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P. 02

EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

No.

Issue

Date:

12/08/98

Rev. Original

Task Code: 852346

Title: Rosenbalin and Monarch Main Cargo Door Inspection

Reference:

N/A

Compliance \

Requested: As scheduled by Maintenance

Planning

Approved by:



Manoower: 3 hrs/aircraft

Priority: N/A

Equipment Affected: Aircraft equipped With Rosenbalm and Monarch Main Cargo Doors

GENERAL:

Weight Change: N/A This ECD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Note: Upon identification of a crack; contact Mx. Control immediately for further disposition.

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. Nakatapina ing Kabupat

A. P. M. Sadden Conference

EMERY WORLDWIDE AIR FLEET CAMPAIGN DIREC EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

					Page 2 of 2	
				1	No. 52-6	
ACCOMPLECIMAEN					Rev. <u>Original</u>	
Note: To gain access to some Rosenbalm cargo-door locks center actuator, cover may require ramoval. To gain access to Monarch dargo door locks, lock cover will require removal. Note: Clean frames as required to facilitate inspection. 1. Using a mirror and flashlight inspect main being door frames in the area of the upper four latch, sustain bolts for cracks and corrosion at each latch location. Note: Pay special attention to frame radiuses located at latch mount both numbrates. Note: Inspection mirror should be inserted into access area directly above latch. Cracks Noted (Circle) YES Location of cracks. Report all defects to Mx. Control and re-install lock covers by removed). Make a log book entry indicating compliance with this FCD and complete the section below. Act. MST Report and Pate: 10 Page: 700 CO						
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4											
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EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

No	5	2-	6

Issue

Date: 12/08/98

Rev. Original

Task Code: 852346

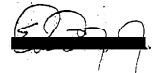
Title: Rosenbalm and Monarch Main Cargo Door Inspection Reference:

N/A

Compliance V

Requested: As scheduled by Maintenance Planning

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Menocwer: 3 hrs/aircraft

Equipment Affected: Aircraft equipped with Rosenbalm and Monarch Main Cargo Doors

Priority: N/A

Publications Affacted: None

Weight Change: N/A

GENERAL: This

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This ECD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Note:	Upon identification	AF a research Caracas as			
1	fireth and and and	of a crack; contact Mx.	Control	immediately	for
<u> </u>	further disposition.				- 1

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Fax:19372646085 Dec 19 '98 EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

Page 2 of 2 No. 52-6

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S (SEND WITH TRIP ENVELOPE)

* 1.4 14 			DWIDE AIRLINES E AUTHORIZATION
			Task Code852330
	Number <u>Al-5233</u>	-04:07 Priority	A Author Richard F. Morano
•	Title AD 93-20	0-02 Main Cargo Door ins	pection
7.00 7.00	<u>.</u>		
	Subject <u>Inspectio</u>	n of Cargo Door Wire Bur	ndle and Latch Rollers
	Equipment/Aircraft A	ffected N7964I N	797AL, N990CF, N993CF, N994CF, N995CF.
	Equipmental work at A	INT SOAL, IN	17 37 AL, N33001 , N33301 , N33401 , N33001 .
$\frac{1}{2}$ $\{x\}$		<u>N105WP</u>	N8#fAL, and N832AL
4.	Drawing #'s Attached	N/A	
	Manuals Affected	N/A	i jes
: .	Est. Man Hours/Elaps	sed Hours <u>½ hr. per ai</u>	rcraft
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•		WEIGHT AND BA	LANCE CHANGES
		Station	Arm Pounds
. •	Add	N/A	N/A N/A
	Flamove	N/A	N/A N/A
	Net Gain/Loss	N/A	N/A N/A
•	Special Notes: R	epetitive inspection requir	ed at Work Accomplished
		50 hour intervals, per FAA	
	of	approval, dated 2/4/92	Aircraft: 105WP
	"	ris M.A. cancels Al-5233-	Date: 12 - 15 - 98
		93-20-02, supersedes	Station: KLAX
	Al	D 92-02-05	Accomp. by: Krun G. Cucwalin
		700	
	Approved by		Date
	Approved by		Date 11-20-98
	FAA Acceptance	ALU	Date
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	MEO24 (Rev. 5 04/11/96)		Page 1 of 3

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MAINTENANCE AUTHORIZATION

	•				Page 2 of 3 No. <u>Al-5233-0</u> -	4:07
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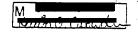
Page 3 of 3 No. <u>AI-5233-04:07</u>

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected could result in loss of the cargo door damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

- Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged, loose, or missing hardware mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.
- 2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.
- 3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.
- 4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 7/32-02









AIRC T MAINTEN	NANCE LOG	WORLDW.	Q.C.	 	39 'YBO
E FLT DATE	STATION GMT FROM TO OUT	BLOCK GA		7132-02 ACFT. NO. N 05 C	NP DCB
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					Task Code _	852330
Number	AI-5233-04:07	Priority	A	A	Author <u>Richard</u>	F. Morano
Title	AD 93-20-02 Mair	Cargo Door Ins	spection			
Subject	Inspection of Card	go Door Wire Bu	ndle and	Latch Ro	llers.	
Equipment	'Aircraft Affected	N796AL, I	N797AL. N	1990CF.	N993CF, N994	CF. N995CF.
		N105WP,	N811AL,	and N83	32AL	
Drawing #'s	: Attached	N/A				
Manuals A	ffected	N/A				
Est. Man H	ours/Elapsed Hours	s <u>½</u> hr. per a	aircraft			
	V	VEIGHT AND B	ALANCE	CHANGI	ES	
		Station	Arm		Pounds	
Add		N/A	N/A		N/A	J. J
Remove		N/A	N/A		N/A	المسائل
Net Gain/Lo	SS	N/A	N/A		N/A	
Special N	150 hour in of approva	inspection requintervals, per FA	A's letter	Aircraf	Work Accomp	
Reference		02, supersedes		Station	12-29-98 n: <u>KOAY</u> np. by: 464	
Approved by		Den D		Date	//-	20-98
Approved by	, <u>A</u>		_	Date	11-2	0-98
FAA Accepta		ALU				

Page 2 of 3 No. <u>Al-5233-04:07</u>

Kit List/Spares N/A Strip List N/A

Page 3 of 3 No. <u>Al-5233-04:</u>07

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second cocurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

 Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged, loose, or missing hardware mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M46493

2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M46493

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M46493

 Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 6200 - 04

M 46423

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	Table		Task Code <u>852330</u>						
Number <u>AI-5233-04:07</u>	Priority	A	Author Richard F. Morano						
Title <u>AD 93-20-02 M</u>	ain Cargo Door Ins	pection							
Subject <u>Inspection of Ca</u>	argo Door Wire Bur	ndle and	Latch Rollers						
Equipment/Aircraft Affected	N796AL, N	1797AL.	N990CF, N993CF, N994CF, N995CF,						
	N105WP.	N105WP, N811AL, and N832AL							
Drawing #'s Attached	N/A								
Manuals Affected	N/A								
Est. Man Hours/Elapsed Hou	ırs ½ hr. per ai	rcraft							
	WEIGHT AND BA	LANCE	CHANGES						
V.	Station	Arm	Pounds						
Add	N/A	N/A	N/A						
Remove	N/A	N/A	N/A						
Net Gain/Loss	N/A	N/A	N/A						
Special Notes: Repetitive	e inspection require intervals, per FAA'	ed at	Work Accomplished						
of approv	al, dated 2/4/92	ĺ	Aircraft: 797AL						
This M.A.	cancels Al-5233-0	08:00	Date: 12-11-98						
Reference: AD 93-20	-02, supersedes		Station: KDAY						
AD 92-02	-05	[Accomp. by: 09980						
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		No. <u>Al-5233-04:07</u>
Kit List/Spares		
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Strip List		
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Page 3 of 3 No. <u>Al-5233-04:07</u>

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

 Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged, loose, or missing hardware mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page. M. Brocke

2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

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3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 09982

4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 6895-10

M 09980

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MAINTENANCE AUTHORIZATION

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Number <u>Af-5233-04</u>	06 Priority A	Task Code <u>85233</u>	
Title <u>AD 93-20-6</u>	2 M∎in Cargo Door Inspection	Author Richard F. Morand	<u> </u>
Subject <u>Inspection o</u>	Cargo Door Wire Bundle and Latch	Rollers.	
Equipment/Aircraft Affec	ted. <u>N796AL N797AL N990C</u>	F, N993CF, N994CF, N9950	<u>)</u>
	N105WP and N811AL		
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Drawing #'s Attached	N/A		
Manuals Affected	N/A		· ·
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Reference: AD 93 AD 92	-20-02, supersedes Date: -02-05 Statio		
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Approved by	Date	6-11-97	
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FAA Acceptance	N A Date		
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Page 3 of 3 No. <u>AI-5233-04:06</u>

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

 Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged, loose, or missing hardware mounting components: Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 80216

2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 85916

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 80911

 Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA.

M 80916

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MAINTENANCE AUTHORIZATION

				Task Code	852330
Number	Al-5233	-04:06	Priority A	Author Richard F	. Merano
Title	AD 93-20	3-02 Main C	argo Door Inspection		
Subject	Inspectio	n of Cargo	Door Wire Bundle and Eatch	Rollers.	
Equipmen	t/Aircraft A	ffected	N796AL N797AL N9906	F, N993CF, N994C	F N995CF
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Drawing #	's Attached		N/A		
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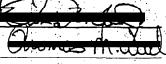
WEIGHT AND BALANCE CHANGES

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Add			N/A :	N/A ·	· N/A
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Net Gain/Loss	:	Ž.	N/A	;N/A	N/A

ſ	Special Notes:	Repetitive inspection required at	Work Accomplished
		150 hour intervals, per FAA's letter of approval; dated 2/4/92 Aircraft	: N 993CE
	Reference:	AD 93-20-02, supersedes Date:_	11/16/98
		AD 92-02-05 Station	
		Accom	p. by: <u>72/26</u>

Approved by

Approved by



Date ____

6-11-97

FAA Acceptance

NA

Date Date R 6.0

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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

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Page 3 of 3 No. Al-5233-04:06

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

1. Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged loose, or missing burdless mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 2260%

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M 22006

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 22006

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4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA.

M 22000

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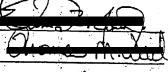
1				Task Code	852330
Number	Al-5233-04:06	Priority	<u>A</u>	Author Richard F	. Merano
Title	AD 93-20-02 Main	Cargo Door Ins	spection		
Subject	Inspection of Cargo	Door Wire Bu	ndie and Latch I	Rollers.	-
		•	12.		
Equipment	/Aircraft Affected	N796AL, N	1797AL N990C	F, N993CF, N994C	F. N995CF.
		N105WP a	ind N811AL		
Drawing #'s	s Attached	N/A			
Manuals A	Affected	N/A			
Est. Man H	ours/Elapsed Hours	½ hr. per a	ircraft		

WEIGHT AND BALANCE CHANGES

<u> </u>	•	•	Station	Arm	Pounds -
Add		2.5	N/A	N/A	N/A
Remove			N/A	N/A	N/A
Net Gain/Loss		, i	N/A	N/A	N/A

Special Notes:	Repetitive inspection required at	Work Accomplished
	150 hour intervals, per FAA's letter of approval, dated 2/4/92	Aircraft: N994CF
Reference:	AD 93-20-02, supersedes	Date: 10 - 20 - 98
	AD 92-02-05	Station: KM5P
		Accomp. by: 13 485

Approved by



Date 6-11-97

Date

6-11-97

FAA Acceptance

Approved by



Date

EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

	MAIN I ENANCE AUTHOR	IZATION
		Page 2 of 3 No. <u>AI-5233-04:06</u>
Kit List/Spares		No. A1-3233-04:06
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Page 3 of 3 No. Al-5233-04:06

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

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M 13685

2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 13685

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 13685

 Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA.

M 13685

AIHCI MAINTENANCE LOG	WORLDWIL	· · · · · · · · · · · · · · · · · · ·	5	852 >					
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			Task Code852330)
Number <u>Al-5233-04:0</u>	7 Priority	A.:	Author Richard F. Morano	
Title <u>AD 93-20-02</u>	Main Cargo Door I	nspection		
	Cargo Door Wire B		Dalla	
		didic and Late	rollers.	
Equipment/Aircraft Affecte	d <u>N796AL</u>	N797AL N990	: CF, N993CF, N994CF, N995C	— <u>;</u> F. '
		N811AL and		
Drawing #'s Attached			•	
Diswill # 2 Witsoued	N/A	··		
Manuals Affected	N/A			
Est. Man Hours/Elapsed Ho	ours ½ hr. per	aircraft		
	WEIGHT AND B	ALANCE CHAI	NGES	
	Station	Arm	Pounds	
Add	N/A	N/A	N/A	
Remove Net Gain/Loss	N/A	N/A	N/A	
-	N/A	N/A	N/A	
150 hou	ve inspection requi r intervals, per FAX val, dated 2/4/92	red at \s letter	Work Accomplished	7
This M.A	. cancels Al-5233-	08:00 I	raft: <u>995 CF</u> :: 12-19-93	
Reference: AD 93-20 AD 92-0	0-02, supersedes 2-05	Stati	on: KBOS	
		Acc.	omp. by:	
approved by		Date	11-20-98	
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AA Acceptance	ALM	Date		_
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MAINTENANCE AUTHORIZATION

Page 3 of 3 No. <u>AI-5233-04:07</u>

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the airplane. Investigation of this occurrence revealed that procedures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

1. Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires; and inspect for damaged, loose, or missing hardware mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 45037

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M 45037

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 45038

4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 1201-03

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ı				lask Code852330					
Number <u>Al-5233-04:07</u>	_ Priority	A		Author <u>Richard F. Morano</u>					
Title <u>AD 93-20-02 Ma</u>	in Cargo Door Ir	spection							
Subject Inspection of Car	Latch	Rollers.							
<u></u>									
Equipment/Aircraft Affected	N796AL.	N796AL, N797AL, N990CF, N993CF, N994CF, N995CF,							
	N105WP.	N811AL	. and N	N832AL					
Drawing #'s Attached	N/A								
Manuals Affected	N/A								
Est. Man Hours/Elapsed Hour	s ½ hr. per a	½ hr. per aircraft							
v	VEIGHT AND B	ALANCE	CHAN	GES					
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Net Gain/Loss	N/A	N/A		N/A					
150 hour in of approva	inspection requi ntervals, per FAA II, dated 2/4/92 cancels AI-5233	A's letter		Work Accomplished aft: N 8 1 (A L) : 1 2 - 2 2 - マ を					
Reference: AD 93-20-0	02, supersedes 05	Station: KNAY Accomp. by:84148							
Approved by			Date	11-20-98					
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FAA Acceptance	ALL		Date						

Page 2 of 3 No. <u>Al-5233-04:07</u>

Kit List/Spares

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Page 3 of 3 No. <u>Al-5233-04:07</u>

1. GENERAL INFORMATION

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M 84148

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M 84148

3. Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 94 11/8

4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 1911 – 19

M 84148

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				Task Code _	852330
Number _/	N-5233-04:07	Priority	<u>A</u>	Author Richard i	F. Morano
Title A	D 93-20-02 Mair	Cargo Door Ins	pection		
Subject <u>In</u>	spection of Carc	o Door Wire Bur	ndle and Latch	Rollers.	
Equipment/Air	craft Affected	N796AL, N	797AL. N990C	F, N993CF, N994C	F. N995CF
			N811AL and N	<u>~</u>	
Drawing #'s At	tached	N/A			
Manuals Affec	ted	N/A	1		
Est. Man Hours	s/Elapsed Hours	½ hr. per air	craft		
•	W	EIGHT AND BAL	ANCE CHANC	BES	
Add		Station	Arm	Pounds	
Remove		N/A	N/A	N/A	
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				N/A	
Special Notes	150 hour inte	spection required ervals, per FAA's dated 2/4/92 ncels Al-5233-08	letter	Work Accomplisher: 932 AL	ed .
Reference:	AD 93-20-02 AD 92-02-05	, supersedes	Station	KBOS	
Approved by	1		Date	112 35	<i>a c c</i>
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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

			Page 2 of 3
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Page 3 of 3 No. <u>Al-5233</u>-04:07

1. GENERAL INFORMATION

Since August 1991, there have been two occurrences of inadvertent inflight openings of the cargo door on Model DC-8-63 series airplanes which had been modified in accordance with Supplemental Type Certificate (STC) SA1802SO. The second occurrence resulted in significant structural damage to the concecures for use of the cargo door warning light system were not included in the Airplane Flight Manual Supplement. In addition, the cargo door wire bundle, which powers the cargo door operating and indicating system could result in a false indication that the cargo door is properly closed and locked. These conditions, if not corrected, could result in loss of the cargo door, damage to the flight control surfaces, and reduced controllability of the airplane.

2. INSPECTION REQUIREMENTS:

1. Inspect the cargo door wire bundle between the exit point of the cargo liner and the attachment point on the cargo door to detect crimped, frayed, or chaffed wires, and inspect for damaged, loose, or missing hardware mounting components. Prior to further flight, repair any damaged wiring or hardware mounting components. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 45033

2. Inspect the cargo door latch rollers (spools) in the lower sill of the cargo door opening of the airplane to ensure that all twelve rollers (spools) can be freely rotated by hand. Prior to further flight, replace any discrepant roller (spool) components found, and repair any rollers (spools) that cannot be rotated freely by hand. Record and correct discrepancies on EWA Non-Routine Maintenance Form MEO9 or log page.

M 45038

3 Ensure that Circuit Breaker labels for "Pump & Valve" are legible and intact.

M 45037

4. Complete the Work Accomplishment section on page 1 of this MA and make a log book entry indicating compliance with this MA. Enter L.P. No. 1177 - 04

M 45038

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ATTACHMENT

NO.4



December 21, 1998

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

EWA has received the Telegraphic Airworthiness Directive T98-26-51, dated December 18, 1998, effective upon EWA's receipt today December 21, 1998.

Following is the EWA aircraft listing effected.

	SERIES/		AIRCRAFT	CARGO	FCD
NO.	MODEL NO.	REG NO.	SERIAL NO.	DOOR STC NUMBER	52-6
1.	DC-8-62	N990CF	46068	SA 1802SO	C/W
2.	DC-8-62	N993CF	46028	SA 1802SO	C/W
3.	DC-8-62	N994CF	45956	SA 1802SO	C/W
4.	DC-8-62	N995CF	46024	SA 1802SO	
5.	DC-8-63	N796AL	46054	SA 1802SO	C/W
6.	DC-8-63	N797AL	46163	SA 1802SO	C/W
7.	DC-8-73	N105WP	46095	SA 1802SO	C/W
8.	DC-8-71	N811AL	46099	SA 1802SO	C/W
9.	DC-8-73	N832AL	46063	SA 1802SO	C/W

NOTE: FQU 52-6 previously complied with on or before December $24,\,1998.$

Mr. Joseph Abramski Page 2 December 21, 1998

EWA's response to this TAD is as follows:

- A. I have notified Operations and they will revise the effected aircraft AFM within the next 24 hours, with a copy of this TAD. Operations will provide AFM sign-off sheet to which will be made record of the AD compliance file.
- B. Quality Control is replacing the previous FCD by issuing a MA A1-5234-04:00 that incorporates all areas as addressed by the TAD inspection criteria.
- C. Starting today, December 21, 1998 EWA will perform the reference MA on the effected EWA's aircraft.
- D. We have currently reviewed the past 30 day history of pilot reports of cabin pressurization anomalies or illuminations of the Main Deck Cargo Door Warning Light, and have only one aircraft N105WP with the prior pilot reports. This aircraft will have the referenced MA performed today prior to flight.
- E. Contacted Bill Cotney, DER, and he will work in concert with the ATL ACO on any repairs in support of any structural findings.
- F. EWA will report inspection findings to ATL ACO and yourself, to include the FCD 52-6 inspections.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

cc: Rene Visscher

Number Al-5234-04:00 Priority A Author Richard F. Morano Title Resembalm Main Cargo Door Inspection Subject Inspects the main cargo door frames and inner skin in the area of the cargo door latch mount belts at each location. Equipment/Aircraft Affected Aircraft equipped with Resembalm Main Cargo Doors Drawing #'s Attached Manuels Affected Est. Man Hours/Elapsed Hours 3 hrs/aircraft WEIGHT AND BALANCE CHANGES Station Arm Pounds Add N/A N/A N/A N/A Remove N/A N/A N/A N/A Net Gain/Loss N/A N/A N/A N/A Net Gain/Loss N/A N/A N/A N/A Special Notes: This MA cancels and supersedes FCD 52-6. This MA to be complied with no later than 12/25/98. Reference: AD T98-26-51 Date: Station: Accomp. by: Date 12/21/98 Acceptance/Approval N/A Date N/A					Task Code <u>852347</u>
Title Rosenbalm Main Cargo Door Inspection Subject Inspects the main cargo door frames and inner skin in the area of the cargo door Jatch mount bolts at each location. Equipment/Aircraft Affected Aircraft equipped with Rosenbalm Main Cargo Doors Drawing #'s Attached Manuals Affected Est. Man Hours/Elapsed Hours 3 hrs/aircraft WEIGHT AND BALANCE CHANGES Station Arm Pounds Add N/A	Number	AI-5234-04:00	Priority A		Author Richard F. Morano
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Page 2 of 3 No. Al-5234-04:00

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Page 3 of 3 No. <u>Al-5234-04:00</u>

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This MA visually inspects the main cargo door	frames and inner skin in the area of the cargo
door latch mount bolts at each latch location.	cargo

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MEMORANDUM

TO: Thomas Wood, Director, Quality Control

FROM: Bob Peck, Manager of Reliability

SUBJECT: DC-8 Main Deck Cargo Door - Telegraphic AD

DATE: 21 December 1998

Dear Sir,

As requested per the Telegraphic AD, a 30 day review of log pages for the nine DC-8 aircraft with Rosenbalm doors was complied with. The scope of the review was to identify write-ups directed to the door warning system, and pressurization problems resulting in corrective action to the main cargo door. The findings of this review are as follows:

- 1. Aircraft N990CF No findings
- 2. Aircraft N993CF No findings
- 3. Aircraft N994CF No findings
- 4. Aircraft N995CF No findings
- 5. Aircraft N796AL No findings
- 6. Aircraft N797AL No findings
- 7. Aircraft N105WP 2 log page write-ups on 26 Nov 1998 Log pages 6913-22 and 6913-24 and log page 6913-25 where structural damage was detected. (See attached package)
- 8. Aircraft N811AL No findings
- 9. Aircraft N832AL No findings

Also along with this memo is a copy of when MA AI-5233-04:07 (Main Cargo Door Inspection) was last complied with on these aircraft.

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2.	DC-8-62	N993CF	1 /22/16	
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5 .	DC-8-63	N796AL	1:05 AM 12/22/98	ALC A
> 6.	DC-8-63	N797AL	1 / 1 / 1	
→ 7.	DC-8-73	N105WP	11.00	
<u>> 8.</u>	DC-8-71			
9.	DC-8-73	N832AL	2:35 AM 12/22/96	

A copy of this AD is to be placed in the AFM of each aircraft listed above directly behind the STC Certificates cover page, in front of the STCs.

Oriector Quality Control
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7. DC-8-73	N797AL		
8. DC-8-71	N105WP		
9. DC-8-73	N811AL		
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A copy of this AD is to be placed in the AFM of each aircraft listed above directly behind the STC Certificates cover page, in front of the STCs.

ATTN: DAVE KIRK

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Driector Quality Control 1-4-99

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1.	DC-8-62	N990CF		
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7.	DC-8-73	N105WP		
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9.	DC-8-73	N832AL		

A copy of this AD is to be placed in the AFM of each aircraft listed above directly behind the STC Certificates cover page, in front of the STCs.

Director Control

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6.	DC-8-63	N798AL		
17	DC-8-73	N797AL		
- - - - -	4	N105WP		
	DC-8-71	N811AL		
9.	DC-8-73		2/21/98 5:00 pm	

A copy of this AD is to be placed in the AFM of each aircraft listed above directly behind the STC Certificates cover page. In front of the STCs.

Director Quality Control



December 28, 1998

Mr. Rany Azzi,
Aerospace Engineer,
Airframe and Propulsion Branch, ACE-117A
FAA, Small Airplane Directorate,
Atlanta Aircraft Certification Office
One Crown Center
1895 Phoenix Boulevard, Suite 450
Atlanta, Georgia 30337-2748

Dear Mr. Azzi:

This letter is in response to the Telegraphic Airworthiness Directive T98-26-51 dated December 18, 1998, in response to item (F).

The inspection was performed on nine (9) aircraft per the TAD item (B) per the requirements of the TAD, and no defects were noted.

Please call if I can be of further assistance.

Sincerely,

Thomas M. Wood

Director Quality Control

TMW/re

MERY WORLDWIDE AIRLINES

MAINTENANCE AUTHORIZATION Task Code 852347 Author Richard F. Morano AI-5234-04:00 Priority Number: Rosenbalm Mata Cargo Door Inspection TALE inspects the men cargo door frames and inner skin in the area of the cargo door Subject latch mouse bolls at each location. Equipment/Alteraft Affected Drawing #'s Attached Manuais Affected 3 hrs/sircraft Est. Man Hours/Elapsed Hours WEIGHT AND BALANCE CHANGES Pounds Sæton Acm NA N/A 21 NA Add NA NA N/A :: Remove N/A Net Gain/Loss Work Accomplished This MA concels and supersedes Special Notes: FCD 52-6. This MA to be complied with no later than 12/25/98: AD T98-26-51 Reference: Date Approved by Approved by Date FAA Acceptance/Approvat

MEO24 (Rev. 5 04/11/96)

Page 1 of 3

MERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

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		Page 2 of 3 No. <u>AL-5234-04:00</u>
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GENERAL: This MA visually inspects the main cargo door frames and inner skin in the area of the cardoor fatch mount bolts at each larch location. MOTE: Upon identification of a crack, contect_Mx. Control immediately for furth disposition. ACCOMPLISHMENT INSTRUCTIONS: Mote: To gain access to some Resembelin cargo door locks, center actuator cover stay require removal. 1. Clean and degreeds the inside structure where the latch fitting bolts attach to the feature are desired visual inspection to detect cracking or any discrepancy of the main deck cargo door in the immediate area of the feature or any discrepancy of the skin in the immediate area of the fastener heads, and locate or missing fastoners. Note: Psy appeals intention to frame radiuses and inner skin located or intention bolt numbers. Note: Inspection minor should be inserted into access area directly above latch. Cracks Noted (Circle) YES NO Location of cracks. 3. Report all defects to Mx. Control and re-install took covers (if removed). 4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished section on page 1 of this MA. Log Page 8: 7015-02	:	
GENERAL: This MA visually inspects the main cargo door frames and inner skin in the area of the callogor fatch mount bolts at each larch location. MOTE: Upon Identification of a crack, contact Mx. Control immediately for furth disposition. ACCOMPLISHMENT INSTRUCTIONS: Mote: To gain access to some Resembelin cargo door locks, center actuator cover stay require removal. 1. Clean and degreeds the inside structure where the latch fitting points attach to the features. 2. Perform an internal detailed visual inspection to detect cracking or any discrepancy of the main deck cargo door in the immediate area of the feature or any discrepancy of the skin in the immediate area of the fastener heads, and locate or missing fasteners. Note: Pay special retention to frame radiuses and inner skin located at latch mount bolt nutplates: Note: Inspection minor should be inserted into access area directly above latch. Cracks Noted (Circle) YES NO Location of cracks: 3. Report all defects to Mx. Control and re-install took covers (if removed). 4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished section on page 1 of this MA. Log Page 8: 705-02		Pere 3 of 3
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MECCA (Rev 5 04/11/95)		Report all defects to Mx. Control and re-install lock covers (if removed). Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this MA.
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	4	Report all defects to Mx. Control and re-install lock covers (If removed). Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this MA. Log Page 3: 17015-02

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EMERY WORLDWIDE AIRL MAINTENANCE AUTHORIZATION

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			MAINTENANCE AUTHORIZATION		
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				Page 2 of 3 No. Al-5234-0	4:00
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MAINTENANCE AUTHORIZATION

Page 3 of 3 No. Al-5234-04:00

GENERAL:

This	IAA visually	inspects the	main ca	ngo do	or fram	ee end	inner skin in	the area of the	e cargo
door	latch moun	t bolts at eact	t latch loc	etton.				•	:
1									

NOTE: Upon identification of a creck, contact Mix. Control immediately for further disposition.

