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**NATIONAL TRANSPORTATION SAFETY BOARD  
WASHINGTON, D.C.**

**ATTACHMENT 6**

**COPY OF ENGINE INSTRUMENTS  
AND OPERATING PARAMETERS  
(5 PAGES)**

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## PILOT'S REFERENCE MANUAL

### OVERVIEW

The MD-88 has two Pratt and Whitney JT8D-219 engines. Each engine has a maximum takeoff thrust rating of 21,750 pounds. Automatic Reserve Thrust (ART) and engine synchronization systems are also installed.

The MD-90 has International Aero V2500 engines and a static thrust of 28,000 pounds. Engine synchronization systems are installed.

The engine nacelles are supported from horizontal pylons by vibration isolating side mount systems. The nacelles are isolated from the fuselage by a firewall within the pylon and by a secondary fire seal at the pylon fuselage interface. The nacelle ventilation system is designed to provide adequate cooling of engine and accessories and to prevent accumulation of combustible mixtures. Each engine is equipped with an acoustic treatment in specific areas for noise suppression.

### ENGINE INSTRUMENTS

For monitoring engine operation, an engine display panel (flat plate) and systems display panel are installed on the center instrument panel. The following indications/readouts are provided:

- Engine Pressure Ratio (EPR).
- $N_1$  and  $N_2$  rpm.
- Exhaust gas temperature (EGT).
- Fuel flow/used.
- Oil pressure and oil temperature.
- Fuel temperature.
- Oil quantity.
- (90) Engine Vibration Display.

**NOTE:**  $N_1$ ,  $N_2$ , EGT, and oil pressure digits will flash if their respective red line limits are exceeded. EGT and fuel flow digits will also flash if a hot start is imminent.

The left and right engine EPR Indicators receive sensing signals from the respective engine air inlet pressure ( $P_{12}$ ) probe and the low pressure turbine discharge pressure ( $P_{17}$ ) probes. Engine pressure ratio ( $P_{17}/P_{12}$ ) is a measure of thrust being developed by the engine, and is displayed on the EPR Indicator.

The left and right  $N_1$  indicators display the respective engine low pressure compressor rotor speed in percent of rpm. The left and right  $N_2$  indicators display the respective engine high pressure compressor rotor speed in percent of rpm.

The left and right EGT Indicators display the temperature of each engine in degrees centigrade. Each indicator receives signals from temperature probes located in the exhaust gas path of each engine.

The left and right fuel flow/used displays normally indicate the rate of fuel flow in pounds per hour for each engine. Pushing the FUEL USED (PPH) Button momentarily changes the display to fuel used in pounds for each engine. A FUEL USED RESET Button resets the fuel used display to zero.

The left and right engine oil pressure is indicated on the systems display panel in psi. Oil temperatures are indicated on this same panel in degrees centigrade. Oil quantity is also indicated for each engine in quarts on the systems display panel.





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APPENDIX A

OPERATING PARAMETERS

OIL QUANTITY

- 8 qt. Minimum - Non-Maintenance Station (12 - 16 qt. Normal).
- 12 qt. Minimum - Maintenance Station.

