RAPCO,	inc.
Air Filter P/N	9-18-1
Changed at:	Next Change:
3405 Hrs.	Hrs.
Date: 2-14-12	Date:
	Aerospace



N588ET. #2 AIRCRAFT LOG

ſ			
ircraft Record Gene			
panufacturer 77	er	Model PA31	-301T N 588 ET
serial	24099	Registration Number	NOBOET
_{pate} of Manufacture			
_{ព្រ} gine(s) currently instal	led:		
_{Nanufacturer}	Model	Se	erial
_{Nanufacturer}	Model	Se	erial
propeller(s) currently inst	alled:		
Manufacturer		Model	
guß Model	Serial	Se	rial
Blade Model	Serial	Serial	Serial
Blade Model	Serial	Serial	Serial

			60				
91	¹ 78	115	60	115.	716	547)

© ASA 1993

tion of Inspections, Tests, Repairs and Alterations ust be endorsed with Name, Rating and Certificate Number of Technician or Re See back pages for other specific entries.)	pair
See back pages for ourse	
rtify that ATC Transponder(s) has/have been inspected, tested and found to mplies with FAR 43, Appendix F. nsponder No. 1 Serial No. 600 2.7 nsponder No. 2 Serial No. 1 Signature: 11/18/07 Signature: 12/18/07 No. 1 Signature: 12/18/07 No. 1	
06 W. Airport Dr., Visana, Ca. 752-7	
ertify that the Altimeter(s) Automatic Pressure Altitude Reporting System(s) d Static Pressure System(s) have been inspected, tested and found to comply	
timeter(s) tested to 25,600 FT	
o-Pilot Altimeter Serial No. Signature: SavCom Avionics Inc. Certificate No. No. Signature: No.	

DATE 19	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE

Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repa

January 1, 2008 Tach/ TT: 3175

Removed the following: PM1000 intercom, KMA24 audio panel, KY197 com, KX170/Max 1700 navcom, GX50 GPS with antenna, KI-203 indicator; Apollo GX-50 GPS system with A-33 antenna and MD-41-724 annunciator, KR-87 ADF, KI-227 ADF indicator, KA-44B ADF antenna, II morrow 612 Loran, CI-121SP Loran antenna and preamp; KNS-80 RNAV. Installed: Garmin GMA 340 audio panel with marker beacon receiver pn 011-00401-10 sn 96280710, and connected to existing marker beacon antenna. Installed new Garmin GNS 430W pn 011-01060-00 sn 23402424 with terrain and WAAS as #1 navcom. Installed serviceable Garmin GNS530 GPS/Com pn 01100550-10 sn 78400121 as #2 navcom. Installed new GA 56 antenna sn 59428241connected to the GNS 530 and installed new GA 35 antenna connected to the GNS 430W. Both antennas installed at forward cabin top. Installed new Mid Continents indicator mn MD200-306 sn H07-11691 connected to the #1 nav, and retained the existing KI 206 indicator sn 12269 connected to the #2 nav. Both nav units were coupled to the autopilot. Installed PCD7100-P CD/MP3 player sn E02872. Previously installed Garmin GTX 327 transponder sn 83707374 was retained and relocated below the Garmin 430W in the manufacturers left avionics rack. RS232 serial data is provided to the #1 GPS/NAV/COM from the transponder, and RS232 serial data is provided to the #2 GPS/NAV/COM from the existing Trans Cal altitude serializer. Installed Comant CI-1125 quadplexer at upper center false firewall to provide VOR/LOC/GS information to the #1 and #2 navcom. The existing cooling fan was routed to #1 and #2 GPS/NAV/COM and transponder. See 337 form and weight and balance revision this date. Removed undocumented GEM engine analyzer and installed new JPI engine analyzer pn EGT 701(6)C sn 26324 in accordance with STC and Master Eligibility List #SA2586NM. See 337 form this date. Tom Regier AP

Facility. (See back pages for other specific entries.)

