

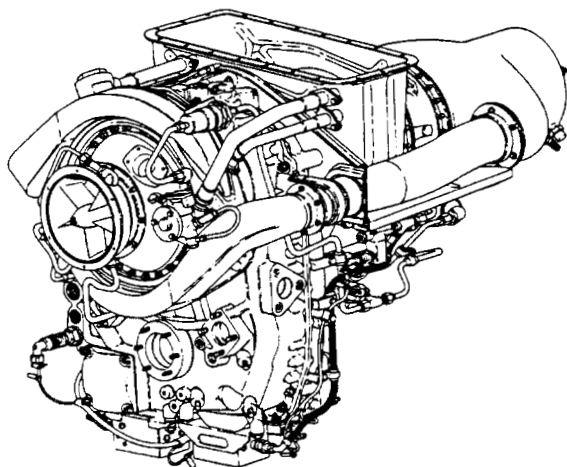


Rolls-Royce

Accident investigation

On-site investigation
report

Rolls-Royce
Model 250-C30P
Engine CAE 895064



Rogers Helicopters
Montello, Nevada

Michael A. Weber

Michael A. Weber
Senior Accident Investigator

Accident date: August 3, 2000
On-Site date: August 5-6, 2000
Report date: August 10, 2000

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1. SYNOPSIS

On August 3, 2000, about 18:57 hours Pacific Standard Time, a Bell 206L-1 helicopter, N1084, with an Allison engine, model 250-C30P, S/N 895064, operated by Rogers Helicopters, was destroyed when the helicopter abruptly rolled to the right and impacted terrain during hover after liftoff. The mishap occurred near Montello, Nevada. The helicopter was operated under 14 CFR Part 91 in support of fire fighting activities as a public use aircraft by the United States Bureau of Land Management. Visual meteorological conditions prevailed and no flight plan was filed for the positioning flight that was planned for Wells, Nevada. The pilot received minor injuries, one passenger seated in the left front cockpit received serious injuries and the remaining passenger who was seated in the left rear cabin facing forward received fatal injuries during the accident sequence.

This report represents the findings of the on-site examination at the Montello, Nevada Helibase.

2. FACTUAL INFORMATION

2.1 History of Flight

The pilot reported that between 1500 and 1830 there were wind gusts of 50 – 60 knots associated with a thunderstorm that passed west of their location. He tied down the main and tail rotor blades before the windstorm, and by 1845, the weather was clear enough to return to Wells and the wind had decreased to 5 – 10 knots. He pre-flighted the helicopter and untied the rotors. He said that the start and pre-takeoff checks were normal. He performed a normal hydraulics off control check. He picked the helicopter up to a 3-foot hover height and glanced down at the engine torque gage. Suddenly, the helicopter did a “violent snap roll” to the right coming to rest upright. He estimated that the whole event only lasted about ¼ second. The pilot added that there was no yaw associated with the event, it was a pure roll, and the engine operated “fine” the entire time. The engine was still running after the accident and he had to shut it off manually. There was no change in audible tone prior to the roll. He did not recall any movement or abnormal feedback in the antitorque pedals or the cyclic and collective controls prior to the accident. The pilot emphasized the violent nature of the roll. It was the pilot’s opinion that even an abrupt, full right cyclic input could not have produced the rare of roll that he experienced. The rotor blade tie down retention straps were found stowed inside the cabin after the accident.

2.2 Personnel Information

Pilot = Kenneth R. Carlton, age 52, commercial pilot license [REDACTED] issued 4/20/98.
Ratings = Airplane single engine land & sea rotorcraft -helicopter, Instrument airplane and helicopter.

2.3 Injuries to Persons

The pilot received minor injuries, one passenger received serious injuries and one passenger received fatal injuries as a result of the accident sequence.

2.4 Aircraft Information

Model	Bell 206L-1
Serial Number	45434
Registration Number	N10864
Airframe Total Time	5532.2
Last Annual Inspection	03/27/2000
Last 300 hour Inspection	5333.4
Last 100 hour Inspection	5521.1
Engine Model	Allison 250-C30P
Rating:	650 Shaft Horsepower
Serial Number	CAE 895064
Engine Total Hours	5541.3
Last 100 Hour Inspection	5530.2

2.5 Meteorological Conditions

Reported observations from the National Weather Service in Boise, Idaho.

