FAA FORM 8130-6, APPLICATION FOR U.S. AIRWORTHINESS CERTIFICATE

Form Approved

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IV. INSPECTION AGENCY VERIFICATION	DATE	_			TITLE							SIGNATURE					
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1	rm 813	0-6 /	10-6	04) Previous	Edition Date	d \$/0√	May be Used	Until Der	leted	Except	t for	Light-Sport Ai	rcraft		NSN	V: 0052-	00-024-7006

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	F. CERTIF	FICATION - I h	ereby certify tha	t I am the register	red owner (or his agent) of t at seq. and applicable Fede	the aircraft o	escribed above; that the	e aircraft is reg	gistered with the Fede	aral Aviation Admini	stration in		
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•	1	Operating Limit as applicable.	ations and Mark	ings in Compliand	e with 14 CFR Section 91.9	9,			ormity, FAA Form 8130-9 (Attach when required)				
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VORTHIN (FAA/DE niy)				etc. (Attach when rec			Previous Airworthi 14 CFR Section	ness Certifica	te Issued in Accordan	nce with (Original A	ftached)		
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VIII. AIRWORTHINESS DOCUMENTATION (FAA/DESIGNEE use only)			Recorded in Ain				14 CFR Section	21 /	of Compliance AAF	(Copy Atta			

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	UNITED STATES OF AMERICA DEPARTMENT OF TRANSPORTATION - FEDERAL AVIATION ADMINISTRATION SPECIAL AIRWORTHINESS CERTIFICATE									
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	orm 8130-7 (07/04)	I I I A I LIOP	DLL III		ERSE SIDE		AL NEO	CENTIONO		-00-693-4000

A	This airworthiness certificate is issued under the authority of Public Law 104-6, 49 United States Code (USC) 44704 and Title 14 Code of Federal Regulations (CFR).
В	The airworthiness certificate authorizes the manufacturer named on the reverse side to conduct production fight tests, and only production flight tests, of aircraft registered in his name. No person may conduct production flight tests under this certificate: (1) Carrying persons or property for compensation or hire: and/or (2) Carrying persons not essential to the purpose of the flight.
С	This airworthiness certificate authorizes the flight specified on the reverse side for the purpose shown in Block A.
D	This airworthiness certificate certifies that as of the date of issuance, the aircraft to which issued has been inspected and found to meet the requirements of the applicable CFR. The aircraft does not meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention On International Civil Aviation. No person may operate the aircraft described on the reverse side: (1) except in accordance with the applicable CFR and in accordance with conditions and limitations which may be prescribed by the Administrator as part of this certificate; (2) over any foreign country without the special permission of that country.
E	Unless sooner surrendered, suspended, or revoked, this airworthiness certificate is effective for the duration and under the conditions prescribed in 14 CFR, Part 21, Section 21.181 or 21.217.



Administration

North FSDO-15

5950 Hazeltine National Dr Suite 500, Citadel International Orlando, Florida 32822 407-812-7730, Fax: 407-812-7710

OPERATING LIMITATIONS EXPERIMENTAL LIGHT SPORT AIRCRAFT

AIRPLANE

REG. NO.	MAKE:	MODEL:	SERIAL NO:
N678TW	Aerotrike	Safari	TWAEROTRIKE2000

These operating limitations form a part of the Special Airworthiness Certificate issued for the airplane described above and must be displayed in the aircraft in accordance with 14 CFR §91.203(b) and be available to the pilot in command of the aircraft..