EGT STARTING LIMITS

- Ground 475° C
- In Flight 625° C

NORMAL IDLE INDICATIONS

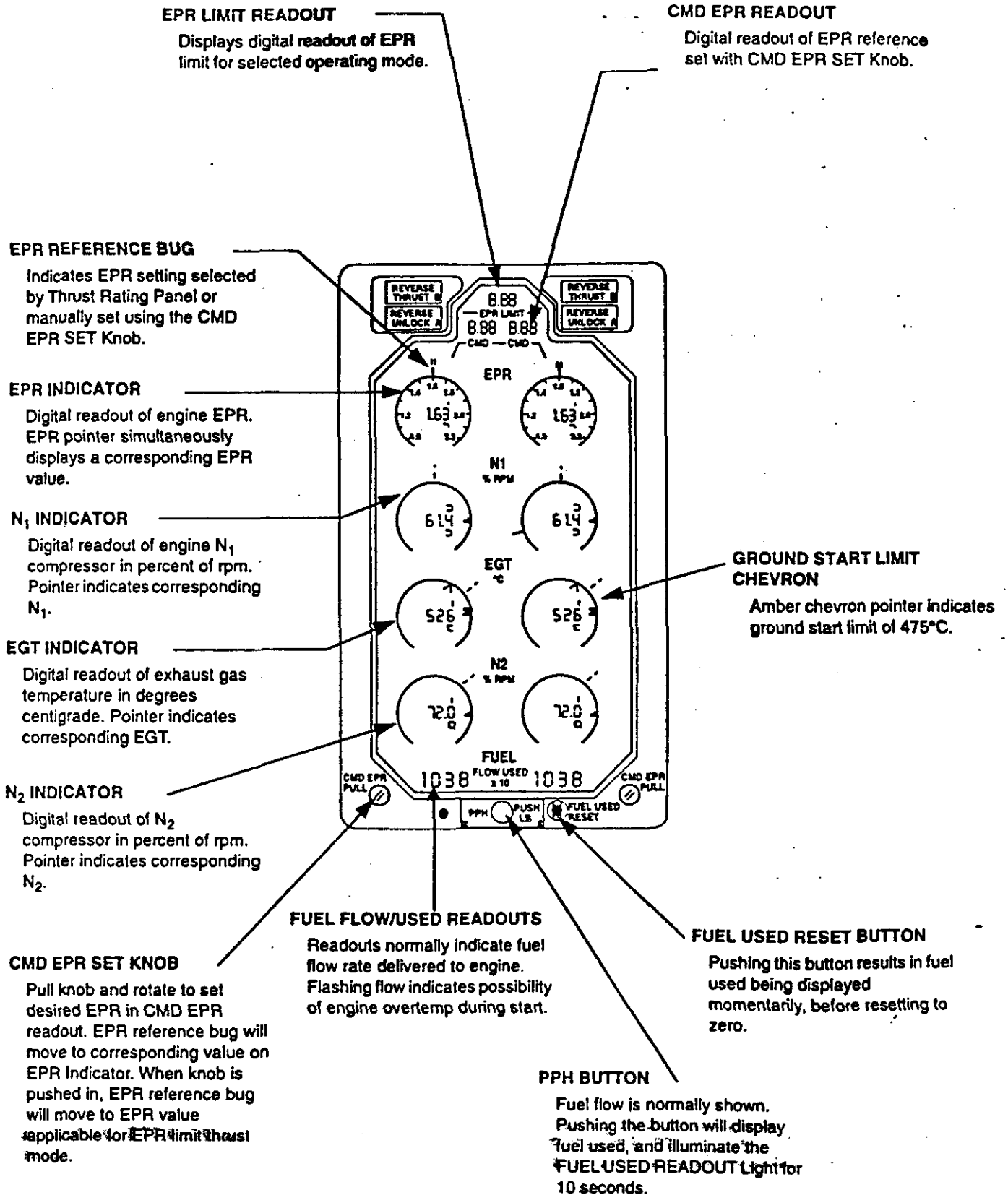
- N<sub>2</sub> 50 - 60%.
- EGT 300 - 480° C.
- Fuel flow 800 - 1200 PPH.
- Oil pressure 45 ± 5 PSI.

AUTOMATIC RESERVE THRUST (ART) OPERATION

- READY N<sub>1</sub>'s, SLATS, ART AUTO, WEIGHT ON WHEELS
- ARMED N<sub>1</sub>'s > 64%.
- FIRES N<sub>1</sub>'s SPLIT OF 30%, DFGC falls or switched or windshear.
- DISARMED N<sub>1</sub>'s < 49% or slat retraction.