		ade
on of Inspections, Tests, Repairs and Alterations to be endorsed with Name, Rating and Certificate Number of Technician or Repair back pages for other specific entries.) CARMIN 430W S/N23402424 FOR REPAIR	4557 S. Frankwood Ave. Reedley, CA 93654 559-859-0374 AIRCRAFT	ons, Tests, Repairs and Alterations n Name, Rating and Certificate Number of Technician or Repair other specific entries.)
VIATION INC. FARRS CPBR431L, WORK ORDER 21427 DISTALLED AFTER REPAIR OF CRACKS IN FAIRINGS PAINTED. REINSTALLED FAIRINGS. 2. Proprietarion of the propriet	N588ET Model: PA32-301T S/N: 32-8024044 Tach: 3380.76 AFTT: 3380.76 ELT DUE: Aug 2012 Lubed all bearings, pulleys, bell cranks, and hinges. Replaced ply, 44-77B nose wheel and wheel bearing sets, #220 SouthContraction. Charles the process of the serviced battery. Checked contractions are serviced by the serviced battery.	
• .		© ASA 1993

N588ET #2 ENGINE LOG

PACIFIC CONTINENTAL ENGINES, INC.

16041 ARMINTA ST.
VAN NUYS, CALIF. 91406
CERTIFIED STATION 4595

N588ET Log #2

ENGINE LOG BOOK

	AIRCRAFT ENGINE LOG FOR	BOOK NO.
ENGINE MAKE Lycomes	MODEL TIO-540-SIAD SE DATE OF MFG.	ERIAL NO. <u>L5741-61A</u> WEIGHTCU. IN
HPAT	RPM_DISPLACEMENTWINTER	
OU CARACITY	QTS. FUEL MIN. OCTANE	,
BREAKER POINT GAP	SPARK PLUG GAP	
OH DRESSURE	OIL TEMP. RANGE	
PROPELIER MAKE	MODELSERIA	AL NO
PITCH	LENGTHREMA	ARKS

NA

	CERTIFIED OVERHAUL TO NEW ENGINE TOLERANCES ZERO-TIME EQUIVALENT
FAA AF	IC CONTINENTAL ENGINES, INC. PPROVED REPAIR STATION #TZ3R885L 041 ARMINTA ST. • VAN NUYS, CA.
Make	TEXTRON LYCOMING
Model	TIO-540-S1AD
S/N	L574 L -61A
HRS. SMO	H 0.00
Federal Avia	was repaired and inspected in accordance with curren ation Regulations and is approved for return to service etails of the repair are on file at this Repair Station unde
Work Orde	NO92918
Date:	1.0/17/91
	Authorized Signature
∨ Total time o	n engine before overhaul2626.62

PERCENTORS, STREETER, ALTER

DATE ALT/GEN STARTER L MAG / DUAL R. MAGNETO CARB / SERVO / PIN RSAIDEDZ THRITL & CNTRL FUEL PUMP FLOW DIVIDER -FUEL REGULATOR TURBOCHARGER CONTROLLER CONTROLLER PR VALUE S/N WASTEGATE NOZZLE ASM COMMENTS: _NONE DATE: 191791 W/O#: 929/8 AUTHORIZED SIGNATURE:

PACIFIC CONTINENTAL ENGINES, INC. FAA APPROVED REPAIR STATION #TZ3R885L 16041 Arminta St. • Van Nuys, CA 91406 RECORD OF ACCESSORIES AND MAJOR PARTS SHIPPED WITH ENGINE NEW/OHC N/O TT _ P/N MZ4222R S/N GUOCOCZO NOTT N/A TSO -0-P/N 10-682560-135/N 72691026/ NOTT -6- TSO -0-_ **(**0)TT <u>-0 -</u> TSO <u>- 0 -</u> P/N RG9080J64 S/N D-2570 NOT -0- TSO -0-N/O TT _____ TSO N/O TT _____ TSO _ __ TSO _ N/O TT ____ N/O TT ____ N/O TT ____ TSO HARNESS A 51 - 1083 SPARK PLUGS RHB32E THESE COMPONENTS WERE REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATIONS AND ARE APPROVED FOR RETURN TO SERVICE, PERTINENT DETAILS OF THE REPAIR ARE ON FILE AT THIS REPAIR STATION

N	TACHOMETER READING	INSPECTION REFERENCE DATA
	· -	
G.TOTAL T	TIME 3552.47	
02003-14-03	gapped fuel and oil ler hoses with	
s determined	to be in	
en e		
Iterations	must be endorse	d by a certificated

lterations must be endorsed by a certificated y with name, rating and cert. no.