Weather at Latitude 41.65 North, Longitude 114.43 West was reported as follows:
At 18:00, temperature was 86°F, relative humidity 25%, wind direction 190 at 8 mph with 15 mph gusts, precipitation 3.04.

2.6 Aids to Navigation

There were no known aids to Navigation. It was reported that the pilot was familiar with the local area/terrain.

2.7 Communications

N/A

2.8 Aerodrome Information

N/A

2.9 Flight Recorder

The aircraft was not equipped with either a flight data recorder or cockpit voice recorder

2.10 Wreckage and Impact Information

On-site examination of the aircraft revealed the following:

The mast and main rotor head was separated from the aircraft and was lying approximately 20 feet northeast of the wreckage. The aircraft windscreen was severely damaged. The left side forward and aft cabin doors were separated from the aircraft. The tail boom was fractured and resting inverted on the ground. The aircraft skid saddles were partially fractured. The tips of both main rotor blades were broken off and found 180° from each other. The main rotor blade tips were thrown approximately 450 feet from the wreckage. The flight control rods and linkage were broken and twisted at the mounting point areas. The flight control links to the main rotor and swash plate areas were removed by the Bell representative and sent to the NTSB laboratory for testing.

Location of the accident was Latitude 41° 16.71' North, Longitude 114° 12.58' West (GPS).

2.11 Engine Examination

On-Site engine examination revealed the following:

There was no indication of external damages to the engine. The right and left side engine mounts were broken as a result of the accident sequence. N1 rotates and checked continuous to the compressor and starter. N2 rotates and is continuous to the 4th stage turbine wheel and the PTO shaft. All pneumatic plumbing was in good condition. There was no damage to the Fuel control, PTG, Fuel pump assembly, or compressor bleed valve assembly. A fuel control system rigging check was completed and no anomalies were noted. The oil tank and oil cooler blower was damaged as a result of the accident sequence. The accident investigation team, consisting of the NTSB IIC, the Bell Helicopter representative and the author decided not to remove the engine for further examination. The engine was producing power.

2.12 Engine Maintenance and Records

Component	Serial Number	Part Number	TSO	Total Time
Engine	895064	23004545	N/A	5541.3
Gearbox	CAG 95064	23005655	N/A	5541.3
Compressor	CAC 90737	23005250	N/A	5541.3
Turbine	CAT 95064	23031925	941.0	5541.3
Fuel Control	BR 52406	23065147	173.0	5541.3
Governor	27539	2524692-11	281.5	5541.3
Fuel Pump	T-200008	6896843	2730.2	5541.3
Fuel Nozzle	AG-83408	6899001	1036.0	5541.3

2.13 Additional Information

None

3. FINDINGS & CONCLUSIONS.

1. The engine was operating during the crash and after final impact.
2. On-Site examination of the engine and accessories did not reveal any evidence of pre-impact mechanical failure.
3. Evidence of engine operation at the time of impact were observed as follows:
 - The main rotor head was separated from the aircraft.
 - The main rotor blade tips were thrown approximately 450 feet after they made contact with the ground.

5. APPENDICES

Appendix A.

Photographs



Photo #1 – Left side of the aircraft.



Photo #2 – Right side of the aircraft.



Photo # 3 – Front of the aircraft.

