FLIGHT EXPRESS INCORPORATED Engine Discrepancy Record

| aft: N | 1: <u>6149B</u> Date: <u>01/3</u> | ////Tach:7/03./ | TSMO: _ | 1258.5 |
|--------|-----------------------------------|----------------------|---------|--------------|
| TEM | DISCREPANCY | CORRECTIVE ACTION | , | |
| 1 | Comply With 50 HR Engine Oil And | CAW SOKE OIL | NAME | |
| Filte | r Change. | CHANGE | CERT. | |
| ····· | | | INSP. | A TIME /3 |
| 2 | Drain Engine Oil | AKAINED ENG OIL | NAME | |
| | | | CERT. | |
| | | | INSP. | has TIME /3 |
| 3 | Remove & Inspect Filter Element | GEMBUEO TINSBUTED | NAME | |
| | | FILTER. NO DEFECTS | CERT. | |
| | | 1900 TED | INSP. | TIME 1/3 |
| 4 | Safety Oil Drain Plug | TWSTALLED + SAFETTED | NAME | 1 |
| | | DAMEN PLUL | CERT. | |
| | | | INSP. | IS TIME /3, |
| 5 | Service Engine Oil | SERVICED 10 973 LSW | NAME | 2 |
| | | SERVICED 10 973 65W | CERT. | |
| | | | INSP. | S TIME /S |
| 6 | Install & Safety New Oil Filter | INSTALLED TOLANED, 9 | NAME | 1 |
| | | SAFETTED NEW FILTER | CERT. | |
| | | 1/n AA48108-2 | INSP. | S TIME /3 |
| 7 | Leak Check Due After Oil Change | LEAK CHECK GOOD. | NAME | 1 |
| | | | CERT. | |
| | | | INSP. | 1 S TIME /3/ |
| | | | NAME | • |
| L | | · | CERT. | |
| | | | INSP. | TIME |
| | | | | TOTAL |

1

FLIGHT EXPRESS, INC. CESSNA 210/T210 ENGINE 100-HOUR INSPECTION CHECKLIST (FAR 135.421a)

AIRCRAFT N 6149B DATE 601 1031 11 TACH 7059. 8 ACTT 12,014. 7 TSMOH 1215.2

| The state of the s |
|--|
| 1 – Check Cockpit Maintenance Report, For Recurring Engine AD's and Enter on Engine Discrepancy Sheets |
| 2 - Calculate engine TSMOH, Enter on Engine Discrepancy Sheet |
| 3 - Check TIS of governor and prop (if less than 90hr to TBO, notify Maintenance Coordinator) |
| TAXI CHECK FLIGHT INSTRUMENTS: |
| 1 - Check for operation: Directional gyro h. Attitude Indicator fund Turn Coordinator fund |
| PRE-INSPECTION RUN-UP: |
| 1 - AT 1700 RPM: Mag drop Lt 75 Mag drop Rt 75 Suction 5.2 "Cycle Prop 400 |
| 2 - At Full Power: (Keep to Minimum) MP 27 RPM2650 F/F 200 Oil Press 50 Oil Temp 65 CHT for Alt Output for EGT FIRE |
| 3 – At Idle: (650-700) from MP 13 Oil Press 45 Oil Temp from CHT from Alternate Vacuum if installed 515 Magneto System Grounding Check from |
| 4 – OPS CHECK BC425 STANDBY ALTERNATOR IF INSTALLED (See Form FX1020BC for op's info) |
| 5 - CUTOFF RISE AT 1,000 RPM: SZD RPM. |
| ACCOMPLISHED BY: Date 01/03/1 |
| POST INSPECTION RUN-UP: |
| 1 - AT 1700 RPM: Mag drop Lt 95 Mag drop Rt 95 Suction 5.2" Cycle Prop 37 |
| 2 – At Full Power: (Keep to Minimum) MP 27 RPM 2675 F/F 21 Oil Press Oil Temp CHT Alt Output EGT EGT |
| 3 - At Idle: (650-700) 700 MP / 3 Oil Press Oil Temp CHT Alternate Vacuum if installed Magneto System Grounding Check |
| 4 – OPS CHECK BC425 STANDBY ALTERNATOR IF INSTALLED (See Form FX1020BC for op's info) |
| 5 - CUTOFF RISE AT 1,000 RPM: |
| |

