

NATIONAL TRANSPORTATION SAFETY BOARD

Office of Aviation Safety Washington, D.C. 20594

April 4, 2018

YIP Simulator Test Plan

OPERATIONAL FACTORS

DCA17FA075

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A. ACCIDENT

Location:	Ypsilanti, Michigan (MI)
Date:	March 8, 2017
Time:	1452 EST (1652Z)
Airplane:	MD-83, N786TW

B. SUMMARY

On March 8, 2017, about 1452 EST, Ameristar Air Cargo Inc. flight 9363, a Boeing MD-83, N786TW, ran off the end of runway 23L after executing a rejected takeoff at Willow Run Airport (YIP), Ypsilanti, Michigan. All 109 passengers and 7 crewmembers evacuated the airplane via emergency escape slides. One passenger was reported to have received a minor injury. The airplane sustained substantial damage (no postcrash fire occurred). The airplane, which had been flown to YIP 2 days before the accident, was operating under the provisions of 14 *Code of Federal Regulations* (CFR) Part 121 as an on-demand charter flight and was destined for Washington Dulles International Airport (IAD), Dulles, Virginia. Daytime visual meteorological conditions prevailed at the time of the accident.

C. SIMULATOR TEST PLAN

1.0 Simulator Setup

Location:	American Airlines Training Center, Dallas TX
Date:	April 11, 2017

Time:	1330 – 1530 CST	
Overall Objectives:		
	• Document simulator syst	ems and alerts
	Document normal Ameri	star taxi procedures
	 Include flight con 	trol check
	Document Ameristar Rej	ected Takeoff (RTO) procedures
Aircraft:	MD-83 simulator (Level C)	
Participants:	Simulator Operator:	Andreas Gruseus (Ameristar)
	Captain (Left) Seat:	Andreas Gruseus (Ameristar)
	FO (Right) Seat:	Pat Hulsey (Ameristar)
	Test Conductor:	David Lawrence (NTSB)

Note: Pat Hulsey was not qualified/current on the MD-83 for Ameristar. Since there was not a full qualified crew in the simulator, actual tests runs to document crew-related procedures for a rejected takeoff/emergency evacuation was not possible.

None¹

Initial Simulator Setup:

- New York LaGuardia (LGA)²
- Cockpit setup: Normal

Observers:

- Weather: (See Appendix)
 - o 10C temp, altimeter 29.80, winds 240/46, clear +10 visibility
- Left seat occupied by Andreas Gruseus, right seat occupied by Pat Hulsey
- Simulator motion disabled

Weight and Balance:

(See Appendix)	
Zero Fuel Weight	114,476
Fuel Weight	31,000
Ramp Weight	145,476
Taxi burn	400
Takeoff Weight	145,076
CG (limits 3.7/22.1)	11.7
Stab Trim	7.0
Flaps	15
V1/Vr/V2 ³	139/142/150

¹ The FAA and Boeing Ops Group members were unable to attend.

² YIP airport was not in the simulator database. LGA was used for runway 22 and similar runway length to YIP 22R.

³ According to the crew interviews, they increased their rotation speed by about 5 knots due to the high winds at YIP.

2.0 Flight Control Check

Procedure

- Faced airplane to the north to simulate taxi to runway 27 at YIP, conducted flight control check.
- Conducted flight control check
 - a. Document procedure:
 - Normal procedure: aileron checked to the left then right, then control column full back, then full forward. While column is full forward, pilot looks for "elevator power on" light on overhead annunciator panel. There was no hydraulic pressure indications, no aural alerts. The power on light tells the crew that pressure was available to the elevator tabs. Aileron check was verified with the "spoiler deployed" light on the overhead annunciator panel. There was no hydraulic pressure indication during this check.
 - Flight control check was part of the taxi checklist. There are no other flight control checks conducted by the pilots. The captain conducted the rudder check, and the FO (right seat) conducted the aileron and elevator check.
 - Inserted "Elevator Control Cables to R Elevator Break" malfunction.⁴ Control check performed again with no noticeable difference in control column movement or resistance. All annunciator indications were identical to normal control check.