ACCOMPLISHMENT INSTRUCTIONS:

Note: To gain access to some Resemberin cargo door locks, center actuator cover may require removal.

1. Clean and degreess the inside structure where the latch fitting bolts attach to the flames:

M 21693

2 Perform an internal detailed visual inspection to detact cracking or any discrepancy of the main deck cargo door in the immediate area of the boits attaching the latch fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and loose or missing fasteners.

152899

Note: Pay special attention to frame radiuses and inner skin located at latch mount bolt nutplates.

Note: Inspection mirror should be inserted into access area directly above latch:

Cracks Noted (Circle) YES NO Location of cracks:

3. Report all defects to Mx. Control and re-install lock covers #f removed).

M 81693

4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this MA.

M 81693

Log Page #: 7/57

				٦	rask Code <u>852347</u>	
Number	AI-5234-04:00	Priority	A	Auth	or Richard F. Morand)
THIS	Rosenbalm Main Curs	o Doer Irs	pection			
1140	Inspects the main cargo door frames and inner skin in the area of the cargo door					
Subject				5L @ (01) 111 m	TO BIES OF MICE STATES	
	latch mount boits at 6					
Equipmen	t/Aircraft Affected	Air rail	ecuipped with	n Rosenbe	im Main Cargo Doors	
Drawing #	#s Attached					
Manuals	Affected					
Est Man Hours/Elapsed Hours 3 hrs/sircraft						
WEIGHT AND BALANCE CHANGES						
	19 11 Classic				Pounds	~
			ration N/A.	Armi N/A	N/A	1 25
	Add	i	N/A	N/A	NA	
	Remove		N/A	N/A	N/A	~
	Net Gain/Loss		1.071			
Specta	Notes: This MA. can	cels and si	.persedes		Vork Accomplished	
	FCD 52-8. T			Air craft	N 994	
				Date:	28 Dec 98	
Refere	nce: AD 198-26-5	1	;	l	KONT	
				Accomp	by: 59069	
Approved	l by	(مرہر استان استار انتخاب		Date	12/21/98	
Approved	by Otoma M	ليدلودي	 	Date	12/21/98	
FAA Acce	eptance/Approval	N'A		Date	N/A	

12/21 198 17:53 NO.641 02/03

EWA QUALITY CONTROL

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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

	: 	Page 2 of 3 No. <u>AJ-5234-Q4:00</u>
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12/21 198 17:53 NO.641 03/03

EMERY WORLDWIDE AIRLINES NIAINTENANCE AUTHORIZATION

Pege 3 of 3

This MA	visually	Inspects	the	main	ರ್ಜಾಡಂ	door	frames	and	inner	skin	ĺΠ	the	ales	of	the	cståo
door lat	ch mount	balts at e	ead!	i laích	location	on.										

		ŗ	No. AI-5234-C4:00
SENEF			
This M. loor la	tch mount	Inspects the main cargo door frames and Inner skin in bolts at each latch location.	
VOTE:	Upen dispes	identification of a creat, contact Mx. Control In	nmediately for further
ACCO		ent instructions:	
Vote:	To gai	n access to some Rosenbalm cargo door locks, cente or cover may require removal.	er
٢.	Clean and	of degrease the inside structure where the latch fitting in to the frames.	M 59069
2.	any discrease of the	n internal dutailed visual inspection to detect cracking apparity of the main deck cargo door in the immediate bolts attaching the latch fittings. Inspect for cracking screpancy of the sidn in the immediate area of the eads, and loose or missing fasteners.	ng
	Note:	Pay special attention to frame radiuses and inn skin located at latch mount bolt nutplates.	er
	Note:	Inspection mirror should be inserted into access and directly above laters.	ea
		oted (Circle) YES (NO)	
	Location	of cracks:	
3.	Report al	I defects to Mx. Control and re-install lock covers	(if M 59069
4.	Make a locomplete	og book entry indicating compliance with this MA at the "Work Accomplished" section on page 1 of t	nd M 59069
	Log Page	#: 7004-02	

						•	Task Code <u>852347</u>
Number	AI-5234-	04:00	Priori	tyA_		Auti	nor Richard F. Morano
Title	Rosenba	im Main Car	go Door	inspection			
Subject	Inspects	the main can	cie doot	frames and	inne	r skin in t	he area of the cargo door
•		unt bolts at e					
Equipmen					with	Rosenba	ılm Main Cargo Doors
Drawing #	ra Attache	a					
Manuals /	Affected						
Est. Man i	Hours/Elaj	osed Hours	3 hrs	/aircraft			
		WE	ight a	ND BALAN	CE C	HANGES	}
				Station		Arm	Pounds
	Add			N/A		N/A	N/A
	Remo	V9		N/A		N/A	N/A N/A
	Net G	ain/Loss	. <u> </u>	N/A		N/A	INA
Special	Notes:	This MA cen FCD 52-8. With no leter AD T98-26-1	This MA than 12	to be compl	s lied	Aircraft: Date: Station:	Vork Accomplished 995 CF 12-23-98 KBOJ by DV 1
Approved Approved FAA Acce	by 🔂	pproval	N/A			Date Date Date	12/21/98 12/21/98 N/A

Page 2 of 3 No. <u>Al-5234-04:00</u>

Kit List/Spares N/A Strip List

Page 3 of 3 No. <u>AI-5234-04:00</u>

GENERAL:

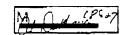
This MA visually inspects the main cargo door frames and inner skin in the area of the cargo door latch mount boits at each latch location.

NOTE: Upon identification of a crack, contact Mx. Control Immediately for further disposition.

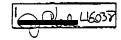
ACCOMPLISHMENT INSTRUCTIONS:

Note:	To gain	access to	\$0in:8	Rosenbalm	cargo	door loo	ks, center
				re removal.	•		

 Clean and degrease the inside structure where the latch fitting bolts attach to the frames



2. Perform an internal detailed visual inspection to detect cracking or any discrepancy of the main deck cargo door in the immediate area of the bolts attaching the latch fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and loose or missing fasteners.



Note: Pay special extention to frame radiuses and inner skin located at latch mount bolt nutplates.

Note: Inspection mirror should be inserted into access area directly above latch.

Cracks Noted (Circle)
Location of cracks:____

ramoved).

YES

NO

Report all defects to Mx. Control and re-install lock covers (if

M 1860-7

4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this MA.

M 186+7

Log Page #: 7001-11

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EMERY WORLDWIDE AIRLINES

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	* : :					Task Code_	
	Number /	AI-5234-04:00	Priority	<u> </u>	A	uthor Richard	F. Merano
	Tile	Rosenbalm Main (Cergo Door In	pection			- 5 % - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1
	Subject	nepects the main	cargo door fra	mas and a	Dec ekin i	n the area of th	e camo do
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BMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

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Page 3 of 3 No. <u>Al-5234-04:00</u>

GENERAL:

This MA visually inspects the main dergo door frames and inner skin in the eres of the cargo deor latch mount bolts at each latch location.

NOTE: Upon identification of a crack, contact Mx. Control immediately for further disposition

ACCOMPLISHMENT INSTRUCTIONS:

Note: To gain access to some Resembalin cargo door locks, center actuator cover may require removal.

1. Clean and degrees the inside structure where the latch fitting bolts attach to the flames

M 337 11

Perform an internal detailed visual inspection to detect cracking or any discrepancy of the main deck cargo door in the immediate area of the boils attaching the letter fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and loose or missing fasteners.

337 11

Note: Pay special attention to frame radiuses and inner skin located at latch mount bolt nutplates.

Note: Inspection mirror should be inserted into access area directly above latch.

Cracks Noted (Circle) YES NO

Location of cracks:

3. Report all defects to Mx. Control and re-install lock covers (if removed).

M 337 11

4. Make a log book entry indicating compliance with this MA and complete the "Werk Accomplished" section on page 1 of this MA.

M 337 11

Log Page #: 6200 0 0

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EMERY WORLDWIDE AIRLINES NAINTENANCE AUTHORIZATION

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Number	AI-523	4-04:00	Priority		Author Richard	d F. Morano
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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

Page 3 of 3 No. Al-5234-04:00

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FOTE: Upon identification of a crack, contact Mx. Control immediately for further disposition.

ACCOMPLISHMENT INSTRUCTIONS:

To gain access to some Rosenbalm cargo door locks, center actuator caver may require removal.

 Clean and degrees the inside structure where the latch fitting boits attach to the frames.

M/6365

2. Perform an internal detailed visual inspection to detect cracking or any discrepancy of the main deck cargo door in the immediate area of the botts attaching the latch fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and locee or missing fasteners.

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Note: Pay special attention to frame rediuses and inner ekin located at latch mount bolt numblates.

Note: Inspection mirror should be inserted into access area directly above latch.

Cracks Noted (Circle)

VEC

NO

Location of cracks;_

3. Report all defects to Mx. Control and re-install lock covers (if removed).

M 16365

4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this MA.

M/6365

Log Page #: 6861-24

MAINTENANCE AUTHORIZATION Task Code 852347 Author Richard F. Morano Priority AI-5234-04:00 Number Rosenbaim Main Cargo Door Inspection Title Inspects the main cargo door frames and inner skin in the area of the cargo door \$ubject latch mount bolts at each location. Aircraft equipped with Rosenbalm Main Cargo Doors Equipment/Aircraft Affected Drawing #'s Attached Manuals Affected 3 hrs/aircraft Est. Man Hours/Elapsed Hours WEIGHT AND BALANCE CHANGES Pounds Arm Station NA NA N/A Add N/A NA N/A Remove N/A N/A N/A Net Galn/Loss Work Accomplished This MA cancels and supersedes Special Notes: FCD 52-8. This MA to be complied Alreraft: 105 WP with no later than 12/25/98. AD T98-26-51 Reference: Station: Accomp. by

 Approved by
 Date
 12/21/98

 Approved by
 Date
 12/21/98

 FAA Acceptance/Approval
 N/A
 Date
 N/A

MEO24 (Rev. 5 04/11/96)

Page 1 of 3

Kit List/Spares



EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

Page 2 of 3 No. Al-5234-04:00

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EMERY WORLDWIDE AIRLINES MAIN ENANCE AUTHORIZATION



Page 3 of 3 No. Al-5234-04:00

GENERAL:

This MA visually inspects the main cargo door frames and inner skin in the area of the cargo

door latch mount bolts at each latch location. Upon identification of a crack, contact Mx. Control Immediately for further NOTE: disposition. ACCOMPLISHMENT INSTRUCTIONS: To gain access to some Rosenbalm cargo door locks, center Note: actuator cover may require removal. Clean and degrease the inside structure where the latch fitting 1. bolts attach to the frames Perform an Internal detailed visual inspection to detect cracking or 2. any discrepancy of the main deck cargo door in the immediate area of the bolts attaching the latch fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and loose or missing fasteners. Pay special ettention to frame radiuses and inner Note: skin located at latch mount bolt nutplates. Inspection mirror should be inserted into access area Note: directly above latch. YES Cracks Noted (Circle) Location of cracks:___ Report all defects to Mx. Control and re-install lock covers (if 3. removed). Make a log book entry indicating compliance with this MA and 4. complete the "Work Accomplished" section on page 1 of this MA.

Log Page #: 7132-18

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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

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EMERY WORLDWIDE AIRLINES MAINTENANCE AUTHORIZATION

Page 3 of 3 No. AL5234-04:00

GENERAL:

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This MA visually inspects the main cargo door frames and laner skin in the area of the cargo

door latch mount bolts at each latch location. The same of the sa Continue of the Continue of the manual of clapestacn. ACCOMPLISHMENT INSTRUCTIONS: 医自由性病 To gain access to some Rosenbalm cargo door locks, center actuator cover may require removal. Clean and degreese the inside structure where the latch fitting 66033 boits attach to the transes. Perform an Internal detailed visual inspection to detect cracking or 2. 6635 any discrepancy of the main deck cargo door in the immediate area of the botts attaching the latch fittings. Inspect for cracking or any discrepancy of the skin in the immediate area of the fastener heads, and loose or missing fasteners. Note: Pay special attention to frame radiuses and inner skin loopted at latch mount bolt nutplates. (1871年) 企业 美的名词 THE ENTE MOTE: inspection mirror should be inserted into access area directly above latch. NO . Cracks Noted (Circle) YES Location of cracks: 3. Report all defects to Mx. Control and re-install lock covers (if M6635 ramoved).

4. Make a log book entry indicating compliance with this MA and complete the "Work Accomplished" section on page 1 of this

0635

Log Page #:

100



March 2,1999

Mr. Rany Azzi,
Aerospace Engineer,
Airframe and Propulsion Branch, ACE-117A
FAA, Small Airplane Directorate,
Atlanta Aircraft Certification Office
One Crown Center
1895 Phoenix Boulevard, Suite 450
Atlanta, Georgia 30337-2748

Dear Mr. Azzi:

This letter is to request written authorization from you to remove the Aircraft Flight Manual Supplement which was installed per Airworthiness Directive T-98-26-51.

This request is to support, by FAA letterhead and signature, your approval provided to me by Email February 8, 1999.

I requested this FAA approval through my FAA Principal Maintenance Inspector SJC FSDO as Airborne did, through the Detroit FSDO, but was informed that I should contact you (see attachment).

Thanking you in advance for your support.

Sincerely,

Thomas M. Wood
Director Quality Control

TMW/csh

cc: Rene' Visscher Edward Jones Abraham Michael

Attachment



U.S. Department of Transportation Federal Aviation Administration San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

February 26, 1999

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

This letter will respond to your request for written authorization from this office to remove the Aircraft Flight Manual Supplement which was installed per Airworthiness Directive T98-26-51, dated December 18, 1998.

It is our opinion that your request should be directed to the Atlanta Aircraft Certification Office for relief of this issue; the address of which, and FAA representative point of contact is stated on page three (3) of the subject airworthiness directive.

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector



February 8, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is to request written authorization from you to remove the Aircraft Flight Manual Supplement which was installed per Airworthiness Directive T98-26-51.

I have attached the e-mail Rany Azzi sent me today, and a copy of the letter provided to Airborne from their FAA DC-8 Partial Program Manager.

Thank you for your support in this matter.

Sincerely,

Thomas M. Wood

Director Quality Control

.TMW/re

Attachments

cc: Rene' Visscher

Wood, Thomas M

From:

Rany Azzi

Sent:

Monday, February 08, 1999 9:12 AM

To:

wood.thomas

Subject:

Re: AD T98 26 51 FLIGHT MANUAL SUPPLEMENT REMOVAL

Tom:

Enclosed is the message I sent to ABX PPM.

Rany

Forward Header

Subject: Re: AD T98 26 51 FLIGHT MANUAL SUPPLEMENT REMOVAL

Author. Rany Azzi at ASOATLACO

Date: 2/2/1999 2:13 PM

Arthur.

Following compliance with Telegraphic AD T98-26-51, it would be acceptable for the Operators to remove the restriction imposed on the AFM Limitation Section by paragraph (A) of subject T-AD.

Rany Azzi -- ACE-117A 770-703-6080

Reply Separator

Subject: AD T98 26 51 FLIGHT MANUAL SUPPLEMENT REMOVAL

Author: Arthur A III Warner at AGL200DTW

Date: 1/22/1999 3:40 PM

MR. BOB ZITNEY, MANAGER OF QUALITY ASSURANCE, AT AIRBORNE EXPRESS CONTACTED ME TODAY AND REQUESTED WRITTEN AUTHORIZATION TO REMOVE THE FLIGHT MANUAL SUPPLEMENT REQUIRED BY AD 98 26 51. AIRBORNE EXPRESS HAS COMPLIED WITH ALL REQUIREMENTS OF THE AD ON ALL SIX OF THE AFFECTED AIRCRAFT WITH NO UNFAVORABLE FINDINGS. THE FLIGHT MANUAL SUPPLEMENT PLACED IN THE FLIGHT MANUAL IS IN DIRECT CONFLICT WITH PROCEDURES FOR CABIN DECOMPRESSION AD 98 26 51 DOES NOT ALTHORIZE THE REMOVAL OF THE CABIN DECOMPRESSION. AD 98 26 51 DOES NOT AUTHORIZE THE REMOVAL OF THE SUPPLEMENT AFTER THE CARGO DOOR INSPECTIONS AND ANY REPAIRS ARE COMPLETED.

I UNDERSTAND YOU HAD A CONVERSATION WITH MR. ZITNEY AND SAID IT WAS OKAY TO REMOVE THE NEW SUPPLEMENT FROM THE AFM AFTER THE OTHER PORTIONS OF AD 98 26 51 WERE COMPLETED.

PLEASE RESPOND WITH YOUR ON THE REMOVAL OF THE NEW SUPPLEMENT SO I MAY PROVIDE AIRBORNE EXPRESS THE REQUESTED LETTER.

ARTHUR A. WARNER ABXA DC-8 PPM **DETROIT FSDO GL23**



Delivering Satisfaction

Worldwide

Transmitting _3 pages including this Cover Page

Date:	Feb. 8 1999	Time:
	Please deliver the following page(s) to:	
Company:	Emery Worldwide	-
Name: Dept:	Tom Wood	
	Fax Number: 937-898-2803	

This document is being sent from:

Bob Zitney Quality Assurance

Direct Fax Number: 1-937-383-4336

If you do not receive all of these pages, please call: (937) 382-5591, ext 2215

For your reference regarding AD T98-26-51.

Author: Rany Azzi at ASOATLACO

Date: 2/2/99 2:13 PM

Priority: Normal

TO: Arthur A III Warner at AGL200DTW

CC: Michael E O'Neil at ANM100L, Greg Dilibero at ANM100L, Rob Jackson at ANM100L,

Edward Garino, Randy Avera, Albert Lam at ANM100L

Subject: Re: AD T98 26 51 FLIGHT MANUAL SUPPLEMENT REMOVAL

Arthur:

Following compliance with Telegraphic AD T98-26-51, it would be acceptable for the Operators to remove the restriction imposed on the AFM Limitation Section by paragraph (A), of subject T-AD.

Rany Azzi --, ACE-117A 770-703-6080

Reply Separator

Subject: AD T98 26 51 FLIGHT MANUAL SUPPLEMENT REMOVAL

Author: Arthur A III Warner at AGL200DTW

Date: . 1/22/1999 3:40 PM

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I UNDERSTAND YOU HAD A CONVERSATION WITH MR. ZITNEY AND SAID IT WAS OKAY TO REMOVE THE NEW SUPPLEMENT FROM THE AFM AFTER THE OTHER PORTIONS OF AD 98 26 51 WERE COMPLETED.

PLEASE RESPOND WITH YOUR APPROVAL OF THE REMOVAL OF THE NEW SUPPLEMENT SO I MAY PROVIDE AIRBORNS EXPRESS THE REQUESTED LETTER.

ARTHUR A. WARNER MEG 8-DG AXEA DETROIT FSDO GL23



U.S. Department of Transportation Federal Aviation Administration DETROIT FLIGHT STANDARDS DISTRICT OFFICE Willow Run Airport - East Side 8800 Beck Rd. Belleville, Michigan 45111

February 3, 1999

Mr. Robert Zitney
Manager, Quality Assurance
and Chief Inspector
ABX Air, Inc.
Airborne Air Park
145 Hunter Drive
Wilmington, Ohio 45177

Dear Mr. Zitney:

This letter will serve as authorization to remove the Aircraft Flight Manual Supplement which was installed per Airworthiness Directive T98-26-51.

Following our telephone conversation of January 22, 1999, I contacted Mr. Rany Azzi, of the Allenca Aircraft Certification Office, and discussed the conflicting information in the Aircraft Flight Manual. This conversation was followed up with a CC Mail message to Mr. Azzi.

Attached is a copy of that message and Mr. Azzi's response stating it would be acceptable for operators to remove the restriction imposed on the AFM Limitation Section by Paragraph (A) of AD T98-26-51, after the requirements of Paragraph (B) and (C) have been accomplished and any defects repaired.

I would suggest that a copy of this letter and the attachment be placed in the aircraft records of the six DC-8 aircraft affected by the AD. This will give Airborne Express a backup should the removal of the AFM Supplement ever come into question.

Sincerely,

Arthur A. Warner III

DC-8 Partial Program Manager

Enclosure

EWA'S RESPONSE TO LETTER DATED [] Sucember 22, 1998

EMERY WORLDWIDE AIRLINES TIME LIMITS MANUAL DC 10 -10/-30

II. DC-10 CHECK INTERVALS

Service Check

The Service Check shall be accomplished prior to aircraft departure if within the preceding twenty-four (24) hours a Service Check or a higher check has not been accomplished. In the event that aircraft departs from a station where no Emery Airline personnel are available, the Service Check must be accomplished within seventy-two (72) hours or prior to departure from a station where maintenance can be performed.

A Check

The A check interval is four hundred and fifty (450) flight hours/12 calendar months (whichever comes first). The A check program consists of four (4) visits (A1 through A4) scheduled sequentially.

C Check

The C Check shall be accomplished each fifteen (15) calendar months/4000 fight hours (whichever comes first). The check program consists of twelve visits (C1 through C12) scheduled sequentially.

Corrosion Program

The DC-10 MSG-3 MRB has incorporated in the corrosion prevention and control program (Report MDC-K4607) into the structural maintenance program (refer to Chapter 7).

SID

The DC-10 Supplemental Inspection Program (SID) is accomplished as outlined in Report L26-012.



October 6, 1999

Mr. William Dime FSDO - SJC FAA 1250 Aviation Ave, Suite 295 San Jose, CA. 95110

Dear Mr. Dime:

This letter is in reference to the bridging and transition of the MSG-2 Maintenance Program to the MSG-3 Maintenance Program for the newly acquired DC-10-10 Aircraft, N68041 S/N 46900, to Emery Worldwide Airlines (EWA) Aircraft Fleet.

Emery Worldwide Airlines, during the process of the cargo conversion, has added the aircraft work scope items necessary to bridge and transition from the previous operators MSG-2 Maintenance Program to Emery Worldwide Airlines MSG-3 Maintenance Program. All systems, Powerplant and Structural/Zonal task, in addition to all CPCP items, have been reviewed and accomplished as necessary to transition into EWA's Maintenance Program.

Advisory Circular number 121-1A, dated 6-26-73, was used in developing this transition/bridging document.

Please find attached the Aircraft N68041 Transition/Bridging document for your review.

Please call if you have any questions. Have a nice day.

Sincerely,

Ronald E. Moody

Manager Quality Prouvance

cc: Thomas Wood Bruce Robbins

REM/lc

AIRCRAFT N68041 (S/N 46900)

T.A.T. 84,955.53 AND T.A.C.30,991 AS OF MAY 19, 1999

1. GENERAL

Bridging/transitioning of aircraft N68041 to the approved Emery Worldwide Airlines (EWA) MSG-3 DC-13 Maintenance Program was accomplished as follows.

A. Systems and Powerplant Tasks

An EWA DC-10 "A4" and "C5" Check package was accomplished during the cargo conversion of aircraft N63041. This accomplished the equivalent of all System and Powerplant tasks within the "A" Check, and those System and Powerplant tasks with a 1C, 2C, 3C, and 6C interval. It should be noted that the freighter modification package requires extensive inspection and systems operational/functional testing throughout the aircraft (see attachment A).

B. Structural/Zonal Tasks

The DC-10 MSG-3 Structures tasks and intervals are calendar driven based on the existing DC-10 CPCP. A review of the Continental structural tasks indicated that CPCP tasks were tracked independently. An EWA "C6" package and work cards 13115503C and 13515503C were accomplished on aircraft N68041. This accomplishes all CPCP tasks with intervals requiring repeat inspection prior to the first scheduled "C" Check (15 months) on the EWA DC-10 maintenance program (no CPCP task interval exceeded). A "C4" Check will be the first check accomplished by EWA. This will ensure no CPCP task intervals are exceeded. All work cards of the EWA MSG-3 DC-10 maintenance program will have been accomplished on aircraft N63041 upon completion of the "C4" check, thus completing the bridging/transition effort. It should be noted that a "D" Check was accomplished on this aircraft in March of 1997 (2 years and 2 months).

2. "A" CHECK

- A. **BRIDGING** The EWA "A4" Check (equivalent of all 4 segments of the EWA DC-10 A Check Program) was accomplished during the freighter conversion/bridging check.
- B. The EWA "A1" Check must be accomplished on this aircraft prior to 85,405 hours (T.A.T.) or by the end of May 2000 (whichever comes first).

3. "C" CHECK

- A. PRORATION The Continental "C" Check was due to be accomplished on 3/28/98. This date has expired. An EWA DC-10 C Check visit must be accomplished prior to placing aircraft in service.
- B. **BRIDGING** The EWA "C6" work card package was accomplished during the freighter conversion/bridging check. This was based on bridging DC-10 CPCP tasks and transitioning to EWA's MSG-3 DC-10 Maintenance Program.

4. "D" CHECK

A. PRORATION - Proration of the "D" Check was accomplished as follows. It was determined that the Continental Check was equivalent of EWA's "C4" Check. Continental "D" Check interval is 2,190 days/ 25,000 hours.