DATE	MAINTENANCE, INSPECTIONS, REPAIRS AND ALTERATIONS
	November 15, 2007 Tach/Total Time: 3168.0 Removed engine for overhaul and sudden stop inspection, disassembled and sent engine components to Lycon Engine Rebuilding Co. FAARS CT3R435 under work order #27972 for cleaning, inspection, and repair of the following as needed in accordance with Lycoming Table of limits manual SSP-1776: fuel pump drive adapter assy., fuel pump drive shaft-gear assy., fuel pump idler gear assy., crankshaft plain idler gear assy., crankshaft heavy idler gear assy., accessory drive gear assy., accessory drive gear assy., accessory drive gear assy., are governor drive gear, prop governor shaft gear, oil pump drive gear assy., vacuum pump shaft, turbo scavenge and thru bolts, right and left engine mount brackets, oil sump assy., and oil filter converter plate. Additionally, Lycon pins and plugs, valves, springs, and keepers; overhauled both crankshaft counterweight assemblies exchange with new pistons, pins and plugs, valves, springs, and keepers; overhauled both crankshaft counterweight assemblies in accordance with SSP-1776 and SI #1143B, overhauled crankshaft in accordance with SSP-1776, main and rod journals are standard, bore marked #1 repaired in compliance with SI #1142B, and crankshaft assembled and balanced in accordance with SI #1012F; overhauled and balanced all connecting rod assemblies in accordance with SSP-1776; inspected accessory with new bushings installed, and exhaust rocker arms were modified in accordance with SI #1468A and re-identified as pn 17F19710; inspected and tested all injector nozzles; overhauled starter pn MZ4218 sn 3E000013 in accordance with Kelly Aerospace OE-A2 overhaul manual; overhauled Bendix D6LN3000 magneto pn 10-682560-13 sn J219813GR (exchange upgraded from D2000 to D3000 series magneto). (continued)

