•	TIME AEROSPACE LLC AEROSPACE SUPPORT SER VICES Note: serves as an equipment list update. W&B change was negligable. Supplemental Equipment List - Weight and Balance Form									
	Referanced W&B report by Gar Date: 12/08/2012	y Johnson dated 10/15/2 Registration: N2521V	003 S/N 18030	T.T. 3050.15						
	Previous	E.W 1324.00	0.0.	Moment 49517.60						
	Removed Equipmer	nt:		0.00 0.00						
0	Installed Equipment PM1000 II	. 0.70	19.00	13.30						
	NOTE: V	New 1324.70 Veight and Balance Char								



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C.G. RANGE CESSNA 170	
2200 2200	7
2100	
2000 NORMA CATEGORY	1
PRONE ALL	-
1900 CATEGOY	
1800 AFT	
FRONT	-
1700 KON	_
36 37 38 39 40 41 42 43 44 45	46
Empty Wt. C.G. Range None	
Maximum Weight 2200 lb. Normal Category	
1900 lb. Utility Category	
No. of Seats 4 (2 at +36), (2 at +70)	
Maximum Baggage 120 lb. (+95)	
Fuel Capacity 37.5 gal. total. 33.5 gal. usable (three 12.5 gal. tanks in wings	
Fuel Capacity 37.5 gal. total, 33.5 gal. usable (three 12.5 gal. tanks in wings a weight of unusable fuel	t +45). See NOTE 1 for
Oil Capacity 2 gal. (-20)	
Control Surface Movements Wing flaps Up Down 30° Ailerons Up 22° Down 14°	
AileronsUp22°Down14°Elevator tabUp10°Down27°	
Elevators Up 28° Down 17°	
Rudder Right 16° Left 16°	
Serial Nos. Eligible 18000 through 18729	
Required Equipment In addition to the pertinent required basic equipment specified	
items of equipment must be installed:	in CAR 3, the following
Landplane: Items 1(a), 103, 104, 201(a), 202(a), 204(a), 402	
Skiplane: Items 1(a), 103, 104, 204(a), 208(a), 402(a), and	I (e) or (d).
Note: For night flying, cabin dome light and instrument lig	hts or equivalent, to
provide illumination of all placards and instruments	
equipment required by CAR 43.	

## Type Aircraft: Cessna 170 Tail Number: N170RH

	Weight	Arm	Moment
Empty Weight	1324.7	37.39	49517.60
Pilot / Front Pax	410	36	14760
Rear Pax	340	70	23800
Baggage Area 1	0		0
Baggage Area 2	0		0
**Zero Fuel Weight	2074	42.47	88077.6
Fuel in Pounds	108	45	4860
Ramp Weight	2182		92937.6
Taxi / Run-Up			
Take-off Weight	2182	42.6	92937.6

#### Weight & Balance

CG Location in inches aft of datum: \_\_\_\_\_42.6

Take-off Distance: Ground Roll\_\_\_837\_, Over 50ft. Obst. \_\_\_2092\_

# OPERATION AND PERFORMANCE DATA

Ibs. gross weight is not to be exceeded – and the useful load in the airplane can be distributed in any way the pilot desires, that is, in baggage (up to the baggage compartment placard limit), gasoline, or passenger load. These are all variables including the pilot's weight, and of course, baggage may be carried in the passenger's location instead of a passenger. It is the responsibility of the pilot and operator to see that the weight and balance are within limitations.

## OPERATIONAL DATA:

## PERFORMANCE INFORMATION

The following operational data are compiled from actual tests with airplane and engine in good condition and using average piloting technique. Data are based upon a gross weight of 2200 lbs. with McCauley propeller installed, and full throttle for take-off and climb. Performance figures are for zero wind velocity and hard surface level runway. Speeds are true indicated airspeeds.

	OUTSIDE AIR TEMPE							
ITEM	ALTITUDE	0°F	20°F	40°F	60°F	80°F	100°F	
Take Off Distance (Ft.) So To clear 50 ft. obstacle 20 Airspeed 76 MPH Take-off Full Throttle, Flaps Up 60	000 Ft. 000 Ft.	1780 2140	1580 1910 2290 2740	2050 2450	2190 2610	2340 2790	2500 3000	

Take Off Correction: Reduce above distances 10% for each 6 MPH Wind Velocity. Ground run approx. 40% of distances shown.

Assessables 71 MDH	Sea Level 2000 Ft. 4000 Ft. 6000 Ft.	1685	1745 1850	1805 1910	1860 1970	2020	2075	
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Landing Correction: Reduce above distances 10% for each 6 MPH Wind Velocity. Roll approx. 45% distances shown.

Normal Rate Climb789Sea Level(Feet Per Minute)862000 Ft.Flaps Up844000 Ft.Full Throttle81	6/0	042	047	000		465
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