	36%	60 mos (EWA)	60 mos (EWA)
72 mos	26 mos	x 36%	- 22 mos Used
	•	21.6 mos Used	38 mos Left

AIRCRAFT N68041 (S/N 46900)

B. **BRIDGING** - Based on "D" Check proration, the C4 check must be accomplished prior to July 2002. It was determined that the EWA "C4" Check will be accomplished on this aircraft prior to 88,955 hours (T.A.T) or by the end of August 2000 (whichever comes first). This is based on bridging DC-10 CPCP tasks.

5. CORROSION PREVENTION AND CONTROL PROGRAM (CPCP)

- A. All CPCP tasks are now an integral part of the EWA DC-10 MSG-3 Maintenance Program. Bridging to the EWA DC-10 maintenance program was accomplished maintaining established CPCP task intervals.
- B. CPCP tasks are integrated into the EWA DC-10 Maintenance Program at the recommended calendar interval. As a result of this integration, the EWA C4 Check must be accomplished on this aircraft prior to 88,955 hours (T.A.T) or by the end of August 2000 (whichever comes first).

6. RESTORATION/DISCARD TASKS

120M2401P - Air Driven Generator (ADG) Restoration
Must be accomplished prior to the end of June 2001.

19012701P - No. 2 and No. 3 Slat Drive Cables Left and Right Wing, Discard Must be accomplished prior to 41,036 T.A.C.

190M3201P - MLG Anti-Skid Manifolds Restoration Must be accomplished prior to 38,491 T.A.C./May 2007.

182M5201P - Cabin Door Pneumatic Bottle Assembly Restoration Must be accomplished prior to the end of May 2005.

Landing Gear:

Continental time limit is 30,000 flight hours. EWA time limit is 8 years or 7,500 cycles. Landing gear proration accomplished as follows:

96 Mos

- 72 Mos

24 Mos Remaining

Landing Gear restoration time was calculated as follows:

7.57 % 7500 CYC 7500 CYC
A. Right MLG - (Flight Hours) 30,000 22,730.27 (TSO) 22,730.27 (TSO) 7500 CYC - 5625 CYC Used 1,875 CYC Remaining

Note: This is the same for the Left MLG and the NLG.

96 Mos
× 75 %

72 Mos

Right MLG Restoration

Must be accomplished prior to the end of December 2001

- B. Left MLG Restoration

 Must be accomplished prior to the end of December 2001
- NLG Restoration.
 Must be accomplished prior to the end of December 2001

7. PRODUCTION CONTROL ITEMS (PCIs)

A. EWA PCI work cards should be scheduled as follows:

181I5201P - Lower Cargo Door Hinge Pin and Hinge Lobe Holes

Must be accomplished prior to 38,491 total aircraft cycles (T.A.C.).

PÂGE 2 OF 3

AIRCRAFT N68041 (S/N 46900)

7. PRODUCTION CONTROL ITEMS (PCIs) - continued

182I5201P - Upper Cargo Door Hinge Pin and Hinge Lobe Holes Must be accomplished prior to 38,491 T.A.C.

19012702P - Left/Right Inboard and Outboard Eievator Actuator Assemblies
Must be accomplished prior to 88,955 total aircraft time (T.A.T.)

190I7201P - Engine Combustion Liner and High Pressure Turbine, G.E. Must be accomplished prior to 31,441 T.A.C.

190M3202P - Main Landing Gear Truck Beam Lube Holes - Endoscope Must be accomplished prior to 85,105 T.A.T.

8. RECORDS SECTION

- A. DC-10 A CHECK Schedule the EWA "A1" Check on aircraft N68041 for accomplishment prior to 85,405 hours (T.A.T.) or by the end of May 2000. Schedule the following A Check visits (A2, A3, and A4) sequentially at 450 flight hours/12 calendar months between segments, whichever comes first (refer to the DC-10 Time Limits Manual).
- B. EWA DC-10 C CHECK Schedule the EWA "C4" Check on aircraft N68041 prior to 83,955 hours (T.A.T) or by the end of August 2000. Schedule the following C Check visits (C2, C3, C4, C5, etc.) sequentially at 4000 flight hours/15 months between segments (whichever comes first).
- C. PCIs Schedule and accomplish the following PCIs as indicated:

120M2401P - Accomplish prior to the end of June 2001, and repeat at a 5 year interval.

181I5201P - Accomplish prior to 38,491 T.A.C., and repeat at a 7500 cycle interval.

182I5201P - Schedule initial compliance prior to 38,491 T.A.C, and repeat at a 7500 cycle interval.

182M5201P - Accomplish prior to the end of May 2005, and repeat at a 6 year interval.

190l2701P - Accomplish prior to 41,036 T.A.C., and repeat at a 10800 cycle interval.

190I2702P - Accomplish prior to 88,955 T.A.T., and repeat at a 4000 hour interval.

19017201P - Accomplish prior to 31,441 T.A.C., and repeat at a 450 cycle interval.

190M3201P - Accomplish prior to 38,491 T.A.C./May 2007 (whichever comes first), and repeat at a 7500 cycle/8 year (whichever comes first) interval.

190M3202P - Accomplish prior to 85,105 T.A.T., and repeat at a 150 hour interval.

D. Landing Gear - Schedule and accomplish landing gear restoration as follows:

Right MLG - Accomplish prior to end of December 2001.

Left MLG - Accomplish prior to end of December 2001.

NLG - Accomplish prior to end of December 2001

CARD NO.	NOTE	DESCRIPTION
JFUV1		WEIGHT OF AIRCRAFT ASSUMED BY LANDING
JFUV2	•	AIRCRAFT REFUELING PROCEDURE
JFUV3		ADF & HF OP CHECK
		MM 34-53-00, 23-11-00
		Verify operational checks are completed.
JFUTN		REINSTALL LH AFT LOWER CARGO DOOR
		MM 52-33-04
		Verify prior operation.
JFUV4	*	DEPRESERVE APU
JFUTT	*	ALTITUDE SYSTEM FUNCTIONAL TEST
		Verify difference between 2 indications are less of 125 ft.
		ATC mode ACS Operational Test
		ATC transponder tests FAR 91.413 Part 43 Appendix F.
JFUTO	**	MAIN BATTERIES INSTALLED
JFUTP		FWD EMERGENCY BATTERY
		Verify prior operation is completed.
JFUTH		OPERATIONS ON MAIN CABIN UNDER FLOOR PANEL FROM STA 396- 2007 ALL CONTROL CABLES
		Verify all operations are completed.
JFUV5	*	COMPASS SWING TO ADJUST THE STANDBY COMPASS
JFU√6	*	ENGINE RUN-UP TEST
		External check MM 71-00-00
		Prestart Check, Engine Starting, After Start Check, Generator Check (24-20-00), Ground Idle Check, Flight Idle, Max Power Assurance, Test High Stage Bleed Air Valve (MM 36-00-00), Engine Shutdown, Engine Leak Check
JFYXX		INSPECTOR WEIGH AIRCRAFT & REPORT TO I.P
JFU∨8	*	PNEUMATIC SYSTEM OPERATIONAL CHECK 1, 2, 3
JFUV9	*	ENGINE FUNCTION CHECK ENGINE 1
	**	MM 24-40-00, 24-50-00, 71.00-00, 36-10-00, 29-00-00, 28-11-01, 26-00-00, 26-10-00, 78-32-11
JFUWB	*	ENGINE FUNCTION CHECK ENGINE 2
	**	SAME AS JFUV9
JFUWC	*	ENGINE FUNCTION CHECK ENGINE 3
	**	SAME AS JFUV9
JFYJW		OPERATIONAL CHECK
		Avionics Compartment, Self Test in cockpit, Operational Test, OPS Test w/Hydraulic Power.
	ĺ	Verify all operational checks are completed.
	İ	Avionics Compt MM References 22-11-05, 27-32-03, 32-61-01, 73-31-02.
		Cockpit - MM 26-10-00, 36-23-00, 27-32-03, 28-41-00, 30-41-01, 32-45-00,

CARD NO.	NOTE	DESCRIPTION
		34-42-00, 77-10-00
		·
		Operational Test - 22-02-01, 23-21-00, 23-22-00, 23-2-00, 23-70-00, 23-50-00, 30-30-00, 30-43-00, 31-21-01, 33-40-00, 30-42-01, 34-43-00, 34-23-00, 34-21-00, 34-28-00, 34-51-00, 34-31-00, 34-52-00, 34-45-00, 34-55-00, 34-48-00, 34-41-00
		OPS Test w/Hyd Pwr - 22-00-02, 22-00-05, 27-60-00
JFUTR	**	BALANCE WEIGHT WARNING LABELS REMOVAL (Not on Aeronavali's List)
JFUTU	**	ELECTRICAL POWER SUPPLY POST CARGO MOD
JFX6S	* #	FUEL TANKS LEAK CHECK
		MM 28-00-02, SRM VOL. II 51-42-01
JFXEK	*	HYDRAULIC POWER SUPPLY POST CARGO MOD.
		Check all actuating cylinders, control cables, and actuating rods on control valves.
		Verify pos of maintenance isolation valves, iaw MM 27-00-00.
		Verify system 3 elevator shutoff valve in aft fuselage compartment OPEN.
JFUTS	**	COCKPIT & LOOSE EQUIPMENT REINSTALLATION
		Reinstall loose equipment in cockpit and cabin.
JFUWE		FAA MANDATORY MARKINGS (MM 11-00-00)
		"Markings required by Federal Regulations"
		11-12-02, 11-15-01, 11-15-03, 11-19-03, 11-21-01, 11-21-03, 11-21-06, 11-31-03, 11-33-03, 11-34-03, 11-37-01*, 11-37-02*, 11-39-04, 11-99-00
		*Note: Verify at slide installation.
JFUTY	*	FINAL RECHECK OF UPPER CARGO DOOR
	**	Aft aircraft complete and engine mounted.
		Check valve adjustment, verify cockpit annunciator not illuminated, open and close vent door for proper operation/unlatch and open vent door. Verify annunciator light illuminates/manually actuate vent door – verify light goes out.
JFUTI	**	PREVENTIVE SURFACE TREATMENT CARGO DOOR BAY
		Corrosion preventive surface treatment on areas reworked per cargo door mod (DPS 4.50-62, 4.50-36)
JFUTJ	**	PAINT TOUCH UP OF REWORKED AREAS AND PARTS
		FR primer touch-up, or complete repaint.
JFUWF		POSTFLIGHT INSPECTION
		No List of inspections refers to SKT: POSTFLIGHT INSP.
		Verify postflight inspection has been completed.
JFUWG		PREFLIGHT INSPECTION
		No List of inspections refers to SKT: PREFLIGHT INSP.
		Verify preflight inspection has been completed.
JFUWD	*	DEPRESERVATION ENGINES 1, 2, & 3

CARD NO.	NOTE	DESCRIPTION
		DPS 3.442-25, DPS 1.151-1
JFUTX		SHORING REMOVAL
		Verify prior operations.
JFUTZ	+	FWD CARGO COMPARTMENT HEAT TEST
JFUT0	*	CABIN PRESSURIZATION SYSTEM TEST
JFUT1	*	PNEUMATIC DECAY TEST
JFUT2	*	AIR DISTRIBUTION TEST
JFUT3	*	CREW OXYGEN SYSTEM ADJUSTMENT/TEST
JFUT4	*	LAV OXYGEN SYSTEM TEST
JFUT5	*	MAIN CABIN SMOKE DETECTOR SYSTEM TEST
JFUT6	*	FWD AND AFT CARGO SMOKE DETECTOR SYSTEM TEST (LOWER)
JFUT7	*	FIREX SYSTEM TEST
JFUT8	*	FUSELAGE LEAK CHECK
JFUT9	*	RAM AIR TEST
		RAM air cabin ventilation valve.
JFUTK	**	FWD DROP CEILING PAINT TOUCH UP
		Paint touch up fwd drop ceiling STA 396 & 519 (between)
JFUUB		FLIGHT/SERVICE INTERPHONE TEST
		Flight/Service test communications interphone.
JFUUC	*	MASTER WARNING & CAUTION SYSTEM TEST
JFUUD	*	GEN 123 GCU CONFIDENCE CHECK
JFUUE	*	EVACUATION SLIDE PACK PRESSURE INDICATOR SYSTEM
		MM 25-61-01
JFUUF	*	LANDING GEAR PROXIMITY SENSOR TEST
JFUUG	*	INSTRUMENT PANEL LIGHTING TEST (MM 33-11-07)
JFUUH	*	CABIN LIGHTING TEST
		DC10-L0054
JFUUI	*	SERVICE AREAS LIGHTING TEST
		DWG SP10330036
JFUUJ	*	EMERGENCY LIGHTING TEST
		DC10-L0073
JFUUL	*	APU TEST
		MM 49-00-00, DC10-L0075
JFUUK	•	PNEUMATIC CONTROLLER BITE TEST
		DC10-L0074
JFÙUN	*	DC POWER SYSTEM TEST
		DWG SP10330036
		Check flight compartment left overhead panel, center instrument panel, flight

CARD NO.	NOTE	DESCRIPTION
		engineer panel standby floodlights come on.
		Check left and right EMER AC BUS OFF, left and right EMER DC BUS OFF lights, and emergency bus summary lights come on.
		Check EMER PWR IN USE lights located on flight compartment overhead panel and at flight engineer panel come on.
		Check left and right EMER AC and DC BUS OFF lights go and cut and emergency bus summary lights remain on.
JFUUO	*	APU FIRE DETECTION TEST
		DC10-L0128
JFUUP	*	FWD L1/R1 CABIN DOOR TEST
JFUUQ	*	LWR FWD CARGO DOOR TEST
JFUUR	*	LWR CENTER CARGO DOOR TEST
		MM 52-32-00
JFUUS	*	LWR AFT CARGO DOOR TEST
		MM 52-33-00
JFUUT	*	ENVIRONMENTAL SYSTEM BITE TEST
		Temperature Control System. DC10-L0111, -L0129, DWG SP10330036
JFUUU	*	ENVIRONMENTAL SYSTEM OP CHECK
		Test of air conditioning. DC10-L0130
JFUUV	*	FLIGHT RECORDER OP CHECK
		MM 31-31-00 pg 501 to 503, Operational test of flight recorder.
JFUUW	*	PITOT STATIC SYSTEM TEST
		Per FAR 91.411, Part 43, Appendix E; DWG SP10250367
		There is also a sign-off on the FAR verification.
JFUUY	*	CARGO DOOR WARNING SYSTEM
JFUUX	*	MAIN CABIN ELECTRICAL OUTLET TEST
		MM 24-55-00, DC10-L0057
JFUUZ	*	AUDIBLE SIGNAL SYSTEM TEST
JFUU0	•	LANDING GEAR POSITION AND WARNING SYSTEM TEST
		MM 32-60-00 PG 501-504 and 504A
JFUU1	•	MAIN EXTERNAL POWER INDICATING LIGHT
		DC10-L0106, -L0112, DWG SP10330036
JFUU2	*	APU GCU CONFIDENCE CHECK
		DWG SP10330036, DC10-L0119
JFUU3	*	RCCB FUNCTIONAL TEST
		MM 24-50-01
JFUU4	*	AUX HYDRAULIC PUMP TEST
		DC10-L0059, DWG SP10330036
JFUU5	•	FAULT ISOLATION PANEL TEST

JFUU6 COMMS (INTERPHONE) TEST JFUU7 CABIN LIGHTING TEST DWG SP10330112 JFUU8 COURIER STATION READING LIGHTS TEST DWG SP10330112 JFUU9 TOURY OXYGEN SYSTEM PURGING DPS 4.901 para. 5.4.5, DQS 4.7-4 JFUVB WASTE SYSTEM FUNCTIONAL TEST DWG SP10380066 JFUTL PROTECTIVE PAINT REMOVAL IN CARGO DOOR AREA Remove protective paint from new skin panels DPS 4.50-169 JFUVC UPPER CARGO DOOR TEST DC10-L0005 JFUVD LDG GEAR SERVICING AND TIRE INFLATION Landing gear system general notes. System fill and bleed instructions.	
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Remove protective paint from new skin panels DPS 4.50-169 JFUVC * UPPER CARGO DOOR TEST DC10-L0005 JFUVD * LDG GEAR SERVICING AND TIRE INFLATION JFUVE * SYSTEM FILL AND BLEED INSTRUCTION Landing gear system general notes.	
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JFUVD * LDG GEAR SERVICING AND TIRE INFLATION JFUVE * SYSTEM FILL AND BLEED INSTRUCTION Landing gear system general notes.	
JFUVE SYSTEM FILL AND BLEED INSTRUCTION Landing gear system general notes.	
Landing gear system general notes.	
System fill and bleed instructions.	,
Checked by hand gear retraction.	
Gear up latch has been checked.	
Special Ferry Permit.	
JFUVF * LANDING GEAR SWING	
Functional check of the landing gear.	
JFUVG * O.A.T.P.: GEAR UP BRAKE APPLICATION	
Functional check of the gear up brake application.	
JFUVH * O.A.T.P.: GEAR FRICTION TEST	
JFUVI * LANDING GEAR CONTROL & POSITION INDICATION	
Functional check - Operational check	
DC10-L102 pg 16.0 – 16.11	
JFUVJ * LDG GR EXTENSION & RETRACTION RATE	
JFUVK * WHEEL BRAKE CHECK	
JFUVL * O.A.T.P. NOSE WHEEL STEERING	
DC10-L0104	
JFUVN * O.A.T.P. AIR DRIVEN GENERATOR	
JFUVO * O.A.T.P. FUEL SHUTOFF SYSTEM	
Check fuel lever on the fuel control unit.	
JFUVP * O.A.T.P. EMERGENCY SHUTDOWN SYSTEM	
JFUVQ • O.A.T.P. THROTTLE SYSTEM	

CARD NO.	NOTE	DESCRIPTION
JFUVR	*	O.A.T.P. SPEEDBRAKE SYSTEM
		Check auto spoiler actuator linkage clear.
		Check speed brake drum in center wheelwell.
		Check speed brake lever will latch in detent.
		Check speed brake handle disarm (DOWN)
JFUVS	*	O.A.T.P. AUTO GROUND SPOILER SYSTEM
		Functional check of auto ground spoiler system per MM 27-60-00
JFUVT	*	O.A.T.P. SLAT CONTROL SYSTEM
JFUVU	*	O.A.T.P. FLAP CONTROL SYSTEM
JFUVV	*	O.A.T.P. AILERON SYSTEM & SPOILER
		Functional Check
JFUVW	*	O.A.T.P. AILERON & SPOILER SYSTEM TEST
JFUVY	*	O.A.T.P. RUDDER SYSTEM
		Functional check of the directional control system.
JFUVX	*	O.A.T.P. HORIZ. STABILIZER & ELEVATORS
		Functional check of: Horizontal Stabilizer, Horizontal Stabilizer Trim System, Elevator Control System
JFUVZ	*	TAKEOFF WARNING SYSTEM
		Functional Check DC10-L0088
JFUV0	*	FIT CHECK CARGO LOADING SYS UPPER CABIN
		Main deck cargo compartment.
		Cargo loading system w/unit load device.
		23 pos w/pallets 88 x 125 and 96 x 125".
		Check rollers and restraints for freedom.
·		Verify all positions are fit checked.
JFUWH		DC1-10 WEIGH PREPARATION
		Verify aircraft is complete and all foreign material is removed.
		Prior operation completed.
JFUWI		WING ANTI-ICE OPERATIONAL CHECK
		MM 30-11-00 During engine run or APU run.
		Verify operational check is completed.

ENARRESPONSETO LETTER CATED January 7, 1999



January 18, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This latter constitutes Emery Worldwide Airlines Inc. (EWA)'s follow-up (initial letter sent 1-15-99 attached) formal response to your letter of investigation (99WP150025), addressed to EWA's President and Chief Operating Officer, dated January 7, 1999.

The Manager of Reliability has prepared a comprehensive fix and action plan to prevent future occurrence of this inadvertent isolated event (See Attachment).

It is equally important to consider and review the overall performance of EWA's Approved Maintenance Program over the past nine (9) years. I have provided a score card (See Attachment) that demonstrates the overall above average performance of EWA's Technical Services Department.

The very achievement of a 98% Mechanical Dispatch Reliability average for the past nine (9) years of an aging fleet reflects the overall achievement of an effective maintenance program under sincere management oversight and leadership.

EWA has maintained an average of two (2) pilot reports per flight hour since 1990. It is important to note that EWA increased its flight hours by 9% in 1998 and decreased the number of PIREP'S per flight hour by 25%.

This performance measurement also demonstrates the effectiveness of EWA's training program and manual system as the mechanics performance is a direct result of EWA's program administration.

Mr. Joe Abramski Page 2 January 18, 1999

I trust EWA's comprehensive fix is satisfactory. Should you wish to discuss this matter more fully, EWA's Senior Management and I would be more than willing to meet with you and address any concerns.

Sincerely,

Dans m. Jac

Thomas M. Wood Director Quality Control

TMW/re

Attachment

cc:

Kent Scott Rene' Visscher QC Managers



MEMORANDUM

TO:

Thomas Wood, Director of Quality Control

FROM:

Bob Peck, Manager of Reliability

SUBJECT: LOI 99WP150025

DATE:

18 January, 1999

The Following is a list of events that contributed to the late reporting of the Mechanical Interruption Summary Reports for the months of September, October, and November 1998, and the detailed corrective actions that have been implemented to prevent this from happening again in the future.

Beginning in September, the completion of the DC-10 Inspection Program became a high priority. The Reliability Section currently has three technical analyst assigned, one Reliability specialist, one Data entry clerk, and the Manager. In September, Reliability had one technical analyst assigned full time to the DC-10 tasking, one assigned 70% of the time to the DC-10 tasking, and 30% to completion of the Fleet Monthly Reliability Report. The third technical analyst was assigned to providing CPCP training to newly contracted Heavy Maintenance facilities, and reviewing completed heavy maintenance packages. The reliability Specialist is responsible for entering data for the Engine Condition Monitoring Program. Mr. Feisley, the technical analyst that was assigned full time to the DC-10 program was lost for 3 weeks in September for surgery on his neck.

In October when preparing for September Monthly Fleet Reliability Report, an unforeseen problem was found with the MERIT data base system. When programs were being run to collect the Pilot Report (PIREP) Data, it was suspected that only about 50% of the reported PIREPs for September were in MERIT. This was determined by comparing previous monthly counts of PIREPs with what was being shown for September. To verify, a complete audit of the log page information to MERIT information was performed. This audit began a snowball affect, and has carried through Decembers data. If the MERIT problem had not been there, the Fleet Reliability Reports would have been on time.

An extensive corrective action program has been initiated to eliminate the MERIT backlog, and prevent this from happening again. The program has two separate actions; one is to assign two people to clear the remaining backlog. These two people are solely responsible for eliminating the MERIT backlog. The second part of this program has two people from aircraft records assigned to monitoring and correcting the current MERIT log

page entries on a daily basis. Also in conjunction with the current month monitoring and correcting, a copy of the log page with the MERIT discrepancy will be forwarded to the Manager of Line Maintenance for his action. The corrective action steps to prevent this from happening in the future, and the current backlog will be completed by 1 March, 1999.

With the addition of the data entry clerk, the Mechanical Interruption Summary Report information will be processed on a daily basis. The MISR information will be reviewed the first full week following the end of the month and forwarded to the FAA/PMI by the end of that week or before, if the current procedure of receiving this information at the end of each month in the Reliability Report is now not acceptable.

Bob Peck

Manager of Reliability



January 15, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA)'s initial formal response to your letter of investigation (99WP150025), addressed to EWA's President and Chief Operating Officer, dated January 7, 1999.

I would like to assure you that your letter has merited EWA's immediate and undivided attention. EWA, as a certificated air carrier and its management and employees are fully appreciative of their responsibilities arising under pertinent laws and under the Federal Aviation Regulations (FAR's) and strive to fulfill these responsibilities in a professional and conscientious manner as successfully demonstrated over the past nine years.

Upon receipt of your letter January 11, 1999, the Manager of Reliability contacted you by telephone and acknowledged receipt of your letter and explained the basic details of the delay of the inadvertent failure of the Reliability Reports being sent to you, that provides you the Mechanical Interruption Summary Reports. This is the first occurrence since the FAA approval of the Reliability Program in 1990.

The Manager of Reliability has completed the September, October and November 1998 MISR reports that are enclosed with this letter.

A comprehensive review was performed by EWA Technical Services Management and a corrective action plan put in place to prevent future non-compliance. A comprehensive Fix will be submitted to you next week.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

Attachments

co: Rene' Visscher Robert Peck

FAA TRAINING AWARDS

EMERY WORLDWIDE AIRLINES FOURTH ANNUAL FEDERAL AVIATION ADMINISTRATION TECHNICAL AWARDS PRESENTATION 1997

EWA Accomplishment Overview

Emery Worldwide Airlines (EWA) is pleased to receive for the fourth consecutive year, the Federal Aviation Administration (FAA) Technical Awards presented to the EWA Mechanics, Technical Service Management, Senior Director Technical Services, and Vice President and General Manager. A chronological history of the awards received to date is presented for your review.

1994 FAA Awards

The awards received during a ceremony on May 11, 1994 were as follows:

1. Mechanical Technical Awards

133 mechanics were presented these awards which represented 42% of the EWA mechanics.

This 42% or 133 mechanics actually represent 96% of EWA's full-time mechanics.

2. Organizational Awards

The highest award, the Diamond Certificate of Excellence was presented to Emery Worldwide Airlines.

3. Master Mechanic Award

This prestigious aviation career accomplishment was presented to Mr. Roy Deeming. The requirement of selection for this award is fifty (50) years of serving as a certificate airframe and powerplant mechanic.

i

1995 FAA Awards

1. Mechanical Technical Awards

EWA employed 304 technicians/mechanics. Out of these, 228 or 75% have received awards. This was a 33% increase in training EWA personnel from the previous year.

This 75% or 228 mechanics actually represent 60% of EWA's full-time mechanics.

2. Organizational Awards

For the second consecutive year, required training percentage achieved by EWA surpasses the requirement stated in the Advisory Circular. The Diamond Certificate of Excellence requires 25% of eligible employees to be trained. Therefore in view of the great achievement of training rendered to its employees, EWA qualified itself to receive again the Diamond Certificate of Excellence.

1996 FAA Awards

1. Mechanical Technical Awards

EWA employment 320 technicians/mechanics. Out of these, 264 or 83% received awards. This is a 14% increase in training EWA personnel from the previous year.

2. Organizational Awards

For the third consecutive year, the required training percentage achieved by EWA surpassed the requirement stated in the FAA Advisory Circular. Therefore, in view of the great achievement of training rendered to its employees, EWA qualified and received the Diamond Certificate of Excellence Award.

1997 FAA Awards

1. Mechanical Technical Awards

EWA employed 338 technicians/mechanics. Out of these, 181 or 54% received awards. This is a 49% decrease in training EWA personnel from the previous year. This decrease reflects the previously accomplished extensive training provided in the previous seven years.

