.5 inches wide with a lap joint visual, was found in the brush on the west side of highway 395.

Note: A wreckage diagram is included as an appendix to this report.

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Two pieces of the top of the center wing were recovered. One piece measured 150-inches long x 79-inches wide (80 inches wide in the entire panel width) with a total of 11 stringers. On this piece a 30-inch section of the outer wing was still attached. The other piece of top of the center wing that was recovered measured 130-inches long x 38-inches wide with a total of 5 stringers.

Right outer wing with rainbow fitting and No. 4 engine nacelle area:

The wing bolts were still secure. The wing was found bottom facing up. The wing exhibited extensive fire damage which included melting of the skin and the fire consumed some areas. The flap plus flap track was found in the area and was separated from the wing.

The empennage / tail section:

The empennage / tail section was found separated from the rest of the aircraft and was lying on its right side. The empennage / tail section was separated in line with the upper escape hatch attachment. The cargo door and the ramp were both found detached from the aircraft structure. The right horizontal stabilizer was fractured at the root of the vertical stabilizer. The left horizontal stabilizer was intact, with the counterweight still attached, and outer tip exhibited impact damage. The entire left horizontal stabilizer tip was bent downwards, the leading edge of the tip was bent aft, and the skin was compressed. The elevator and rudder boost packs were still attached to empennage. The vertical tail was intact but exhibited impact damage. A hole was observed on the right side middle manual of the rudder. This hole did not pass through to the left side. The lower rudder surface near the horizontal stabilizers was pushed forward and deformed.

2.11 Engine Examination

Engine No. 1

As numbered left to right aft looking forward, SN AE100843, PN 6846583. The last overhaul according to the data plate took place on June 12, 1984 at 8,342 hours on the engine. The torque meter was intact but separated from the inlet housing. A section of the reduction gearbox housing and the drive gears were still attached to the torque meter. The accessory and reduction gearbox housings were fractured and the gears were exposed. The compressor and turbine cases exhibited no breaches. All of the 1st stage compressor blades were intact and none exhibited any bending. The engine was still attached to the pylon and the pylon exhibited fire damage and blistering of the paint. The engine was covered in soot forward of the heat shield.

Engine No. 2

SN AE101504, PN 6897151 as identified by the data plate. The last overhaul according to the data plate took place on October 23, 1978 at 1,229 hours on the engine. The engine was still attached to the pylon. Examination of the 1st stage compressor blades and the 4th stage turbine blades revealed that all the blades were intact and that the blade tips were slightly bent in the direction opposite rotation. The accessory gearbox remained still attached to the engine however; the left side of the accessory gearbox was fractured exposing the accessory gears. The reduction gearbox was fractured exposing the reduction gears. Fire/heat distress was observed forward of the heat shield but none aft of the heat shield. The torquemeter assembly was not attached to the inlet housing.

Engine No. 3

SN AE101267, PN 6897151 as identified by the data plate. The last overhaul according to the data plate took place on May 5, 1984 at 8,454 hours on the engine. The torquemeter was intact but separated from the inlet housing. A portion of the reduction gearbox and drive gear was still attached to the torque meter. The reduction gearbox housing was fractured and the gears were separated from the pinion gear. The compressor and turbine cases exhibited no breaches and the engine remained attached to the pylon. All of the 1st stage compressor blades were intact and some were bent in the direction opposite rotation (the engine rotates counterclockwise from the rear). The remaining 1st stage blades were normal. No signs of fire damage were observed on the engine. Heavy sooting and fire damage was noted on the inlet housing assembly.

Engine No. 4

SN A101267, PN 6897151 as identified by the maintenance records. The 1st and 2nd stage compressor blades were heavily bent in the direction opposite rotation to the point that the blades were laying one on top of the other. The inlet housing, accessory gearbox, and reduction gearbox were all consumed in the post-crash fire. Examination of the 4th stage turbine blades revealed that they were bent over in the direction opposite rotation. The engine was still attached to the pylon and no fire damage was observed aft of the heat shield. The compressor and turbine cases exhibited no breaches.

PROPELLERS

All sixteen-propeller blades were accounted for with six of the blades still attached to the propeller hub and the remaining blades scattered in and around the general area where the wings, propeller assemblies, and engines were found. All of the propeller blades found separated from the propeller assembly exhibited varying degrees of damage which included tip fractures, multiple airfoil fractures, blade butt fractures, airfoil impact marks, airfoil scrape marks, bent airfoils, and twisted airfoils. All the blades were found imbedded into the soil and some exhibited fire damage to the de-icing boot and/or were covered with soot. Three of the propeller blades found separated from the propeller assemblies had numbers on the airfoil. Those numbers were N231, N239471, and N239471.

All four-propeller assemblies were accounted for and were found in and around the general area where the wings and engines were found but were all separated from their respective engine.

2.12 Engine Maintenance and Records.

Engine Position #1

Component	Serial Number	Part Number	Total Time	TSO
Engine	AE 100843	6846583	10100.4	1757.1
Gearbox	Unknown	Unknown	Unknown	Unknown
Power Section	Unknown	Unknown	Unknown	Unknown
Torquemeter	Unknown	Unknown	Unknown	Unknown
Fuel Control	Unknown	Unknown	Unknown	Unknown
Temp Datum Valve	Unknown	Unknown	Unknown	Unknown
Coordinator	Unknown	Unknown	Unknown	Unknown
Fuel Pump	Unknown	Unknown	Unknown	Unknown
Speed Control	Unknown	Unknown	Unknown	Unknown
Speed Valve	Unknown	Unknown	Unknown	Unknown
Torquemeter Pickup	Unknown	Unknown	Unknown	Unknown

Engine Position #2

Component	Serial Number	Part Number	Total Time	TSO
Engine	AE 101504	6897151	9943.5	3802.5
Gearbox	Unknown	Unknown	Unknown	Unknown
Power Section	Unknown	Unknown	Unknown	Unknown
Torquemeter	Unknown	Unknown	Unknown	Unknown
Fuel Control	Unknown	Unknown	Unknown	Unknown
Temp Datum Valve	Unknown	Unknown	Unknown	Unknown
Coordinator	Unknown	Unknown	Unknown	Unknown
Fuel Pump	Unknown	Unknown	Unknown	Unknown
Speed Control	Unknown	Unknown	Unknown	Unknown
Speed Valve	Unknown	Unknown	Unknown	Unknown
Torquemeter Pickup	Unknown	Unknown	Unknown	Unknown