FAARS #X9PR865N WO #3979 on 11/15/06. Camshaft reground by Aircraft Engine and Accessory Co. Inc. FAARS #U11R533K, WO #142562. All hydraulic valve lifters were inspected, reground and cleaned by Aircaft Specialties Service FAARS #DD2R74K, WO #153380. Crane Lear Romec fuel pump pn RG9080J6A/M sn B-551 was overhauled and modified by B&S Aircraft Parts and Accessories, Inc FAARS #NE2R028L wo #R06-01669, reference Crane Lear Romec manual RG9080 series revision #3, 1/6/95; SB RG9080-73-001 revision #3, 7-28-03; SB #539A, 11-22-00; and complies with AD 2003-14-03(a). Harrison oil cooler pn 85353111 sn 80A-1541 overhauled provided by Aircraft Parts and Accessories, Inc FAARS #NE2R028L wo #R06-01669, reference Crane Lear Romec manual RG9080 series revision #3, 1/6/95; SB RG9080-73-001 revision #3, 7-28-03; SB #539A, Oil Cooler Service, Inc FARS RF3R813L wo #56868 in accordance with Hydrison Space LMSS (2017 RD).	#U12R226L. Bendix fuel injector RSA10ED2 pn 2524791-6 sn 77733 overhauled by Airflow Performance Inc. FAARS #X9PR865N WO #3979 on 11/15/06. Camshaft reground by Aircraft Engine and Accessory Co. Inc. FAARS #U11R533K, WO #142562. All hydraulic valve lifters were inspected, reground and cleaned by Aircraft Specialties Service FAARS #DD2774K, WO #153380. Crane Lear Romec fuel pump pn RG908016A/M sn B-551 was overhauled and modified by B&S Aircraft Parts and Accessories, Inc FAARS #NE2R028L wo #R06-01669, reference Crane Lear Romec manual RG9080 series revision #3, 1/6/95; SB RG9080-73-001 revision #3, 7-28-03; SB #539A, 11-22-00; and complies with AD 2003-14-03(a). Harrison oil cooler pn 85353111 sn 80A-1541 overhauled by Pacific Oil Cooler Service, Inc FARS RF3R813L wo #56868 in accordance with Harrison Spec. HES-62-77F. Ignition harness overhauled by Aero-Lite FAARS AE3R368L wo #6K974 in accordance with Skytronics Inc. PMA Manufacturing Procedure #10 Rev. H dated 11-08-04. Hartzell propeller governor pn F-4-11BZ sn D374UJ completely disassembled, cleaned, inspected, resealed, bench tested and set I.A.W. Hartzell 130B, Rev 7 11-03 by Santa Monica Propeller Service Inc. FAARS ZY2R362L wo #G-210-06. Complied with SB 240S (mandatory parts replacement at overhaul) by installing the following new parts: oil filter bypass spring/spacer/sleeve/valve, oil filter converter plate gasket, all pipe plugs and prop governor idler gear setscrew, all standard main and rod bearings, all LW-12596 connecting rod bolts and nuts, oil pump #05K23463 with "N" impeller gears pn 51C23290 and 51B23274, all rocker shaft pn AEL72626; oil pressure relief spring pn 61084 and ball, tachometer shaft, all cylinder head drain tubes and interconnect hoses, vernatherm valve, magneto drive cushions, all gaskets, seals, induction hoses, clamps and tach drive snap ring. Installed new oil sump haffle plate. Complied with AD 23 of height by the MD23 of the feature of the proper of the		DAJE M/17/01 BRT.	GROUND TIME	AIR TIME	TOTAL	SIGNATURE OF PERSON MAKING ENTRY	TACHOMETER READING	INSPECTION REFERENCE DATA
				#U11R Service overhat Crane 1 11-22-(Oil Co- harness Manufa complet Santa M replacer converte 12596 c rocker s and inte	8 #X9PR865N 533K, WO #I FAARS #D uled and modi Lear Romec n 00; and compli- oler Service, overhauled acturing Proce- tely disassemb Monica Propell ment at overha- er plate gasket connecting rod whaft pn AEL7 erconnect hose- pap ring. Insta	I WO #3979 of 142562. All It D2R74K, Wo fied by B&S Ananual RG908 ies with AD 2 Inc FARS RI by Aero-Lite Edure #10 Repoled, cleaned, ler Service Incaul) by install and pipe plug It bolts and nut 2626; oil press, vernatherm lled new oil service Incaul)	on 11/15/06. Camshaft reground by Aircraydraulic valve lifters were inspected, rego #153380. Crane Lear Romec fuel Aircraft Parts and Accessories, Inc FAAR 0 series revision #3, 1/6/95; SB RG9080 003-14-03(a). Harrison oil cooler pn 853 F3R813L wo #56868 in accordance with FAARS AE3R368L wo #6K974 in v. H dated 11-08-04. Hartzell proper inspected, resealed, bench tested and seeds. FAARS ZY2R362L wo #G-210-06. Ging the following new parts: oil filter be a sand prop governor idler gear setscrew, a seed the proper inspected in the following new parts: oil filter be a sand prop governor idler gear setscrew, a seed the proper inspected in the following new parts: oil filter be a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand prop governor idler gear setscrew, a seed the proper inspected with a sand proper inspected with	overhauled by Airflow Peaft Engine and Accessory Caround and cleaned by Airpump pn RG9080J6A/M S #NE2R028L wo #R06-0-73-001 revision #3, 7-28-53111 sn 80A-1541 overhath Harrison Spec. HES-62-accordance with Skytron ller governor pn F-4-11B at I.A.W. Hartzell 130B, R Complied with SB 240S (mypass spring/spacer/sleeve/stl standard main and rod bear gears pn 51C23290 and one ten seals, initially like and the seals are seals and the seals and the seals and the seals are seals are seals and the seals are	tems approval rformance Inc. o. Inc. FAARS caft Specialties sn B-551 was 669, reference 03; SB #539A, uled by Pacific 77F. Ignition cs Inc. PMA Z sn D374UJ ev 7 11-03 by andatory parts valve, oil filter urings, all LW- 51B23274, all and drain tubes amps and tach
TOTALS			TOTALS	•		:			
	garage and terri. no.		IOIALS	:	:	:	All repairs, inspections, alteration mechanic or repair agency with	ons must be endorsed	y a certificated