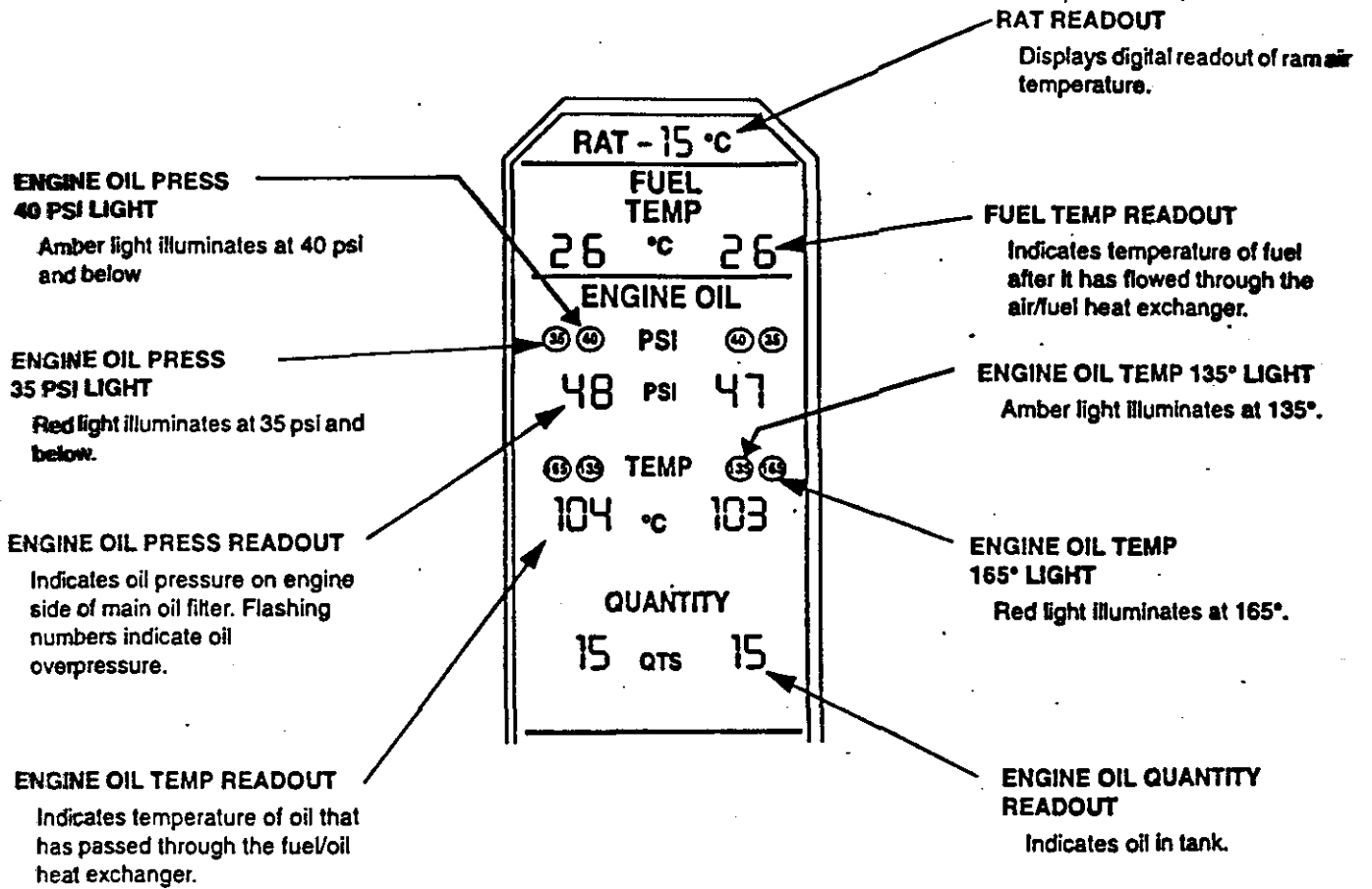
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**PILOT'S REFERENCE MANUAL**