Engine Position #3

Component	Serial Number	Part Number	Total Time	TSO
Engine	AE 101267	6897151	10875.0	2420.0
Gearbox	Unknown	Unknown	Unknown	Unknown
Power Section	Unknown	Unknown	Unknown	Unknown
Torquemeter	Unknown	Unknown	Unknown	Unknown
Fuel Control	Unknown	Unknown	Unknown	Unknown
Temp Datum Valve	Unknown	Unknown	Unknown	Unknown
Coordinator	Unknown	Unknown	Unknown	Unknown
Fuel Pump	Unknown	Unknown	Unknown	Unknown
Speed Control	Unknown	Unknown	Unknown	Unknown
Speed Valve	Unknown	Unknown	Unknown	Unknown
Torquemeter Pickup	Unknown	Unknown	Unknown	Unknown

Engine Position #4

Component	Serial Number	Part Number	Total Time	TSO
Engine	AE 100814	Unknown	Unknown	2612.9
Gearbox	Unknown	Unknown	Unknown	Unknown
Power Section	Unknown	Unknown	Unknown	Unknown
Torquemeter	Unknown	Unknown	Unknown	Unknown
Fuel Control	Unknown	Unknown	Unknown	Unknown
Temp Datum Valve	Unknown	Unknown	Unknown	Unknown
Coordinator	Unknown	Unknown	Unknown	Unknown
Fuel Pump	Unknown	Unknown	Unknown	Unknown
Speed Control	Unknown	Unknown	Unknown	Unknown
Speed Valve	Unknown	Unknown	Unknown	Unknown
Torquemeter Pickup	Unknown	Unknown	Unknown	Unknown

Note: Individual component times and history were not documented. The above times were provided to the author from the owner.

2.13 Additional Information

Focus of the investigation was pointed toward aircraft structural issues based on this aircraft models' previous history. It was the decision of the NTSB IIC and the investigation team that investigation of the engines would be limited to visual on-site observations only.

3. FINDINGS & CONCLUSIONS.

- Visual examination on-site of all four engines did not reveal any evidence of any pre-impact mechanical failures. All damages to the engines were the result of impact forces caused by the crash and the post crash fire.
- All four engines were operating at the time of impact with terrain. Evidence of engine operation at the time of impact was blade bending in the opposite direction of engine rotation noted on the compressor and turbine sections as well as missing gear teeth on accessory gears.

Appendix A.