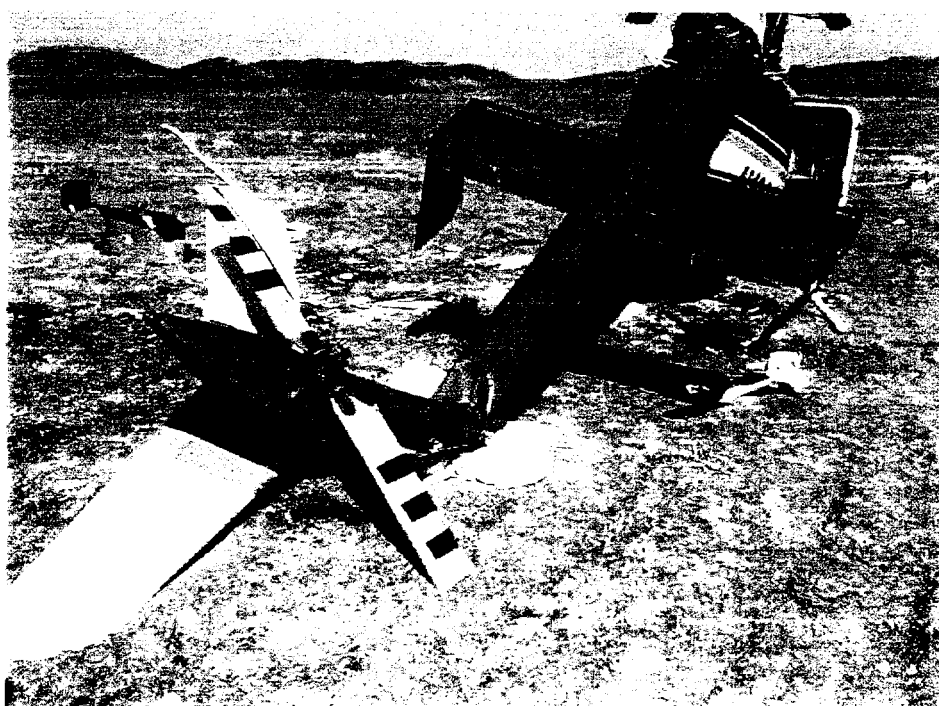


Photo # 4 – Rear of the aircraft.

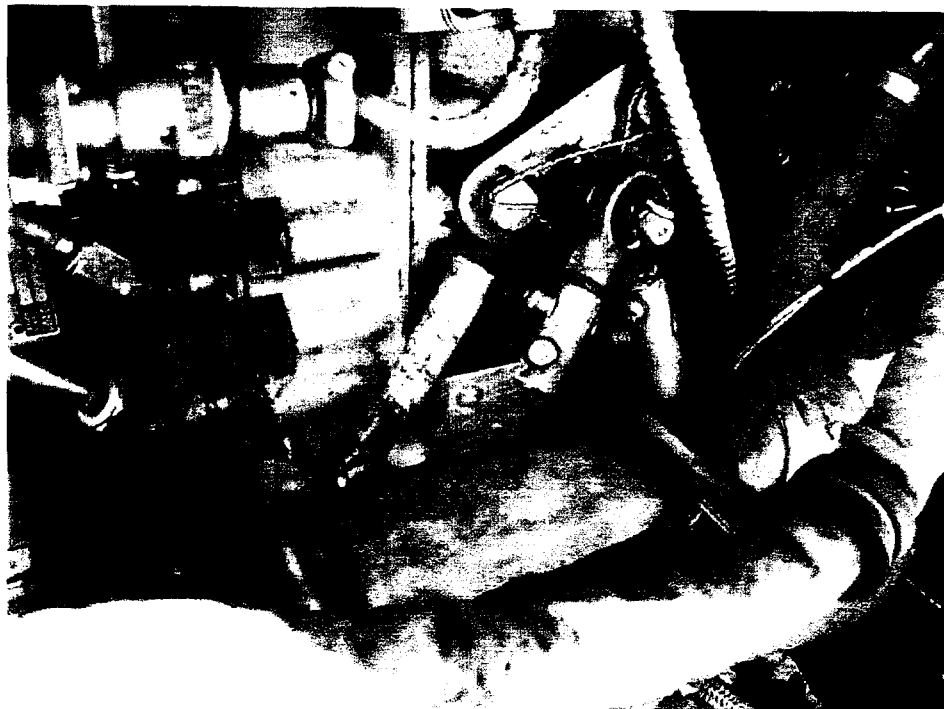


Photo # 5 – Right engine mount broken.