7	TACHOMETER READING	INSPECTION REFERENCE DATA
77733 ov by Aircraft ected, regree ec fuel pu c FAARS a RG9080-7 er pn 85353 ance with 1974 in ac ill propelled d and set 110-06. Cc 1 filter byp tscrew, all N" impeller ball, tachor all gaskets, ith AD 02.	Aerospace Power Systerhauled by Airflow Per Engine and Accessory Cound and cleaned by Aircomp pn RG9080J6A/M s #NE2R028L wo #R06-01 3-001 revision #3, 7-28-0 111 sn 80A-1541 overhau Harrison Spec. HES-62-coordance with Skyronion governor pn F-4-11B2 I.A.W. Hartzell 130B, Romplied with SB 240S (moss spring/spacer/sleeve/vestandard main and rod bear gears pn 51C23290 and meter shaft, all cylinder he seals, induction hoses, cl. 12-07 by installing 06B2 ance with SB 543B. (control of the seals of the seals) (control of the sea	of Inc. FAARS aft Specialties n B-551 was 669, reference 3; SB #539A, alled by Pacific 77F. Ignition cs Inc. PMA Z sn D374UJ ev 7 11-03 by andatory parts alve, oil filter urings, all LW- 51B23274, all and drain tubes amps and tach 3072 oil filter
alteration	ns must be endorsed name, rating and cert	by a certificated , no.

DATE	MAINTENANCE, INSPECTIONS, REPAIRS AND ALTERATIONS	
11/15/07	(continued) Assembled engine using the Avco Lycoming Direct Drive Overhaul Manual, Table of limits #SSP1776 and IPB #PC-315-5 rev. 12-96 as a guide. Used SI 1343A as a guide to secure propeller idler shaft set screw; used SI 1060J to maintain proper .028080 inch dry tappet clearance; used SI 225B to install proper rocker arm thrust washers; used SI 1324A as a guide to install crankshaft oil seal; use SI 1059B as a guide to pre-lubricate internal engine parts prior to assembly; used SI 1029D as a guide to tighten thru-studs and bolts. Installed magneto with new capacitor assemblies, and timed to engine. Installed new air conditioner compressor assembly pn 101173-006 and new inlet and outlet air conditioner hoses from compressor-to-firewall certified and supplied by PHT Inc. Tulsa, OK. Installed new alternator and air conditioner drive belts pn 73965-009 and 73695-010. Kelly Aerospace Turbocharger pn 406610-9026 sn JBO152 overhauled by Approved Turbo Components, Inc. FAA certificate #07TR2100 W.O. #27306 in accordance with Kelly Aerospace Ovhl. Manual P/N 400600-0000 and including all new heat blankets/shields installed. Garrett P.R. Valve pn LW14445-10 sn JAO110 also overhauled by Approved Turbo Components, Inc. wo #27307 in accordance with Control System Ovhl. Manual #TP20-0129\TP30-4003. Installed new TIT exhaust probe pn 25-86230. Installed and rigged new waste gate control cable pn LW-16019. Exhaust system repaired by Knisely Welding, Inc FAARS #NJ3R712L under WO # 124163 which included new left and right stack assemblies, new tailpipe, new wastegate elbow. Wastegate/transition pn LW16858 overhauled by Aerospace Welding Minneapolis, Inc. FAARS #UWDR792L wo 100459. Complied with SB448A by setting waste gate gap at full power. Complied with SB 484 and AD 91-21-01R1 by inspection of exhaust system. Installed three new fuel injector lines pn AEL12098-100 and AEL12098-170 and one serviceable injector line at cylinders #2, #4,#5,. Installed all new injector line clamps and inspected in accordance with AD 02-2	