On-Site

Photographs

At

Walker, California



Photo #1 – Engine wreckage field



Photo #2 – Engine wreckage field

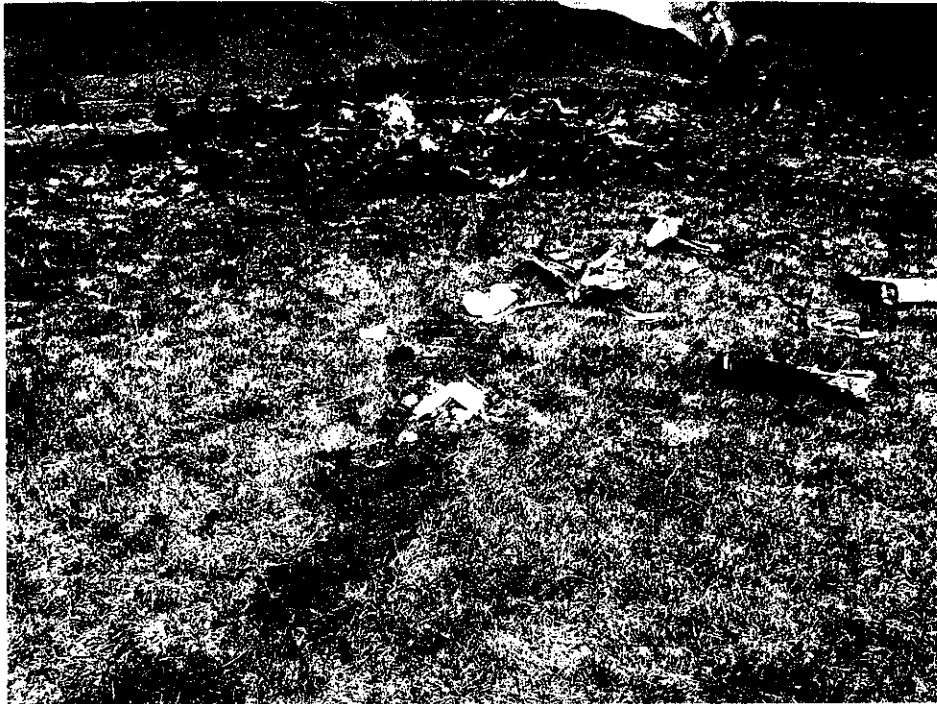


Photo #3 – Aircraft wreckage field



Photo #4 – Aircraft wreckage field



Photo #5 – Aircraft parts



Photo #6 – Aircraft debris

Engine S/N AE100843, AE 101504, AE 101267, A100814 --- June 17, 2002



Photo #7 – Flight path

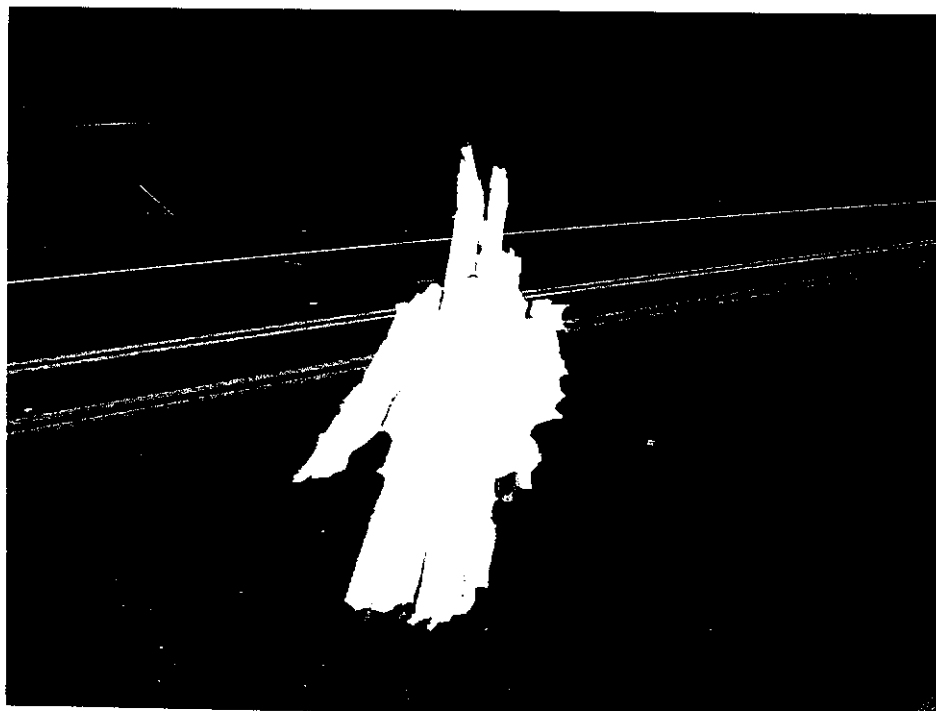


Photo #8 – Debris on road



Photo #9 – Aircraft wing



Photo #10 – Prop and gearbox assembly



Photo #11 – Engine inlet



Photo #12 – Engine inlet



Photo #13 – Engine rear turbine area

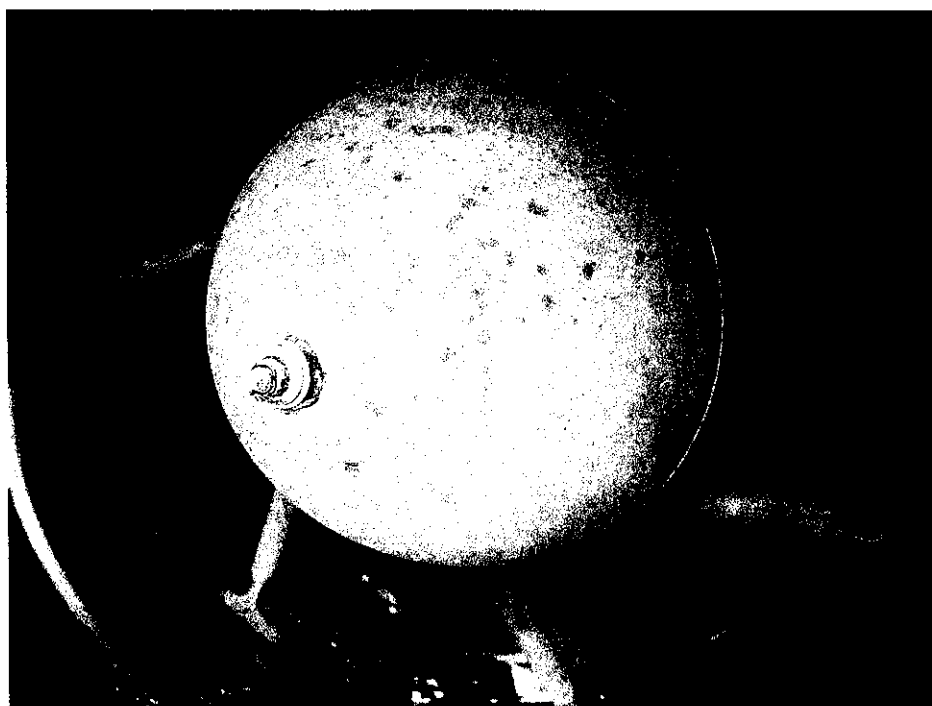


Photo #14 – Engine rear turbine area



Photo #15 – Engine compressor



Photo #16 – Engine compressor



Photo #17 – Engine debris



Photo #18 – Engine debris



Photo #19 -- Engine debris

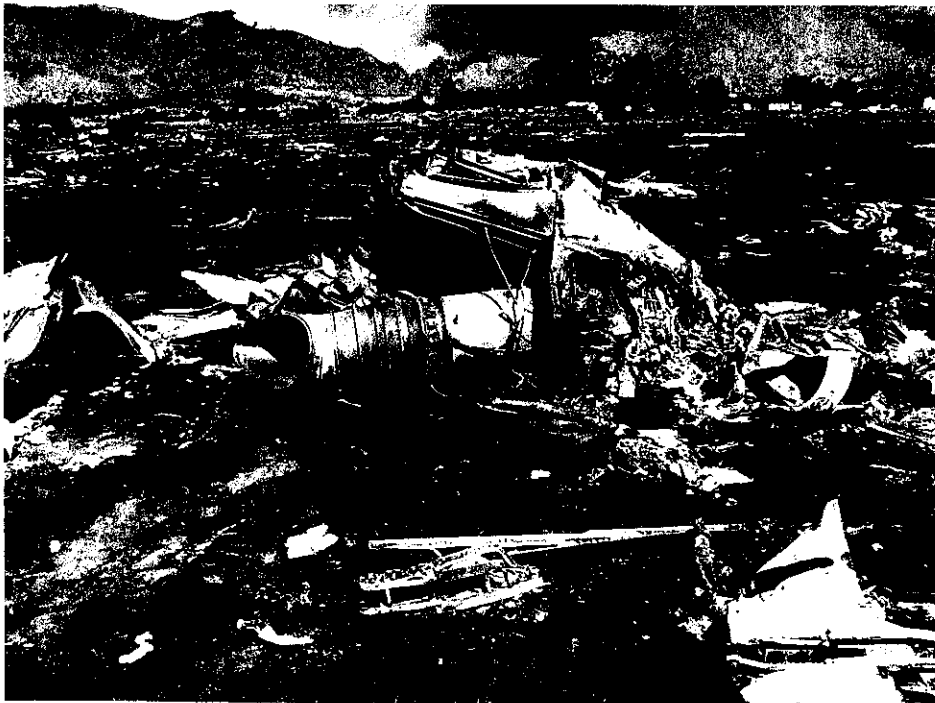


Photo #20 -- Engine debris



Photo #21 – Engine debris



Photo #22 – Engine debris



Photo #23 – Prop and gearbox assembly



Photo #24 – Prop and gearbox assembly

Appendix B.