**ENGINE DISPLAY PANEL (MD-88)**

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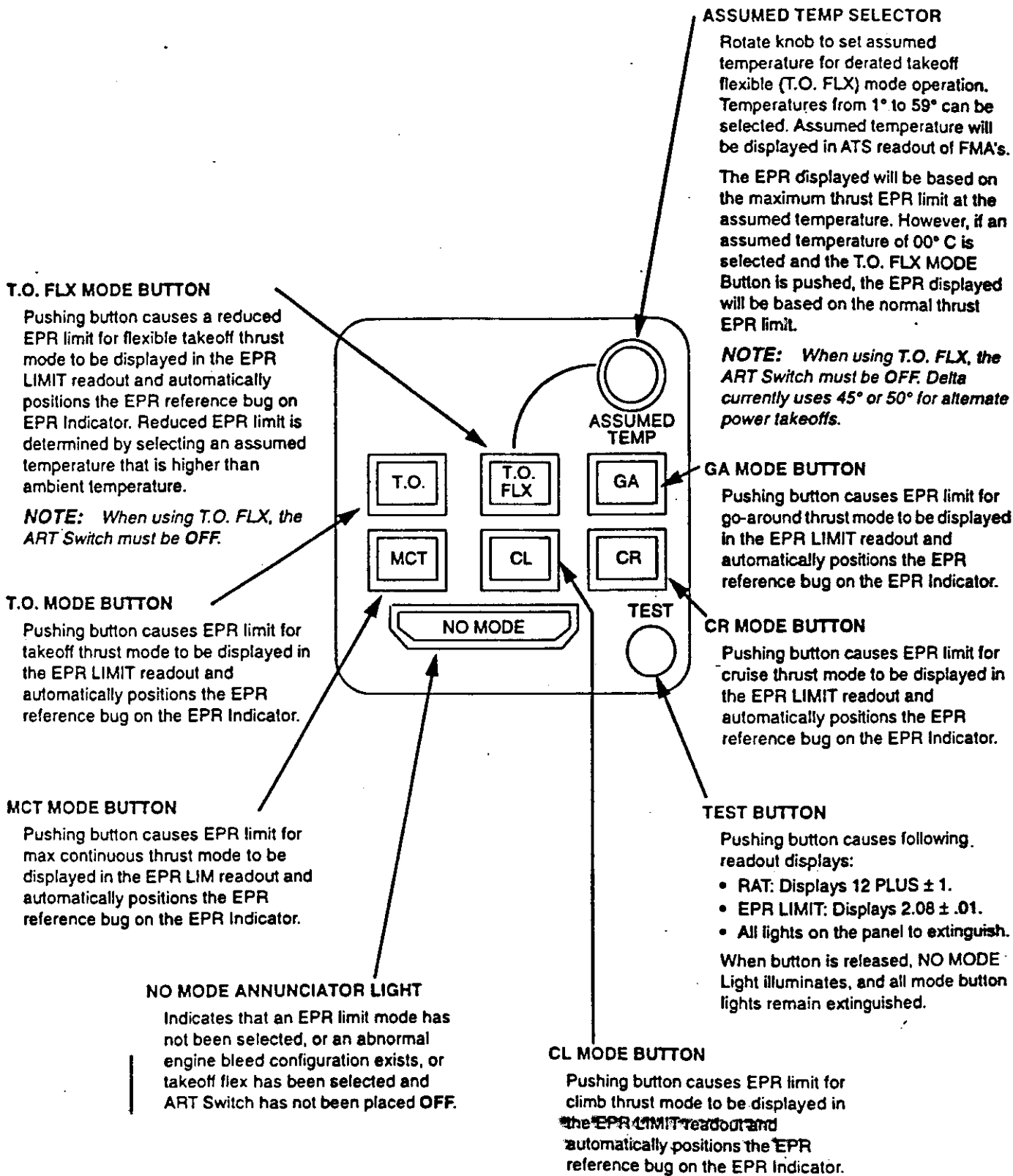


**SYSTEMS DISPLAY PANEL**

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# PILOT'S REFERENCE MANUAL



THRUST RATING PANEL