AIRCRAFT N <u>1e149</u>B

TACH <u>7059.</u>8

ENGINE:

| 1 – RECORD ON DISC. SHEET: Compression Test_/80 Lengine Oil Drained Replace & Safety Oil Filter Linstall plug & safety drain plug Service engine oil |
|--|
| 2a – MAGNETOS: Check for oil leaks, condition & security of Mags & "P" leads Timing check 22 degrees BTC Ignition harness for condition & security |
| 2b -MAGNETOS: 500 Inspection: (TCM & Slick) Inspect I/A/W Overhaul Manual |
| 3 – COWLING – Check for cracks, dents & other damage Security & condition of cowl fasteners Baffle Seals for condition & attachment Baffle Seals for condition & attac |
| 4 - ENGINE MOUNT - Engine shock mounts, Mount structure & ground strap |
| 5 - CRANKCASE, OIL SUMP AND ACCESSORY SECTION - Inspect for cracks and evidence of oil Leakage Check bolts and nuts for looseness Check breather tube lines for general condition Obstructions & security Front crank shaft seal for leaking |
| 6 - ENGINE CYLINDERS - Check for cracks on cylinder head & cooling fins Valve cover gaskets & push rod housings for leaks. |
| 7 - ENGINE BAFFLES, INTER-CYL BAFFLES & SEALS - Check for condition & security of attachment |
| 8 - ENGINE CONTROLS & LINKAGE - Check general condition & freedom of movement Check for travel, security & attachment for evidence of wear Check friction locks or vernier for operation |
| 9 - GOVERNOR - Condition, security, control operation, cable bracket wear, operation & oil leaks |
| 10 – FUEL INJECTION SYSTEM – Condition & security of mixture control unit, flow divider & fuel injector lines Fuel injector nozzles: Inspect & clean on condition (or) 300 hours of operation Check fuel control screen assy: (IO-520L) (Gasket P/N 649983) Pressurize system and leak check |
| 11 - ENGINE DRIVEN FUEL PUMP - Check for evidence of leakage, security, attachment & condition |
| 12 - INDUCTION SYSTEM -Inspect for evidence of leakage, security of clamps, tubes & ducting |
| 13 - B & C L60-2 ALTERNATOR BELT - Replace alternator belt P/N Gates 7300, every 200 hrs (Belt tension, 3/8' deflections with 12 lbs. pressure applied at center of belt). |
| 14 - ALTERNATORS AND REGULATORS: |
| A – Aft mounted alternators (B & C L60-2) (Ford 95 amp) Check belt for condition & tension Electrical connections, wires & support brackets Bottom attach bolt torque 41 ft. lbs. plus Check bolt for non-rotation at torque |
| B – (550P) Fwd Mounted Alternator: Check condition & attachments Oil leaks Wiring harness |
| C-(B & C B425) Standby Alternator: Check for security & condition Wiring Wiring |

| 15 - OIL COOLER - Check for condition, security & leaks held Oil temp probe for leaks and security |
|---|
| 16 - EXHAUST SYSTEM (COLLECTORS, MUFFLERS AND TAIL PIPES) Inspect for cracks and security Cabin heat shroud and ducting Cabin heat box & valve Inspect mufflers for cracks & burned out flame arresters Tail pipe supports |
| 17 - ELECTRICAL WIRING - (Inspect for condition & security) at firewall CHT Les Oil Temp EGT Probe |
| 18 – STANDBY VACUUM SYSTEM – Check for condition & security |
| 19 - FIREWALL - Check for condition, deformation wrinkles, & cracks |
| 20 - HOSES - Check for condition, chaffing, routing, leaking & security: Main fuel 6 Oil pressure 6 Manifold Pressure 6 Fuel flow Fuel Pump Vapor return to firewall fitting |
| 21 - STARTER - Check for security & condition full Solenoid & electrical connections |
| 22 - VACUUM/PRESSURE SYSTEM - Check for condition of hoses Condition, security & leakage at pump Vacuum pressure relief valve for security & condition Replace relief valve filter (Every 100 hrs) Air/oil separator and hoses for condition & security |
| 23 – SPARK PLUGS: Cleaned & Gapped Tested Install and Torque all plugs When all engine discrepancies are cleared, install all leads |
| I certify this ENGINE has been inspected in accordance with a 100 HR Inspection and was determined To be in airworthy condition. |
| Signature: |
| Signature: Date: 0/ /03/// |