2. Organizational Awards

For the Fourth consecutive year, the required training percentage achieved by EWA surpassed the requirements stated in the FAA Advisory Circular. Therefore, in view of the great achievement of training rendered to its employees, EWA qualified for and received again the Diamond Certificate of Excellence Award.

Awards Summary:

This training is a direct contribution to the continued success of EWA. We have experienced for the past nine years an average of 98% Mechanical Dispatch Reliability performance, a standard desired by many Air Carriers.

These FAA awards exemplify EWA's professional approach to lead its employees to produce the highest level of safety possible and the most cost effective process to provide the customer the best product.

Continues to Produce Successful Results

Emery Worldwide Airlines Maintenance Program is tested by other means than it's Mechanical Dispatch Reliability that has maintained 98% average over the past nine years. EWA has gone through several very in-depth FAA/DOD/Outside Firms inspections over the past nine (9) years. The successful results of these inspections continued to reveal EWA's ratings to be higher than the Industry performance of the 121 Air Carriers and average to excellent ratings from the Department of Defense (DOD).

In 1992, EWA went through a very in-depth FAA NASIP Inspection to which EWA rated 64% higher than the Industry performance of the 121 Air Carriers. EWA received honorable recognition for this achievement from the San Jose FAA Certificating Holding Office Manager.

In 1995, EWA received a specific FAA inspection that was administered by FAA Washington, DC to be accomplished on all 121 Air Carriers in 1995. This inspection was titled a Regional Aviation Safety Inspection Program (RASIP). This inspection lasted ten days and covered the Operations/Maintenance Departments. On June 22, 1985, the FAA RASIP team provided EWA Senior Management a debrief of their findings. The team reported that their inspection did not reveal any major discrepancies and overall EWA was above average in performance.

In 1997, EWA received a comprehensive Internal Evaluation performed by the SH&E International Air Transport Consultancy. This evaluation was performed based on the FAA NASIP items to ensure EWA has adequate systems and controls in place to support the growth of the airline. A report was provided to EWA Senior Management from the SH&E team that reflected an excellent rating of the Technical Services Organization. Their report specifically reflected that all aspect of the necessary systems and controls were in place and performing excellent ratings.

EWA Technical Services Department has gone through four Department of Defense (DOD) inspections in the past nine years. We received average to excellent ratings on all inspections.

EWA's Maintenance Program success is a direct result of true team effort promoting synergy.

Another indicator for EWA's performance is reflected by the low number of FAA Enforcement Actions received. The following data provides an analytical summary of this performance.

EMERY WORLDWIDE AIRLINES MAINTENANCE PERFORMANCE BASED ON FAA SAFETY INSPECTION/ENFORCEMENT HISTORY

YEAR	#ADMIN ENFORCEMENT'S	FAA <u>NPTRS</u>	FLEET SIZE	FLT HOURS	CYCLES	PILOT REPORTS
1990 1991 1992 1993 1994 1995	4 3 3 2 1 2	Ref. Total Ref. Total Ref. Total Ref. Total Ref. Total Ref. Total Ref. Total	7 20 29 29 37 37 39	11,070 28,095 40,606 42,473 52,465 55,178 57,994	4,732 12,565 20,559 20,718 23,704 25,169 23,960	3,679 10,512 17,196 15,443 16,667 16,230 15,234
1997 1998	0 1	Ref. Total Ref. Total	43 43	62,405 68,140	28,127 32,561	14,760
TOTALS	17	4,944		418,426	192,095	22,061 131,832

EWA PERFORMANCE FACTORS SUMMARY - 1/90 THROUGH 12/98

- During the nine (9) year period of Air Carrier Operations, EWA Technical Department experienced the following:
 - FAA Administrative Enforcement's compared to # of Safety Inspections = .3%
 - FAA Administrative Enforcement's compared to # of Flight Hours = .004%
 - FAA Administrative Enforcement's compared to # of Flight Cycles = .008%
 - FAA Administrative Enforcement's compared to # of Pilot Reports = .01%
- EWA's FAA Administrative Enforcement's are minor in numbers as represented during 1990 thru 1998.
- EWA increased its fleet size by 22% in 1994 and decreased its number of PIREP's per flight hour by 5%.
- EWA increased its fleet size by 6% in 1996 and decreased its number of PIREP's per flight hour by 3%.
- EWA increased its fleet size by 10% in 1997 and decreased its number of PIREP's per flight hour by 10%.
- EWA increased its flight hours by 9% in 1998 and decreased its number of PIREP'S per flight by 25%.

FAA/SPOT RAMP INSPECTION RESULTS 1998

EWA incorporated an airline industry standard "FAA Spot/Ramp Inspection Procedures" into our Maintenance Policy and Procedure Manual (MPP) in 1995.

The purpose of this program was to enhance EWA's Continuing Analysis and Surveillance System (FAR 121.373) for the continuing analysis and surveillance of the performance and effectiveness of its inspection program and the program covering other maintenance, preventative maintenance, and alterations and for the correction of any deficiency in those programs.

It also provides direct support to FAR 119.59 to assure that EWA properly handles FAA Inspector contacts, and expedites the handling of any FAA request for information.

In 1998, 78 FAA Station Inspections of the EWA's 43 line stations were reported. A total of 173 minor findings was noted and corrected. This number of findings reflected 70% of the inspections resulted in an average of 2 write-ups per visit, and 30% no findings.

This audit performance continues to reflect EWA's compliance of FAA regulations and company policies and procedures.



U.S. Department of Transportation

Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

RECEIVED
MAR 2 2 1999
KENT T. SCOTT

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028 EIR 99WP150025 is incorporated into EIR 99WP150028 EIR 99WP150029 is incorporated into EIR 99WP150028 EIR 99WP150032 is incorporated into EIR 99WP150028 EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W

EWA'S RESPONSE TO LETTER DATE! UNIVERY 27, 1999



February 11, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to my letter dated February 8, 1999, formal response to your LOI #99WP150028 dated January 27, 1999, and Robert Peck's letter to you dated February 2, 1999 advising you of the scheduled EWA Reliability Meeting dates.

I am forwarding you the completed September, October and November Reliability Reports that will be covered in the scheduled meeting tomorrow, February 12, 1999.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc: Rene' Visscher Robert Peck



February 8, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines Inc., (EWA) formal response to your letter of investigation (99WP150028) addressed to EWA's President and Chief Operating Officer, dated January 27, 1999 (See Attachment).

At the outset, I would like to assure you that your letters have merited EWA's immediate and undivided attention. EWA, as a certificated air carrier, and its management and employees are fully appreciative of their responsibility arising under pertinent laws and under the Federal Aviation Regulations (FAR's) and strive to fulfill these responsibilities in a professional and conscientious manner.

Upon immediate receipt of your referenced letter February 2, 1999, EWA's Manager of Reliability Robert Peck discussed this with you by telephone, and then responded in writing the same day by a letter faxed to you. (See Attachment).

Mr. Peck discussed during this conversation on February 2, 1999, the previous phone call on January 15, 1999 to which he informed you the Reliability Meetings had not been held due to the MERIT data system problem. This discussion was previously held due to the LOI file number 99WP150025 (See Attachment) concerning the late submission of the Mechanical Interruption Summary reports that is included in the monthly Reliability Report. Mr. Peck's January 18, 1999 letter provided you the root cause of this isolated problem and a comprehensive fix that is in place to date (See Attachment). My letters dated January 15, 1999 and January 18, 1999 provided the Delay Reports for September, October and November, and a comprehensive fix and action plan represented in Mr. Peck's January 18, 1999 letter.

The two letters of investigation 99WP150025 and 99WP150028 pertain to the same subject matter with regard to EWA's procedure of control. EWA can appreciate the FAA's voicing of its possible concerns with the EWA Reliability Meetings not held, we question whether two separate letters of investigation and two separate files, each bearing a different File Number, were required or even appropriate. This is not, and should not be an enforcement matter, rather it is a technical issue. I have enclosed EWA Septembers Fleet Reliability Report which reflects the information referenced (See Attachment).

A detailed review of EWA's Maintenance Reliability Program Document No. EWA-51990, dated June 13, 1997 Revision #7, and Temporary revision #7a, dated 1-2-98 was performed to evaluate compliance of EWA continuous airworthiness maintenance program during this reference period of not producing the report and having the meetings.

Mr. Joseph Abramski Page 2 February 8, 1999

All aspects of this program to ensure continuous airworthiness was performed on a daily basis as is referenced in the Document Chapter 4, Data Collection System. In addition daily/weekly scheduled meetings are held with the Technical Services Management to address all aspects of EWA's continuous maintenance program including daily corrective actions.

Chapter 6, "Corrective Action System" was in non-compliance as is referenced on page 4, item D. System Over-Par Reports.

Based on the day-to-day interactive support of this system, no information was relayed that would or did cause non-compliance to the continued airworthiness program.

I trust that this has been responsive to your letter. Should you wish to discuss this matter more fully, EWA's Senior Management and I would be more than willing to meet with you an address your concerns. If this is the case, please contact me to arrange this meeting.

Sincerely,

DWINDS III. WOOD

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc:

Kent Scott Rene' Visscher Robert Conlon Michael Dworkin



San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5443

January 27, 1999

File Number: 99WP150028

CERTIFIED MAIL - RETURN RECEIPT REQUESTED

Mr. Kent Scott
President & Chief Operating Officer
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Scott:

This letter is to inform you that Emery Worldwide Airlines, Inc., (EWA), the holder of Air Carrier Certificate Number RRXA558B, may be in violation of Federal Aviation Regulations, in that EWA has not conducted the monthly Maintenance Reliability Program meeting and review of its continuous airworthiness maintenance program as required in its D74 Operations Specifications Document EWA-51990, for the months of September, October, and November, 1998; and that this matter is under investigation by the Federal Aviation Administration.

We offer you the opportunity to submit a written statement to this office regarding this matter, which should be accomplished within ten (10) working days following receipt of this letter. Your response should contain all pertinent facts and extenuating or mitigating circumstances that you believe may have a bearing on this matter. Should you elect not to respond within the specified time, our report will be processed without the benefit of your statement.

Sincerely,

ORIGINAL SIGNED BY

Joseph A. Abramski
Principal Maintenance Inspector

cc: Rene P. Visscher - EWA Thomas M. Wood - EWA 

February 2, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski

This letter is the initial response to your Letter of Investigation, File Number: 99WP150028 which was received today. The topic of the Maintenance Reliability Program meeting was discussed during our telephone conversation on the 15th of January, 1999. At that time we were discussing the Letter of Investigation, File Number: 99WP150025, concerning the late submission of the Mechanical Interruption Summary reports for the same time frame, September, October, and November 1998. During that conversation you asked about the Reliability meetings, and I advised you that they had not been held do to the problem that we were experiencing with the MERIT data system.

The cause for not having the Maintenance Reliability Program meetings is the same as for the late submission of the Mechanical Interruption Summary Reports; Reliability being heavily involved in the development of the DC-10 Inspection Program, and the unforeseen problem with the MERIT data system. The MERIT problem being the driver.

Emery Worldwide Airlines, elected to go to the U.S. Airways Maxi Merlin data system (MERIT). This data system ties all aspects of airline operation together. With the implementation of a new system, there will always be unforeseen problems. The problem experienced with the September data is a prime example of the type of problems that can occur. The problem has been aggressively attacked, and by February 5th all backlog should be eliminated.

The failure to hold the monthly Maintenance Reliability Meeting has not hindered the EWA Reliability program, as can be evidenced by the improving trend in the Mechanical Dispatch Rate for these three months. The rate for the three months was 96.1%, 97.1% and 97.5% respectively.

At the present time all data has been entered in the MERIT data system, has been checked, and all necessary download runs have been received. Septembers Monthly Report is at the copy center, and completion of October and Novembers Reports should be by Friday, February the 5th. A Meeting has been scheduled for the 12th of February to review September, October and November 1998 Reliability Program, and a second meeting is scheduled for February 26th to review December and January data. This will get us back on schedule.

Bob Peck

Manager of Reliability

cc: Rene P. Visscher, Vice President Technical Services Thomas M. Wood, Director of Quality Control



303 CORPORATE CENTER DR. VANDALIA, OH 45377

FAX: (937) 898-2803 PHONE: (937) 454-3940

THOMAS M. WOOD **DIRECTOR QUALITY CONTROL**

FACSIMILIE COVER SHEET DATE February 3, 1999 SEND TO FAX#: 264-6072 **DELIVER IMMEDIATELY TO:** TELEPHONE NO: 264-6501 Kent Scott NAME: COMPANY / DEPARTMENT: Emery Worldwide Airlines This is page 1 of 3 pages sent. Kent: Here is the initial response on the LOI for the Reliability Meeting. I will formally respond this week. **FACSIMILE MESSAGE FROM:** Thomas M. Wood/re NAME:



MEMORANDUM

TO:

Thomas Wood, Director of Quality Control

FROM:

Bob Peck, Manager of Reliability

SUBJECT:

LOI 99WP150025

DATE:

18 January, 1999

The Following is a list of events that contributed to the late reporting of the Mechanical Interruption Summary Reports for the months of September, October, and November 1998, and the detailed corrective actions that have been implemented to prevent this from happening again in the future.

Beginning in September, the completion of the DC-10 Inspection Program became a high priority. The Reliability Section currently has three technical analyst assigned, one Reliability specialist, one Data entry clerk, and the Manager. In September, Reliability had one technical analyst assigned full time to the DC-10 tasking, one assigned 70% of the time to the DC-10 tasking, and 30% to completion of the Fleet Monthly Reliability Report. The third technical analyst was assigned to providing CPCP training to newly contracted Heavy Maintenance facilities, and reviewing completed heavy maintenance packages. The reliability Specialist is responsible for entering data for the Engine Condition Monitoring Program. Mr. Feisley, the technical analyst that was assigned full time to the DC-10 program was lost for 3 weeks in September for surgery on his neck.

In October when preparing for September Monthly Fleet Reliability Report, an unforeseen problem was found with the MERIT data base system. When programs were being run to collect the Pilot Report (PIREP) Data, it was suspected that only about 50% of the reported PIREPs for September were in MERIT. This was determined by comparing previous monthly counts of PIREPs with what was being shown for September. To verify, a complete audit of the log page information to MERIT information was performed. This audit began a snowball affect, and has carried through Decembers data. If the MERIT problem had not been there, the Fleet Reliability Reports would have been on time.

An extensive corrective action program has been initiated to eliminate the MERIT backlog, and prevent this from happening again. The program has two separate actions; one is to assign two people to clear the remaining backlog. These two people are solely responsible for eliminating the MERIT backlog. The second part of this program has two people from aircraft records assigned to monitoring and correcting the current MERIT log

page entries on a daily basis. Also in conjunction with the current month menitoring and correcting, a copy of the log page with the MERIT discrepancy will be forwarded to the Manager of Line Maintenance for his action. The corrective action steps to prevent this from happening in the future, and the current backlog will be completed by 1 March, 1999.

With the addition of the data entry clerk, the Mechanical Interruption Summary Report information will be processed on a daily basis. The MISR information will be reviewed the first full week following the end of the month and forwarded to the FAA/PMI by the end of that week or before, if the current procedure of receiving this information at the end of each month in the Reliability Report is now not acceptable.

Bob Peck

Manager of Reliability



San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7581 FAX: (408) 279-5443

January 7, 1999

File Number: 99WP150025

CERTIFIED MAIL - RETURN RECEIPT REQUESTED

Mr. Kent Scott
President & Chief Operating Officer
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Scott:

This letter is to inform you that Emery Worldwide Airlines, Inc., the holder of Air Carrier Certificate Number RRXA558B, may be in violation of Federal Aviation Regulations, in that this office has not received the required Mechanical Interruption Summary Reports for the months of September, October, and November, 1998; and that this matter is under investigation by the Federal Aviation Administration.

We offer you the opportunity to submit a written statement to this office regarding this matter, which should be accomplished within ten (10) working days following receipt of this letter. Your response should contain all pertinent facts and extenuating or mitigating circumstances that you believe may have a bearing on this matter. Should you elect not to respond within the specified time, our report will be processed without the benefit of your statement.

Sincerely,

ORIGINAL SIGNED BY

Joseph A. Abramski Principal Maintenance Inspector

cc: Rene P. Visscher - EWA Thomas M. Wood - EWA .• .



January 15, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA)'s initial formal response to your letter of investigation (99WP150025), addressed to EWA's President and Chief Operating Officer, dated January 7, 1999.

I would like to assure you that your letter has merited EWA's immediate and undivided attention. EWA, as a certificated air carrier and its management and employees are fully appreciative of their responsibilities arising under pertinent laws and under the Federal Aviation Regulations (FAR's) and strive to fulfill these responsibilities in a professional and conscientious manner as successfully demonstrated over the past nine years.

Upon receipt of your letter January 11, 1999, the Manager of Reliability contacted you by telephone and acknowledged receipt of your letter and explained the basic details of the delay of the inadvertent failure of the Reliability Reports being sent to you, that provides you the Mechanical Interruption Summary Reports. This is the first occurrence since the FAA approval of the Reliability Program in 1990.

The Manager of Reliability has completed the September, October and November 1998 MISR reports that are enclosed with this letter.

A comprehensive review was performed by EWA Technical Services Management and a corrective action plan put in place to prevent future non-compliance. A comprehensive Fix will be submitted to you next week.

Sincerely,

Thomas M. Wood

Director Quality Control

TMW/re

Attachments

cc: Rene' Visscher Robert Peck



January 15, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

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Sincerely,

Thomas M. Wood

Director Quality Control

TMW/re

Attachments

cc: Rene' Visscher

Robert Peck



San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone, (408) 291-7531 FAX: (408) 279-5448

January 7, 1999

File Number: 99WP150025

CERTIFIED MAIL - RETURN RECEIPT REQUESTED

Mr. Kent Scott
President & Chief Operating Officer
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Scott:

This letter is to inform you that Emery Worldwide Airlines, Inc., the holder of Air Carrier Certificate Number RRXA558B, may be in violation of Federal Aviation Regulations, in that this office has not received the required Mechanical Interruption Summary Reports for the months of September, October, and November, 1998; and that this matter is under investigation by the Federal Aviation Administration.

We offer you the opportunity to submit a written statement to this office regarding this matter, which should be accomplished within ten (10) working days following receipt of this letter. Your response should contain all pertinent facts and extenuating or mitigating circumstances that you believe may have a bearing on this matter. Should you elect not to respond within the specified time, our report will be processed without the benefit of your statement.

Sincerely,

ORIGINAL SIGNED BY

Joseph A. Abramski Principal Maintenance Inspector

cc: Rene P. Visscher - EWA Thomas M. Wood - EWA

			DELAY	Y SUMMARY		
DCS	FLEET				Septe	mber 1998
Tail # N105WP	Aircraft Type DC8-73	Fit Date 9/18/98	Flt # EB024	Fit Leg KDAY -to- KONT	Delay Length 0 Hr. 34 Min.	ATA 3415
Discrepan	ıcy:			Corrective Action:		3413
TAS HEA	ATER LIGHT ILLU	MINATED.		PLACARD INSTA LIGHT OPS CHEC	TO DMI #809238 IAW M LLED DD 9/28/98. TAS CKS GOOD IAW DC3 M D. THIS CLEARS DMI # VED.	SYS AND M 34-11-01 NO
Tail # N105WP Discrepan	Aircraft Type DC8-73	Fit Date 9/19/98	Fit# EB023	Fit Leg KONT -to- KDAY Corrective Action:	Delay Length 0 Hr. 19 Min.	ATA 3511
ON PREF	LT NORMAL AMO AMOUNT DOWN	OUNT PRIOR TO TO 1100 PSI.	O ENG START		OXYGEN REG PIN PUS OXY LK CK OK.	SHING EMER
Tail # N105WP	Aircraft Type DC8-73	Flt Date 9/26/98	Fit # EB029	Fit Leg KLAX -to- KDAY	Delay Length 1 Hr. 21 Min.	ATA 3311
Discrepand	cy:			Corrective Action:		
	IOKE IN COCKPIT DNSOLE AND SUIT		OM F/O'S	BY F/O'S LEG. TR. CHAP 33-1 CAT "C INSTALLED. CLO REPLACED BALL.	UELD BALLAST BURN ANSFER TO DMI #8094 TO DUE DATE 10/6/98, P SED 9/27/98 AT LAX. F AST ASSY, SYS OPS NO LEARS DMI #809410.	10 IAW MEL LACARD REMOVED ANI
Tail # N2674U	Aircraft Type DC8-73F	Flt Date 9/1/98	Fit# EB361	Fit Leg EBBR -to- KDAY	Delay Length Cancelled	ATA 2811
Discrepano	;y:			Corrective Action:		
EDGE OF PYLON H	EFILLING ACFT FO LEFT WING INBO AS FUEL LEAKING AS FUEL LEAKING	ARD #2 ENGING OUT OF IT. #	E. #1 ENGINE	AND AOG TANK S	UND AND REPAIRED I ERVICES PER MM 28 F UEL LEAK CHECK PER DEFECTS NOTED.	OSITIVE,
Tail # N2674U	Aircraft Type DC8-73F	Flt Date 9/24/98	Flt # EB102	Fit Leg KDAY -to- KPPX	Delay Length 0 Hr. 58 Min.	ATA 5351
Discrepane	y:	•	_	Corrective Action:		
Discrepanc		7124170	EBIOE		O FEL 36 WITH.	

ON PREFLT FOUND PAINT CHIPPED TO FIBERGLASS REMOVED AND REPLACED RADOME DUE TO IN NUMEROUS AREAS ON RADOME. DELAMINATION LAW MM 53. Tail# Aircraft Type Flt Date Flt# Fit Leg Delay Length ATA N500MH DC8-71F 9/5/98 EB029 KLAX -to- KDAY 0 Hr. 40 Min. 3243 Discrepancy: Corrective Action: #6 BRAKE INOP AND EVIDENCE OF HYDRAULIC LEAK. REMOVED AND REPLACED #6 BRAKE IAW MM CHAP 32. OPS AND LEAK CHECKS GOOD.

EMERY WORLDWIDE AIRLINES

	DELA	Y SUMMARY	-	
DC8 FLEET			Septe	mber 1998
Tail # Aircraft Type Flt Da	ite Fit#	Fit Leg	Delay Length	ATA
N500MH DC8-71F 9/10/9	98 EB430	KDAY -to- KLAX	0 Hr. 44 Min.	2341
Discrepancy:		Corrective Action:		
#3 MAIN FUEL QTY INDICATOR UNE 0.0 - 40.0 LBS INFLIGHT AND VERY E	RELIABLE READ ERRATIC.	DUE DATE 9-21-98 9/21/98 AT PDX. R MAIN TANK LOW	UNDER 23-11 CAT °C 3, PLACARD INSTALL: ŒINSTALLED FWD 01 ER SUPPORT, CHECK 61, PLACARD REMOV	ED. CLOSED 3 PROBE IN #3 S GOOD THIS
Tail # Aircrast Type Flt Da	te Fit#	Fit Leg	Delay Length	ATA
N500MH DC8-71F 9/15/9	8 EB439	KDAY -to- KYYZ	1 Hr. 56 Min.	5612
Discrepancy:		Corrective Action:		
F/O SIDE WINDOW WILL NOT OPEN.		F/O SIDE WINDOW	ADJUSTED, OPS NOF	IMAL.
			<u> </u>	
Tail # Aircraft Type Fit Dat	e Flt#	Flt Leg	Delay Length	ATA
N500MH DC8-71F 9/17/9	B EB011	KMCI -to- KDAY	1 Hr. 08 Min.	2811
Discrepancy:		Corrective Action:		
AT THE COMFLETION OF FUELING F FUEL WAS LEAKING FROM #4 ALT T. AREA.	OR DEPARTURE, ANK REAR SPAR		BACKSHELL TO BOOS ED 3 TURNS. OPS AND	
Tail # Aircraft Type Flt Dat	e Fit#	Fit Leg	Delay Length	ATA
N500MH DC8-71F 9/27/98	EB315	KPHL -to- KDAY	1 Hr. 34 Min.	2434
Discrepancy:		Corrective Action:		
BATTERY WILL NOT HOLD CHARGE.			PLACED MAIN AIRCE	
		BATTERY, OPS AN	D VOLT CHECK GOOD).
Tail # Aircraft Type Flt Date	: Flt #	Flt Leg	Delay Length	ATA
N500MH DC8-71F 9/28/98	EB115	KPHL -to- KDAY	1 Hr. 26 Min.	2721
Discrepancy:		Corrective Action:		
(BTB) RUDDER BINDING AND CLICK AND CONTROL MOVED.	ING WHEN TRIM	PRESSURE SEALS I CLEANED CABLES	IATION ON RUDDER ON GYRO COMPARTMI PERFORMED OPS CH NDING OPS CHECK N	ENT. ECK PER MM
Tail # Aircraft Type Fit Date	: Fit #	Fit Leg	Delay Length	ATA
N602AL DC8-73F 9/3/98	EB124	KDAY -to- KMSP	0 Hr. 59 Min.	2734
Discrepancy:		Corrective Action:		
EP WILL NOT SHOW ELEVATOR PAST WHEN CONTROL COLUMN IS MOVED	NEUTRAL FULL FORWARD.	PERFORMED OPER TABS, AND INDICA NOTED.	ATIONAL CHECK OF TORS. NORMAL, NO	•

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Tail# N602AL	Aircraft Type DC8-73F	Fit Date	Fit#	Fit Leg	Delay Length	ATA
		9/22/98	EB107	KBSM -to- KMCI	Cancelled	2761
Discrepanc	•			Corrective Action:		
L/H OB SP CHECK.	OILER DID NOT I	DEPLOY ON W	EEKEND	OUTBD CTRL AR CRANK BROKEN CRANK PER MM	ID CTRL ARM BROKE; M BROKEN; LEFT OUT REPLACED BOTH CT 27-62-1; CHECKED RIC THIN LIMITS; NO ADJ	TBD SPOILER RL ARMS AND FER MM 27-60-
Tail #	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA
N602AL	DC8-73F	9/25/98	EB004	KDAY -to- KCLT	0 Hr. 30 Min.	2752
Discrepancy	y:			Corrective Action:		
(BTB) WH WARNING	EN SELECTING F HORN CAME ON	LAPS FOR TA CONTINUOU	KE-OFF THE SLY.		ESEATED TAKE-OFF WING SYSTEM CHECKS	
Tail #	Aircraft Type	Flt Date	Flt#	FIA I	Delevit	
N604AL	DC8-73F	9/2/98	EB224	Fit Leg KDAY -to- CYMX	Delay Length 1 Hr. 22 Min.	7721
Discrepancy		,,,,,,	LULL	Corrective Action:	1 11.22 Wint.	7721
	ON'T DEVELOP R. IG EGT LIMITS O				EPLACED #4 ENGINE I OWER RUN UP CHECK	
Tail#	Aircraft Type	Flt Date	Fit#	Flt Leg	Delay Length	ATA
N604AL	DC8-73F	9/19/98	EB173	EBBR -to- KDAY	5 Hr. 20 Min.	5521
Discrepancy	: :			Corrective Action;		
RT ELEVA	TOR O/B TAB LO	WER SURFAC	E BUBBLED.	IAW MM 27-00-34.	ND FOUND TO BE IN 1 REMOVED AND REPL -30-1, OPS CHEKS OK.	.ACED R/H
Tail #	Aircraft Type	Flt Date	Flt #	Flt Leg	Delay Length	ATA
N606AL	DC8-73F	9/24/98	EB321	KDFW -to- KDAY	Cancelled	2751
Discrepancy	:			Corrective Action:		
	D ADJUST FLAP R SERVO CONTRO	· - · ·	HROW AND		O RUDDER AND ELEV S IAW MM 27-50-0 COE	
Tail # N606AL	Aircraft Type DC3-73F	Fit Date 9/29/98	Flt # EB521	Fit Leg KDRW -to- KDAY	Delay Length 3 Hr. 41 Min.	ATA 2741
Discrepancy	:			Corrective Action:		
	ЛТСН FOR WHEE HANDLE INOP (C			REMOVED AND RI SIDE. SYSTEM OP	EPLACED STAB TRIM S NORMAL.	MOTOR LH