- (1) No person may operate this aircraft for other than the purpose of meeting the requirements of § 91.319(b) during phase I flight testing and, for the purpose of operating light-sport aircraft, after meeting these requirements as stated in the program letter (required by § 21.193) for this aircraft. In addition, this aircraft must be operated in accordance with applicable air traffic and general operating rules of part 91 and all additional limitations herein prescribed under the provisions of § 91.319(e). These operating limitations are a part of Form 8130-7, must be carried in the aircraft at all times, and must be available to the pilot in command of the aircraft.
- (2) This aircraft must display the word "experimental" in accordance with § 45.23(b).
- (3) This aircraft does not meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation. The owner/operator of this aircraft must obtain written permission from another CAA before operating this aircraft in or over that country. That written permission must be carried aboard the aircraft together with the U.S. airworthiness certificate and, upon request, be made available to an ASI or the CAA in the country of operation.
- (4) Application must be made to the geographically responsible FSDO or MIDO for any amendment to these operating limitations. During phase I flight testing to meet the requirements of § 91.319(b), or as a result of the incorporation of a major change, all flights must be conducted within the assigned geographic area. 25 nautical mile radius of Greystone airport Lat29-16.815N, Long82-07.489W
- (5) Flight testing required for phase I operations or as a result of the incorporation of a major change will be conducted in the assigned test area. Flight test operations will only be conducted under VFR day conditions, with the pilot as the sole occupant of the aircraft. This aircraft must be operated for at least (5) hours in the assigned geographic area. Following the satisfactory completion of the required number of flight hours in the flight test area, the pilot must certify in the aircraft records that the aircraft has been shown to comply with § 91.319(b) with a statement that includes the following information: "I certify that the prescribed flight test hours have been completed and the aircraft is controllable throughout its normal range of speeds and throughout all maneuvers to be executed, has no hazardous operating characteristics or design features, and is safe for operation. The flight test was completed under the following conditions: maximum operating weight, style/set of wing or sail, maximum demonstrated airspeed, and minimum demonstrated stall speed." All major changes or modifications will be listed in the aircraft records and the compliance statement will be restated with the changes listed. The aircraft may not be operated in excess of the weights and speeds demonstrated.

- (6) Any change to the flight test area location or size must be coordinated with the geographically responsible FSDO where the aircraft is based, with FAA concurrence received in writing.
- (7) Except for takeoffs and landings, this aircraft may not be operated over densely populated areas or in congested airways.
- (8) This aircraft is prohibited from operating in congested airways or over densely populated areas, unless directed by air traffic control, or unless sufficient altitude is maintained to effect a safe emergency landing in the event of a power unit failure, without hazard to persons or property on the ground.
- (9) This aircraft is to be operated under VFR day only.
- (10) After completion of phase I flight testing, unless appropriately equipped for night and/or instrument flight in accordance with § 91.205, this aircraft is to be operated under VFR day only.
- (12) No person may operate this aircraft for carrying persons or property for compensation or hire.
- $(13) \, n/a$
- $(14) \, n/a$
- (15) The pilot in command of this aircraft must advise the passenger of the experimental nature of this aircraft and that it does not meet the certification requirements of a standard certificated aircraft.
- (16) This aircraft must contain the placards and markings as required by § 91.9. In addition, the placards and markings must be inspected for legibility and clarity, and the associated systems inspected for easy access and operation, to ensure they function in accordance with the manufacturer's specifications during each condition inspection.
- (17) This aircraft is prohibited from aerobatic flight, that is, an intentional maneuver involving an abrupt change in the aircraft's attitude, an abnormal attitude, or abnormal acceleration not necessary for normal flight.
- (18) n/a
- (19) The pilot in command of this aircraft must hold at least—
 - (a) A student pilot certificate with a weight shift control category, single engine land class, and Aerotrike, Safari make/model privilege endorsement by an authorized instructor; or
 - (b) A sport pilot certificate, with a weight shift control category, single engine land class, and Aerotrike, Safari make/model privilege within that set of aircraft (reference § 61.1(b)(14)); or
 - (c) A recreational pilot certificate or higher with sport pilot privileges, with a weight shift control category, single engine land class, and Aerotrike, Safari make/model privilege within that set of aircraft (reference § 61.1(b)(14)); or
 - (d) A recreational pilot certificate or higher.