FLIGHT EXPRESS, INC. B & C SPECIALTY PRODUCTS OPS CHECK B425 ALTERNATOR

AIRCRAFT NO: 61496 DATE: 01/03/11 TACH: 7059.8 ACTT: 12,014.7

- ASSURE THAT THE "STBY ALT" AND "STBY ALT SENSE" CIRCUIT BREAKERS AND "STBY ALT" MASTER SWITCH ARE IN THE "ON" POSITION.
- 2 REDUCE SYSTEM ELECTRICAL LOADS TO APPROX 10 15 AMPS
- 3 SET ENGINE TO 2000 RPM MINIMUM.
- 4 SWITCH PRIMARY ALTERNATOR FIELD SWITCH TO OFF.
- 5 CHECK THAT THE "STBY ALT ON" ANNUNCIATOR LIGHTS
- INCREASE THE ELECTRICAL LOAD TO OVER 20 AMPS. THE "STBY ALT ON" ANNUNCIATOR LIGHT SHOULD BE BLINKING. REDUCE THE ELECTRICAL LOAD TO LESS THAN 20 AMPS. THE "STBY ALT ON" ANNUNCIATOR SHOULD BE ON STEADILY.
- SWITCH THE PRIMARY ALTERNATOR FIELD SWITCH TO ON. THE "STBY ALT ON" ANNUNCIATOR SHOULD GO OFF.
- 8.— RETURN THE ENGINE TO IDLE RPM. A.S.

| AUTHORIZED | / | | |
|------------|---|-------------------|--|
| SIGNATURE: | | CITID Marries and | |
| _ | • | CERTIFICATE#: | |

FLIGHT EXPRESS INCORPORATED

CPS



| | _ | | |
|-----------|---|------------------------------|------------------------|
| Aircraft: | N: <u>6149B</u> Date: <u>01</u> | 03/1) Tach: 7059.8 | _TSMO: <u>/2/5.2</u> |
| ITEM | | CORRECTIVE ACTION | |
| 1 | Compression Results | #1-72 #2-74 | NAME |
| - | | 3-14 4-72 | CERT. |
| | | 5-76 6-74 /80 | DINSP. LA TIME OF |
| 2 | Drain Engine Oil | DRAWEN BULLE | NAME |
| | | 012 | CERY. |
| | | | INSP. RD TIME C/D: |
| 3 | Pemove & Inquest Eiles Ei | 1- | |
| - | Remove & Inspect Filter Element | FEMOUED & DUSTECTED | NAME CEAT. |
| | | NO DEFELTS NOTED | |
| | | | INSP. Jus TIME 6/09 |
| 4 | Safety Oil Drain Plug | Installed oil drain plug | NAME |
| | | with new gasket and | CERT. |
| <u></u> | | so kelled. | INSP. AS TIME 1/04/11 |
| 5 | Service Engine Oil | Serviced engine oil with | NAME |
| | | Nots of Aeroshell WG. | CERT. |
| | | Visit Procedure Confe | INSP. AS TIME 01/04/11 |
| | L. t. 11 0 C C t. N. C. | | |
| 6 | Install & Safety New Oil Filter | | NAME |
| | | element, torqued to 175/1/15 | 1777 |
| | | and saletico. | INSP. AND TIME 1/04/11 |
| 7 | Service Spark Plugs | Cleaned, inspectal, capped | NAME |
| | | | CERT. |
| <u> </u> | | Spark plugs. Torqued to 30th | INSP. ho TIME % |
| 8 I | nstall Spark Plug Leads | Installed sponk olucites (5) | NAME |
| | | The state of the production | CERT. |
| | | | INSP. Aug TIME 1/04/1/ |
| | | | TOTAL |
| X 1221 / | 7/02) | | TIME |