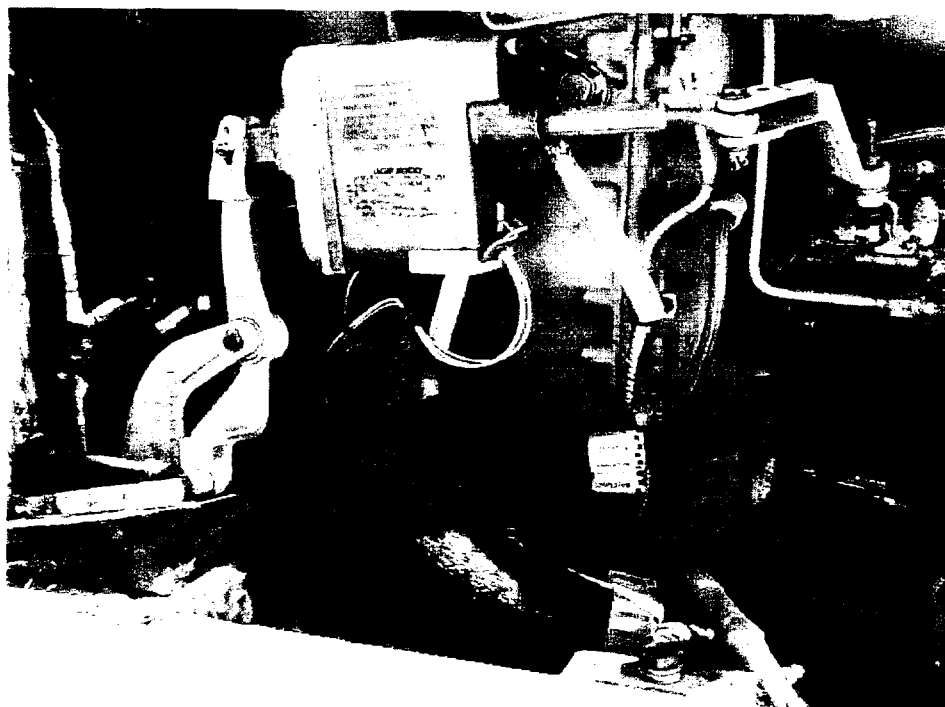


Photo # 6 – Left engine mount broken.

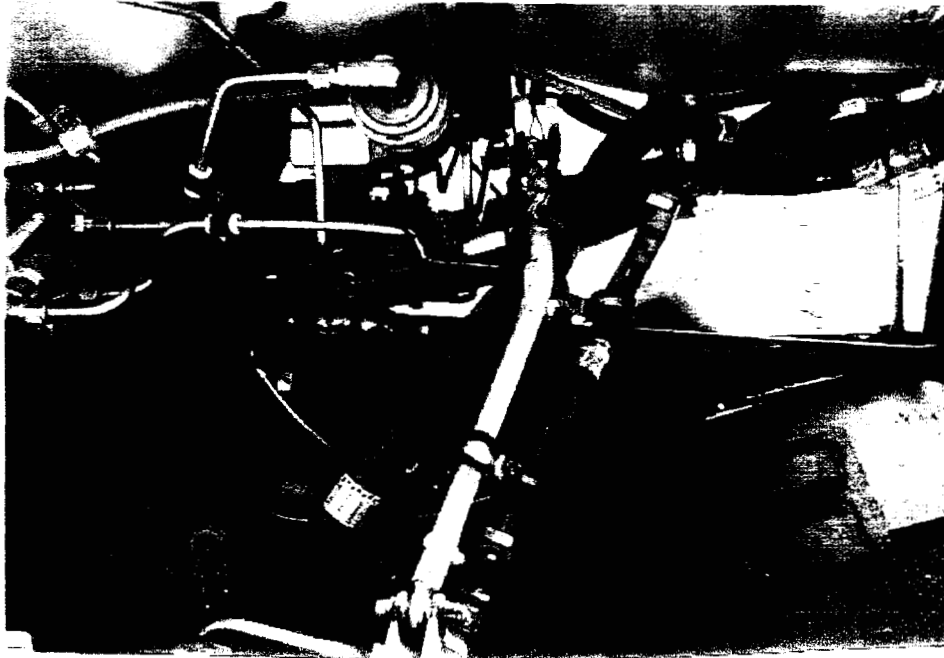


Photo # 7 – Rear engine mount bent.



Photo # 8 – Main rotor blade system.



Photo # 9 – Red main rotor blade tip.



Photo # 10 – White main rotor blade tip.

