FUEL VAF

The engine fuel system can become susceptible to fuel vapor formation on the ground during warm weather. This will generally occur when the outside ambient air temperature is above 85°F. The situation is further aggravated by the fact that the engine fuel flows are lower at idle and taxi engine speeds. When vapor occurs as evidenced by idle engine speed and fuel flow fluctuations, the following procedures are recommended.

IRES

- 1. With the mixture full rich, set the throttle at 1800 RPM to 2000 RPM. Maintain this power setting for 1 to 2 minutes or until smooth engine operation returns.
- 2. Retard the throttle to idle to verify normal engine operation.
- Advance the throttle to 1200 RPM and lean the mixture a described under FUEL SAVINGS PROCEDURES FOR FLIGHT TRAINING OPERATIONS.
- 4. Just prior to TAKEOFF, apply full throttle, for approximately seconds to verify smooth engine operation for takeoff.

NOTE

When the engine is operated above 1800 RPM, the resulting increased fuel flow also makes for lower fuel temperatures throughout the engine fuel system. This increased flow purges the fuel vapor and the cooler fuel minimizes vapor formation.

STALLS

The stall characteristics are conventional and aural warning provided by a stall warning horn which sounds between 5 and 1 knots above the stall in all configurations.

Power off stall speeds at maximum weight for both forward aft C.G. positions are presented in Section 5.

STALL SPEEDS AT 2550 POUNDS

Conditions: Power Off

MOST REARWARD CENTER OF GRAVITY

	FLAP SETTING	ANGLE OF BANK								
		0°		30°		45°		60°		
			KCAS	KIAS	KCAS	KIAS	KCAS	KIAS	KCAS	
Γ	UP	48	53	52	57	57	63	68	75	
ı	10°	42	50	45	54	50	59	59	71	
	30°	40	48	43	52	48	57	57	68	

MOST FORWARD CENTER OF GRAVITY

	ANGLE OF BANK								
FLAP SETTING	0°		30°		45°		60°		
0217110		KCAS	KIAS	KCAS	KIAS	KCAS	KIAS	KCAS	
UP 10°	48 43	53 51	52 46	57 55	57 51	63 61	68 61	75 72	
30°	40	48	43	52	48	57	57	68	

NOTES:

- Altitude loss during a stall recovery may be as much as 230 feet.
 KIAS values are approximate.

Figure 5-3. Stall Speeds

ICLOCK / O.A.T. INDICATOR

An integrated clock / O.A.T. / voltmeter is installed in the upper left side of the instrument panel as standard equipment. For a complete description and operating instructions, refer to the Supplements, Section 9.

STALL WARNING SYSTEM

The airplane is equipped with a pneumatic type stall warning system consisting of an inlet in the leading edge of the left wing, an air-operated horn near the upper left corner of the windshield, and associated plumbing. As the airplane approaches a stall, the low pressure on the upper surface of the wings moves forward around the leading edge of the wings. This low pressure creates a differential pressure in the stall warning system which draws air through the warning horn, resulting in an audible warning at 5 to 10 knots above stall in all flight conditions.

STANDARD AVIONICS

Standard avionics for the Model 172S airplanes include the following equipment:

KX-155A	Nav/Com		with	ΚI	208	or	ΚI	209A		
	Indicator F									
KT-76C	Transponder									
KMA-26	Audio Par	iel								
3000-11	Emergeno	y Locato	or Trai	nsmi	tter (E	LT)				

For complete operating instructions on the standard and optional avionics systems, refer to the Supplements, Section 9.

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