		DATE 11/15/b)	GROUND TIME	AIR TIME	TOTAL TIME	SIGNATURE OF PERSON MAKING ENTRY	TACHOMETER READING	INSPECTION REFERENCE DATA
		BRT. FWD.	with 1 entry (shock installe used a	11.29 hours TIS 03-13-02). Ins c mounts pn J38 lled CH48103 fil as a guide to pe	IS (see logbook astalled new vac 804-40 and new Titer. Installed the erform turboche 1966 hours TIS	the engine compartment, including the oil kentry 10/01/04). Reinstalled vacuum pucuum hose from vacuum pump to firew vattach bolts and nuts. Serviced engine vacuum Brackett air inlet filter. Test run and arger system operational test. Set idle rpreper SI 1009AR. I certify that this enginermined to be in airworthy condition. Total	wall. Reinstalled engine w with 12 quarts Chevron 100 I test flew aircraft. Kelly A m and mixture. Next engi- ne has been inspected in acc	rith all new Lord 0 mineral oil and erospace SB #23 ne overhaul due
(Tach/ Drained installe	pary 1, 2008 n/ TT: 3175 ed engine oil ar led new CH481 ir. Tom Regier	103-1 oil filter.	ilter with no metal found. Added 12 qu. Removed turbo scavenge pump and re	uarts Chevron 100 straight einstalled after resealing b	mineral oil and —— y Lycon Engine
			:	:	:(
			:	:	:			
	Tr	OTALS	:	:	:	All repairs, inspections, alterat mechanic or repair agency wit	tions must be endorse th name, rating and c	d by a certificated ert. no.
						modia		
						Nu. 12		4 .
	,				· ////	•		
								• *. .

BSW/TT/ AIR 4557 S. Frankwood Ave. Reedley, CA 93654 559-859-0374 ENGINE

N588ET Model: <u>TIO-540-S1AD</u> S/N: <u>L5741-61A</u>
Tach: <u>3380.76</u> TT: <u>3380.76</u> SMOH: <u>212.76</u> STOH: <u>N/A</u>

Changed engine oil, 12 qts. Aeroshell 100 Plus. Replaced CH4103-1 filter and cleaned screens. Cleaned and gap spark plugs. Set mag timing. Checked compressions #1 76, #2 76, #3 73, #4 78, #5 77, and #6 79. Replace 99072-00 pulley, 553-230 bolt, 62833-82 shims, 95061-151 bushing,BA-7 air filter, Operated engine and leak checked. All work performed IAW current FARs and manufactures maintenance manuals. The above work and engine has been inspected in accordance with an Annual inspection and engine was determined to be in airworthy condition, and is approved for return to service. AD compliance report attached to aircraft permanent records. Return to service.

a:	C	arl E. Smith IA				
Signature.			Date:	September 14, 2011		
	·		:		,	
	:	:	:			
TOTALS	:	:	:	All repairs, inspections, alteration mechanic or repair agency with a	ns must be endorsed name, rating and cert	by a certificated . no.

OF PERSON

ENTRY

INSPECTION

REFERENCE

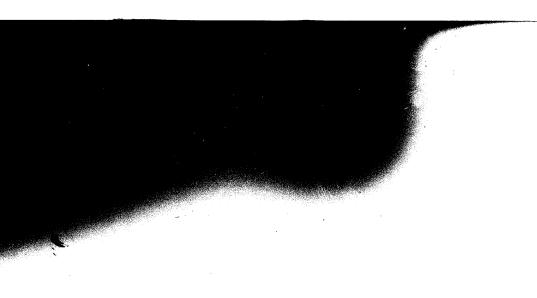
DATA

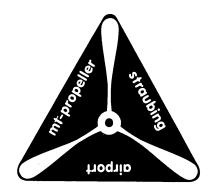
TACHOMETER

READING

	TACHOMETER READING	INSPECTION REFERENCE DATA
	,	
alteration	ns must be endorsed name, rating and cei	by a certificated