	FI	

Tail =	Aircraft Type	Flt Date	Fit#	Fit Leg	Delay Length	ATA
N606AL	DC3-73F	9/30/98	EB004	KDAY -to- KCLT	0 Hr. 51 Min.	2731
Discrepanc	y:			Corrective Action:		
SLOWLY A	ERIENCED LIGH ACCELERATING TLY BELOW BOT MACH .77 NI 86%	THRU 302-305 TOM OF FAIR	KNOTS LAS ED POS FL 280	RIGGED ELEVATO	EPLACED ELEVATOR OR GEARED CONTROU LEVATORSERVO IAW	TABS AND
Tail#	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N791FT	DC8-73F	9/5/98	EB017	KRNO -to- KDAY	Cancelled	2925
Discrepancy	y:			Corrective Action:		
PRESSURE	ACH, WITH GEAJ E READS 2600 PSI, ING TO 2000.			BLED SPOILER RE ON GROUND. DUF GAUGE IN COCKP REMOVED AND R TRANSMITTER NO SPOILER PUMP. L REMAINS. REMO' RESERVOIR IAW N REMOVED AND R	T TOP OF ACCUMULA SERVOIR. PRESS CHE LING 5 MIN DURATION TO TOROPPED BELOW 2 EPLACED SPOILER SY DELP. REMOVED AN EAK CHECK GOOD BUVED AND REPLACED SMM 27-61-1 LEAK CHE EPLACED CASE DRAIN (S FILTER, SYS OPS A)	CKS NORMAL I PRESSURE ISOO PSI. IS PRESSURE ID REPLACED IT PROBLEM SPOILER CK OK. N AND
Tail # N792FT	Aircraft Type DC8-73F	Fit Date 9/3/98	Fit # EB017	Fit Leg KRNO -to- KSLC	Delay Length 0 Hr. 32 Min.	ATA 7721
Discrepancy	/:			Corrective Action:		
MAXED AT OTHER 3 E	OVERTEMP AFT T 913 ° EGT FOR 1 ENGINES WERE A E POWER AT 400'	10 SEC, 91.8 N BOUT 865 ° E	1 34 ° C OAT.		GINE IAW CFM 56 MM D DEFECTS NOTED.	CHAPTER 72-
Tail #	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N795FT	DC8-73F	9/9/98	EB014	KDAY -to- KBOS	3 Hr. 56 Min.	3263
Discrepancy	/ :			Corrective Action:		
LANDING ON PREFL	GEAR WARNING IGHT.	HORN DOES	NOT SOUND		EPLACED LANDING G OPS CHECKS GOOD.	EAR
Tail #	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N795FT	Dx23-73F	9/10/98	EB013	KBOS -to- KDAY	4 Hr. 37 Min.	2621
Discrepancy	y:			Corrective Action:		
FIRE AGEN	CRAFT ARRIVAL NT WARNING LIC INSPECTED FOU TLE WITH ZERO I	HTS ILLUMIN IND RIGHT W	IATED,	REMOVED AND R BOTTLE IAW MM CHECK GOOD PER	EPLACED R/H INBOAF CH 26-20-1. INSPECTI R MM CH 26-20-1.	RD FIRE ON AND

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Tail # N795FT	Aircraft Type DC8-73F	Fit Date 9/15/98	Fit # EB003	Fit Leg KCLT -to- KDAY	Delay Length 0 Hr. 36 Min.	ATA 2421
Discrepano	y:			Corrective Action:		2421
#4 BUS PC WITH EXT	OWER FAIL LIGHT FERNAL OFF.	CON AFTER EX	NGINE START	PLACED #4 GEN (INSTALLED CAT KDAY, REPLACEI M/M 24-10-00. PE UP HANDBOOK, #	#4 GEN AND OPS CHEC ON DMI #809217, PLAC "C" IAW MM 24-1. CLO D #4 GEN. & RECONNE RFORMED ENG RUN P I4 ENG OPS CKS GOOD EARS DMI # 809217, PL	ARD DSED 09/19/98 CTED CSD LAW ER EWW RUN- NO DEFECTS
Tail #	Aircraft Type	Fit Date	Fit #	Flt Leg	Delay Length	ATA
N795FT	DC8-73F	9/25/98	EB173	EBBR -to- KDAY	2 Hr. 05 Min.	2811
Discrepanc	y:			Corrective Action:		
ENG PYLO	LIGHT FOUND FU ON, INBOARD OF CENTER OF FUS A.	LEFT MAIN GE	AR DRIPPING	NORMAL. CLEAN RESIDUAL FROM REPLACED DAMA	T TO SEEPING SCREW TED AREA, NO LEAKS PANEL LEAK ON TOP GED O-RING AT #1 M ER, LEAK CHECKS OK	NOTED. OF WING. AIN TANK
Tail #	Aircraft Type	FIt Date	Flt #	Fit Leg	Delay Length	ATA
N796AL	DC8-63	9/3/98	EB118	KDAY -to- KEWR	0 Hr. 57 Min.	5234
Discrepancy	γ:			Corrective Action:		
MAIN CAR INTO POSI	GO DOOR AFT LOTION.	OCKS WILL NO	OT LOCK	REMOVED AND R STC SA1802X0 OPS	EPLACED AFT TORQU S CHECK GOOD.	E TUBE IAW
Tail #	Aircraft Type	Flt Date	Flt #	Fit Leg	Delay Length	ATA
N796AL	DC8-63	9/10/98	EB335	KORD -to- KDAY	Cancelled	7721
Discrepancy	<i>r</i> :			Corrective Action:		
° EGT. RE	ON ROTATION FROM LRD #3 ENG EGT EXCEEDED 590 * EGT. RETARDED POWER TO 1.75 EPR TO MAINTAIN EGT LIMITS.				AND ACFT FOR FERR THIS ACFT AND FOUN TENDED. REMOVED A FOR MEO 77. LK AND NIR Y-0004 AND MEO	VD IT TO BE ND REPLACED OPS CHECK
Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N796AL	DC8-63	9/15/98	EB017	KRNO -10- KDAY	0 Hr. 26 Min.	3441
Discrepancy	<i>י</i> :			Corrective Action:		
	REEN PULSATES THE GAIN WILL				EPLACED RADAR TRA	NSCEIVER

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Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N796AL	DC8-63	9/15/98	EB017	KRNO -to- KDAY	0 Hr. 30 Min.	3131
Discrepanc	•			Corrective Action:		
	FER BLOCK OUT P WITH SHIPS PO		RDER PITOT	CYCLED CIRCUIT POWER RELAY ST POWER OPS CHEC	BREAKERS FOUND F FARTED ENGINE #4 A CKS NORMAL.	FOR EXTERNAL ND PUT ON A.C
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N796FT	DC8-73F	9/15/98	EB379	KLAX -to- KDAY	Cancelled	5752
Discrepancy	<i>/</i> :		1	Corrective Action:		
FLAP BUS	CABLE BRACKE	T BROKEN RA	H W/W.	REPLACED BRAKI RIGGED FLAP SYS CHECKS GOOD.	E AND CABLES FOR F ITEM AND RUDDER S	LAP BUS AND YSTEM, ALL
Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N797AL	DC8-63	9/2/98	EB023	KDAY -to- KRDU	Cancelled	7712
Discrepancy	r:	•		Corrective Action:		
#1 AND #4 O'CLOCK F	ENG EXHAUST E POSITION.	OUCT CRACKE	D BRACKET 6		#4 IN OUT BI-DUCTS OSITION LAW DC-8 SRI	
Tail # N797AL	Aircraft Type DC8-63	Fit Date 9/4/98	Flt# EB020	Fit Leg KDAY -to- KLRD	Delay Length 1 Hr. 41 Min.	ATA 5234
Discrepancy	:			Corrective Action:		
MAIN CAR	GO DOOR WILL 1	NOT CLOSE.		OF IMPROPER SILI CARGO DOOR LAT DOOR LOCK ACTU	ILL GUARD SUPPORT GUARD REMOVAL) CH SPOOL, CAUSING IATOR, REMOVED AN ED DOOR SEVERAL TI	LODGED IN FAILURE OF D REPLACED
Tail # N797AL	Aircraft Type DC8-63	Flt Date 9/5/98	Flt ∉ EB021	Fit Leg KDFW -to- KDAY	Delay Length 1 Hr. 58 Min.	ATA 7721
Discrepancy:	:			Corrective Action:		
READS 45 F ENGINE EG READS 39 F EGT AT IDI PSI, OIL TE	EGT AT IDLE RE PSI OIL TEMP RE. FT AT IDLE READ PSI, OIL TEMP 71 LE READS 360 ° C MP 69 ° C, OAT 4 S 350 ° C, OIL PR C, OAT 41 °.	ADS 70°C OA PS 360°, OIL P! °C, OAT 41°. ', OIL PRESSUI 1°. #4 ENGIN	T 41°. #2 RESSURE #3 ENGINE RE READS 44 E EGT AT		'D FOR FLIGHT BY M. ITT & WHITNEY MM	

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Tail # N797AL	Aircraft Type DC3-63	Fit Date	Fit#	Fit Leg	Delay Length	ATA		
		9/5/98	EB322	KDAY -to- KDFW	0 Hr. 25 Min.	7611		
Discrepani	•			Corrective Action:				
#2 ENGIN	E THROTTLE IS VIENT.	ERY STIFF CO	OMING OUT OF	ADJUSTED FRICT	TON LOCK ARM, OP C	HECKS GOOD.		
Tail#	4 incom 9 Tr	51. 5						
N797AL	Aircraft Type DC8-63	Fit Date 9/17/98	Fit # EB234	Fit Leg KDAY -to- KDFW	Delay Length	ATA		
Discrepano		3,1,1,50	LD234		Cancelled	7111		
•	•			Corrective Action:				
CRACKE	FLT FOUND #1 EN D AND BROKEN. C INLET RAMP CRA	N POST FLT	RAMP FOUND #2	ASSY. NO OTHER GOOD. REMOVED	EPLACED #1 ENGINE: DEFECTS NOTED, OP DAND REPLACED #2 E OTHER PROBLEMS NO	S CHECKS ING NOSE		
Tail#	Aircraft Type	Flt Date	Fit#	Flt Leg	Dalay Langth			
N801GP	DC8-71F	9/2/98	EB440	KDAY -to- KYYZ	Delay Length Cancelled	ATA 2755		
Discrepanc	y:	••		Corrective Action:		2733		
FLAPS INI	W/FLAPS SELECTED FULL DOWN ON APP AT REF +5 FLAPS INDICATED 40 ° ON ROLL OUT FLAPS INDICATED 45 °.				WHILE T/S FOUND INBD AND MIDWING LOCKOUT CYLINDER BYPASSING AND STA 98 FLAP ACTUATION BYPASSING. REMOVED AND REPLACED INBD MIDWING FLAP LOCKOUT AND STA 98 FLA ACTUATION. OP AND LEAK CHECKS GOOD.			
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA		
N8076U	DC8-71F	9/12/98	EB003	KCLT -to- KDAY	0 Hr. 28 Min.	3311		
Discrepancy	<i>y</i> :			Corrective Action:				
ALL AC LO VOLTS, FR	DAD AND DC LOAI EQ GAUGE LIGHT	D, CSD TEMP S ON FE PAN	, AC-DC EL INOP.	09/22/98, PLACARD DAY. TIGHTENED LIGHTS SW OPS CH	EL 33-1 CAT "C" DMI # INSTALLED. CLOSEI KNOB ON F/E INST PA IECKS GOOD IAW DCS 8 PLACARD REMOVE	0 09/12/98 AT WEL WHITE 3 MM. THIS		
Tail #	Aircraft Type	Fit Date	Fit #	Flt Leg	Delay Length	ATA		
N8076U	DC8-71F	9/16/98	EB323	KMSP -to- KDAY	1 Hr. 42 Min.	3234		
Discrepancy	:			Corrective Action:				
	DING GEAR DOOR LEFT SIDE).	COMES OPE	N ON	LUBED LEFT MAIN OP CHECKS GOOD	GEAR DOOR LATCH PER MM CH 32.	& ACTUATOR,		
Tail # N8076U	Aircraft Type DC8-71F	Flt Date	Fit#	Fit Leg	Delay Length	ATA		
_		9/16/98	EB124	KDAY -to- KMSP	0 Hr. 48 Min.	3234		
Discrepancy	:			Corrective Action:				
GEAR UPLA HOLD DOO	ATCH ON LEFT SIE R CLOSED ON GR	DE OF MAIN V OUND.	WILL NOT		OPS CHECKS GOOD TI COULD NOT DUPLICA			

	DELAY SUMMARY
DC8 FLEET	
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Tail#	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA
N8076U	DC8-71F	9/25/98	EB380	KDAY -to- KLAX	0 Hr. 23 Min.	3248
Discrepand	cy:			Corrective Action:		
READS 70	BRAKE ACCUMUL 30 PSI WITH EITHE SURE ON SYSTEM	R BRAKES PU	URE GAUGE JMPED DOWN	#809405 PLACARI DAY. SERVICED	O DMI LIST MEL# 32-1. D INSTALLED. CLOSEI ACCUMULATOR, LEAI ARS DMI #809405, PLA	O 9/25/98 AT K & OP CHECK
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N8079U	DC8-71F	9/3/98	EB342	KDAY -to- KEWR	Cancelled	3441
Discrepano	ey:			Corrective Action:		
(ATB) AT	FL31.0 RADAR PA	UNTS GROUN	D ON RT SIDE.	DATE 9-13-98 PLA AT KDAY. REMO ANTENNA MOTOR RADAR SYS OPS C	9042 IAW MEL34-21 CA CARD INSTALLED. CL VED AND REPLACED R DRIVE ASSY WMA 70 CHECKS GOOD IAW DO 42 PLACARD REMOVE	OSED 09-03-98 RADAR NC. WEATHER -8 MM THIS
Tail #	Aircraft Type	Flt Date	Flt#	Fit Leg	Delay Length	ATA
N8079U	DC8-71F	9/5/98	EB341	KDAY -to- KEWR	Cancelled	7311
Discrepano	;y:			Corrective Action:		
SHUTDOV	DUCHDOWN #1 EN VN UNTIL N2, N1 S CONDITIONS NORM	POOLED DOV	VN, ALL		EPLACED R/H FUEL M. OPS CHECK NORMAL	
Tail # N8079U	Aircraft Type DC8-71F	Fit Date 9/10/98	Fit# EB314	Flt Leg KDAY -to- KBOS	Delay Length 1 Hr. 51 Min.	ATA 2725
Discrepanc	y:			Corrective Action:		
SPOILER TO ON EXTER	ITCHING TO SHIPS TO NORMAL POILI RNAL POWER SEL IP, PUMP OPS CHE	ER PUMP C/B ECTING OVER	POPS WHEN		EPLACED SPOILER SYS OPS CHECKS GOOD LE	
Tail #	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA
N3084U	DC8-71F	9/4/98	EB524	KROC -to- KDAY	Cancelled	2611
Discrepanc	y:			Corrective Action:		
INTERMIT	E FIRE WARNING ITENTLY WITH EN VN FIRE WARNING	IGINE RUNNI	NG OR	FIRE LOOPS AND I FIRE LOOPS. #2 ET GOOD STATIC AN	EPLACED R/H AND L/H R/H AND L/H LOWER C NGINE FIRE WARNING D ON HIGH POWER RU .OSED AND SECURED	ORE COMPT OPS CHECKS NIAW MM

DC8 FLEET				- COMMANI	September 1998		
Tail # N3087U	Aircraft Type DC8-71F	Flt Date 9/16/98	Fit # EB259	Fit Leg KDAY -to- KMKE	Delay Length 0 Hr. 34 Min.	ATA 2515	
Discrepano	:y:			Corrective Action:		-512	
CAPTAIN	S SEAT WILL NO	T ADJUST UP	AND DOWN.	REMOVED AND R CHECK GOOD.	REPLACED CAPTAIN'S	SEAT OPS	
Tail # N8091U	Aircraft Type DC8-71F	Fit Date 9/24/98	Fit # EB380	Fit Leg KDAY -to- KLAX	Delay Length 0 Hr. 29 Min.	ATA 3233	
Discrepanc	y:			Corrective Action:			
ON POST I LEAKING	LIGHT FOUND L HYDRAULICS.	EFT MLG ACT	TUATOR	REMOVED AND REPLACED LEFT MLG ACTUATOR SYSTEM, OPS CHECK AND LEAK CHECK GOOD IAW DC8 MM 32-31-19.			
Tail # N8091U	Aircraft Type DC8-71F	Fit Date 9/29/98	Flt # EB002	Fit Leg KMHR -to- KDAY	Delay Length 0 Hr. 29 Min.	ATA 2621	
Discrepancy	r.	•		Corrective Action:			
UPON GEA	R EXTENTION L	EFT AGENT LO	OW PRESSURE	REMOVED AND RECHECKS OK PER M	EPLACED L/H I/B FIRE	BOTTLE. OP	
Tail # N811AL	Aircraft Type DC8-71F	Fit Date 9/1/98	Fit# EB017	Fit Leg KSLC -to- KDAY	Delay Length 0 Hr. 30 Min.	ATA 2912	
Discrepancy	•			Corrective Action:			
HYDRAUL! ON POST F	IC FLUID LEAK N LIGHT IN SLC	OSE GEAR AF	REA FOUND	RESEALED ONE FI VALVE NO LEAKS FULL	ITING ON NOSE GEAI NOTED HYD SYSTEM	R CONTROL I SERVICED TO	
Tail # N811AL	Aircraft Type DC8-71F	Flt Date 9/4/98	Fit# EB011	Fit Leg KMCI -to- KDAY	Delay Length 1 Hr. 07 Min.	ATA 7331	
Discrepancy:	:	•		Corrective Action:			
REF: DMI #308472 #2 F/F INOP.				REMOVED AND REPLACED @3 F/F AMPLIFIER. OPS CHECKS GOOD ON ENGINE RUN. THIS CLEARS DMI #808472. PLACARD REMOVED.			
Tail # N811AL	Aircraft Type DC8-71F	Flt Date 9/22/98	Fit # EB043	Fit Leg KLAX -to- KDAY	Delay Length 1 Hr. 20 Min.	ATA 3611	
Discrepancy:				Corrective Action:			
#2 ENGINE	HIGH STAGE OPE OUT CRUISE.	N LIGHT REN	IAINS ON	REMOVED AND RE	PLACED H.S. VALVE. RMAL PER MM 36-00		

EMERY WORLDWIDE AIRLINES

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DC8	FLEET				Septe	mber 1998
Tail # N311AL	Aircraft Type DC8-71F	Flt Date 9/27/98	Flt # EB152	Fit Leg KDAY -to- KMEX	Delay Length 2 Hr. 19 Min.	ATA 2421
Discrepan	cy:			Corrective Action:		
LIGHT O	LIGHT FOUND #4 JT AND NO PRESS LLELED OR BUS F	S TO TEST FEA	ATURE ON	GENERATOR CON	OR P1-1323 LOOSE OF OTROL PANEL. TIGHT O PERFORMED ENGIN OD.	ENED
Tail # N8177U	Aircraft Type DC8-71F	Fit Date 9/1/98	Fit# EB137	Fit Leg KYYZ -to- KDAY	Delay Length Cancelled	ATA 7721
Discrepand	y:			Corrective Action:		
#3 ENGIN STABLE	E EGT AND FUEL	flow errat	TIC, NI AND N2	MAIN ENG FUEL F R/R LOW PRESS F	AM 73-11-10 ENG RUN PUMP AS PER MM 73-1 UEL PUMP AS PER MA CIT SENSORS ENGINE	11-10 NO FEX. 4 73-11-10.
Tail # N8177U	Aircraft Type DC8-71F	Flt Date 9/4/98	Flt # EB019	Flt Leg MMMY -to- KDAY	Delay Length Cancelled	ATA 3423
Discrepano	: v :			Corrective Action:		
FIRST OF	FICERS ADI HAS C IMES.	3 AND FD FLA	GS IN VIEW		EPLACED #2 VERTICA S CHECKED NORMAL	
Tail # N8177U	Aircraft Type DC8-71F	Fit Date 9/24/98	Fit # EB038	Flt Leg KDAY -to- KATL	Delay Length 0 Hr. 22 Min.	ATA 2841
Discrepano	y:			Corrective Action:		
	QTY FUEL GAUGE C AT 15,000.	SHOWS 5,000	LBS WITH	DUE DATE 10-4-98 9/25/98 AT MSP. SU 28 CHECKED DRIP READING 7000 LBS	QI DMI 809379 MEL 28: PLACARD INSTALLE! JMPED #4 MAIN TANI STICK INBD READS 2 6 OP CHECKS NORMA 19 PLACARD REMOVE	D. CLOSED K LAW MM CI 1.5 INDICATO L. THIS
Tail # N8177U	Aircraft Type DC8-71F	Fit Date 9/30/98	Fit# EB328	Fit Leg KMSP -to- KDAY	Delay Length 2 Hr. 03 Min.	ATA 7721
Discrepanc	y:			Corrective Action:		
	E EGT IS INDICAT AND CRUISE ALL			REMOVED AND RI CHECKS GOOD IA	EPLACED #3 LH T HAF W CFM MM 77.	RNESS. OPS
NORMAL.						
NORMAL. Tail # N870TV	Aircraft Type DC3-73F	Fit Date 9/10/98	Fit # EB 173	Fit Leg EBBR -to- KDAY	Delay Length 3 Hr. 20 Min.	ATA 2841

SUMPED TANK CENTER WING AUX FUEL QTY IND OPS CHECKS GOOD.

CTR WING AUX FUEL GUAGE ROLLS FROM MIN TO MAX UNIT EMPTY. IST AUX FUEL GAUGE STUCK AT 14850 WHEN FILLED TO MAX 11000 LBS AND ONCE TRANSFER STARTS GAUGE DOESN'T WAVE.

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Tail # N870TV	Aircraft Type DC8-73F	Flt Date 9/23/98	Flt # EB223	Fit Leg KYMX -w- HDAY	Delay Length Cancelled	ATA 2811
Discrepan	icy:			Corrective Action:		2311
#3 ENGIN	7E FUEL LEAK ON	ENGINE STAI	RT.	REPLACED "O" RI	NG AT FUEL FLOW TO ND RUN ENGINE NO I	RANSMITTER LEAKS.
Tail# N921R	Aircraft Type DC8-63F	Fit Date 9/16/98	Fit# EBI17	Fit Leg KEWR -to- KDAY	Delay Length Cancelled	ATA 7721
Discrepan	cy:			Corrective Action:		
	BORTED TAKE-OFF NO MORE THAN		4 ENGINE EGT	ENGINE INOP IAW PERFORMED ENG COMBUSTION CH MANUAL 72-00-00 AND REPLACED # PERFORMED ENG	OR FERRY FLIGHT WITE EWA JOB CARD MECTON BORESCOPE ON A AMBERS AND HPT IA NO DEFECTS NOTED A ENGINE IAW DC-8 MINE RUN ADJ/RS4 LAW DD, LK CHECK GOOD	D30 METHOD 3 44 ENGINE W JT-3-7 D. REMOVED M 71-00-00, V DC-8 MM 71-
Tail # N950R	Aircraft Type DC8-63	Fit Date 9/17/98	Fit # EB316	Fit Leg KDAY -to- KPHL	Delay Length 0 Hr. 12 Min.	ATA 2311
Discrepano	cy:			Corrective Action:		
C-PIT AU ON TAXI	DIO PANEL C/B OP OUT.	ENED WOULI	D NOT RESET	REMOVED AND RECHECKS GOOD.	EPLACED FO'S C-PIT S	SPEAKER, OPS
Tail # N930R	Aircraft Type DC8-63	Flt Date 9/22/98	Fit # EB020	Fit Leg KDAY -to- KLRD	Delay Length Cancelled	ATA 7112
Discrepand	ey:			Corrective Action:		
L/H AFT E	FLIGHT WALKAR(BI-DUCT CRACKED MENT POINT.				IG FASTENERS INST E OPS CHECKS GOOD.	
Tail # N950R	Aircraft Type DC8-63	Flt Date 9/24/98	Fit # EB123	Fit Leg KMSP -to- KDAY	Delay Length 3 Hr. 19 Min.	ATA 7112
Discrepano	y:			Corrective Action:		
#1 ENGIN	E OUTBOARD FW	BI-DUCT DAN	IAGED.	FWD BI-DUCT IAW	EPLACED #1 ENGINE (DC8 MM CH 78, OPS D PART FROM N998CH	CHECK
Tail # N951R	Aircraft Type DC8-63	Flt Date 9/2/98	Flt # EB521	Fit Leg KMKE -to- KDAY	Delay Length 6 Hr. 35 Min.	ATA 7611
Discrepand	ıy:			Corrective Action:		
#! THROT	TTLE BINDS DURIN	IG CLIMB.			: AT THROTTLE QUAI S THROTTLE DOES N OTED.	