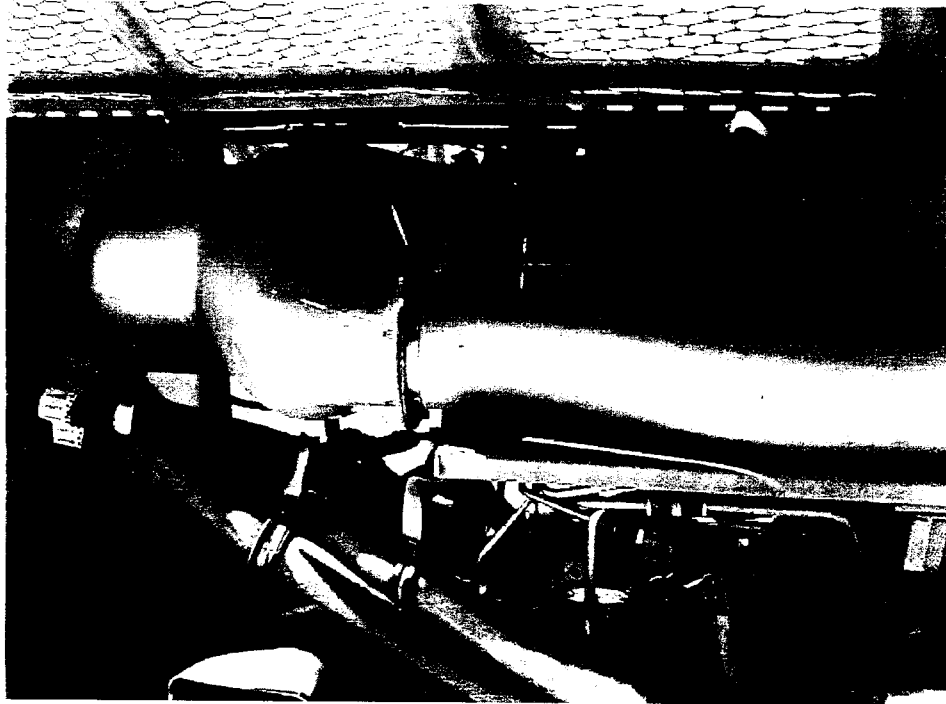


Photo # 11 – Right side of the engine.

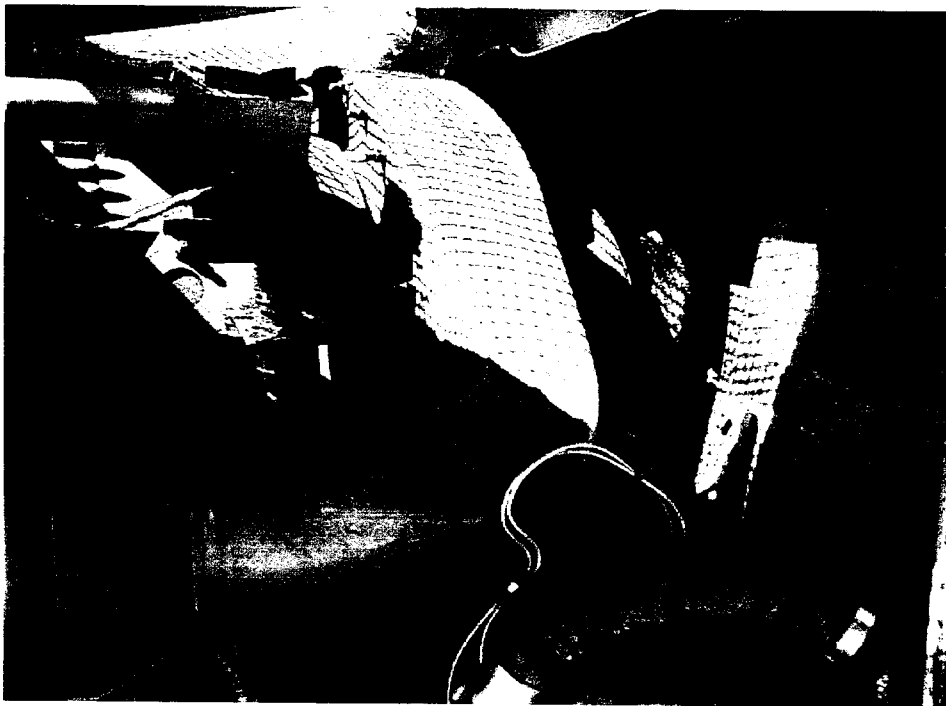


Photo # 12 – Right front of engine.



Photo # 13 – Front of compressor.



Photo # 14 – 3rd Stage Power Turbine wheel.