3.0 Rejected Takeoff Procedures

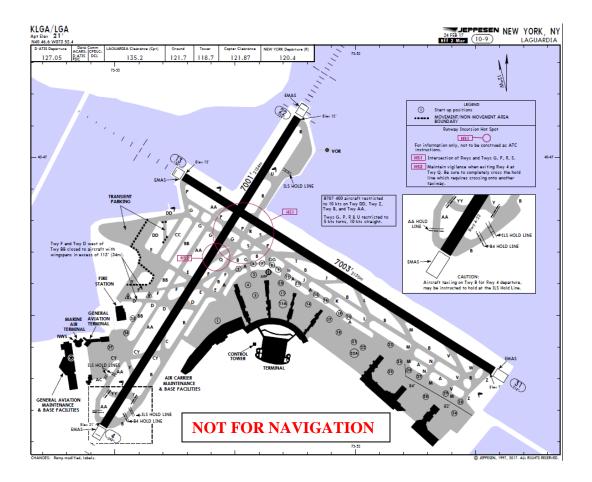
Procedure

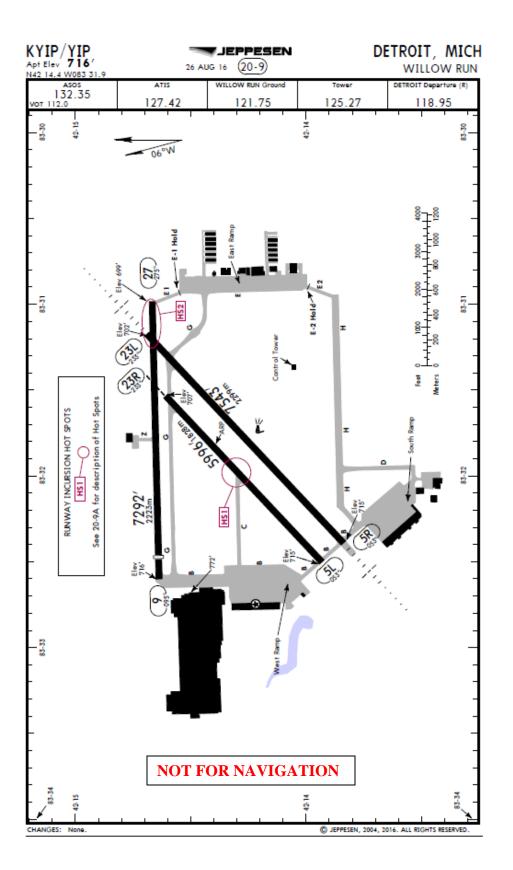
- Reposition airplane to end of runway 22 (LGA)
- Configured airplane (flaps 15, weights, weather)
- Pre-select Engine Fire (L or R) at 134 knots (purpose of run was to document the rejected takeoff procedures, not replicate the accident).
- Pilot performed rejected takeoff to full stop.
- Rejected takeoff made by pm, captain had hands on throttles for takeoff, removed hands at V1 call.
 - a. Pilot reduced thrust levers to idle, simultaneously disconnecting the autothrottles. According to the pilot, he initiated maximum braking as well.
 - b. Pilot then pulled on the ground spoilers lever full aft.
 - c. Pilot then initiated reverse thrust
- Ameristar procedures are to call "V1" at the actual V1 speed. Pilots do not call "V1" at V1-5 knots.

⁴ See Appendix for malfunction page.