- (20) This aircraft must not be used for banner towing operations or intentional parachute jumping.
- (21) The pilot in command of this aircraft must notify air traffic control of the experimental nature of this aircraft when operating into or out of airports with an operational control tower. When filing IFR, the experimental nature of this aircraft must be listed in the remarks section of the flight plan.
- (22) Aircraft instruments and equipment installed and used under § 91.205 must be inspected and maintained in accordance with the requirements of part 91. Any maintenance or inspection of this equipment must be recorded in the aircraft maintenance records.
- (23) No person may operate this aircraft unless within the preceding 12 calendar months it has had a condition inspection performed in accordance with the scope and detail to appendix D to part 43, or other FAA-approved programs, and was found to be in a condition for safe operation. As part of the condition inspection, cockpit instruments must be appropriately marked and needed placards installed in accordance with § 91.9. In addition, system-essential controls must be in good condition, securely mounted, clearly marked, and provide for ease of operation. This inspection will be recorded in the aircraft maintenance records.

(24) n/a

- (25) Condition inspections must be recorded in the aircraft maintenance records showing the following, or a similarly worded, statement: "I certify that this aircraft has been inspected on [insert date] in accordance with the scope and detail of appendix D to part 43 or the manufacturer's inspection procedures, and was found to be in a condition for safe operation." The entry will include the aircraft's total time-in-service, and the name, signature, certificate number, and type of certificate held by the person performing the inspection.
- (26) An experimental LSA owner/operator as a repairman for this aircraft under § 65.107 or an appropriately rated FAA-certificated mechanic may perform the condition inspection required by these operating limitations.
- (27) These Operating Limitations and associated FAA Form 8130-7, Special Airworthiness Certificate expire on unlimited

Jay B. Kurtz DAR

Date issued: 2007

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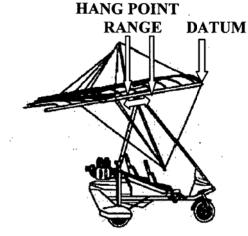
Jay B. Kurtz DAR

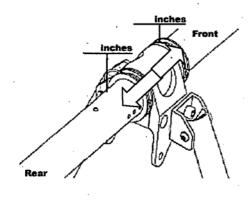
Date issued: 8-29

WEIGHT-SHIFT-CONTROL (TRIKE) WEIGHT AND LOADING

Make: HEROTRIKE Date: 8-23-01
Model: SAFARI Registration #: N678TW
Empty Weight: 440 Serial #: TWAEROTRIKE 2000
Gross Weight: 950 Wing Make: NORTH WANGModel: 19 FT
19 FT
Hang Point Range
The tricycle carriage is suspended from the wing hang point at a range of 56 3/8 inches and
5 % 6/0 inches back from the datum at the front of the keel tube.

Trim Hang Point Position for Payload Carried (Check and complete 1 or 2)
The wing can be trimmed to adjust for heavy or light payloads and speed by adjusting the trike
hang point forward or rearward within the hang point range.
1. All positions within the hang point range are operational at any payload up to the aircraft gross weight.
2. For payloads up to pounds use hang point position inches to inches from
datum. For payloads from pounds up to aircraft gross weight, use hang point position
inches to inches from datum.
Load Distribution (Check and complete 1 or 2)
1. Any payload up to gross can be distributed between one or both seats.
2. Specify any payload distribution considerations between the front and back seat
T. A. A. T.





Effect of Changes to Hang Point

		Maneuverability	Speed	Wing Efficiency	Pitch Stability	Use for Payload
Hang point moved Forward	→	7 Increase		\leftrightarrow	الا Decrease	Heavy
Hang point moved Rearward	+	☑ Decrease	☑ Decrease	\leftrightarrow		Light

Notes:

- *Empty Weight:* The total weight of the complete airframe, engine, wing and all permanently installed accessories and equipment. Includes unusable fuel and oil.
- Payload: Gross weight Empty weight = Payload. The weight of pilot, passenger, usable fuel, oil and baggage.
- Gross Weight: Empty weight plus the weight of passengers, fuel, oil, baggage, etc. all that is carried in the plane. Specified by the manufacturer.