

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

U.S.A. N95473

Nationality and Registration Mark

6-21-13

Date

Removed right landing gear assembly. Removed right wheel and brake assembly. Removed fabric from right landing gear assembly. Removed bent landing gear streamline strut tube from right landing gear and welded new direct replacement Univair streamline strut tube, p/n UB-A51, in accordance with AC 43.13-1B chg. 1, section 5, welding and brazing. Repaired small dent in forward upper landing gear tube by welding a cover patch of .049 4313 steel tube according to AC 43.13 chg. 1, section 5, welding and brazing. Sand blasted and inspected welds. Installed new fabric on right gear leg according to Poly-Fiber covering manual no. 1 dated July 2001 or later FAA approved revision. Completed per STC SA1008WE. Painted white. Patched holes in left side of rudder, left aileron at last outboard rib and bottom of fuselage in accordance with AC 43.13-1B, section 4, repairs in fabric covering. Repaired 6" of broken wood stringer on bottom of fuselage just aft of right landing gear according to AC 43.13-1B chg. 1, section 4. Replaced right shock cord assemblies with Univair shock cords, p/n 9010HD. No change in aircraft weight and balance.

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☐ Additional Sheets Are Attached