D. APPENDIX

1.0 Charts





2.0 V-Speeds

D-83	08/15/10		JT8D-219
Gross Weight	146,	000 lb	S.
TAK	EOFF	LAND	ING
Use Airport Speeds and C	Analysis for V orrections	FLAP SLAT	MIN MAN
V1	139	0°/RET	248
VR	142	0°/EXT	194
V2	150	11°/EXT	169
0/EXT	APO 8-15 V2+5_	15/EXT	167
0/EXT FI	AD6 -2 V2+15	28°/EXT	155
0/RET	199	VREF 28°	144
CLEAN	248	VREF 40°	140

3.0 Weather⁵

KYIP 081722Z 0818/0918 25032G48KT P65M FEW060 FM082100 26031G46KT P65M SCT070 FM082300 26024G36KT P65M SKC FM090100 27014KT P65M SKC FM090800 25008KT P65M BKN150 FM091600 26008KT P65M SCT060 OVC080=

METAR KYIP 0819532 A2980 RMK AO2 SLPNO \$= METAR KYIP 0818532 A2977 RMK AO2 SLPNO ?= METAR KYIP 0817532 A2979 RMK AO2 PK WND 24046/1656 SLPNO 58012 \$=

END 0004 WEATHER/NOTAM REPORTS 000 GRAPHIC 004 NON-GRAPHIC 000 NOTAM

4.0 Weight and Balance

WEIGHT & BALANCE (maximum certificated weights in bold)		
Basic Operating Weight (+ crew weight)	87,304	
Flight Attendant weight (4)	720	
Passenger weight ⁶	21,450	
Baggage/Cargo Weight	5.002	

⁵ This weather was the observation at the time of the accident.

⁶ Weight was assumed 195 pounds each (winter weight) to include carry-on bags.

Zero Fuel Weight	114,476
Maximum Zero Fuel Weight	122,000
Fuel Weight (pounds)	31,000
Ramp Weight	145,476
Maximum Taxi Weight	161,000
Taxi Fuel Burn	400
Actual Takeoff Weight	145,076
Maximum Takeoff Weight ⁷ (landing	146,400
limited)	
Estimated Fuel Burn (IAD)	10,276
Estimated Weight on Landing (IAD)	134,800
Maximum Landing Weight	139,000
Takeoff CG (% of MAC) ⁸	11.7
CG Limits (FWD/AFT)	3.7/22.1
Stab Trim	7.0
Takeoff Flaps	15
V1/VR/V2 (146,000 pound card) ⁹	139/142/150
Landing VREF (40 degrees)	135

⁷ The maximum structural takeoff weight is 160,000 pounds.
⁸ Mean Aerodynamic Chord.
⁹ According to the crew interviews, they increased their rotation speed by 5 knots due to the high winds at YIP.

5.0 Ameristar Checklists

AMERISTAR AIR CARGO, INC.

DC-9 AIRCRAFT OPERATING MANUAL NORMAL CHECKLISTS - AMPLIFIED Revision: 25 02/01/17

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Normal Checklist MD-83

BEFORE	START
(F/O)	(CAPT)
Briefing/EFB (B)	COMPLETED/SET
Briefing/EFB (B) Logbook & Gear Pins Cockpit Preparations (B) .	ON BOARD
Cockpit Preparations (B) .	COMPLETED
Preflight Inspection (B)	COMPLETED
Preflight Inspection (B) Windows (B)	CLOSED /LOCKED
Circuit Breakers (B)	CHECKED
Oxygen & Interphone (B)	
Smoke Goggles (B) Pressurization	CHECKED
Pressurization	AUTO (up) and SET
Air Cond Auto Shutoff	AUTO
Anti-Skid	ARMED
EGPWS. FD CMD, CADC, EFIS &R	TESTED
FD CMD, CADC, EFIS &R	ADIO NAV Selectors
	NORMAL
EMER Lights Seat Belt & No Smoking S Flt Guidance Ctrl Pnl (B)	ARMED
Seat Belt & No Smoking S	ignsON
Flt Guidance Ctrl Pnl (B)	CHECKED & SET
Clocks & Altimeters (B) FMS (B) Engine Oil Quantity	SET
FMS (B)	CHECKED & SET
Engine Oil Quantity	CHECKED
Takeoff Warning	CHECKED
Rudder Power	POWER
Radar	OFF
Transponder	STANDBY
Radios (B) Rudder & Aileron Trim	SET
Rudder & Aileron Trim	FREE & ZERO
Parking BrakesSET, PR	RESSURE CHECKED
Parking BrakesSET, PR Fuel Pumps & X Feed	ON & CLSD
Fuel Quantity (B)	MIN ,FOB
V Speeds (B)	,
Stabilizer Trim (B)	CHECKED & SET
V Speeds (B) Stabilizer Trim (B) PMS (B)	CHECKED & SET