DATE	MAINTENANCE, INSPECTIONS, REPAIRS AND ALTERATIONS	
REP AA3 AND ES-4 OIL OPE BAT	LACED VACUMME PUMP WITH NEW TEMPEST 215CC S/N 814217. INSTALLED NEW VACUMME FILTER RELEIF FILTERS. OP CHECKED SYSTEM OK. REPLACED 8103-2 OIL FILTER WITH NEW UNIT. DID NOT REPLACE REPLACED COPILOT SEAT LINKAGE ROD P/N 763422-00. RATIONAL CHECK OF SEAT OK. REPLACED G-35 TERY WITH NEW UNIT. TOP CHARGED AND INSTALLED. OUND RUN UP AND LEAK CHECK OK. RAY HARKINS AP	





mt-propeller

The Winner's Propeller

PROPELLER LOGBOOK

EASA DE.21G.0008 / EASA.21J.020

MTE - 119 Juni 2005

Owner Name	
Address	
Propeller Model MTV - 9 - B	
Blade Model	
Hub Serial No. 070165	
Blade Serial No. 1 YB - 29709	Propeller Diameter: 798 cm + 95 cm
2 48 - 29-110	Low Pitch: 12,5 ° ± 0,2 ° TCI: 72 Month
3 YB-29-11-1	High Pitch: <u>3 '</u> ° ± <u> </u>
4	Feathering:° ± °
5	Feathering: ° ± ° Reverse: ° ± °
6	at radius station: 74 cm

MAINTENANCE & SERVICE HISTORY

Date	Total Propeller Time	Time Since Overhaul	Descript	Authorized Signatures Repairman and Station Certificate Numbers	
19,0307	.00:00.		Manufac.	tured:	
_ I certif inspecti _ conditio	M588ET Made model: 198-52 Black Tach: 3380.7 y this propeller has boon as per FAR 43 Apon. AD compliance rejucto service.	AND	170165 -29110, <u>TB-29111</u> DH: <u>N/A</u> ance with an Annual o be in an airworthy		
Signatu	re:		ctory Service Bullet	ins On Pages 11 & 12	

REVISED WEIGHT AND BALANCE FORM

--Equipment Change--

Revision Date: January 1, 2008

Supercedes weight and balance supplement dated November 15, 2007

Aircraft Information

* Make/Model:

Piper PA32-301T

* Serial:

32-8024044

* Registration:

N588ET

Maximum Ramp Weight:

3617 pounds

Maximum Takeoff or Landing Weight:

3600 pounds

New Empty Weight:

2277

pounds

New Useful Load:

1340

pounds

New EWCG:

+83.77

inches

Equipment Added/Removed	Weight	Arm	Moment
Aircraft: Revision dated 11/15/2007	+2289.68	+ 83.9	+192109.77
Remove II Morrow 612 Loran	- 3.8	+ 68.5	- 260.3
Remove CI-121SP Antenna/Preamp	- 1.0	+ 97.5	- 97.5
Remove KX170/Mac 1700 Tranceiver	- 7.3	+ 61.6	- 449.68
Remove KI-203 Indicator	- 1.7	+ 65.5	- 111.35
Remove KY-197 Tranceiver	- 4.2	+ 63.7	- 267.54
Remove KNS-80 Rnav	- 7.0	+ 62.3	- 436.1
Remove KR-87 ADF	- 3.2	+ 61.8	- 197.76
Remove KI-227 ADF indicator	7	+ 65.55	- 45.89
Remove KA-44B ADF antenna	- 4.2	+192.75	- 809.55
Remove KMA 24 audio panel	- 1.7	+ 60.4	- 102.68
Remove Apollo GX-50 GPS	- 1.0	+ 61.7	- 61.7
Remove MD-41 annunciator/switch unit	75	+ 64.65	- 48.49
Remove A-33 GPS antenna	25	+ 87.4	- 21.85
Add GNS 430W Nav/GPS/Com/WAAS	+ 6.2	+ 62	+ 384.4
Add GNS 530 Nav/GPS/Com	+ 8.6	+ 62	+ 533.2
Add GA 56 GPS antenna	+ .25	+ 91	+ 22.75
Add GA 35 GPS/WAAS antenna	+ 1.0	+100	+ 100
Add GMA 340 audio panel	+ 1.7	+ 64	+ 108.8
Add Mid-Continent MD-200 indicator	+ 1.4	+ 65	+ 91
Add PCD7100P CD/MP3 player	+ 2.2	+ 63.4	+ 139.48
Add JPI EGT-701 Temp. Indicator	+ 3.2	+ 65	+ 208
Total	+2277.43	XXX	+190787.01

