

ATTACHMENT 2

POWERPLANT GROUP CHAIRMAN'S FACTUAL REPORT ENG-11-IA-051

14 CFR Part 33.77 Foreign Object Ingestion

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Part 33 AIRWORTHINESS STANDARDS: AIRCRAFT ENGINES			
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Sec. 33.77

[Foreign object ingestion.]

- [(a) Ingestion of a 4-pound bird, a piece of tire tread, or a broken rotor blade, under the conditions set forth in paragraph (f) of this section, may not cause the engine to--
- (1) Catch fire;
- (2) Burst (penetrate its case);
- (3) Generate loads greater than those specified in Sec. 33.23; or
- (4) Lose the capability of being shut down.
- (b) Ingestion of 3-ounce birds, 1 1/2-pound birds, or mixed gravel and sand, under the conditions set forth in paragraph (f) of this section, may not cause more than a sustained 25 percent power or thrust loss or require the engine to be shut down.
- (c) Ingestion of water, ice, or hail, under the conditions set forth in paragraph (f) of this section may not cause a sustained power or thrust loss or require the engine to be shut down.
- (d) For an engine that incorporates a protective device, compliance with this section need not be demonstrated with respect to foreign objects sought to be ingested under the conditions set forth in paragraph (f) of this section, if it is shown that--
- (1) Such foreign objects are of a size that will not pass through the protective device:
- (2) The protective device will withstand the impact of the foreign objects; and
- (3) The foreign object or objects stopped by the protective device will not obstruct the flow of induction air into the engine.
- (e) In showing compliance with paragraphs (a) and (b) of this section, the engine need be tested by ingesting only that foreign object specified in paragraph (a) which the applicant shows has the most severe effect on the engine and by ingesting the mixed gravel and sand specified in paragraph (b) and either the 3-ounce birds or the 1 1/2-pound birds, as specified in paragraph (f) of this section.
- (f) The prescribed foreign object ingestion conditions are as follows:

Foreign object	Test quantity	Speed of	Engine	Ingestion
·				

		foreign object	operation	
Birds:				
3-oz size	One for each 50 sq. in. of inlet area or fraction thereof up to a maximum of 16 birds. 3-oz bird ingestion not required if a 1 1/2-lb. bird will pass the inlet guide vanes into the rotor blades.	Liftoff speed of typical aircraft.	Takeoff	In rapid sequence to simulate a flock encounter.
1 1/2-lb. size	first 300 sq. in.	Initial climb speed of typical aircraft.	do	Do.
4-lb. size	One if it can enter the inlet.	Maximum climb speed.	Maximum cruise	Aimed at critical area.
Ice	Maximum accumulation on inlet cowl and engine face resulting from a 30- second delay in actuating anti-icing system.	Sucked in	Maximum cruise.	To simulate an intermittent maximum icing encounter at 25 degrees F.
Hail: (0.8 to 0.9 specific gravity).	For subsonic and supersonic engines: With inlet areas of not more than 100 sq. in.: one 1-in. hailstone. With inlet area of more than 100 sq. in.: one	typical aircraft.	cruise at	In a volley to simulate a hailstone encounter. One half the number of hailstones aimed at random areas over the face

	1-in. and one 2-in. hailstones for each 150 sq. in. of inlet area or fraction thereof.			of the inlet area and the other half aimed at the critical face area.
	For supersonic engines (in addition): 3 hailstones each having a diameter equal to that in a straight line variation from 1 in. at 35,000 ft. to 1/4 in. at 60,000 ft. using diameter corresponding to the lowest supersonic cruise altitude expected.	cruise velocity. Alternatively use subsonic velocities with	Maximum cruise	Aimed for critical engine face area.
Water	4 percent of engine airflow by weight.	Sucked in	Takeoff and flight idle.	For 3 minutes at each engine operation condition as spray to simulate rain.
Mixed gravel and sand (one part stones with diameter not less than 3/16nor more than 1/4 in. and 7 parts sand.)		do	Takeoff	Over a 15- minute period.
Broken rotor blade: (The heaviest compressor or turbine blade, broken at the outermost retention groove or member or at least 80% of	One		do	Release from rotor followed by 15 second delay prior to initiating shutdown.1

an integral blade.)				
Tire tread:				
(Having width and length equal to full width of tread.)	do	do	do	

¹Blade containment must be demonstrated with a complete engine to evaluate secondary effects of blade loss and to determine blade fragment trajectories, except that in fan engines, the fan assembly may be tested separately for blade containment if it is demonstrated that fan blade or vane debris would not enter the compressor after a fan blade failure.]

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Comments

▼ Document History

Notice of Proposed Rulemaking Actions:

Notice of Proposed Rulemaking. Notice No. 71-12; Issued on 04/26/71.

Final Rule Actions:

Final Rule. Docket No. 11010; Issued on 09/20/74.

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