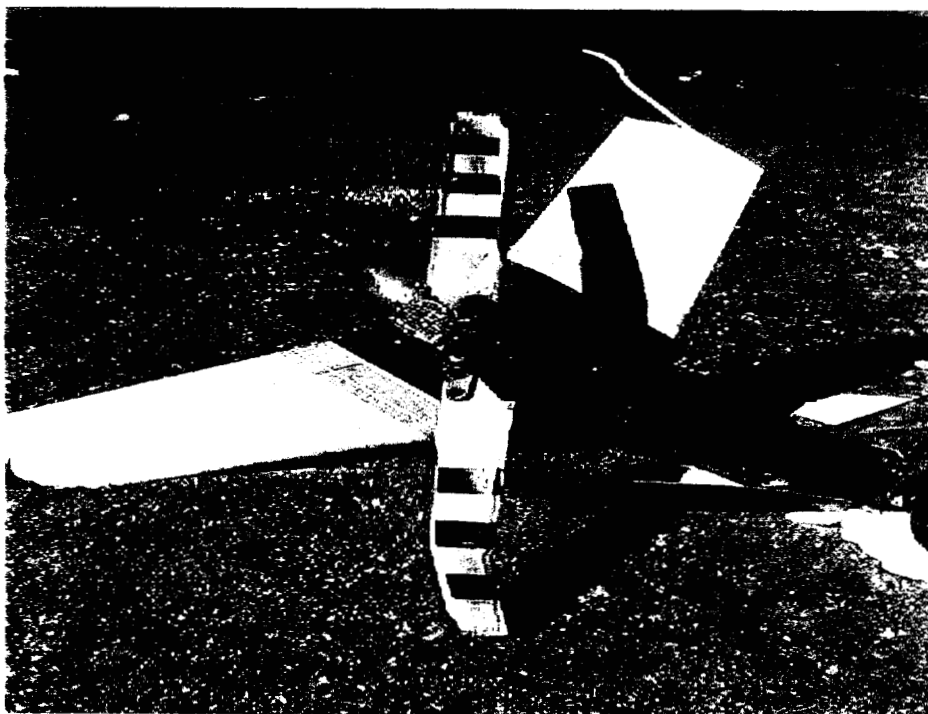


Photo # 15 – Tail rotor system.

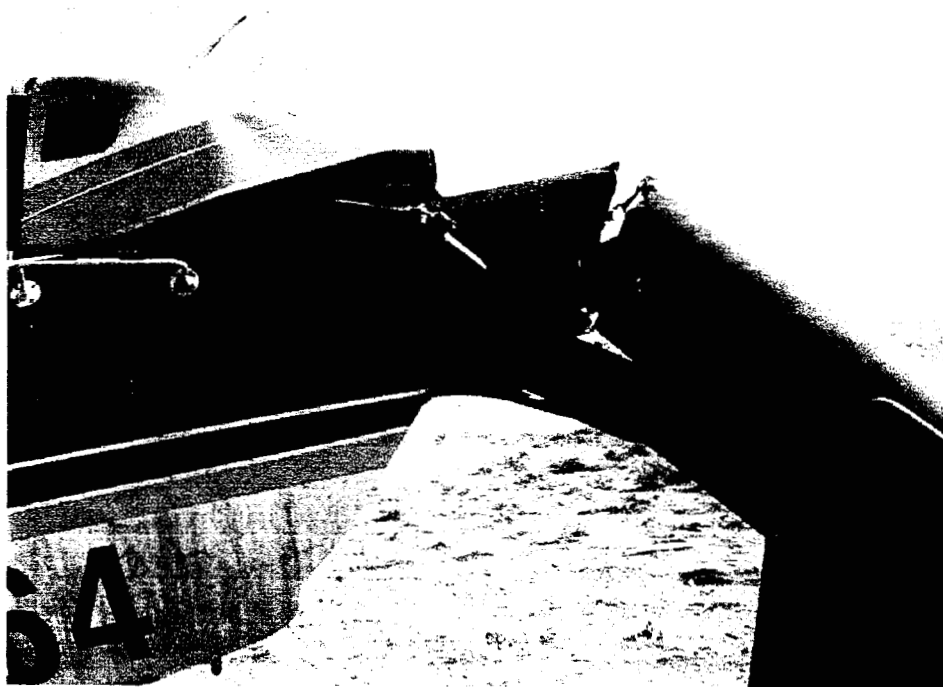


Photo # 16 – Tail boom/Tail rotor drive shaft.

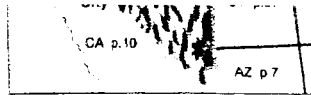
Appendix B.
Investigation Participants

Appendix B --- Investigation Participants.

The subject personnel were in attendance for the on-site investigation at Montello, Nevada.

