SAFE OPERATING AIRSPEEDS

NOTE

All airspeeds in this section are indicated airspeeds in Knots (KIAS) and assume zero instrument or installation error. You should make sure your system has been correctly calibrated and account for those errors as necessary.

Max Demonstrated X-WIND component	20
Rotate Speed (V _R) with flaps 24 deg.	65-80
Best Angle of Climb (V _X)	85
Best Rate of Climb (V _Y)	105
Cruise Climb	140-160
Stall Speed clean (V _S)	76
Stall Speed Landing config. (V _{SO})	61
Landing Gear Operating Speed (VLO)	150
Landing Gear Extended Speed (V _{LE})	165
Approach Flaps Extended	160
Full Flaps	140

PRE-FLIGHT INSPECTION

COCKPIT CHECK

Aircraft Status Log	CHECKED
Required Forms/Certificates	
All Electrical Switches	
Circuit Breakers	CHECK IN
Gear Handle	
Battery Master Switch	ON
Aircraft Battery	
	CHECK QUANTITY, BALANCE & RESET
Fuel Selector	FULLEST TANK
Oxygen Quantity	CHECK
Trim Servos	CHECK
Flaps	DOWN
Pitot Heat Cover	REMOVE/STOWED
Pitot Heat	ON (10 SEC)

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Pitot Tube	VERIFY WARM AND CHECK CONDITION
Exterior Lights	
Pitot Heat	
Battery Switch	
	ON FOR 30 SECONDS, CHECK BLADES FOR HEAT
LEFT FUSELAGE	
Main Entry Door	CLOSE AND CHECK
	EXTERNALLY
Step	
Rear Window	
Upper and Lower Antennas	CHECK CONDITION
Static Port	CLEAR
A/C Vent	SCREEN CLEAR
Horizontal Stabilizer/Elevator	ATTACH POINTS, FREE
Elevator Trim Tab	SECURE, ATTACH POINTS, FREE
Rudder	CONDITION, ATTACH POINTS, FREE
RIGHT FUSELAGE	
Horizontal Stabilizer/Elevator	ATTACH POINTS, FREE
Baggage Door	CLOSED & LOCKED
Static Port	CLEAR
Windows	CHECK CONDITION
RIGHT WING	
Right Main Gear Door	CONDITION, ATTACH POINTS
Right Main Gear Mount	
	HYDRAULIC & BRAKE LINES/PADS, EXTENSION (4" MINIMUM)
Right Main Tire	CONDITION, INFLATION
Right Flap	ATTACH POINTS, MOVEMENT
Aileron Push Rod	
Right Aileron	
Nav / Strobe Lights	
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Fuel Tank Vent	CLEAR
Right Fuel Tank Cap	VISUALLY CHECK
	FUEL & SECURE
Leading Edge/Stall Strips	
Underwing Panels	
Fuel Sump Drain	SAMPLE
Wing Root Fairing	SECURE
NOSE	
Cowling	SECURE
	CONDITION, SECURE, LEAKAGE, FREE
Propeller Spinner	SECURE
	CONDITION, COVERS REMOVED
Engine Intakes	
Nose Gear Strut	CONDITION, EXTENSION (4" MINIMUM)
Nose Gear Tire	CONDITION, INFLATION
Landing Light	
Oil Level	
Oil Cap & Door	SECURE
Windshield	
LEFT WING	
Wing Root Fairing	SECURE
Fuel Sump Drain	
Underwing Panels	
Leading Edge/Stall Strips	
	VISUALLY CHECK FUEL AND SECURE
Pitot Tube	CONDITION, SECURE, WARM
Fuel Tank Vent	
Nav/Strobe Lights	
Left Aileron	
Aileron Trim Tab	
Left Flap	ATTACH POINTS, MOVEMENT
Left Main Tire	

cannot be neglected. The repaired contour of any repair should be similar to the original contour to remain as close as possible to the same airfoil as before thus maintaining the same "lift" on each blade. In addition the repair must result in the nick being fully removed and the blade surface polished. Give your propeller care, respect its overhaul periods, and it will pull you through many hours of flight. If in doubt, have it inspected by a certified propeller repair facility.

FUEL SYSTEM

The fuel system feeds fuel to the engine through a fuel screen pickup, fuel selector, gascolator/fuel filter, electric boost pump, and finally a fuel control unit. The fuel tanks should be sumped at regular intervals.

CAUTION

After flight when parking and securing the aircraft ensure the fuel selector is either in the left or right tank position. If the selector is "in between" the fuel in the higher wing will drain to the lower wing causing a dangerous imbalance condition. Also, fuel may dump through the vent system onto the ramp until the higher wing is completely empty.