FLIGHT EXPRESS INCORPORATED

CPS



Engine Discrepancy Record

| | Page of | | |
|--|------------------------------|-------------|------------------|
| Aircraft: N: 61493 Date: | 01/03/11 Tach: 7059. 8 | _TSMO: | 1215.2 |
| ITEM DISCREPANCY | CORRECTIVE ACTION | | |
| 9 VACULEM RELIEF | - Replaced vacuum relieb | NAME | <u> </u> |
| FINTER DUE CHAPLE | E liter. | CERT. | |
| | | INSP. | AD TIME 1/04/11 |
| | | | 4 |
| 10 INDUCTION AND | RIZZZ Replaced Induction Air | NAME | |
| ADBC CLAMP BLOK | B) filter Adel clamp. | CERT. | |
| | | INSP. | Pers TIME /01/11 |
| | | | |
| 11 RETOKRYE FULL BA | 1 1/1 | NAME | |
| CASE THEN BOLTS OF | , (| CERT. | |
| TKS BRACKET | The Brocket | INSP. | AS TIME OI/AI/U |
| | | NAME | |
| /21 | | CERT. | |
| | | | TUAF |
| | | INSP. | TIME |
| 13 | | NAME | *** |
| | | CERT. | <u> </u> |
| ************************************** | | INSP. | TIME |
| | | | |
| 14 | | NAME | |
| | | CERT. | |
| | | INSP. | TIME |
| | | | |
| 15 | | NAME | |
| | | CERT. | |
| | | INSP. | TIME |
| | | T | |
| 16 | | NAME | |
| | | CERT. | |
| <u> </u> | | INSP. | TIME |
| | | | TOTAL TIME |
| FX 1221 (7/93) | | | |

FLIGHT EXPRESS, INC.

ENGINE LOG

CESSNA T210 SERIES

N6149B

CONTINENTAL IO-550-P(8)B

S/N: 821115 - R

INSTALLED: 03/18/09

TACH: 5844.6

AIRCRAFT TOTAL TIME: 10799.5

TIME IN SERVICE:

| /NOTES: | |
|--|----|
| DATE LAST O/H:01/24/09 TIME SINCE NEW:0000.0 | |
| Removee 3-8-11 Hobbs 7148 | 1. |
| Act 12,103.0 T50 1303.5 | |
| | |
| | |
| | |
| | |

FLIGHT EXPRESS, INC. ACCESSORY REMOVAL & INSTALLATION LOG

| AIRCRAFT NO: 61498 L/H R/H | |
|---|-------------|
| Serialized Numbered Parts, Maintenance Log and Parts Used Sheet | |
| This form will be used on any and all parts that have serial numbers, cand/or is subject to warranty. | ores |
| REMOVAL | |
| Date Removed 03 / 17 / 09 Tach Time Removed 5844. | 6 |
| Part Number <u>70550 P2</u> S/N <u>82/036 R</u> | |
| Description of Part <u>Engine</u> Reason for Removal <u>Timed out</u> | |
| | |
| /7: | |
| (Signature) Certificate # (Please make sure that the core is returned to the Parts Department with completed green | n to a |
| Please call the Parts Department in Orlando if not sure on some cores) | u tag. |
| (If the part was returned to stock as serviceable, then attach a copy of the yellow tag) | |
| , | |
| INSTALLATION | |
| | |
| Date Installed 63 / 17 / 08 Tach Time 5844.6 | |
| Part Number <u>7055P8B</u> S/N <u>821115</u> | |
| | 09 |
| Log Book Entry: Kemoved + Keplaced Enjoye As pe | 2 |
| MANUFACTURERS Spees of Flight Express INC C210 | PL |
| Engine Builday / And Installation Log Book Entry | |
| Kan, Adusted and healled checked bood - End | |
| Mechanic Installing Part (Please Print) Son Brown | |
| Mechanic Installing Part (Please Print) Son Brown (Signature) Certificate # | |
| Originaturo) Certificate # | • |
| INSPECTED BY: | |
| Please attach completed warranty card & 8130 or yellow tag, etc. if applicab | le) |
| Location: (√) ORL BNA CPS LUK TPA | |
| | |
| FX 1006A (05/03) | |