DC8	FLEET				September 1998		
Tail # N951R	Aircraft Type DC8-63	Flt Date 9/16/98	Flt # EB137	Fit Leg	Delay Length	ATA 7831	
Discrepan	ıcy:			Corrective Action:	7 12 Ivim.	7331	
	NE THRUST REVES		E HANDLE	#809237 CAT "C" I KRNO. INSTALLE	O DMI AS PER MEL 78 DUE 9/26/98. CLOSED ED SERVICEABLE THE DLE. OPS CHECK GOO ARS DMI #809237.	9/26/98 AT RUST	
Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA	
N957R	DC8-63F	9/4/98	EB028	KDAY -to- KRDU	1 Hr. 06 Min.	3441	
Discrepan	cy:			Corrective Action:			
	X RADAR WOULD MODE TEST PATTE				CHECK OF WEATHER ECTS NOTED SYSTEM IM.		
Tail # N957R	Aircraft Type DC8-63F	Fit Date 9/9/98	Flt # EB003	Flt Leg KCLT -to- KDAY	Delay Length 0 Hr. 36 Min.	ATA 7831	
Discrepano	cy:			Corrective Action:		•	
	ER HYD INDICATO PRESSURE.	R INDICATES	GENERAL	SYSTEM FOUND T TOUCHING - SEPA STOW CATCH SWI	IRUST REVERSER / AL WO WIRING TERMIN, RATED ON THRUST R ITCH SYSTEM OPS CH 3 MM 78-30-1 REF ACM	AL EVERSER ECKS	
Tail # N957R	Aircraft Type DC8-63F	Fit Date 9/9/98	Fit # EB324	Flt Leg	Delay Length	ATA	
Discrepano		9/9/96	EB324	KDAY -to- KMSP Corrective Action:	1 Hr. 14 Min.	7531	
	E STALLS AND WO		OOL UP	REMOVED AND REPLACED #1 ENGINE PRBC OPS CHECK GOOD ON GROUND RUN.			
Tail # N957R	Aircraft Type DC8-63F	Fit Date 9/9/98	Fit # EB599	Fit Leg KMCN -to- KDAY	Delay Length Cancelled	ATA 2926	
Discrepano	;y:			Corrective Action:			
REVERSE OPERATII	R PUMP INDICATI NG	ES 2500 PSI WI	HEN		EPLACED THRUST RE P ASSY IAW DC MM 78 IAL, 3000 PSI.		
						· · · · · · · · · · · · · · · · · · ·	
Tail# N957R	Aircraft Type DC8-63F	Flt Date 9/12/98	Flt # EB138	Fit Leg KDAY -to- KYYZ	Delay Length 0 Hr. 00 Min.	ATA 3245	

REPLACED TERMINAL AT LEFT MAIN ANTI-SKID

TERMINAL BLOCK AND TIGHTENED ALL NUTS ON TERMINAL BLOCK. PERFORMED REJECTED TAKE-

OFF INSPECTION IAW WORK CARD SP006 AND SP001.

(BTB) AT THE INITIATION OF THE TAKE-OFF ROLL

(APPROX 40 KTS) THE ANTI-SKID INOP LIGHT CAME ON WITH THE LFO INDICATING A MANUAL

CONDITION, REJECTED THE TAKE-OFF.

September 1998

	DELAY SUMMARY
DC8 FLEET	

Tail # N957R	Aircraft Type DC3-63F	Flt Date 9/25/98	Fl: ≠ EB439	Fit Leg KYYZ -to- KDAY	Delay Length 0 Hr. 31 Min.	ATA 2741
Discrepan	icy:		25437	Corrective Action:	5 11. 51 Will.	2141
	UDINAL TRIM CIR OT RESET.	CUTT BREAK	ER POPPED,	STAB TRIM SERV	/O ACTUATOR REPLA I CHECKED SERVICEA	
Tail # N959R	Aircraft Type DC8-63F	Fit Date 9/1/98	Flt # EB038	Flt Leg KDAY -to- KATL	Delay Length Cancelled	ATA 2811
Discrepan	cy:			Corrective Action:		2011
FUEL LEZ LEAK FR	AKING FROM AFT OM AFT END OF #	END OF #4 PY I PYLON.	YLON. FUEL	FUEL LEAKS JUS & 474.000 LEAK C AREA FOUND NO	NED AND RESEALED I FWD OF MID SPAR A HECK OK. INSPECTEI EVIDENCE OF LEAK I RWIND PANELS NO LE	T XRS542.000 D#1 PYLON N PYLON.
Tail # N959R	Aircraft Type DC8-63F	Flt Date 9/26/98	Flt # EB332	Fit Leg KDAY -to- KFLL	Delay Length 1 Hr. 06 Min.	ATA 2741
Discrepano	cy:			Corrective Action:		
FOUND S FORWAR	TAB SUITCASE HA .D.	ANDLE STICK	ING		LONGITUDINAL TRINCK GOOD IAW MM CH	
Tail # N959R	Aircraft Type DC8-63F	Flt Date 9/30/98	Flt # EB316	Fit Leg KDAY -to- KPHL	Delay Length Cancelled	ATA 2811
Discrepano	cy:			Corrective Action:		
ON PREFI	LT FOUND FUEL D LT FOUND FUEL D .EA FWD OF BOTH	RIPPING FRO	M CENTER		EPLACED DEFECTIVE C CHECK GOOD IAW M	
Tail# N961R	Aircraft Type DC8-73F	Fit Date 9/1/98	Flt # EB173	Fit Leg EBBR -to- LEMD	Delay Length Cancelled	ATA 77211
Discrepano	y:			Corrective Action:		
	VERTEMP ON TAK D POWER 5% TO M				ND IAW 77-10 GROUNE UP PERFORMED PER	
Tail # N964R	Aircraft Type DC8-63	Flt Date 9/1/98	Fit # EB037	Fit Leg KATL -to- KDAY	Delay Length Cancelled	ATA 2611
Discrepand	ıy:		•	Corrective Action:		
	ENGINE FIRE WAI RED BOTH BOTTL				EPLACED SHORTED F IR DC-8 M/M CHAP 26-	

	FLEET			SUMEMLARY	Septe	mber 1998
Tail #	Aircraft Type	Flt Date	Flt#	Flt Leg	Delay Length	ATA
N964R	DC8-63	9/2/98	RB037	KATL -10- KDAY	3 Hr. 00 Min.	7721
Discrepano	cy:			Corrective Action:		
	URING TAKE-OFF DURING TAKE-OFF		T EXCEED	REMOVED AND R ENGINE CHANGE	EPLACED #4 ENGINE PAPERWORK, OPS C	PER MEO77 HECK NORM
Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N991CF	DC8-54F	9/9/98	EB106	KROC -to- KDAY	Cancelled	2844
Discrepano	cy:			Corrective Action:		
OIL AND	DESCENT CIRCUIT FUEL PRESSURE I B AND POPPED AC	NDICATORS F	OPPED,	PRESSURE WORK: FUEL PRESSURE I MEL 28-8 DMI# 809 PLACARDS INSTA REPAIRED SHORT PYLON HARNESS	IN FUEL PRESSURE S S NORMALLY, TRANS NDICATOR TO DMI LI 9113 DUE 9-19-98 FAA LLED. CLOSED 09/09/ ED WIRE 3E12D24 PIN DISCONNECT, OPS CH 113, PLACARD REMO	FERRED #3 ST AS PER CAT "C" '98 AT ROC. I D AT R5-36 ÆCK GOOD
Tail#	Aircraft Type	Flt Date	Flt#	Flt Leg	Delay Length	ATA
N991CF	DC8-54F	9/13/98	EB314	KDAY -to- KROC	0 Hr. 26 Min.	7831
Discrepano	ey:			Corrective Action:		
ENGINE V	JRING TAKE-OFF P WAS POWERED UP OR LIGHT CAME ON	TO 1.5 EPR #3		REATTACHED #2 POPS CHECK GOOD	ENGINE T/R DOOR LIC ON ENGINE RUN.	энт ѕwітсн,
-	Aircraft Type	Flt Date	Flt#	Flt Leg	Delay Length	ATA
Tail #						AIA
	DC8-54F	9/17/98	EB106	KDAY -to- KROC	2 Hr. 02 Min.	7931
N991CF		9/17/98		-	• •	
N991CF Discrepanc			EB106	KDAY -to- KROC Corrective Action: TRANSFERRED TO: "B" ASSIGNED CO! INSTALLED. NOT! VERIFIED BEFORE KMCN. REMOVED TRANSMITTER IA! CHECK GOOD IAW	• •	7931 MEL 79-1 CAT ARDS JALLY SED 9/22/98 A' DIL QNTY 11-1, OPS 1-1 REF ACMI
N991CF Discrepance (BTB) DU GAL.	ey: JRING TAXI #2 OIL Aircraft Type	QTY DROPPE	EB106 ED BELOW 6 Fit #	KDAY -to- KROC Corrective Action: TRANSFERRED TO "B" ASSIGNED COI INSTALLED. NOTE VERIFIED BEFORE KMCN. REMOVED TRANSMITTER IAV CHECK GOOD IAW N/R Y0015. THIS CI	2 Hr. 02 Min. DDMI LIST IAW EWA INTROL #809262, PLACE: QTY MUST BE VISUE ACH FLIGHT. CLOS AND REPLACED #2 CW DACO DC8 MM 79-31 LEARS MEL/CDL 8092 Delay Length	7931 MEL 79-1 CAT ARDS JALLY SED 9/22/98 A' OIL QNTY 61-1, OPS -1 REF ACMI 62. ATA
N991CF Discrepanc (BTB) DU GAL.	ey: IRING TAXI #2 OIL	, QTY DROPPE	EB106	KDAY -to- KROC Corrective Action: TRANSFERRED TO "B" ASSIGNED COI INSTALLED. NOTE VERIFIED BEFORE KMCN. REMOVED TRANSMITTER IAV CHECK GOOD IAW N/R Y0015. THIS CI	2 Hr. 02 Min. DDMI LIST IAW EWA INTROL #809262, PLACE: QTY MUST BE VISUE ACH FLIGHT. CLOS AND REPLACED #2 CW DACO DC8 MM 79-31 LEARS MEL/CDL 8092	7931 MEL 79-1 CAT ARDS JALLY SED 9/22/98 A' OIL QNTY 61-1, OPS -1 REF ACMI 62.

SYS, OPS CHECK AND LEAK CHECK GOOD IAW DC8

MM 32-31-19.

LEAKING HYD.

EMERY WORLDWIDE AIRLINES

DC8	FLEET			SUMMARY		mber 1998
Tail# N991CF	Aircraft Type DC8-54F	Fit Date 9/23/98	Fit # EB266	Fit Leg KMEM -to- KDAY	Delay Length Cancelled	ATA 2912
Discrepanc	y:			Corrective Action:		
HYDRAUI LANDING	LIC FLUID LEAK GEAR DOORS.	FOUND BETW	EEN MAIN	& SPOILER SYS RE	ITY VALVE TO RT H EVERSE FLOW CK VA LEAK CHECK GOOD.	LVE ON RT
Tail # . N992CF	Aircraft Type DC8-54F	Flt Date 9/4/98	Fit # EB106	Fit Leg KDAY -to- KROC	Delay Length 0 Hr. 23 Min.	ATA 5231
Discrepanc	y:			Corrective Action:		
B PIT DOC	R WONT SHUT O	FF BELLY DO	OR LIGHT.	2-99 I PLACARD IN REMOVED AND RE	L 33-3 CAT D CONTR STALLED. CLOSED 9 EPLACED STRIKER, C ARS DMI #809071, PL	9/5/98 AT ROC. PPERATIONAL
Tail # N992CF	Aircraft Type DC8-54F	Flt Date 9/16/98	Flt # EB040	Flt Leg KDAY -to- KMEM	Delay Length 0 Hr. 19 Min.	ATA 8011
Discrepancy	/ :			Corrective Action:		
25% N1 GR	ENG #2 DID NOT COUND CREW AD DISCONTINUED	VISED OF EXT	TERNAL FIRE		RE, NONE FOUND. M S STARTED NORMAL D.	
Tail # N992CF	Aircraft Type DC8-54F	Flt Date 9/17/98	Flt # EB040	Fit Leg KDAY -to- KMEM	Delay Length 0 Hr. 22 Min.	ATA 3611
Discrepancy	r:			Corrective Action:		
ENG#1 & # CLIMB PO' ON LEFT N OFF MANTI	OFF POWER ENG 22 MANIFOLD PR WER ENG RELIEF LANIFOLD PRESS FOLD PRESS 38 P TI-ICE CIRCUIT F	ESS 48 PSI BO VALVE LIGH 44 PSI, WITH SI. ON POSTFI	TH SIDES. AT T FOR ENG #2 #2 BLEED LIGHT FOUND	REGULATOR OPS O NO LEAKS OR DEFI PLUG AND MELTEI	PLACED #2 PNUEMA HECSK GOOD ON EN ECTS NOTED. REPLA D WIRING ON #2 A/I N ECKS GOOD ON GROU OTED.	IG RUN #2 ENG CED CANNON IOSE COWL.
Tail # N993CF	Aircraft Type DC3-62	Flt Date 9/16/98	Flt # EB350	Fit Leg KEWR -to- KMCO	Delay Length 0 Hr. 47 Min.	ATA 2411

Corrective Action:

REMOVED AND REPLACED #1 CSD OIL TEMP IND AND ENG TEMP BULB. OPS CHECKS NORMAL. ACCOMPLISHED ENGINE RUN.

Discrepancy:

FREQ'S NORMAL.

#1 CSD OUTLET TEMP ERRATIC. FLUCTUATES 60-140 RAPIDLY AND INTERMITTENTLY. RISE TEMP FLUCTUATES OCCASIONALLY 10-20 °. VOLTS AND EDGO'S NORMAL

EMERY WORLDWIDE AIRLINES

		E		RLDWIDE AIRLINI	23	
DC8	FLEET		DELA	Y SUMMARY	Sente	mber 1998
	···				Осрес	111001 1998
Tail #	Aircraft Type	Flt Date	Flt#	Flt Leg	Delay Length	ATA
N993CF	DC8-62	9/13/93	EB350	KEWR -to- KMCO	1 Hr. 19 Min.	3441
Discrepano	ry:			Corrective Action:		
TILTS LE	#809282 RADAR V FT, RIGHT AND D NT WORKS WITH	OWN WITH A	I RADAR NY A/C	ANTENNA, SYS O & R IAW MM 34-4 TEST EQUIPMENT 9/25/98 AT KMCO. NORMAL ON GRO	EPLACED WEATHER PS CHECKS NORMAL I-I. STAB REQS OPER NOT AT STA, DMI OF SWAPPED VG'S OPS OUND, TIS CLEARS ME RD CONTROL #80928	ON GROUND ATIONAL TES PEN. CLOSED CHECKS IL 34-21.
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N993CF	DC8-62	9/19/98	EB228	KDAY -to- KMCO	0 Hr. 22 Min.	2321
Discrepane	y:			Corrective Action:		
(BTB) F/O	'S RADIO INOP.			JACK #2 COMM A2	NOT TO BE SEATED, ND #1 COMM FO'S SID ED AND REPLACED # DIAW DC8 MM.	E OPS CHECK
Tail #	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA
N993CF	DC8-62	9/27/98	EB310	KDAY -to- KDEN	4 Hr. 10 Min.	7111
Discrepancy	y:			Corrective Action:		
FOUND ON ASSEMBL	V PREFLIGHT, #1 Y LOOSE.	ENGINE NOSE	BULLET	MOUNTING STUB	NOSE BULLET ASSEN BROKEN, REMOVED A OPS CHECK GOOD.	
Tail #	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N994CF	DC8-62	9/5/98	EB024	KDAY -to- KONT	0 Hr. 00 Min.	5234
Discrepancy	/ :			Corrective Action:		
	OT VERIFY CARG L VIEW PORT.	O DOOR LOCK	CED FROM		LOSED, LATCHED AN , NO DEFECTS NOTED	
Tail # N994CF	Aircraft Type DC8-62	Flt Date 9/9/98	Flt# EB023	Fit Leg KONT -to- KDAY	Delay Length 0 Hr. 46 Min.	ATA 5270
Discrepancy	: :			Corrective Action:		
	OR OPEN LIGHT I	ILLUMINATEI	WITH ALL		TRIKER OPS CHECKS	GOOD, LIGH
Tail # N994CF	Aircraft Type DC8-62	Fit Date 9/30/98	Flt # EB108	Fit Leg KDAY -to- KBSM	Delay Length 0 Hr. 23 Min.	ATA 7713

Corrective Action:

PERFORMED HIGH POWER RUN ON #1 ENG ALL PARAMETERS CHECK GOOD.

Discrepancy:

RETARDED #1 POWER LEVER .03 EPR BELOW MCT POWER SETTING TO MAINTAIN EGT BELOW LIMITS ON CLIMB WITH REDUCED POWER T/O.

DC3 F	LEET		DELAI	SUMMARY	September 1998		
Tail ≠ N996CF	Aircraft Type DC8-62F	Flt Date 9/11/98	Flt # EB228	Fit Leg KDAY -to- KMCO	Delay Length 0 Hr. 59 Min.	ATA 7230	
Discrepancy	/:			Corrective Action:			
• •	OMPRESSOR ST D AT BRAKE RE		WER WAS	DEBRIS AND PERI	INLET OF #3 ENGINE FORMED ENGINE FOD ED. ACFT OK FOR CO	INSPECTION	
Tail # N997CF	Aircraft Type DC8-62F	Fit Date 9/10/98	Fit # EB117	Fit Leg KEWR -to- KDAY	Delay Length 2 Hr. 52 Min.	ATA 3424	
Discrepancy	: :			Corrective Action:			
#1 CAPT C	RS SELECTOR IN	OP ALL POSIT	TONS		EPLACED CAPT INSTI HECK NORMAL AS P		
Tail # N997CF	Aircraft Type DC8-62F	Flt Date 9/22/98	Flt # EB228	Flt Leg KDAY -to- KMCO	Delay Length Cancelled	ATA 3211	
Discrepancy	<i>r</i> :	••		Corrective Action:			
ON POST-F	LIGHT FOUND R	/H MLG STRU	T FLAT.	REPACKED R/H MAIN GEAR STRUT. NO LEAKS NOTED IAW MM 32-11-2.			
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg KBOS -to- KIND	Delay Length 1 Hr. 24 Min.	ATA 3611	
N998CF	DC8-62F	9/10/98	EB737	Corrective Action:	1 111. 24 14111.	3011	
Discrepancy #1 BLEED ON HIGH.	WILL NOT GO OF	N HIGH WHEN	switch is	FOUND C/P ON PR DEFECTIVE. REM	ESSURE REGULATOR OVED AND REPLACE ATOR VALVE, OPS C 1 CH 36-12-1.	D C/P,	
Tail #	Aircraft Type	Flt Date	Flt#	Flt Leg	Delay Length	ATA	
N998CF	DC8-62F	9/11/98	EB737	KBOS -to- KIND	Cancelled	7112	
Discrepancy	<i>r</i> :			Corrective Action:			
ENG AFT E	UAL INSPECTION BI-DUCT 6 O'CLO VISUAL INSPEC VD BI-DUCT 7 O'C	CK POSITION. TION, FOUND	UPON DAMAGE TO	DC8 MM CHAP 78	EPLACED AFT BI-DUC -10-3. REMEOVED AN SY IAW DC8 MM 78-10	ID REPLACED	
Tail #	Aircraft Type	Flt Date	Flt#	Fit Leg	Delay Length	ATA	
N998CF	DC8-62F	9/19/98	EB738	KMSP -to- KIND	Cancelled	8011	
Discrepane	y .			Corrective Action:			
PNEUMAT WITH THE A DUCT P AND R/H !	TART UP #2 STA TIC PRESSURE DI E PNEUMATIC CF RESSURE SPLIT MANIFOLD OF A TIC LEAK IN THE	ROPS TO ABO ROSS FIELD VA EXISTS BETW BOUT 6 PSI SU	UT 15 PSI. ALVE CLOSED EEN THE L/H ISPECT	LEAK AND OPS C	LEPLACED STARTER (HECK GOOD. GROUN IO NOTEABLE PRESS(ID RUN 32 PS	

DC8 1	FLEET		DELA	SUMMARY	Octo	ober 1998		
Tail # N105WP	Aircraft Type DC3-73	Fit Date 10/22/98	Fn # EB012	Fit Leg KDAY -to- KSAN	Delay Length 3 Hr. 00 Min.	ATA 2811		
Discrepancy	<i>y</i> :			Corrective Action:		2511		
ON PRE-FI MAIN GE	LIGHT INSP FOUN AR DOOR	ND FUEL LEAK	FWD OF R/H	R/R GANGLA SEA WHEEL WELL L	L ON COUPLING JUST : EAK CHECKED GOOD.	FWD OF RT		
Tail # N105WP	Aircraft Type DC8-73	Flt Date 10/27/98	Fit # EB011	Fit Leg KSAN -to- KDAY	Delay Length 4 Hr. 22 Min.	ATA 5612		
Discrepancy	<i>r</i> :			Corrective Action:				
REF: NON- SEVERELY	MEL #1096 CAPT SCRATCHED AN	SLIDING WINI ID IMPAIRS VI	OOW IS SION.	R/R CAPT SLIDING CHECKS NORMAL	G WINDOW IAW M/M 5 _ THIS CLEARS NON-N	6-13. OPS MEL 1096.		
Tail# N500MH	Aircraft Type DC8-71F	Flt Date 10/13/98	Fit # EB430	Fit Leg KDAY -to- KBOS	Delay Length Cancelled	ATA 2711		
Discrepancy	:			Corrective Action:				
A/C STILL CONDITIO TRIM.	FEELS RIGHT WI NS, REQUIRES 1	NG HEAVY AL 1/2 TO 2 UNITS	L FLIGHT OF LEFT AIL	TRAVEL CK AILERONS FOUND R/H CONTROL TAB OU OF RIG. RIGGED R/H CONTROL TAB AND R/R AILERON BUS CABLES. OPS CK GOOD.				
Tail # N500MH	Aircraft Type DC8-71F	Fit Date 10/21/98	Fit # EB521	Fit Leg KDFW -to- KDAY	Delay Length 0 Hr. 41 Min.	ATA 2841		
Discrepancy	:			Corrective Action:				
#4 MAIN FT FLIGHT.	JEL GAUGE REAI	OS 4000 LBS HI	GH IN	REMOVED AND REPLACED #4 MAIN FUEL QUANTITY INDICATOR. OPS CHECKS GOOD.				
Tail# N500MH	Aircraft Type DC8-71F	Fit Date 10/28/98	Fit # EB332	Fit Leg KDAY -to- KFLL	Delay Length 0 Hr. 40 Min.	ATA 2821		
Discrepancy: #4 MAIN T. WILL NOT	ank intermedl	ATE FILL VALV	Æ INOP,	Corrective Action: R/R #4 FUEL LEVEL CONTROL VALVE. #4 FUEL SYSTEM OPS & LEAK CHECKS GOOD NO DEFECTS NOTED.				
Tail # N603AL	Aircraft Type DC8-73F	Fit Date 10/6/98	Fit # EB028	Fit Leg KDAY -to- KRDU	Delay Length 0 Hr. 53 Min.	ATA 3245		
Discrepancy:				Corrective Action:				
ILLUMINAT	ED ANTI-SKID AN IED. NOTE ON ST ID ILLUMINATED	IONTROL		L/H GEAR WELL IN	RMINAL STUD ON LFO VBD TOP STUD. TIGHT M STUD. C?S CK GOO	ENED		