Engine #1

Logbook

T-130 #1

T56A-9D

S/N AE100843

The Standard



The Standard

ENGINE LOG

ASA-SE-1

Engine Record General Information

Manufacturer

Allison

Model

T-56A-9D

Serial

AE 8400 100843

Type Certificate

This engine is currently installed in aircraft

T-130 #1

Minimum Octane Fuel

Oil Grade: Summer

Winter

Magnetos Time

Point Setting

Firing Order

Spark Plug Gap

Manufacturer recommended overhaul at

hours

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The Standard Engine Log
SE-1

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ASA-SE-1

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P N R A S A - S E - 1



9 781560 271659

DATE 19 <u>94</u>	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
4/20			545.4	<p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4) and was determined to be in airworthy condition.</p> <p>Total Time: <u>Unknown</u> Date: <u>4/20/94</u></p> <p>Signed: <u>[Signature]</u></p> <p>HAWKINS & POWERS AVIATION, INC. FAA CRS BZBR701C GREYBULL, WYOMING</p>
4/20/94				<p>INSTALLER ENG # 1 Position T-130 TAT2008295</p>

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DATE 19 <u>95</u>	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
11 5 APR 1995			788.09	<p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>788.09</u></p> <p>Signed: <u>[Signature]</u> Date: <u>11 5 APR 1995</u></p> <p>HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> GREYBULL, WYOMING</p>

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DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
12/11 JUN 1996				<p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>879.5</u></p> <p>Signed <u>[Signature]</u> Date <u>12/1 JUN 1996</u></p> <p>HAWKINS & POWERS AVIATION, INC. FAA CRS BZBR701C GREYBULL, WYOMING</p> <p style="text-align: right;">TAT 20417.05 WO 96-0035</p>
14 JUN 96			879.5	<p>Removed + Replace turbine assy FAW TO 25-T56-16 Leak Check Good, Perform seal Break-in TAT 20417.05 TSO ENG 879.5 S/U installed 0428 Ref WO # 96-053 from [redacted]</p>

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DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
MAR 24 1997			9446.5	<p>Total time and time and date of last overhaul verified from military records on file at Hawkins and Powers Aviation, Inc. FAA CRS [redacted]</p> <p>Signature: <u>[Signature]</u></p> <p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>9446.5</u></p> <p>Signed <u>[Signature]</u> Date <u>MAR 24 1997</u></p> <p>HAWKINS & POWERS AVIATION, INC. FAA CRS [redacted] GREYBULL, WYOMING</p> <p style="text-align: right;">TT: 9446.5 TSO: 1104.90</p>
MAR 24 1997			9446.5	<p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>9446.5</u></p> <p>Signed <u>[Signature]</u> Date <u>MAR 24 1997</u></p> <p>HAWKINS & POWERS AVIATION, INC. FAA CRS [redacted] GREYBULL, WYOMING</p> <p style="text-align: right;">"B" CHECK TSO: 1104.96 TAT: 206424 WO # 96-2211</p>

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[illegible]

DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
APR 15 1999			1253.31	<p>I certify this <u>ENGINE</u> has been inspected in accordance with a <u>C</u> inspection and was determined to be in airworthy condition. Signature <u>[Signature]</u> Certificate No. <u>[Redacted]</u> Date <u>APR 15 1999</u> C/Eng. Total Time <u>20890.86</u> HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> Work Order No. <u>6107C</u> Greybull, Wyoming</p>
AUG 30 1999			1482.77	<p>REMOVED ENG T-130#1 POSITION METAL ON R.G.B MAG PLUG ENG TSD: 1482.77 ENG TT 9824.37 TAT. 21021.92 <u>[Signature]</u> HAWKINS, POWERS <u>[Redacted]</u></p>

DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
SEP 17 1999		1482.71		REPLACED ROLLER BEARING ON MAIN DRIVE BEAR I.A.W 2J-TSG-13 REF WO# 7534. HAWKINS, Powers Signature [Redacted]
OCT 25 1999		1482.71		OP'S CHECKED ENGINE I.A.W IC-130A-2-4, IC-130B-2-11, C/W COMPRESSOR WASH, INSP ECTED R.G.B. 3 POWER SECTION MAG PLUGS 3 OIL SCREENS C/W PERFORMANCE RBN, OPS 3 LEAK CK GOOD ENG PERFORMANCE 100% REF WO# 7660 HAWKINS, Powers Signature [Redacted]

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DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
AUG 12 2001		1482.71		INSTALLED #1 POSITION T-130 TAT 21592.3 ENGINE TT 9825.97 I.A.W IC-130A-2-4 REF WO# 12800 HAWKINS, Powers Signature [Redacted]
SEP 1 2001		1580.09		I certify this <u>ENGINE</u> has been inspected in accordance with a <u>Check</u> inspection and was determined to be in airworthy condition. Signature <u>[Signature]</u> Certificate No. [Redacted] Date <u>SEP 1 2001</u> A/C/E. Total Time <u>21695.62</u> HAWKINS & POWERS AVIATION, INC. FAA CRS [Redacted] Work Order No. <u>13311</u> Greybull, Wyoming

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DATE 19__	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
AUG 11 2001	2748.44			A Ck CW TAT: 21598 ³ WO# 12776 [redacted] <i>Shaggy</i>
AUG 30 2001	1530.09			A Ck CW TAT: 21645 ⁵ WO# 130334 [redacted] <i>Shaggy</i>
SEP 20 2001	1554.88			A Ck CW TAT: 21672 ¹⁰ WO# 13230 [redacted] <i>Shaggy</i>
NOV 12 2001	1590.57			A Ck CW TAT: 21706 ⁴ WO# 13376 [redacted] <i>Shaggy</i>
MAR 29 2002	1632.38			A Ck CW TAT: 21779 ¹ WO# 14439 [redacted] <i>Shaggy</i>
APR 21 2002	1649.09			A Ck CW TAT: 21763 ⁴ WO# 14544 [redacted] <i>Shaggy</i>
MAY 12 2002	1675.51			A Ck CW TAT: 21791 ¹⁰ WO# 14806 [redacted] <i>Shaggy</i>
JUN 2 - 2002	1703.22			A Ck CW TAT: 21821 ¹⁰ WO# 15111 [redacted] <i>Shaggy</i>

Appendix C.