Mr. Richard Parker	NTSB	IIC	[REDACTED]
Mr. Matt Rigsby	Bell Helicopter	Party	[REDACTED]
Mr. Bob Galloway	Office of Aircraft Services U.S. Department of the Interior	Party	[REDACTED]
Mr. Steve Rauch	Office of Aircraft Services U.S. Department of the Interior	Party	[REDACTED]
Mr. Mike Weber	Rolls Royce	Party	[REDACTED]

Appendix C.
Map of the Vicinity



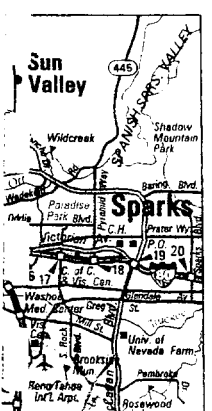
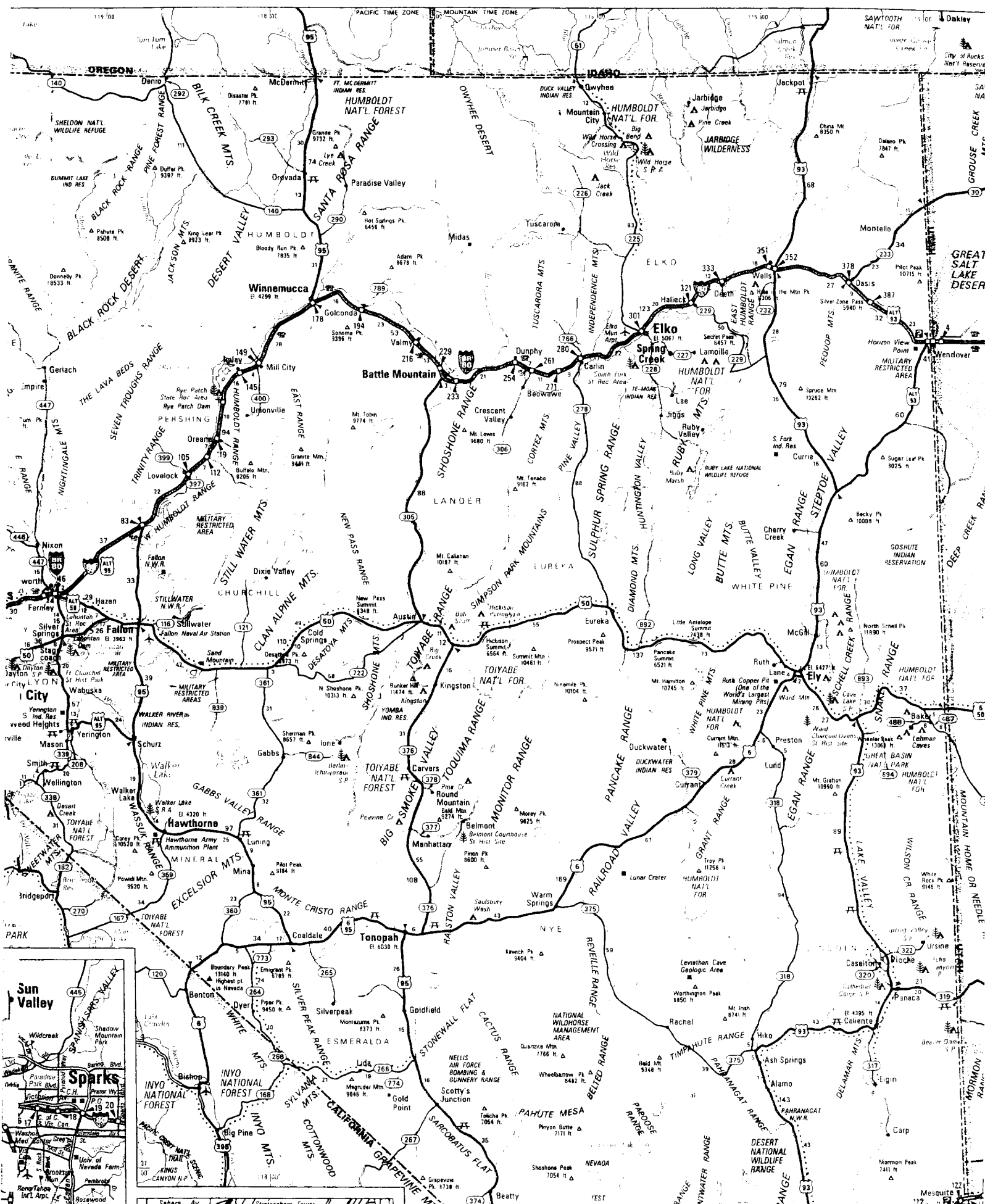
Between Principal Cities

Ely	146	217	259	270	318	190	256	271	207	245	339	345	246	120	317	346	167	121	271
Las Vegas	324	413	116	25	435	436	245	384	312	453	94	540	82	165	445	437	208	367	466
Reno	171	217	329	470	32	289	317	61	133	468	445	539	238	525	437	60	237	400	164