READY FOR START
(F/O) (CAPT)
Documents ON BOARD
Cabin SECURE
Cockpit DoorLOCKED
Pneumatic CrossfeedsOPEN
Hydraulics ON, HIGH & CHECKED
Anti-Collision LightsON
Annunciator/Door Lights CHKD & LTS OUT
Air Cond Supply SwitchesOFF
Fuel PumpsSET
Pitot/Windshield Heat CAPT/ON
IgnitionON Pneumatic PressureCHECKED
Pneumatic PressureCHECKED

AFTER	START			
(F/O)	(CAPT)			
Pneumatic Crossfeeds	CLOSED			
Transponder	XPDR XPDR 1/2			
Hydraulics O	N, HIGH & CHECKED			
APU Air Switch				
Engine Anti-Ice	ON/OFF			
Fuel Heat	ON/OFF			
Ignition Electrical System	OFF			
Electrical System	CHECKED			
Air Cond Supply Switche	sAUTO			
Yaw Damper	ON			
Annunciators	CHECKED			
Shoulder Harness (B)				
Ground Equipment (B)	CLEAR			
TAXI				
(F/O)	(FO)			
Flaps/Slats (B) °.	 •. TO BLUÈ LT 			
Flaps/Slats (B)	(B) CHECKED			
APU				
Fuel Heat				
Takeoff Briefing (B)				

AFTER START

BEFORE TAKEOFF

(F/O)	(FO)
(F/O) Flaps/Slats (B)	, TO BLUE LT
V Speeds (B)	RECHKD
Stab Trim (B)	SET
Annunciators	
Flight Attendants	SEATED
Departure Runway (B)	CONFIRMED
Ignition	ON
	ON
Ignition . Radar/EGPWS (Terrain Display). Transponder.	ON
Ignition . Radar/EGPWS (Terrain Display). Transponder.	ON
Ignition Radar/EGPWS (Terrain Display)	ON ON TA/RA ON



6.0 Photos (taken by Ops Group Chairman)

Photo 1: Center Quadrant set up for flight control check.



Photo 2: Fuel panel indications.

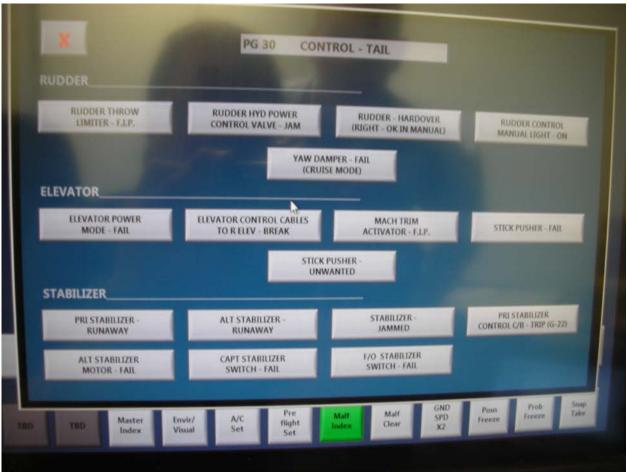


Photo 3: Simulator Malfunction Page (Control – Tail).



Photo 4: photo of control column full forward.



Photo 5: Overhead annunciator indications with control column full forward. "Elevator Pwr On" light indicated by arrow.