Engine #2

Logbook

100-ADD 1-130 #2
AE101504

ENGINE LOGBOOK

JEPPESAN
SANDERSON

EA-ELB

ENGINE LOGBOOK

Logbook No. 1
From FEB 1998
To _____

MANUFACTURER ALLISON MODEL T56
SERIAL NUMBER AE101504 DATE OF MANUFACTURE 3 FEB, 1978
RATED BHP/SHP/THRUST _____ RPM 13820 MANFOLD PRESSURE _____
FUEL REQUIRED OCTANE/PERFORMANCE _____ ALTERNATE FUEL _____
PROPELLER SER. NO. _____ DATE INSTALLED _____
REGISTERED OWNER HAWKINS 3 Powers AVIATION
STREET & NO. 2441 HIGHWAY 14
CITY, STATE, ZIP GREYBULL WY 82426
DATE PURCHASED _____ DATE SOLD _____
REGISTERED OWNER _____
STREET & NO. _____
CITY, STATE, ZIP _____
DATE PURCHASED _____ DATE SOLD _____
TBO HOURS _____ LAST OVERHAUL DATE/HOURS 14 MAY 1979

2

[illegible]

1

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
FEB 12 1998	9115.1	This QEC has been modified for the installation of HP-C-130A Chip Detector Circuit per 337 for the following aircraft: T-131-337 dated 6/6/97, T-130-337 dated 5/10/97, T-133-337 dated 5/21/97. <i>[Signature]</i>
FEB 12 1998	9115.1	Total time and date of last overhaul verified from military records on file at Hawkins and Powers Aviation, Inc. FAA CTS <i>[Signature]</i> Engine TSO: 2974.1 <i>[Signature]</i>

16

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
FEB 12 1998	9115.1	C/W HAWKINS & POWERS FAA APPROVED HP-C-130A INSPECTION GUIDE "C CHECK" REPLACED ALL (6) FUEL NOZZLES, REAR BEARING SUPPORT, 1 IN & EXCITER & AC GEN INSTALLED ENGINE #2 POSITION T-131. TAT 19235.02 ENG TSO 2974.1 OP'S & LEAK CHECK GOOD I.A.W IC-130A-2-4 & IC-130A-2-11 ENGINE PERFORMANCE 102.2% FOR ADDITIONAL INFORMATION REF WOT 3909 <i>[Signature]</i> HAWKINS & POWERS
AUG 1 1998	9176.10 9282.91	REPLACED BOTH IGNITORS AND LEADS I.A.W 2J-T56-16 <i>[Signature]</i> TAT 19282.91 TSO 3021.96 HAWKINS & POWERS
AUG 8 1998	9176.10	Engins Removed T-131 #2 POSITION TAT 19296.29 FOR CHIPS ON R.G.B MAG PLUG ENG TSO 3035 REF WOT 5194. <i>[Signature]</i> HAWKINS & POWERS

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
SEP 11 1998	9176.10	INSTALLED R.G.B. S/N AG-021272 I.A.W 2J-T56-16. R.G.B. TSO 3549.3 ENG TSO: 3035.10 OP'S 3 LEAK CK GOOD I.A.W IC-130A-2-4 REF WO# 5207 OLD R.G.B S/N A 20282A HAWKINS 3 POWERS AVI 11-1-98 ██████████
SEP 15 1998	9176.10	INSTALLED ENGINE # 3 POSITION N 8230H TAT 14872.1 ENG TSO 3035.10 I.A.W IC-130A-2-4 REF WO 5155 ██████████ HAWKINS 3 POWERS ██████████
APR 14 1999	9184.3	REMOVED ENGINE F.O.M TAT 14880.3 # 3 POSITION N 8230H REF WO# 6573. Eng TSO 3043.3. ██████████ HAWKINS 3 POWERS ██████████

18

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
AUG 26 1999	9184.3	INSTALLED ENG T-130 TAT 2005.14 #2 POSITION OP'S LK CHECKS GOOD I.A.W IC-130A-2-4 REF WO# 7473C. ENG TSO 3043.3 HAWKINS 3 POWERS ██████████
OCT 4 1999	9238.83	REPLACED T.D AMP ENG TSO 3097.83 TAT: 21059.67 REF WO# 7778C. HAWKINS 3 POWERS ██████████
APR 18 2000	9323.59	ACK CMW TAT 2189.53 WO# 8005 ██████████ TSO: 3128.06 ██████████
MAY 9 2000	9326.22	ACK CMW TAT 2109.67 WO# 8061C ██████████ ENG TSO 3130.69 ██████████
MAY 27 2000	9344.49 P.D. 9289.96	ENGINE REMOVED T-130 #2 POSITION TAT 21112.98 ENG TSO: 3148.98 FOR CRACKED OUTER COMBUSTION LINER REF WO# 9000 HAWKINS 3 POWERS ██████████

19

DATE: 15 1992 TIME: 2002

INSPECTION OR MAINTENANCE PERFORMED
AGENCY & CERTIFICATE NO.

has been inspected TSO: 1447.86
in a C inspection
and was found
ned to be in airworthy condition.
Certificate No. [REDACTED]
A/C Eng. Total Time 20890.86
S & POWERS AVIATION, INC.
IC Work Order No. 6107C
eybull, Wyoming

15 1992 2002

FUEL NOZZLES CD'S?
TAT: 20890.86
TSO: 1447.86 REF WO# 6107C
TEMP DATUM VALUE
W 15-130A-2-4 TAT 20929.5' ENG TSO:
1484.55 ER WO# 7009C
HAWICINS? Powers

DATE: 18 2000 TIME: 1010

INSPECTION OR MAINTENANCE PERFORMED
AGENCY & CERTIFICATE NO.

WO# 8005

18 2000 1010

WO# 8867

9.81

has been inspected ENG TSO: 1742.0
with a check inspection
ned to be in airworthy condition.
Certificate No. [REDACTED]
A/C Eng. Total Time 21185.09
& POWERS AVIATION, INC.
IC Work Order No. 9405C
eybull, Wyoming

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
12 2 2000	9289.96	REPLACED OUTER COMBUSTION LINER ALL FUEL NOZZLES (6 EA) I.A.W 2J-T56-16 314 OPS CHECK GOOD I.A.W IC-130A-2-4 REF WO#9014N ENG TSO 3148.96. HAWKINS 3 POWERS WILLIAM J. HAWKINS [REDACTED]
28 2000	9289.96	RAN ENGINE ENGINE PERFORMANCE 100.4 LEAK CHECK 3 OPS CK GOOD I.A.W IC-130B-2-11 3 IC-130A-2-4 CHG TSO 3148.96 REF WO#9074N- HAWKINS 3 POW WILLIAM J. HAWKINS [REDACTED]

20

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
UL 21 2000	9289.96	INSTALLED ENGINE #2 POSITION T-130 TAT 21219.22, ENG TSO 3148.96 IAW IC-130A-2-4 OP'S 3' LEAK CK GOOD IAW IC-130A-2-4 3' IC-130A-2-11 ENG PERFORMANCE 99.7% RGR W/O# 9541C HAWKINS 3' Powers Signature
UL 06 08 2000	9382.52	I certify this ENGINE has been inspected ENG TSO: 2641.2 in accordance with a Recheck inspection and was determined to be in airworthy condition. Signature Signature Certificate No. Number Date 06 08 2000 A/C Eng. Total Time 21125.69 HAWKINS & POWERS AVIATION, INC. FAA CRS Number Work Order No. 9405C Greybull, Wyoming