					Octo	
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N603AL	DC8-73F	10/21/98	EB044	KDAY -to- KLAX	0 Hr. 20 Min.	3421
Discrepano	As			Corrective Action:		
OFF FLAGS IN VIEW DURING POST FLT CHECK CAPTS HIS, F/O'S RMI. #2 COMPASS SYSTEM PRECESSING ON GND COMPARATOR LIGHT ON.				GOOD IAW DC8 M	NAL GYRO. SYSTEM O' IM R/R #2 DG AND SW FOR T/S. SYSTEM OPS IM	APPED
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N606AL	DC8-73F	10/22/98	EB332	KDAY -to- KFLL	0 Hr. 55 Min.	2912
Discrepanc	y:			Corrective Action:		
HYDRAU ON GROU	LIC LEAK AFT OF ND.	PYLON ENG #3	DRIPPING	TIGHTENED B-NU CHECKS GOOD.	T ON #3 ENG REV SYS	OPS AND LEA
		•				
Tail #	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N791FT	DC8-73F	10/31/98	EB825	KMIA -to- MSLP	0 Hr. 00 Min.	3222
Discrepanc						
ON WALK	y: LAROUND FOUND ROD BROKEN, AL LENT POINTS AT 1	LSO SKIN DAM	AGE AROUND	INSTALLED SERV IAW MM 32-22-1, A	EN NLG RH AFT DOOR ALT DOOR ROD ASSY ADDITIONALLY INSPEC	P/N 4648253 CTED LH AFT
ON WALK	LAROUND FOUND ROD BROKEN, AL	LSO SKIN DAM	AGE AROUND	REMOVED BROKE INSTALLED SERV LAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY FA	ALT DOOR ROD ASSY LDDITIONALLY INSPECTAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA LETWEEN STA-20 TO 12 O DAMAGE BY NLO DO L-8-53-601-98 DATED 11 I FAA FORM 8110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETE	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED 2 AND LONG OOR ROD IAW -1-98 PROVED & WASHI DESIG
ON WALK SLPPORT ATTACHN	LAROUND FOUND ROD BROKEN, AI MENT POINTS AT I	SO SKIN DAM DOOR HINGES	AGE AROUND	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED 1 MM 32-30-0 NORM	ALT DOOR ROD ASSY ADDITIONALLY INSPECTAND FOUND TO RUSTI EPLACED BY SERV LH AW MM 32-22-1. REPA IETWEEN STA-20 TO 12 O DAMAGE BY NLG DO 1-8-53-601-98 DATED 11 N FAA FORM 8110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETE AL COND.	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED 2 AND LONG OOR ROD LAW -1-98 PROVED & INASHI DESIG RACTION IAW
ON WALK SLPPORT ATTACHN	LAROUND FOUND ROD BROKEN, AI MENT POINTS AT 1	SO SKIN DAM DOOR HINGES Fit Date	AGE AROUND	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED 1 MM 32-30-0 NORM	ALT DOOR ROD ASSY LDDITIONALLY INSPECTAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA LETWEEN STA-20 TO 12 O DAMAGE BY NLG DO L-8-53-601-98 DATED 11 IF FAA FORM \$110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETE AL COND. Delay Length	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED 2 AND LONG OOR ROD LAW -1-98 PROVED & INASHI DESIG RACTION IAW
ON WALK SLPPORT ATTACHN Tail # N792FT	AROUND FOUND ROD BROKEN, AI MENT POINTS AT Aircraft Type DC8-73F	SO SKIN DAM DOOR HINGES	AGE AROUND	REMOVED BROKE INSTALLED SERV LAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COC DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED 1 MM 32-30-0 NORM Fit Leg KEWR -to- KDAY	ALT DOOR ROD ASSY ADDITIONALLY INSPECTAND FOUND TO RUSTI EPLACED BY SERV LH AW MM 32-22-1. REPA IETWEEN STA-20 TO 12 O DAMAGE BY NLG DO 1-8-53-601-98 DATED 11 N FAA FORM 8110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETE AL COND.	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED 2 AND LONG OOR ROD LAW -1-98 PROVED & INASHI DESIG RACTION IAW
ON WALK SUPPORT ATTACHN Tul # N792FT Discrepance	Aircraft Type DC8-73F TC LEAK R/H MAY	SO SKIN DAM DOOR HINGES Fit Date 10/7/98	Fit # EB117	REMOVED BROKE INSTALLED SERV LAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY F. # SW 810 DATED I MM 32-30-0 NORM Fit Leg KEWR -to- KDAY Corrective Action:	ALT DOOR ROD ASSY LDDITIONALLY INSPECTAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA LETWEEN STA-20 TO 12 O DAMAGE BY NLG DO LS-53-601-98 DATED 11 IF FAA FORM \$110-3 MA L-1-98, CHECKED RETE L-98, CHECKED RETE L-98 CHECKED MIN. Delay Length 6 Hr. 08 Min. RED, REINSTALLED RA	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED AND LONG OOR ROD LAW -1-98 PROVED & NMASHI DESIG RACTION IAW ATA 3614
ON WALK SLPPORT ATTACHN Tail # N792FT Discrepance	Aircraft Type DC8-73F TC LEAK R/H MAY	SO SKIN DAM DOOR HINGES Fit Date 10/7/98	Fit # EB117	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED I MM 32-30-0 NORM Fit Leg KEWR -to- KDAY Corrective Action: REMOVED, REPAIL	ALT DOOR ROD ASSY LDDITIONALLY INSPECTAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA LETWEEN STA-20 TO 12 O DAMAGE BY NLG DO LS-53-601-98 DATED 11 IF FAA FORM \$110-3 MA L-1-98, CHECKED RETE L-98, CHECKED RETE L-98 CHECKED MIN. Delay Length 6 Hr. 08 Min. RED, REINSTALLED RA	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED AND LONG OOR ROD LAW -1-98 PROVED & NMASHI DESIG RACTION IAW ATA 3614
ON WALK SUPPORT ATTACHN Tail # N792FT Discrepance	Aircraft Type DC8-73F TC LEAK R/H MAY	SO SKIN DAM DOOR HINGES Fit Date 10/7/98	Fit # EB117	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COO DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED I MM 32-30-0 NORM Fit Leg KEWR -to- KDAY Corrective Action: REMOVED, REPAIL	ALT DOOR ROD ASSY LDDITIONALLY INSPECTAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA LETWEEN STA-20 TO 12 O DAMAGE BY NLG DO LS-53-601-98 DATED 11 IF FAA FORM \$110-3 MA L-1-98, CHECKED RETE L-98, CHECKED RETE L-98 CHECKED MIN. Delay Length 6 Hr. 08 Min. RED, REINSTALLED RA	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED AND LONG OOR ROD LAW -1-98 PROVED & NMASHI DESIG RACTION IAW ATA 3614
ON WALK SUPPORT ATTACHN Tail # N792FT Discrepance PNEUMAI PANEL/FL	Aircraft Type DC8-73F TIC LEAK R/H MAY	SO SKIN DAM DOOR HINGES. Fit Date 10/7/98	Fit # EB117	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COC DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED 1 MM 32-30-0 NORM Fit Leg KEWR -to- KDAY Corrective Action: REMOVED, REPAIL MANIFOLD SUPPL	ALT DOOR ROD ASSY ADDITIONALLY INSPECAND FOUND TO RUSTI EPLACED BY SERV LH AW MM 32-22-1. REPA ETWEEN STA-20 TO 12 O DAMAGE BY NLO D 8-53-601-98 DATED 11 I FAA FORM 8110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETT AL COND. Delay Length 6 Hr. 08 Min. RED, REINSTALLED RA Y DUCT.	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED AND LONG OOR ROD IAW -1-98 PROVED & INASHI DESIG RACTION IAW ATA 3614 H PNEUMATIC
ON WALK SUPPORT ATTACHN Tuil # N792FT Discrepance PNEUMAI PANEL/FL Tuil #	Aircraft Type DC3-73F Aircraft Type DC3-73F	FR Date 10/7/98 NIFOLD BELOV	Fit # EB117 V F/O	REMOVED BROKE INSTALLED SERV IAW MM 32-22-1, A DOOR ROD ASSY REMOVED AND R ROD P/N 4648253 I FUSELAGE SKIN B 31R TO 34R DUE T COOPESA EO #COC DOCUMENTED ON SUBMITTED BY F, # SW 810 DATED I MM 32-30-0 NORM Fit Leg KEWR -to- KDAY Corrective Action: REMOVED, REPAIL MANIFOLD SUPPL	ALT DOOR ROD ASSY ADDITIONALLY INSPECAND FOUND TO RUSTIF EPLACED BY SERV LH AW MM 32-22-1. REPA ETWEEN STA-20 TO 12 O DAMAGE BY NLO D 8-53-601-98 DATED 11 N FAA FORM 8110-3 API AA DER ROBERT M MA 1-1-98, CHECKED RETE AL COND. Delay Length 6 Hr. 08 Min. RED, REINSTALLED RAY Y DUCT.	P/N 46-18253 CTED LH AFT ED. IT WAS AFT DOOR IRED AND LONG OOR ROD IAW -1-98 PROVED & NASHI DESIG RACTION IAW ATA 3614 H PNEUMATIC

DC8	FLEET				Octo	October 1998		
Tail # N796FT	Aircraft Type DC8-73F	Fit Date 10/30/98	Fh # EB025	Fit Leg KSEA -to- KDAY	Delay Length 1 Hr. 53 Min.	ATA 2761		
Discrepance	À:			Corrective Action:				
rt wing	FLT SPOILERS CR	LANK ARMS BE	roken.	REPLACED 2 EA C GOOD IAW MM 27	RANKS AND DRIVE RI -60-0.	NGS. CHECK		
Tail#	Aircraft Type	Fit Date	Fn #	Fit Leg	Delay Length	ATA		
N797AL	DC8-63	10/13/98	EB324	KDAY -to- KMSP	0 Hr. 37 Min.	2724		
Discrepanc	•			Corrective Action:				
STANDBY FLIGHT	RUDDER ACCUM	IULATOR REA	DS '0' ON PRE-	LEAK CHECK GOO	DDER ACCUMULATOR D, R/R GAUGE WITH N PS AND LEAK CHECK	ŒW		
Tail#	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA		
N801GP	DC8-71F	10/11/98	EB313	KBOS -to- KDAY	2 Hr. 40 Min.	3241		
Discrepanc	y .	•		Corrective Action:				
#7 TIRE B	LOWN PRIOR TO	VI ABORTED T	CAKEOFF.	REMOVED AND REPLACED #7 MAIN TIRE IAW MM 32- 40-1.				
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA		
N8076U	DC8-71F	10/12/98	EB019	MMMY -to- KDAY	0 Hr. 19 Min.	552		
Discrepanc	y;			Corrective Action:				
BOTH MO RADAR O	i.0' AIRCRAFT TOX DE 'C' X-PONDERS PERATES INTERM ING STRIKE.	REPORTED O	UT BY ATC.	ALTI-CODER SWIT DEACTIVATED FIR PER UNITED MM E BEING DEACTIVAT CHECK OF BOTH # ALTI-CODER SWIT DEACTIVATED PL MM. PLACARD RES	O NOT FIND ANY DEFI CH IN FO'S POSITION. UST OFFICERS ALTI-CO USE TO FIRST OFFICER FED. PERFORMED OPI I AND #2 TRANSPONI ICH IN CAPTAIN'S POSI ACARD INSTALLED IA MOVED. BOTH TRANS AW DC-8 MM. GROUN	DDER SWITCH L'S ALTI-CODI ERATIONAL DERS WITH ITION. W UNITED PONDERS OP		
Tail #	Aircraft Type	Fit Date	Fh #	Fit Leg	Delay Length	ATA		
N8076U	DC8-71F	10/15/98	EB430	KDAY -to- KONT	0 Hr. 00 Mir.	2847		
Discrepanc	y:	•		Corrective Action:				
3100 A DE	EL QTY INDICAT CREASE OF 5900 I FUEL LEAK OR IN	DIFFERENCE II	VI HR	NOTED VERIFIED	AFT AFTER REFUEL N 18200 BY DRIP STICK I D #4 ALT QTY IND GO	REMAINING		
T '' 4	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA		
Tail#	DC8-71F	10/16/98	EB439	KDAY 40- CYYZ	1 Hr. 31 Min.	2811		
1 211 # N8076U								
	y:			Corrective Action:				

			DELAY	SUMMARY		
DC8	FLEET				Octo	ber 1998
T±11# N3076U	Aironaft Type DC8-71F	Fit Date 10/20/98	F.t ≠ EB820	Fit Lag	Delay Longh Cancelled	ATA 7721
Discrepanc	76			Corrective Action:		
(ATB) AF REDUCE ACCOMP	TER VI #4 EGT MO ED NI 1% TO MAIN LISHED INFLIGHT TURNED TO MMM	TAIN EGT BEI SHUTDOWN, I	LOW LIMITS. DUMPED		ATOR AS REQUIRED. (OPS CHECKS
Tail#	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA
N8076U	DC8-71F	10/21/98	EB155	MMMX -to- KDAY	Cancelled	7721
Discrepanc	y:			Corrective Action:		
START, SI HIGHER T REDUCED MOMENT EGT BELO SHUTDOV	LOW TO ACCELER LOW TO CUTBACK (HAN 1,3 & 4. AFT.) NI BY 1% & EGT ARILY. REDUCED OWN AND RETURNE LING TIME 21 MIN	FUEL FLOW ER VR EGT WE CLIMBED TO N1 BY 2% TO MPLIHSED IN D TO MMMX	& EGT ENT TO 905 ° 908 ° MAINTAIN		IC RIG CHECK ON TH M.E.C. TO ENGINE #2. AW MM 73-21-10.	
Tail # N8076U	Aircraft Type DC3-71F	Fit Date 10/24/98	Fit #	Flt Leg KDAY -to- KCLT	Delay Length 0 Hr. 00 Min.	ATA 2611
1480700	DC3-71F	10/24/98		Corrective Action:	o H. oo Mile	2011
WARNING REJECTIN	OFF ROLL JUST PR G WENT OFF, REJE IG TAKEOFF FIRE D AT 118 KNOTS.	CTED TAKEOR	F, WHILE	BLOWN OUT, CAUS LOOP DETECTOR, AND L/H REVERSE ASSY, L/H REVERS CHECKS GOOD. PE	E BLEED AIR CAP ON I SING DAMAGE TO #4 I TURBINE COOLING TO R HALF, R/R COOLING ER HALF AND DUCTIN ERFORMED REJECTED IAW SP006 PARAGRA	JH AFT FIRE JBING ASSY TUBING IG CAP. OPS TAKEOFF
Tail #	4: A T	Th D	m. 4	The fire	Delectoreth	ATA
N8079U	Aircraft Type DC8-71F	Flt Date 10/16/98	Fit# EB117	Fit Leg KEWR -to- KDAY	Delay Length 0 Hr. 00 Min.	7223
Discrepance	v:			Corrective Action:		
#3 ENG AL	BRADABLE CARTE POSITION (8" PEEL		OF ABOUT 8"	REPAIR LAW CFM5	6-2C ESM SECTION 72- TRIDGE RECONDITION ACTORY.	
Tail # N8084U	Aircraft Type DC8-71F	Fit Date 10/15/98	Fit # EB035	Fit Leg KORD -to- KDAY	Delay Length 0 Hr. 02 Min.	ATA 3243
Discrepance				Corrective Action:		
•	TT-SKID INOP LIGI	ht on with s	YSTEM	TRANSFERRED TO CONTROL #810240 INSTALLED. CLOS CONTROL BOX. A	DMI PER MEL REF 32- DUE DATE 10-25-98 PI ED KDAY 10-15-98 R/I NTI-SKID INOP OPS NO 40 PLACARD REMOVI	LACARDS R ANTI-SKID DRMAL THIS

DC8	FLEET			SUMMERKI	Octo	ber 1998
Tail # N8084U	Aircraft Type DC3-71F	Flt Date 10/28/98	Fit # EB321	Fit Leg KDFW -to- KDAY	Delay Length 3 Hr. 32 Min.	ATA 2511
Discrepanc	À:			Corrective Action:		
	TAXI INTERMITTI IRCUIT LIGHT FL			R/R #2 ENG RIGH FUNCTIONS NOR REMOVED.	T FAN FIRE DETECTOR MAL DMI 810420 CLE	CASSY SYSTE ARED PLACAS
Tail# '	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N8085U	DC8-71F	10/28/98	EB004	KDAY -to- KCLT	I Hr. 04 Min.	5613
Discrepanc	À:			Corrective Action:		
FO SIDE W O'CLOCK	VINDOW (AFT) CR POSITION 2 1/4", 8	ACKED 2 PLAC 3 O'CLOCK 1".	CED 6	PG 603 CODE 2. E CLOSED CYMX 10	TO BE WITHIN LIMIT: NTERED ON NON MEL 1-29-98 R/R F/O'S SIDE ' CLEARED NON MEL #4	LIST #67. WINDOW AFT
Tail #	Aircraft Type	Flt Date	Fit#	Fit Leg	Delay Length	ATA
N8087U	DC8-71F	10/21/98	EB124	KDAY -10- KMSP	0 Hr. 26 Min.	2153
Discrepanc	y .	•		Corrective Action:		
PACK COO DOOR OP! TOP OF DI	ACK WAS TURNE DLING FAN START ENED SUPRA LOG ESCENT #1 & #4 E PECT GROUND CO	TED AND PACE 5589-08 (TTEM NGS WENT TO	COOLING #2) ALSO AT GROUND	NORMAL FOR GR	WITCH RH PACK OPS (OUND AND FLIGHT M(IDLE & FLIGHT IDLE (DDE. #1 & #4
Tail # N3087U	Aircraft Type DC3-71F	Flt Date	Fit#	Fit Leg	Delay Length	ATA
		10/21/93	EB323	KMSP -to- KDAY	1 Hr. 17 Min.	2153
Discrepancy	r:			Corrective Action:		
FAN AND R/II GROU	O WHEN TURNING COOLING DOOR (IND CONTROL RE EMS WORKS NOR	CAME ON AND Lay c/b a/c t	OPEN, PULL	AND RESEATED C VERIFIED OPERAT	TO GRD SHIFT SWITCH ONNECTOR ON RT SW TION OF SWITCH AND ELAY. SYSTEM OPER- N.	TTCH. R2-46 RT AIR
Tail#	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N8087U	DC8-71F	10/22/98	EB002	KDAY -to- KPIA	1 Hr. 34 Min.	0
Discrepancy	<i>r</i> .			Corrective Action:		
	ENGINE N1 RPM FLIGHT IDLE AT			R2-63 SYSTEM OP:	EPLACED GROUND CO S CK GOOD AT THIS TI OPERATES GOOD AT	ME. #1 AND #

DC8 I	FLEET				Octo	her 1998
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N3087U	DC3-71F	10/22/98	EB002	KDAY +to- KPLA	Cancelled	7711
Discrepanc	y:			Corrective Action:		
EXCESSIV WENT BE	FLIGHT SHUTDOW FE VIB AND NOISE LOW MIN FLIGHT #4 ENG WENT TO 1.	. TOP OF DESC IDLE TO 32% I	CENT #1 ENG N1. TOP OF	SUSPECT PACK W ALSO MALFUNCT GOOD. REPLACEI CHECK GOOD ON TEAM. REPLACEI S1-69 NOSE OLEO	AND PACK SECTION (AS RUNNING WHEN # ION, SYS LEAK AND O D S1-68 OLEO SWITCH ENGINE RUN BY MAII D R2-44 RIGHT GROUN SWITCH, SYS OPS CHE MAINTENANCE TEAM.	FENG WAS P CHECKS , SYS OP TENANCE D RELAY AND CKS GOOD O
Tail#	Aircraft Type	Fit Date	Fit #	Flt Leg	Delay Length	ATA
N8087U	DC8-71F	10/22/98	EB105	KROC -to- KDAY	3 Hr. 01 Min.	7611
Discrepanc				Corrective Action:		
	TTLE OUT OF RIG . , #3 1.9, #4 2.4.	AT FL270 79% i	NI,#1 READ	MEC, SECURED. R NOTED ENGINE P. NO THROTTLE SP	OR FOR PMC HARNES LAN ENGINES #2 & #3 & ARAMETERS (N1 & N2 LIT, ACCOMPLISHED # RUN-UP HANDBOOK A	AT 50% N1 ANI) EQUAL WITH #2 ENGINE PM
Tail # N8087U	Aircraft Type DC8-71F	Fit Date 10/23/98	Fh # EB012	Flt Leg KDAY -to- KSAN	Delay Length 0 Hr. 34 Min.	ATA 2822
				Corrective Action:		
Discrepance #1 FEED F	PUMP INOP			#810356, DUE DAT	LAW MEL 28-4 CAT C F TE 11-2-98 PLACARD IN 25-98 R/R #1 MAIN FE PS CHECK NORMAL T ARD REMOVED.	STALLED. ED PUMP PER
T-11#	1: A T	Fit Date	Flt#	Fit Leg	Delay Length	ATA
Tail # N8091U	Aircraft Type DC8-71F	10/2/98	EB017	KRNO -to- KDAY	1 Hr. 23 Min.	73311
Discrepano				Corrective Action:		
•	es to zero with	H LOW POWER	SETTINGS.	R/R F/F XMTR. R.ª M/M.	AN ENG SYS CHECKS (600D PER 77-1
Tail#	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N8091U	DC8-71F	10/24/98	EB508	KDAY -to- KSJU	Cancelled	3423
Discrepan	ey:			Corrective Action:		
AD FLIC	O.'S HORIZON SHO GHT DIRECTOR FI NOSE UP.	OWS INTERMIT LAGS IN VIEW	TENT GYRO AND DRIFTS 5	R/R #2 VERTICAL LAW DC8 MM.	GYRO. SYSTEM OPS	CHECKS GOOL

October 1998

DC8 FLEET

	Aircraft Type	Fit Date	Fh. #	Fit Leg	Dalay Longth	474
MSHAL	DC3-71F	10/1/98	EB456	KRDY -to- KDAY	1 Hr. 45 Min.	2131
Discrepano	ry:			Corrective Action:		
RAPID DI CABIN AI	ecompression at LT.	T FL350 UNAB	LE TO HOLD		AMP LOOSE ON RAH AC ZATION CHECK GOOD	
Tail# ·	Aircraft Type	Flt Date	Flt #	Fit Leg	Delay Length	ATA
N811AL	DC8-71F	10/9/98	EB124	KDAY -to- KMSP	0 Hr. 41 Min.	2740
Discrepano	.y:			Corrective Action:		
AND ELE DISCREP. WHEN US STAB TRI NOSE UP AS BAD (SPEED IS THIS OC(ICE LOG PAGE 714 VATOR TRAVEL C ANCY ON PREVIOU SING PICKLE SW O IM A/C YAWS TO R OR DOWN. INTEN ON CLIMBOUT AS I INCREASED YAW TURS WITH YAW E OTIDEABLE W/YAW TO ROLL	HECK IN REFE US PAGE. LOG R SUITCASE H LIGHT WHEN T ISITY OF MOV DURING DESC MOTION IS M DAMPER ON O	RENCE TO PAGE 7141-05 ANDLES FOR TRIMING EMENT NOT ENT. AS ORE SEVERE. R OFF, IT IS	7141-05 - RAN #2 & DOWN WITH AILE	K GOOD AT THIS TIME & #3 ENG MOVED STAE FRON AND RUDDER PO NOTED. OPS CK GOOI	UP AND
	Aircraft Type DC8-71F	Fit Date 10/13/98	Fh # EB107	Fit Leg KBSM -to- KDAY	Delay Length 1 Hr. 08 Min.	ATA 5234
N811AL	DC8-71F				• -	
N811AL Discrepand ON OPEN	DC8-71F ry: TING MAINCARGO	10/13/98	EB107	KBSM -to- KDAY Corrective Action:	• -	5234
N811AL Discrepance ON OPEN ROD ENT	DC8-71F Fy: TING MAINCARGO D BROKE.	10/13/98	EB107	KBSM -to- KDAY Corrective Action:	1 Hr. 08 Min.	5234
N811AL Discrepance ON OPEN ROD ENT	DC8-71F ry: TING MAINCARGO	10/13/98	EB107	KBSM -to- KDAY Corrective Action: REPLACED ROD E	1 Hr. 08 Min. ND OPS CHECKED NO	5234 RMAL
N811AL Discrepance ON OPEN ROD ENT	DC8-71F TY: TING MAINCARGO DROKE. Aircraft Type DC8-71F	10/13/98 DOOR, LATCH	EB107 CYLINDER Fit #	KBSM -to- KDAY Corrective Action: REPLACED ROD E	1 Hr. 08 Min. ND OPS CHECKED NO! Delay Length	5234 RMAL.
N811AL Discrepance ON OPEN ROD ENT Tail # N811AL Discrepance	DC8-71F TY: TING MAINCARGO DROKE. Aircraft Type DC8-71F	10/13/98 DOOR, LATCH Fit Date 10/17/98	EB107 I CYLINDER Fh # EB011	KBSM -to- KDAY Corrective Action: REPLACED ROD E Fit Leg KTUS -to- KDAY Corrective Action:	1 Hr. 08 Min. ND OPS CHECKED NO! Delay Length 0 Hr. 32 Min.	5234 RRMAL. ATA 3263
N811AL Discrepance ON OPEN ROD ENT Tail # N811AL Discrepance	DC8-71F TNG MAINCARGO DBROKE. Aircraft Type DC8-71F Fy:	10/13/98 DOOR, LATCH Fit Date 10/17/98	EB107 I CYLINDER Fh # EB011	KBSM -to- KDAY Corrective Action: REPLACED ROD E Fit Leg KTUS -to- KDAY Corrective Action: REPLACED LAND	1 Hr. 08 Min. ND OPS CHECKED NO! Delay Length 0 Hr. 32 Min.	5234 RRMAL. ATA 3263
N811AL Discrepanc ON OPEN ROD ENT Tail # N811AL Discrepanc	DC8-71F TNG MAINCARGO DBROKE. Aircraft Type DC8-71F Ty: G GEAR/FLAP WAR	10/13/98 DOOR, LATCH Fit Date 10/17/98	EB107 ECYLINDER Fh # EB011 NOPERATIVE.	KBSM -to- KDAY Corrective Action: REPLACED ROD E Fit Leg KTUS -to- KDAY Corrective Action: REPLACED LAND: SWITCH, SYS OPS	1 Hr. 08 Min. ND OPS CHECKED NO. Delay Length 0 Hr. 32 Min. ING GEAR/FLAP WARN CK GOOD.	S234 RMAL ATA 3263 TING HORN
ROD ENT Tail # N811AL Discrepand LANDING Tail #	DC8-71F TY: TING MAINCARGO D BROKE. Aircraft Type DC8-71F TY: D GEAR/FLAP WAR Aircraft Type DC8-71F	10/13/98 DOOR, LATCH Fit Date 10/17/98 ENING HORN E	EB107 ECYLINDER Fh # EB011 NOPERATIVE. Fh #	KBSM -to- KDAY Corrective Action: REPLACED ROD E Fit Leg KTUS -to- KDAY Corrective Action: REPLACED LAND SWITCH, SYS OPS	1 Hr. 08 Min. ND OPS CHECKED NO. Delay Length 0 Hr. 32 Min. ING GEAR/FLAP WARN CK GOOD. Delay Length	S234 RMAL ATA 3263 IING HORN ATA

D 00 -			DELA	Y SUMMARY		
DC8 1	FLEET			October 1998		
Tail# N311AL	Aircraft Type DC3-71F	Fit Date 10/31/98	Fit # EB234	Fit Leg KDAY -to- KDFW	Delay Length 2 Hr. 08 Min.	ATA
Discrepancy	y:			Corrective Action:	2 III. Oo WILL	2321
LEVER FO	NG THE #2 FUEL RWARD W/BOOS ID TO 20 PSL	, air & hyd si T Pump on, fi	HUTOFF JEL PRESS	R/R FUEL FIRE SH	IUTOFF VALVE RIG, O W DC3 MM 28-22-01.	PS CK GOOD, NO
Tail # · N8177U	Aircraft Type DC8-71F	Fit Date 10/10/98	Fh # EB381	Fit Leg KPLA -to- KOAK	Delay Length 4 Hr. 00 Min.	ATA 7721
Discrepancy				Corrective Action:		
#2 EGT ER INTERMIT	RATIC. READS F TENTLY.	ROM 0 ° TO MA	XIMUM	R/R LEFT AND RIC OPS NOW CHECK RUN.	THT EGT HARNESS ON NORMAL OP CHECKE	I #2 ENGINE. D OK ON ENG
Tail #	Aircraft Type	Fit Date	Fit#	Fh Leg	Delay Length	ATA
N8177U	DC8-71F	10/15/98	EB011	KSAN -to- KDAY	0 Hr. 27 Min.	3463
Discrepancy	:	•		Corrective Action:		
ON CREW : EQUIPMEN	MAN-UP, FOUND T.	NO POWER TO	CAPT NAV	RE-SET #1 FMS DA	DC O/B OP CHECKS N	ORMAL.
Tail #	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA
N832AL	DC8-73F	10/13/98	EB106	KDAY -to- KROC	1 Hr. 05 Min.	3421
Discrepancy:				Corrective Action:		
1. #2 COME VIEW. 2. C FLAG IN VI	PASS SYSTEM PR APT'S ADI INDIC EW.	ECESSING HDO ATE 3 ° NOSE I	G FLAG IN ON HORIZON	CHECK, OPS CHEC VERTICAL GYRO	DNAL GYRO. PERFORN KS GOOD IAW DC8 M ALSO SWAPPED #1 & # .ESHOOTING. BOTH S	4. 2. R/R #1 2 ADI'S FOR
Tail#	Aircraft Type	Fit Date	Fh #	Flt Leg	Delay Length	ATA
N832AL	DC8-73F	10/14/98	EB052	KDAY -to- KFLL	·1 Hr. 13 Min.	5234
Discrepancy:				Corrective Action:		
NSTEAD O	SING CARGO DO F LOCKING. CAI KS ON THE HAN	GO DOOR HY	D VALVE	ELECTRICAL & HA NON MEL CNTRL # SILL SWITCH. ADJ	ED MAIN CARGO DOO RDWARE PORTION OF 3. R/R SW-03 SWITCH USTED MANUAL NEED OD. HAND VALVE NO SES NON MEL #3.	F DOOR ON AND DOOR DLE VALVE
Tail#	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA
N832.AL	DC8-73F	10/30/98	EB001	KOAK -to- KDAY	Cancelled	3463
Discrepancy:				Corrective Action:		
NAVIGATIO	EFLIGHT, WE CA IN SYSTEM WOR BOTH WORK ON	K INDEPENDE	VILY. WE		GNED, BUILT, AND AI	PPROVED PER