Performed by: Certificate Number: Marine Brown

Tom Regier AP 1

Professional Pilots Inc. 12304 West 165th Avenue Lowell, IN 46356

FAA APPROVED

AIRPLANE FLIGHT MANUAL SUPPLEMENT For Piper Model PA-32-301T NORMAL CATEGORY ONLY

Reg. No.	N588ET	
Ser. No.	32-8024044	

This supplement must be attached to the FAA Approved Airplane Flight Manual dated January 9, 1980 when a MT-Propeller MTV-9-B/198-52 constant speed propeller is installed. The information contained herein supplements or supercedes the basic manual only in those areas listed herein. For limitations, procedures, and performance information not contained in this supplement, consult the basic airplane flight manual.

Approved: Charles L. Smalley, Manager, Systems and Flight Test Branch Chicago, Aircraft Certification Office

1. GENERAL:

1.5 Propeller

3 Blade Propeller

(a) Number of propellers1(b) Propeller ManufacturerMT-Propeller(c) Blade Model198-52(d) Number of blades3(e) Hub ModelMTV-9-B(f) Propeller diameter (centimeters)

 (1) Minimum
 193

 (2) Maximum
 198

(g) Propeller type Constant speed, Hydraulically actuated

2. LIMITATIONS:

2.7 Power Plant Limitations

3 Blade Propeller

(a) Number of engines1(b) Engine manufacturerLycoming(c) Engine model numberTIO-540-S1AD

(d) Engine operating limits

(1) Maximum Horsepower300(2) Max. Rotation Speed (RPM)2700(3) Max Oil Temperature (F)245

(e) Oil Pressure

(1) Minimum (red line) 25 PSI (2) Maximum (red line) 100 PSI

(f) Fuel Flow / Pressure

Maximum (red line) 35 gal/hr; 14 PSI

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(g) Fuel (Minimum Grade)

100 Green, or

100LL Blue aviation grade

(h) Number of propellers

(i) Propeller manufacturer

MT-Propeller MTV-9-B

(j) Propeller hub and blade model

198-52

(k)Propeller diameter (centimeters)

Minimum

193

Maximum

193

(l) Blade angle limits

Low pitch stop

12.4 +/- 2 degrees

High pitch stop

32.0 +/- 1 degrees

ALL OTHER SECTIONS UNCHANGED

AA Approved:	.1	2	FEB	2007

US Department of Transportation Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form A	/ppi	ove	d
OMB	No.	212	0-002

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43. Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

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Continuation N588ET January 4, 2007

8) Diagrams:

Not applicable.

9) Special Inspection Requirements:

No Change. See MT-Propeller Operation and Installation Manual E-124 for details. Special inspections are required following events such as overspeed, lightening strike, foreign object / ground strike, or fire / heat damage. See manual for complete list and details.

10) Application of protective treatments:

No Change. See MT-Propeller Operation and Installation Manual E-124 for details.

11) Data:

No Change. See MT-Propeller Operation and Installation Manual E-124 for all details relative to proper torque values.

12) List of Special Tools:

Not applicable.

13) Commuter category aircraft:

Not applicable.

14) Recommended overhaul periods:

See MT-Propeller Operation and Installation Manual E-124 for details. MT-Propeller currently recommends overhaul every 1800 hours of operation, or 72 calendar months, whichever comes first.

15) Airworthiness limitations section:

No additional airworthiness limitations.

16) Revision:

This ICA may be revised by submitting a letter to the local FSDO along with a revised FAA Form 337, and ICA for consideration.

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