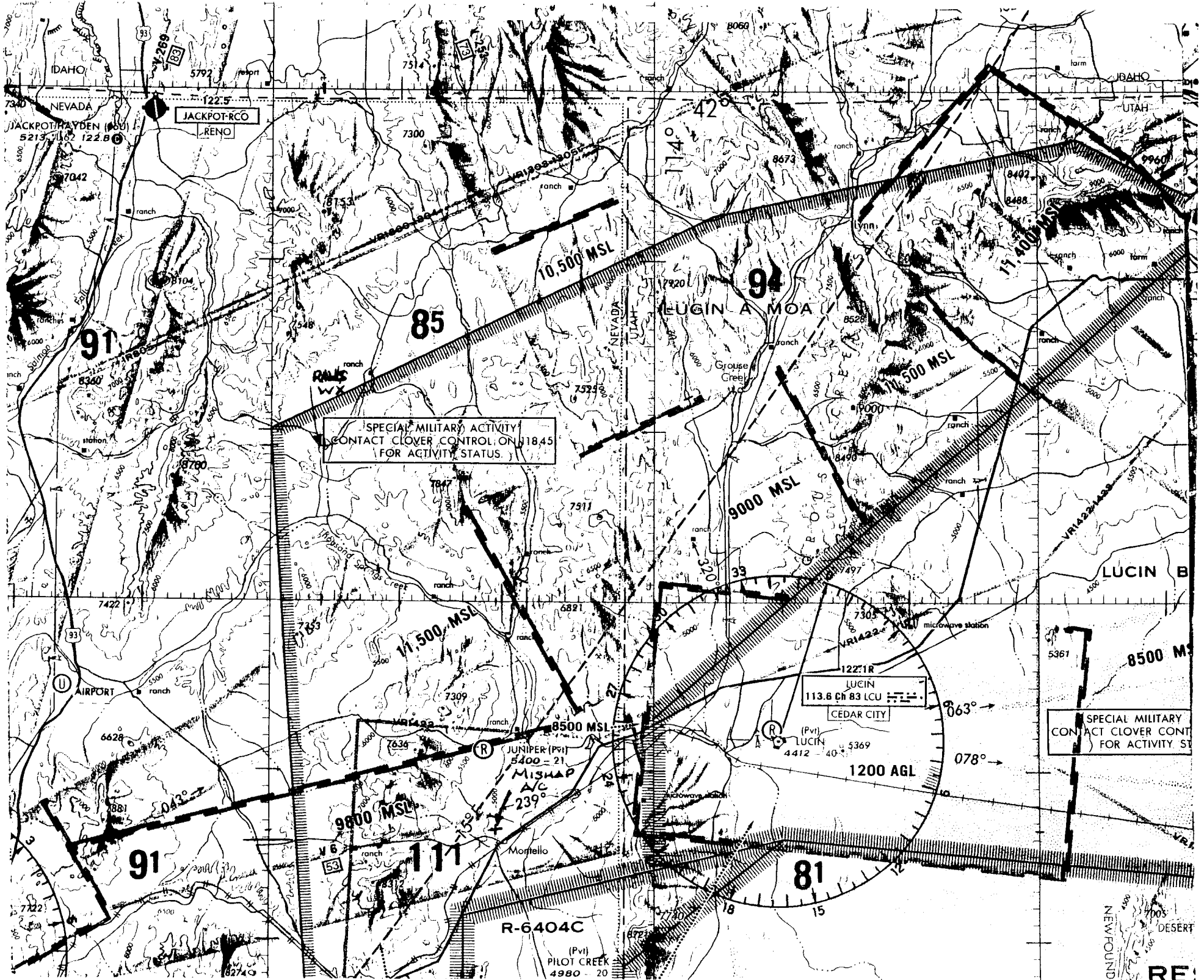
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Mileages in red between red arrowheads, in black, between intersections.



Sahara Av. Stargazer Tower



JACKPOT-RCO
RENO

SPECIAL MILITARY ACTIVITY
CONTACT CLOVER CONTROL ON 11845
FOR ACTIVITY STATUS

LUCIN
113.6 CH 83 LCU
CEDAR CITY

SPECIAL MILITARY
CONTACT CLOVER CONT
FOR ACTIVITY ST

(Pvt)
LUCIN
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(Pvt)
PILOT CREEK
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