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
MAR 8 2001	9598.77	I certify this <u>ENGINE</u> has been inspected <u>TSO: 3331.08</u> in accordance with a <u>C Check</u> inspection and was determined to be in airworthy condition. Signature <u>[Signature]</u> Certificate No. <u>[Redacted]</u> Date <u>MAR 8 2001</u> A/C Log. Total Time <u>2146134</u> HAWKINS & POWERS AVIATION, INC. FAA CRS BZBR701C Work Order No. <u>101107C</u> Greybull, Wyoming
APR 28 2001	9637.95	REPLACED ALL FUEL NOZZEL'S (6) I.A.W IC-130A-2-4 TAT 21440.52 ENG TSO: 3370.26 REF WO# 11666C. <u>[Signature]</u> HAWKINS & POWERS <u>[Redacted]</u>

22

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
SEP 1 2001	9893.05	I certify this <u>Engine</u> has been inspected <u>TSO: 3625.36</u> in accordance with a <u>B Check</u> inspection and was determined to be in airworthy condition. Signature <u>[Signature]</u> Certificate No. <u>[Redacted]</u> Date <u>SEP 1 2001</u> A/C Log. Total Time <u>21695.62</u> HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> Work Order No. <u>13311C</u> Greybull, Wyoming
MAR 29 2002	9930.8	Replaced All 6 Fuel Nozzels IAW TO IC-130A-2-4 TAT 21747.91 ENG TSO 3677.65 Ref WO# 13965C HAWKINS & POWERS <u>[Signature]</u> <u>[Redacted]</u>

23

FINANCE RECORD

DATE	TIME	LOCATION	AGENCY & CERTIFICATE NO.
21 2000	22	ACK CON TAT 121	#9260K
25 2000	22	ACK CON TAT 121	954C
2001	22	ACK CON TAT 121	978C
2001	22	ACK CON TAT 121	#7894C
2001	22	ACK CON TAT 121	#10311C
2001	22	ACK CON TAT 21344	#10676C
2001	22	ACK CON TAT 21344	#1794C
3 2001	22	ACK CON TAT 21344	#11992C
12 2001	22	ACK CON TAT 21344	#12155C
2001	22	ACK CON TAT 21344	17385
2001	22	ACK CON TAT 21344	0#12581
2001	22	ACK CON TAT 21344	10#1279C
2001	22	ACK CON TAT 21344	0#13033C
2001	22	ACK CON TAT 21344	#13250
2001	22	ACK CON TAT 21344	13370
2002	22	ACK CON TAT 21344	14439
2002	22	ACK CON TAT 21344	14544
2002	22	ACK CON TAT 21344	1086

ENANCE RECORD

[illegible]

Appendix D.

Engine #3

Logbook

T56-A9D

AE-101267

#3

ENGINE LOGBOOK

in the USA

EA-ELB

ENGINE LOGBOOK

DEC 84

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Ship to: 7383 6WN Road, Casper, WY 82604-1835
(800)443-9250 • (307)266-3838 • FAX: 307-472-5106

ENGINE LOG TABLE OF CONTENTS

SECTION SEQUENCE	DESCRIPTION OF SECTION	PAGE NUMBER
1st SECTION	ENGINE DESCRIPTION	3
2nd SECTION	INSTALLATION	4
3rd SECTION	LIFE LIMITED PARTS OR OVERHAUL REQUIREMENTS	6
4th SECTION	AIRWORTHINESS DIRECTIVE COMPLIANCE	8
5th SECTION	FACTORY SERVICE BULLETIN COMPLIANCE	12
6th SECTION	INSPECTION AND MAINTENANCE RECORD	16

ENGINE DESCRIPTION

MANUFACTURER Lycoming Allison MODEL T56-A9

SERIAL NUMBER AE 101267 DATE OF MANUFACTURE _____

RATED SHP/SHP/THRUST _____ RPM _____ MANFOLD PRESSURE _____

FUEL REQUIRED OCTANE/PERFORMANCE _____ ALTERNATE FUEL _____

PROPELLER SER. NO. _____ DATE INSTALLED _____

REGISTERED OWNER HAWKINS & POWERS AVIATION

STREET & NO. 2441 HIGHWAY 14

CITY, STATE, ZIP GREYBULL WY 82426

DATE PURCHASED _____ DATE SOLD _____

REGISTERED OWNER _____

STREET & NO. _____

CITY, STATE, ZIP _____

DATE PURCHASED _____ DATE SOLD _____

TBO HOURS _____ LAST OVERHAUL DATE/HOURS _____

SECRET

[illegible]