HYDRAULIC SYSTEM

A self-contained hydraulic system is used to operate the landing gear. The pump is electrically powered. When the "gear up" position is selected the pump is activated and 2000 psi is provided to the up side of the landing gear actuators raising the gear. This pressure is maintained although the electric pump is disabled by a limit pressure switch and the pressure holds the gear in its retracted position. Upon selecting the "down" position, 2000 psi is provided to the down side of the actuators and lowers the landing gear until it is down and locked.

As with any hydraulic system proper servicing is required. Use only MIL-L-5606 "red" hydraulic fluid.

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The airframe will withstand the storage quite well under almost any circumstances since it is of high temperature materials however the upholstery, instruments and avionics will suffer from excessive heat and exposure to the sun so a cover is recommended. Elastomers such as tires also need to be protected from exposure to ultraviolet to limit their deterioration.

Fuel tanks should be filled or drained completely, the control surfaces locked, the aircraft electrically grounded, a pitot cover installed, the static port (or ports if installed on both sides) covered, the engine and cabin cooling air intake (NACA inlet) covered or plugged and the battery removed.

PREPARATION FOR SERVICE

Following storage, the aircraft preparations for flight should include the following:

- · Remove all taped openings, plugs and control locks.
- Clean and thoroughly inspect the aircraft checking the gear, tires, controls pitot and static ports.
- · Install a serviced battery.
- The fuel tanks should be checked for water accumulation and purged as required.
- Following a short but thorough engine ground check the aircraft should be flown for 30 minutes maximum and given a very thorough post flight inspection.

FUEL SERVICING

The fuel should be clean and water free. The firewall gascolator drain should be checked on preflight inspections for evidence of water, and the filter checked for solid foreign material. It is good practice to leave the tanks full to minimize the amount of combustible fuel/air vapor present in the tanks. This also helps minimize the amount of water vapor in the fuel system.

	RECOMMENDED SERVICING
	Interval - Preflight
	Check & service engine oil
	Drain water trap
	Verify fuel quantity
	Interval – First 25 Hours
	Service oil. Inspect for rub and wear inside cowling
	Inspect fuel lines for security
	Check bleed air lines for security
	Check battery fluid
	Check brake lines
	Check all gear doors (nose and main gears)
	Check control surface hinges and control rods and connections – if hinges show signs of corrosion or excessive wear replace the hinge.
	Jack the aircraft and retract gear/check gear door fit
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ENGINE FIRE/MECHANICAL FAILURE AIRBORNE

Pitch to Glide Attitude	110 KIAS
Propeller Control Lever	FEATHER
Fuel Condition Lever	CUTOFF
Power Control Lever	LOW IDLE
Fuel Selector Valve	OFF

NOTE

If smoke is present in the cabin, shut off all equipment operated by engine bleed air.

Perform Forced Landing Procedure

AIRSTART PROCEDURES

WARNING

Do not attempt to restart a failed engine caused by a known mechanical failure (Ng - 0%) or engine fire if Ng is above 50%.

ENGINE FLAMEOUT IF Ng IS ABOVE 50% (HOT AIR START)

Check Fuel Quantity	SWITCH TO FULLEST TANK
Power Control Lever	IDLE
Ignition	ON
Fuel Condition Lever	CHECK ON
Ng / ITT	MONITOR

WHEN ENGINE RELIGHTS (ABOVE 51% Ng AND 400°C ITT)

Ignition	OFF
Power Control Lever	AS REQUIRED
Land at Nearest Suitable Airfie	ld & Investigate

WARNING

During airstarts above 14,000' or with Ng<10%, starting temperatures tend to be higher and caution is required, if Ng is below 50%.

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FORCED LANDING

The use of gear UP versus gear DOWN is a function of the type of landing site. If the site is relatively hard and smooth, a gear DOWN landing is recommended. Conversely, if the site is soft or rough, a gear UP landing is recommended. This procedure can be used for practice, and actual engine failure or a precautionary landing.

NOTE

For feathering, a minimum oil pressure of 15 psi should be registered if propeller is windmilling.

Landing Gear	UP
Flaps	UP
Propeller Control Lever	
Airspeed	110 KIAS

The above configuration should give maximum glide performance with approximately 500 fpm descent and an 18:1 glide ratio. This should result in approximately 3.5 nm glide distance per 1000' of altitude lost.

Enter Forced Landing Pattern Overhead at high/low key whichever altitude permits, using an initial aim point 1/3 of the way down the runway/intended landing site. Use approximately 2500' AGL for High Key altitude and approximately 1300' AGL for Low Key altitude with the propeller feathered. If unable to feather the propeller, use 3500 AGL for High Key and 1700 AGL for Low Key.

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