LIFE LIMITED PARTS OR OVERHAUL REQUIREMENTS

NAME OF PART	ENGINE NUMBER	SERIAL NUMBER	APPLICATOR DATE/ENG. HOURS	DATE ON PART AT INSTAL	LIFE LIMIT OR T.B.O.	SCHED. REMOVAL DATE/ENG. HOURS	REMOVAL DATE/ENG. HOURS	AGENCY CERT. NO.
1. BIRD		ADT-10000	154					
2. 100-20074		ADT-10000	154					
3. 100-20074		ADT-10000	154					
4. 100-20074		ADT-10000	154					
5. 100-20074		ADT-10000	154					
6. 100-20074		ADT-10000	154					
7. 100-20074		ADT-10000	154					
8. 100-20074		ADT-10000	154					
9. 100-20074		ADT-10000	154					
10. 100-20074		ADT-10000	154					
11. 100-20074		ADT-10000	154					
12. 100-20074		ADT-10000	154					
13. 100-20074		ADT-10000	154					
14. 100-20074		ADT-10000	154					
15. 100-20074		ADT-10000	154					
16. 100-20074		ADT-10000	154					
17. 100-20074		ADT-10000	154					
18. 100-20074		ADT-10000	154					
19. 100-20074		ADT-10000	154					
20. 100-20074		ADT-10000	154					
21. 100-20074		ADT-10000	154					
22. 100-20074		ADT-10000	154					
23. 100-20074		ADT-10000	154					
24. 100-20074		ADT-10000	154					
25. 100-20074		ADT-10000	154					
26. 100-20074		ADT-10000	154					
27. 100-20074		ADT-10000	154					
28. 100-20074		ADT-10000	154					
29. 100-20074		ADT-10000	154					
30. 100-20074		ADT-10000	154					
31. 100-20074		ADT-10000	154					
32. 100-20074		ADT-10000	154					
33. 100-20074		ADT-10000	154					
34. 100-20074		ADT-10000	154					
35. 100-20074		ADT-10000	154					
36. 100-20074		ADT-10000	154					
37. 100-20074		ADT-10000	154					
38. 100-20074		ADT-10000	154					
39. 100-20074		ADT-10000	154					
40. 100-20074		ADT-10000	154					
41. 100-20074		ADT-10000	154					
42. 100-20074		ADT-10000	154					
43. 100-20074		ADT-10000	154					
44. 100-20074		ADT-10000	154					
45. 100-20074		ADT-10000	154					
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47. 100-20074		ADT-10000	154					
48. 100-20074		ADT-10000	154					
49. 100-20074		ADT-10000	154					
50. 100-20074		ADT-10000	154					
51. 100-20074		ADT-10000	154					
52. 100-20074		ADT-10000	154					
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60. 100-20074		ADT-10000	154					
61. 100-20074		ADT-10000	154					
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64. 100-20074		ADT-10000	154					
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67. 100-20074		ADT-10000	154					
68. 100-20074		ADT-10000	154					
69. 100-20074		ADT-10000	154					
70. 100-20074		ADT-10000	154					
71. 100-20074		ADT-10000	154					
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88. 100-20074		ADT-10000	154					
89. 100-20074		ADT-10000	154					
90. 100-20074		ADT-10000	154					
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94. 100-20074		ADT-10000	154					
95. 100-20074		ADT-10000	154					
96. 100-20074		ADT-10000	154					
97. 100-20074		ADT-10000	154					
98. 100-20074		ADT-10000	154					
99. 100-20074		ADT-10000	154					
100. 100-20074		ADT-10000	154					

INSPECTION & MAINTENANCE RECORD

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
22 Dec 89	446.3	I certify this <u>FAIR</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.169 (f) (5), and was determined to be in airworthy condition. Total Time: <u>447.3</u> Signed <u>[Signature]</u> Date <u>12-22-89</u> HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> GREYBULL, WYOMING
		11-July-1990 Total Times 562.3 hrs. I certify this engine has been inspected in accordance with an FAA approved inspection program and determined to be airworthy. <u>[Signature]</u> FAA CRS# <u>[Redacted]</u> Hawkins & Powers Aviation Greybull, WY.

16

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
		I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.169 (f) (5) and was determined to be in airworthy condition. Total Time: <u>1389.25</u> Signed <u>[Signature]</u> Date <u>3-18-91</u> HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> GREYBULL, WYOMING
		I certify this <u>ENGINE</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition. Total Time: <u>1451.649</u> 23 9775.02 Signed <u>[Signature]</u> Date <u>3-20-92</u> HAWKINS & POWERS AVIATION, INC. FAA CRS <u>[Redacted]</u> GREYBULL, WYOMING

17

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
22 APR 93	9841.89	<p>I certify this <u>ENG</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>9841.87</u></p> <p>Signed <u>[Signature]</u> Date <u>22 APR 93</u></p> <p>HAWKINS & POWERS AVIATION, INC.</p> <p>FAA CRS <u>[Redacted]</u></p> <p>GREYBULL, WYOMING</p>
4-20-94	9877.91	<p>I certify this <u>Eng</u> has been inspected in accordance with the procedures of an FAA Approved Inspection Program per FAR 91.409 (f) (4), and was determined to be in airworthy condition.</p> <p>Total Time: <u>[Redacted]</u></p> <p>Signed <u>Robert C. Whit</u> Date <u>4-20-94</u></p> <p>HAWKINS & POWERS AVIATION, INC.</p> <p>FAA CRS <u>[Redacted]</u></p> <p>GREYBULL, WYOMING</p>

18

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
16 AUG 1994	10016.92	EN6 Removed for Vibration checks in rear turbine support Hsein 116.8 CAS 116.8 TSD 1162.5
9 JUL 1996	10016.92	REPLACED TURBINE ASSEY 1.A.W TO: 2J-T56-16 S/N AT-T901389, REPLACED ALL (6EA) FUEL NOZZLES, ENG STARTER, FUEL CONTROL, T.D AMP, TD VALVE & DC GEN SEE HAWKINS' POWERS WO# 96-1225 FOR DETAILS 116.8 HAWKINS' POWERS T.S.D 1163.8 TT 9618.8
2 AUG 1996	9618.8	Engines and drive end of test overhauled and that from military records on file at Hawkins and Powers Aviation, Inc. P.O. Box 1225 Hawkins, Texas 75767 116.8 116.8

18

DATE	TOTAL TIME	INSPECTION OR MAINTENANCE PERFORMED AGENCY & CERTIFICATE NO.
8 2001	10385.68	<p>I certify this <u>ENGINE</u> has been inspected TSO: 1958.28 in accordance with a <u>C check</u> inspection and was determined to be in airworthy condition. Signature <u>[Signature]</u> Certificate No. <u>[Redacted]</u> Date <u>8 2001</u> A/C Reg. Total Time <u>21401.34</u> HAWKINS & POWERS AVIATION, INC. FAA CRS BZBR701C Work Order No. <u>10407</u> <u>Greybull, Wyoming</u></p>
11 2001	10582.37	<p>REPLACED #3 ENGINE R.G.B. TAT: 21598.0 ENG TSO: 2154.97 RGB TSO: 2138.7 R.G.B. SM ON AGO 20078 OP'S AND LEAK C/K GOOD IAIW 1C-130A-2-4 REF WO# 12729C. <u>[Signature]</u> HAWKINS & POWERS <u>[Redacted]</u></p>

24

[illegible]

Appendix E.

Engine #4

Logbook

T-130 #4

ENGINE LOGBOOK

EA-ELB

ENGINE LOGBOOK

Logbook No. ONE
From 11/1998
To _____

ENGINE DESCRIPTION

MANUFACTURER ALLISON MODEL T56 A9D

SERIAL NUMBER AE100814 DATE OF MANUFACTURE 3

RATED BHP/SHP/THRUST _____ RPM _____ MANFOLD PRESSURE _____

FUEL REQUIRED OCTANE/PERFORMANCE _____ ALTERNATE FUEL _____

PROPELLER SER. NO. _____ DATE INSTALLED _____

REGISTERED OWNER HAWKINS & POWERS AVIATION INC

STREET & NO. 2441 HIGHWAY 20 WEST

CITY, STATE, ZIP GREYBULL WY 82426

DATE PURCHASED _____ DATE SOLD _____

REGISTERED OWNER _____

STREET & NO. _____

CITY, STATE, ZIP _____

DATE PURCHASED _____ DATE SOLD _____

TBO HOURS _____ LAST OVERHAUL DATE/HOURS 9-19-1983

INSTALLATION

[illegible]

APR 09 2001
12 2002
INSTALLED ENGINE #4 POSITION T-130 TAT
IC-130A-2-4, IC-130B-2-11
C/W AND LEAK CK GOOD ENGINE PERFORMANCE
C/W C/W REF WO# 14499C
HAWKINS? Powers
[Signature]

APR 09 2001
12 2002
R.G.B Thrust Seal IAW TO 2J-756-16
TD AMP IAW IC-130A-2-4 S/N ON
C/W and LEAK checked good on both
C/W REF WO# 14509C ? 14538C
HAWKINS? Powers
[Signature]

INSPECTION & MAINTENANCE RECORD

INSPECTION OR MAINTENANCE PERFORMED
AGENCY & CERTIFICATE NO.

APR 09 2001
12 2002
INSTALLED TURBINE (HOT SECTION)
C/W ADT 901149, DISASSEMBLED R.G.B Rep-
ARMS AS REQUIRED, INSPECTED POWER
SECTION, REPLACED ALL FUEL NOZZLES
IN UNITS. REPLACED EXCITER BOX,
FUEL FLOW TRANSMITTER, COORDINAT-
OR COOLER & AC GEN. C/W HAWKINS?
Powers "C" SERVICE AND INSPECTION GUIDE.
C/W AND LEAKED CHECKED GOOD MAINT
C/W I.A.W IC-130A-2-4, IC-130A-19
IC-16, IC-130B-2-11 REF WO# 12912N.
HAWKINS? Powers
[Signature]

INSPECTION & MAINTENANCE RECORD

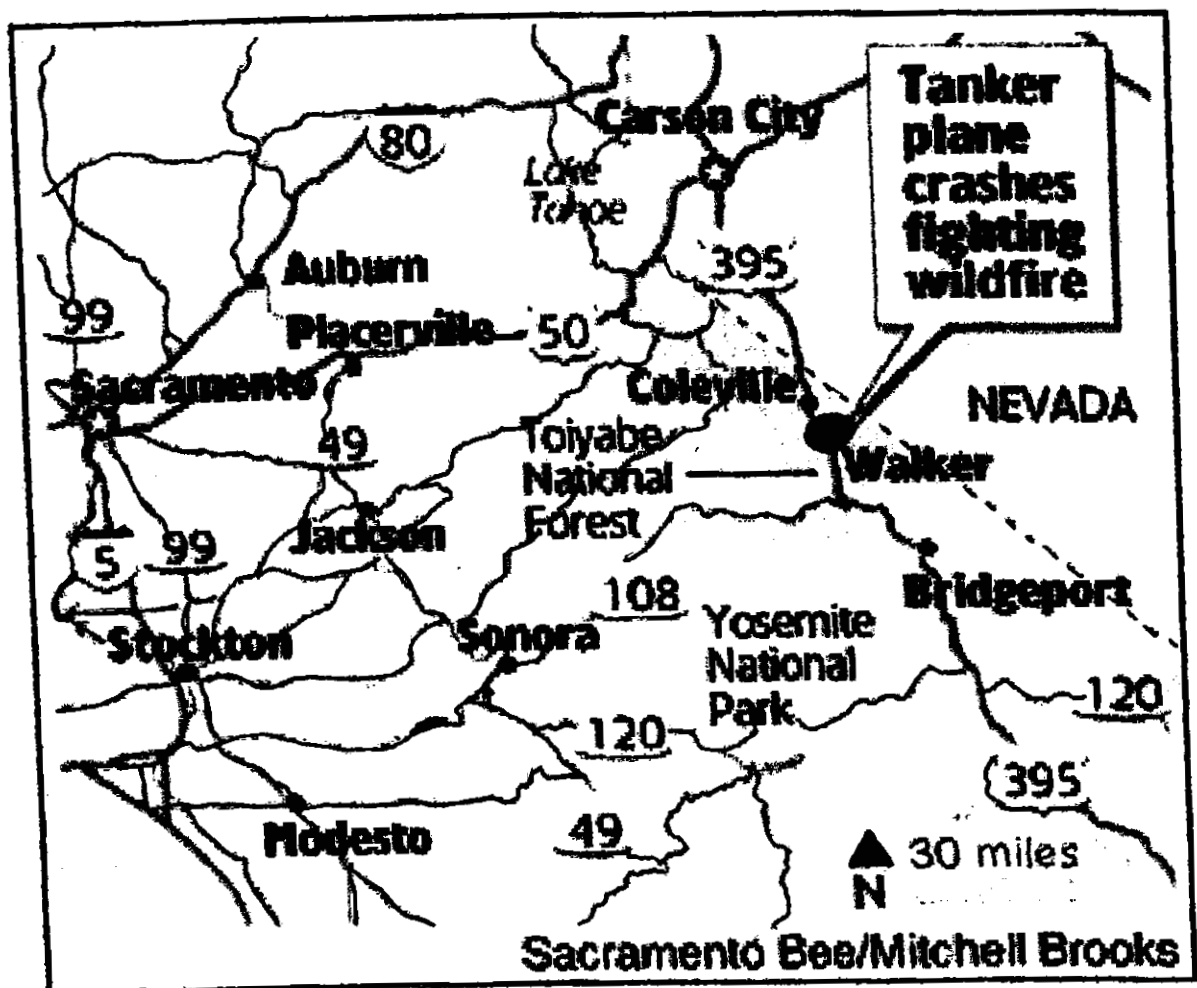
[illegible]

Appendix F.

Map

Of the

Area




Appendix G.

Participants

Appendix G. — Participants

Walker, California, June 18-20, 2002

Mr. George Peterson	NTSB	IIC	
Mr. Pierre Scarfo	NTSB	Party	
Mr. Mike Weber	Rolls-Royce	Party	
Mr. George James	Lockheed Martin	Party	
Mr. Carl Meyer	Hawkins & Powers	Party	Unknown
Mr. Bill Bulger	Forest Service	Party	Unknown
Mr. William Kunger	FAA	Party	

Appendix H.

Wreckage Diagram

LAX02GA201