DC8	FLEET				Oct	ober 1998
Tail # N950R	Aircraft Type DC8-63	Fit Date 10/8/98	Fit # EB151	Fit Leg MMMCK -to- KDAY	Delay Length Cancelled	ATA 3230
Discrepan	cy:			Corrective Action:		
BY DOOR	N GEAR WOULD N R NOT LATCHED, (IR DOOR WARNIN URES.	JEAR NOT LAT	TCHED AND	4802330-1 IAW DC	EN PIN AND INSTALLE -8 SERVICE BULLETIN 972 AND MDC-BOEIN -10-98.	₹32-101 R-5
Tail#	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N950R	DC3-63	10/14/98	EB321	KDFW -to- KDAY	0 Hr. 25 Min.	2811
Discrepano	-y:			Corrective Action:		
	AK O/B LEADING E HT WALK AROUNI		NG FOUND ON	FOUNDFUEL PROP SEALED FUEL PRO	BE COVER LOOSE. SE BBE PANEL	CURED AND
Tail #	Aircraft Type	Flt Date	Fit #	Flt Leg	Delay Length	ATA
N950R	DC8-63	10/21/98	EB233	KDAY -to- KPLA	Cancelled	2311
Discrepano	. y:	÷		Corrective Action:		
PREFLIGI	BOTH AND AROU HT FOUND HYD LI RD SLOT ACTUAT	EAK IN RIGHT		LINE FOR OUTBD INSTALLED FLEXI OPS/LEAK CHECK PERFORMED IAW ENTERED ON NON CLOSED KPDX 10-	VMM 28-10-01. FOUNT SLOT ACT RETRACT I BLE HYD LINE IN ITS GOOD. THIS INSTALI EWA MM CH 7 SECTIC FMEL LIST UNDER ITI 24-98 REMOVED FLEX LINE CLEARS NON M	LEAKING, PLACEATION ON I AND EM #231. CLINE AND
Tail #	Aircraft Type	Fit Date	Fit#	Flt Leg	Delay Length	ATA
N950R	DC8-63	10/29/98	EB316	KDAY -to- KPHL	1 Hr. 37 Min.	8011
Discrepano	ry:			Corrective Action:		
ENGINE I	IED INFLIGHT ENG AW EWW QRH DU ON FULLY ON DES 15 MIN / POS OIL P	E TO START V SCENT. ENGIN	ALVE #3 ENG	ACTION, START V. NOTED. LOG PAG WIRING FROM WI	AGE 7362-15 FOR COR ALVE OPS CK GOOD, I E 7362-15-P02 CA: REF NG TO #3 ENG PYLON T VALVE OP CHECKS	NO DEFECT PLACED PERFORMED
Tail#	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N951R	DC8-63	10/6/98	EBIIS	KPHL -to- KDAY	0 Hr. 48 Min.	3441
Discrepano	y:			Corrective Action:		
	ED TO RAMP DUE (ING TO THE LEFT			RESET CANNON PI ON GROUND.	LUG ON #1 DG. OPS C	HECK GOOD
	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
Tail #	COMMISSION AND LANDS	I'I' Date		KDAY -to- KPHL	Cancelled	5321
Tail # N951R	DC8-63	10/18/98	EB316	ADAI -10- AINE	Catiocited	2221
	DC8-63	10/18/98	EB316	Corrective Action:	Calcula	. 3321

DC8	THE TOTAL			SUMMARY		
	FLEEI			October 19		
Tail ≠	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N951R	DC3-63	19/20/98	EBlié	KDAY -to- KPHL	0 Hr. 47 Min.	3423
Discrepan	cy:			Corrective Action:		
	APT HORZ SHOW I TAXI OUT.	IVA NWOCI ° D) 3 ° RT TURN	FAULTS FOUND. AMPS FOR T/S AN	RATIONAL CHECK OF SWAPPED #1 AND #2 D D RE-ERECTED #1 VEI ADIS OP CHECK GOOD	NSTRUMENT RTICAL GYRO
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N951R	DC8-63	10/23/98	EB033	KDAY -to- KATL	5 Hr. 21 Min.	2811
Discrepano	:y:			Corrective Action:		
FUEL LEAK AT #1 ENGINE PYLON AND INBD OF #1 ENGINE PYLON.					EPLACED DEFECTIVE K CHECK GOOD IAW N	
Tail # N957R	Aircraft Type DC8-63F	Fit Date 10/13/98	Fit # EB011	Fit Leg KSAN -to- KDAY	Delay Length 1 Hr. 11 Min.	ATA 3261
Discrepano	ry:			Corrective Action:		
	DEPARTURE CHEC G GEAR WARNING				EPLACED BAD SWITCH STEM OPS CHECKS GO	
Tail # N957R	Aircraft Type DC3-63F	Fit Date 10/23/98	Fit # EB152	Fit Leg KDAY -to- MMMX	Delay Length Cancelled	ATA 3441
	DC3-63F			•		
N957R Discrepand	DC3-63F	10/23/98	E8152	KDAY -to- MMMX Corrective Action:	Cancelled LOOSE, TIGHTENED I	3441
N957R Discrepand	DC8-63F -y:	10/23/98	E8152	KDAY -to- MMMEX Corrective Action: FOUND RADAR RT	Cancelled LOOSE, TIGHTENED I	3441
N957R Discrepance (BTB) R4 Tail #	DC8-63F ADAR WILL TEST E Aircraft Type DC8-63F	10/23/98 BUT NO INPUT. Fit Date	EB152	KDAY -to- MMMEX Corrective Action: FOUND RADAR RT RACK, OPS CHECK	Cancelled LOOSE, TIGHTENED IS GOOD. Delay Length	3441 RADAR RT IN ATA
N957R Discrepance (BTB) R.4 Tail # N959R Discrepance	DC8-63F ADAR WILL TEST E Aircraft Type DC8-63F Ty:	IO/23/98 BUT NO INPUT. Fit Date 10/6/98	EB152 Flt # EB338	KDAY -to- MMMEX Corrective Action: FOUND RADAR RT RACK, OPS CHECK Fit Leg KDAY -to- KATL Corrective Action: OK FOR CONTINUE	Cancelled LOOSE, TIGHTENED IS GOOD. Delay Length	3441 RADAR RT IN ATA 7721
N957R Discrepance (BTB) R.4 Tail # N959R Discrepance #3 ENGIN	DC8-63F ADAR WILL TEST E Aircraft Type DC8-63F Ty:	IO/23/98 BUT NO INPUT. Fit Date 10/6/98	EB152 Flt # EB338	KDAY -to- MMMEX Corrective Action: FOUND RADAR RT RACK, OPS CHECK Fit Leg KDAY -to- KATL Corrective Action: OK FOR CONTINU WHITNEY MM CH	Cancelled LOOSE, TIGHTENED IS GOOD. Delay Length 0 Hr. 32 Min.	3441 RADAR RT IN ATA 7721
N957R Discrepance (BTB) R.4 Tail # N959R Discrepance #3 ENGIN TO 370 °.	DC8-63F Aircraft Type DC8-63F TY: Aircraft Type DC8-63F Aircraft Type DC8-63F	IO/23/98 BUT NO INPUT. Fit Date 10/6/98 HEN SLOWLY I	Fit # EB338 NCREASES	KDAY -to- MMMEX Corrective Action: FOUND RADAR RT RACK, OPS CHECK Fit Leg KDAY -to- KATL Corrective Action: OK FOR CONTINU WHITNEY MM CH AND TEST.	Cancelled LOOSE, TIGHTENED HES GOOD. Delay Length 0 Hr. 32 Min. ED SERVICE AS PER PR 72-0 PAGE 510 ENGINE	3441 RADAR RT IN ATA 7721 RATT & E ADJUSTMENT

DC8	FLEET		DELAY	Y SUMMARY	0.4	. 1 . 1000
		 .			Ucti	ober 1998
Tuil #	Aircraft Type	Flt Date	Fit #	Fit Leg	Delay Length	ATA
N919R	DC3-63F	10/13/98	EB037	KATL -to- KDAY	2 Hr. 55 Min.	3213
Discrepano	ry:			Corrective Action:		
GROUND SHIFT MECH IS BROKEN.				R/R NOSE GEAR E VALVE, RIG & OP	MERGENCY STEERIN S CHK GOOD.	G SHUTOFF
Tail#	Aircrast Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N959R	DC8-63F	10/15/98	EB316	KDAY -to- KPHL	0 Hr. 30 Min.	2723
Discrepano	ry:			Corrective Action:		
(BTB) YA	LW DAMPNER W/N	I ENGAGE.		EWA MEL 22-2 CA INSTALLED. CLO REPORT YAW DAI	AW DAMPNER TO DMI T C DUE DATE 10-25-9: SED KPHL 10-15-98 PE MPNER OPS CHECKS N 252 PLCARD REMOVE	8 PLACARD R CREW ORMAL THIS
Tail #	Aircraft Type	Flt Date	Flt #	Fit Leg	Delay Length	ATA
N959R	DC8-63F	10/29/98	EB036	KDAY -to- KORD	6 Hr. 56 Min.	2811
Discrepanc	y :			Corrective Action:		
ON POST ENGINE.	FLIGHT FOUND F	UEL LEAKING	FROM #3		EL CONTROL PERFOI ING OPS CHECK & LEA VDBOOK	
Tail ≠	Aircraft Type	Fit Date	Fh #	Fh Leg	Delay Length	ATA
N961R	DC8-73F	10/6/98	EB341	KEWR -to- KDAY	2 Hr. 29 Min.	2822
Discrepanc	y .			Corrective Action:		
#1 ALT TA VALVE.	NK SELECTOR V	ALVE WILL NO	T CLOSE	LUBED CABLES AT VALVE, OPS CHEC	ND PULLEYS FOR #1 A K GOOD.	LT TANK
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N961R	DC8-73F	10/20/98	EB820	MIMIMEY -to- KLAY	Cancelled	7721
Discrepancy	r.			Corrective Action:		
(BTB) PRI REJECTED	OR TO 100 KTS, #2 D TAXEOFF AT 100	ENG EGT WE KTS AND STE	NT TO 905 ° LL CLIMBING	CHECK NORMAL C	AND VSVS IAW MM 73 ON ENGINE RUN. EGT OFF POWER SETTING.	
Tail#	Aircraft Type	Fit Date	Fh #	Flt Leg	Delay Length	ATA
N961R	DC8-73F	10/27/98	EB014	KDAY -to- KBOS	0 Hr. 26 Min.	3611
Discrepanc,	z:			Corrective Action:		
PNEUMAT ENGINE S	TC BLEED LEAK F TART.	ROM LEFT SID	E AFTER		COOP VALVE OPEN CI ECTION ON VALVE. O E.	

DC8	FLEET		DELAI	SUMMARY	Octo	ctober 1998	
Tuil ≠	Aircraft Type	Fit Date	Fit ≠	Fit Leg	Delay Length	ATA	
N961R	DC8-73F	10/23/93	EB013	KEOS -to- KDAY	0 Hr. 43 Min.	2421	
Discrepano	ry:			Corrective Action:			
PERFORM PREFERE	GENERATOR COM I PREFERENTIAL NTIAL CHECK #1 US TIE AT THE SA FF LINE.	CHECK DURIN AND #2 GENER	NG LATORS PICK	DEFERRED LAW DC8 MEL 24-1, CAT C, DUE DATE 11-0 98. DMI #810414 ASSIGNED. CLOSED KBSM 10-29-98 INSPECTED GENERATORRELAY & BUS TIE RELAY. RESET CANNON PLUGS ON GEN CONT PANEL. OPS CHECK NORMAK ON ENGINE RUN UP LAW RUN UP HANDBOOK. THIS CLEARS DMI #810414 PLACARD REMOVED.			
Tail#	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA	
N961R	DC8-73F	10/30/98	EB123	KMSP -to- KDAY	1 Hr. 00 Min.	2421	
Discrepano	y:			Corrective Action:			
	GEN WILL NOT PE COULD NT GET F				NES. RESET #1 CSD. S ENGINES, ALL SYSTEM		
Tail #	Aircraft Type	Fit Date	Fit #	Flt Leg	Delay Length	ATA	
N990CF	DC8-62	10/11/98	EB341	KEWR -to- KDAY	Cancelled	2822	
Discrepano		20.21.22		Corrective Action:	2		
found # on prefi	2 MAIN FUEL TAN LIGHT.	K BOOST PUM	P PRESSURE '0'	REMOVED AND REPLACED BOOST PUMP WIRE HARNESS IN #2 MAIN FUEL TANK. PERFORMED BY CONTRACT SERVICE.			
Tail#	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA	
N991CF	DC8-54F	10/4/98	EB305	KROC -to- KDAY	0 Hr. 49 Min.	79333	
Discrepano	y:			Corrective Action:			
START, O	IL PRESS LIGHT S' IL PRESS NORMA ED THROTTLE DII	, oil temp N		REMOVED AND CHECKED OIL FILTER - CHECKS OK, NO CONT.MINATES NOTED. FOUND CONNECTOR LOOSE AT OIL PRESS SWITCH - TIGHTENED CONNECTOR AND SAFETIED. GROUND RUN ENG #3 OIL LIGHT CHECK GOOD. OIL FILTER LEAK CHECK GOOD.			
Tail #	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA	
N991CF	DC8-54F	10/17/98	EB109	KDEN -to- KDAY	1 Hr. 07 Min.	7933	
Discrepano	y:			Corrective Action:			
#3 ENGINE LOW OIL PRESSURE LIGHT DID NOT GO OUT AFTER ENGINE START UP. OIL PRESSURE NORMAL				FOUND LOOSE CANNON PLUG AT #3 OIL PRESS SWITCH CLEANED AND TIGHTENED SAME. OPS CK NORMAL DURING ENG RUN.			
Tail #	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA	
N991CF	DC8-54F	10/24/98	EB107	KBSM -to- KDAY	0 Hr. 22 Min.	2721	
Discrepano	. y:			Corrective Action:			
LOWER R	UDDER SUPPORT D.	BEARING NEE	DS TO BE		ER REMOVED AND RE ALLED RUDDER IAW		

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Tail #	Aircraft Type	Fit Date	Fit ≠	Fit Leg	Delay Longth	A.T.A	
N991CF	DC8-54F	10/28/98	EB110	KDAY -to- KDEN	0 Hr. 33 Min.	2611	
Discrepanc	y:			Corrective Action:			
#3 FIRE W	ARNING WILL N	OT TEST.		CLEANED CONNE #3 FIRE WARNING	ECTION AT L'H TOP AI G SYSTEM CHECKS GO	T FIRE LOOP OD.	
Tail#,	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA	
N991CF	DC8-54F	10/29/98	EB127	KIAH -to- KDAY	0 Hr. 33 Min.	2611	
Discrepanc	y:			Corrective Action:			
ON PREFL TEST.	IGHT FOUND #3 !	ENG FIRE TEST	WOULD NOT	OPENED #3 ENG C CLEANED FWD CO BOX. OPS CK OK.	OWLING INSPECTED ONNECTOR ON MAIN	LOOP AND JUNCTION	
Tail #	Aircrast Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA	
N992CF	DC3-54F	10/6/98	EB108	KDAY -to- KBSM	1 Hr. 39 Min.	7611	
Discrepancy	<i>y</i> :			Corrective Action:			
	TLE IS 2 FULL KN ES DURING ALL I			THROTTLE TO BE	HPOWER ENG #2. FOU 3/4 KNOB BEHIND OT DER WITHIN LIMITS L CCTS NOTED.	HER,	
Tail #	Aircraft Type	Fit Date	Flt #	Fh Leg	Delay Length	ATA	
N992CF	DC8-54F	10/8/98	EB128	KDAY -10- KIAH	1 Hr. 23 Min.	2921	
Discrepancy	.			Corrective Action:			
AUX PUME	PFAILED DURING	PRE-FLIGHT.		R/R MAIN SYSTEM HYDRAULIC AUX PUMP AND CLEAN PRESSURE RELIEF VALVE. OPS AND LEAK CHECKS GOOD.			
Tail #	Aircraft Type	Fit Date	Fh#	Fit Leg	Delay Length	ATA	
N992CF	DC8-54F	10/9/98	EB128	KDAY -to- KIAH	0 Hr. 42 Min.	7611	
Discrepancy	:			Corrective Action:			
#1 THRUST ON PREFLI	T LEVER MOVEM IGHT.	ENT IS VERY S	TIFF FOUND		ROTTLE LINKAGES, O D, NO DEFECTS NOTE		
Tail #	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA	
N992CF	DC8-54F	10/31/98	EB127	KIAA -to- KDAY	4 Hr. 10 Min.	2421	
Discrepancy	:			Corrective Action:			
ON ENG ST	TART #4 ENG GEN	VOLTS AND FI	REQS LOW.		L PANEL AND FREQ & 'S OPS NORMAL PER M		

DC8	FLEET		DELA	Y SUMMARY October 1998		
Tall # N993CF	Aircraft Type DC8-62	Fit Data 10/7/98	Fit # E3123	Fit Leg KLAY 46- KLAH	Delay Length 0 Hr. 36 Min.	ATA 7111 - S
Discrepano	Y:			Corrective Action:		
#1 ENG N	CSE COWL LSD IN	ILET CRACKE	D.	R/R #1 ENG NOSE	COWL OPS CHECK G	00D.
Tail #	A A .T.					
N993CF	Aircraft Type DC8-62	Fit Date 10/21/98	Fht # EB511	Fit Leg KSLC 40- KDAY	Delay Length 1 Hr. 10 Min.	ATA 2811
Discrepance	v:			Corrective Action:		2311
•	K RT INBOARD FI	LAP AREA			. #3 FUEL TANK ACCES LEAKS NOTED.	S PANEL LAW
Tail#	Aircraft Type	Fit Date	Fit#	Fit Leg	Delay Length	ATA
N993CF	DC8-62	10/30/98	EB323	KMSP -to- KDAY	1 Hr. 42 Min.	2321
Discrepance	λ <u>.</u>			Corrective Action:		
	ND #2 RADIOS INC MOUNT OF WATE A AREA				ON #2 POS, SWAPPED & FE MICS, OPS CHEC	
Tail#	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA
N996CF	DC8-62F	10/9/98	EB309	KDEN -to- KDAY	0 Hr. 45 Min.	2321
Discrepancy	y:			Corrective Action:		
#1 VHF X-	MITTER WILL TRA VERY WEAK ANI	Ansmit but v D broken.	VILL NOT		MIKE SWITCH STUCK AL POSITION. SYS OP	
Tail #	Aircraft Type	Flt Date	Fh#	Fit Leg	Delay Length	ATA
N997CF	DC8-62F	10/27/98	EB039	KMEM -to- KDAY	0 Hr. 34 Min.	2++1
Discrepancy	r.			Corrective Action:		
PRIOR TO	DEPARTURE GRO	UND POWER I	NOP.	#810389 CAT C DU CLOSED KDAY 10- RELAY. GROUND	DEFERRED IAW MEL 2 E 11-6-98 PLACARD IN: 27-98 R/R EXTERNAL POWER OP CHECKS G S DMI #810389 PLACAR	STALLED. POWER OOD IAW DC8
Tail#	Air u a ft Type	Flt Date	Fit #	Fit Leg	Delay Length	ATA
N997CF	DC8-62F	10/30/98	EB006	KDAY -to- KJFK	0 Hz. 42 Min.	3423
Discrepancy				Corrective Action:		
#I ADI INT	TERMITTENTLY P	RECESSES.		FEMOVED AND RE	EPLACED #1 VERTICAL	. GYRO. ≉I

DC8	FLEET				October 1998		
T±ll # N998CF	Aircraft Type DC3-62F	Fit Daic 10/16/98	हा: ∉ EB006	Fit Log KDAY 40- KJFK	Delay Length 0 Hr. 00 Min.	ATA 7230	
Discrepano	aði:			Corrective Action:			
(BTB) ENG #2 COMPRESSOR STALL AT T/O PWR ABORTED T/O.				REMOVED FOD FI NONE FOUND. OI	ROM ENG, INSPECTED PS CKECKS GOOD.	FOR DAMAGE	
Tail # N998CF	Aircraft Type DC8-62F	Flt Date 10/29/98	Fit # EB039	Fit Leg KMEM -to- KDAY	Delay Length 5 Hr. 50 Min.	ATA 3233	
Discrepanc	y:			Corrective Action:			
PRIOR TO LEAKING	DISPATCH FOUN	D L/H MILG PIV	OT GLAND	R/R L/H MLG PIVO CK NORMAL	PT GLAND IAW MM CH	32-31-01 OPS	
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA	
N998CF	DC8-62F	10/31/98	EB306	KDAY 40- KROC	0 Hr. 42 Min.	3423	
Discrepanc	y .			Corrective Action:			
	ATT FLAG INTERN I FREEZES IN THA		OMES INTO		AIRED FO VG CANNON EPLACED FO VG, SYS (

DC	3	FL	E	E	7	ľ
DC	3	FL	E	E	7	

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Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N500MH	DC8-71F	11/11/98	EB0046	KDAY -to- KELP	Cancelled	2311
Discrepand	cy:			Corrective Action:		
FUEL LE	AKING AT FLAPS	BETWEEN #1	AND #2 ENG.	PERFORMED TAN UNITED DC-8 MM	VK UNIT CONDUIT RE I 28-42-09 STEP 4 NO L	PAIR IAW EAKS NOTED.
Tail # N603AL	Aircraft Type DC8-73F	Fit Date 11/13/98	Fit # EB0101	Fit Leg KPDX -to- KDAY	Delay Length Cancelled	ATA 5753
Discrepano	y:			Corrective Action:		
R.H. FLAP	STATION 219 OU	TBD FITTING	CRACKED	REPLACED FITTI	NG OPS CKS OK	
Tail#	Aircraft Type	Flt Date	Fit#	Flt Leg	Delay Length	ATA
N603AL Discrepance	DC8-73F	11/28/98	EB332	KDAY -to- KFLL Corrective Action:	1 Hr. 02 Min.	3421
#2 COMPA	ASS SYSTEM DRIF DICATOR STAYS P	TS 1/2 ° PER M ARKED ON TH	INUTE. THE IE "X" AND		IPASS INDICATOR, O M.	P CHECKS
Tail # N604AL	Aircraft Type DC8-73F	Flt Date	Fit# EB430	Fit Leg KDAY -to- KLAX	Delay Length 9 Hr. 05 Min.	ATA 2811
Discrepance		11/4/20	<i>LB</i> 430	Corrective Action:	9 FL. U. Min.	2311
	KS #1 AND #4 ENG	G PYLONS.		RESEALED SEVER MM 28-10-01, LEAF RESEALED PROBE	AL HIGHLOCKS ON # CCKD GOOD IAW MM CONNECTOR ON #4 F CCHECK GOOD IAW N	128-10-01. PYLON LAW
Tail #	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N604AL	DC8-73F	11/28/98	EB331	KFLL -to- KDAY	1 Hr. 38 Min.	2811
Discrepancy	•			Corrective Action:		٠
	AKAGE DISCOVER YLON TO WING LI			SPAR APPROX ONE ENG. ALSO NOTEI POINT FOR #1 ENG	UND FUEL SEEP CLAS E HALF WAY BETWEE D FUEL SEEP CLASS 'E STINGER AFT SPAR. LINGTS EWWA MM CF	N #1 AND #2 I' AT ATTACH OK FOR
Tail # N606AL	Aircraft Type DC8-73F	Flt Date 11/4/98	Flt # EB0238	Fit Leg KPIA -to- KDAY	Delay Length 0 Hr. 26 Min.	ATA 2421
Discrepancy		111 1179	250230	Corrective Action:	o tu. 20 Min.	4741
#4 BTR LO	CKED OPEN ON E	NGINE SHUTE EFFECTIVE.	OWN. BUSS 44 BUSS		N CONTROL PANEL.	#4 BUSS OPS

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DC8 FLEET

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Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Langth	ATA
N606AL	DC3-73F	11/4/98	EB0011	KDAY -to- KLAX	3 Hr. 26 Min.	2421
Discrepand	:y:			Corrective Action:		
	#4 ENG SHUTDOW CT PWR APPLIED I		,		EPLACED #4 GENERA ERATOR CONTROL P.# THIS TIME.	
Tail# ·	Aircraft Type	Fit Date	Flt #	Flt Leg	Delay Length	ATA
N606AL	DC8-73F	11/4/98	EB0048	KLAX -to- KDAY	0 Hr. 28 Min.	2811
Discrepanc	y:			Corrective Action:		
	LKING ON BOTTOI ORS, AND HYD LE A			GROUP "A" SLOW NO IMMEDIATE A SERVICE. INSPEC NUT AT STAB TRI	LAW EMM FOUND LE SEEP ACTION CODE CTIONOK FOR CON' TED HYD LEAK, FOUN M MOTOR. TIGHTENE REA OPS CK GOOD NO	2, REQUIRING TINUED ND LOOSE B ID B-
Tail #	Aircraft Type	Fit Date	Fit#	Flt Leg	Delay Length	ATA
N606AL	DC8-73F	11/16/98	EB223	CYMEX -to- KDAY	Cancelled	2781
Discrepano	y:			Corrective Action:		
	SERVICE CHECK P EDGE SLOT BETV EAKING.			CHECK NO LEAK KSAN 11-18-98 IN:	LINE WITH FLEX LIN S. PLACED NON MEL STALLED NEW RIGID DRMAL. THIS CLEARS	#26. CLOSED LINE, OPS AN
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg KSAN -to- KDAY	Delay Length 0 Hr. 25 Min.	ATA 3421
N606AL	DC8-73F	11/19/98	EB011		0 m. 25 Min.	3421
Discrepano	y:			Corrective Action:		
COMPASS RMI.	S FLAG IN VIEW O	n fo's CDI Al	ND CAPTAIN'S	REMOVED AND R GYRO). OPS CHEC	EPLACED F/O'S (#2 DI) CK NORMAL.	RECTIONAL
Tail #	Aircraft Type	Fit Date	Flt.#	Flt Leg	Delay Length	ATA
N791FT	DC8-73F	11/4/98	EB0152	KDAY -to- MMEX	Cancelled	7531
Discrepano	y:			Corrective Action:		
BACK PO POWER #: FLAMED ALL ENG! CAME BA	VELING OFF AT FI WER FROM CLIMI 3 ENGINE COMPR OUT. IGNITION O INES POSITION AN ICK TO NORMAL O DNE MINUTE.	B POWER TO (ESSOR STALL VERRIDE WA (D#3 ENGINE	CRUISE .ED AND S PLACED TO . RELIT AND	IAW M/M 71-00-00 LEAKING INTERN ASSY IAW MM 75	TROUBLESHOOTING , FOUND VBV FUEL G ,AL. R&R VBV FUEL G -31-10. LOW POWER E DOD AT THIS TIME.	EAR MOTOR SEAR MOTOR

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Tail #	Aircraft Type	Fit Date	FIt #	Fit Leg	Delay Length	ATA	
N791FT	DC8-73F	11/13/98	EB0011	KSAN -to- KDAY	0 Hr. 20 Min.	2745	
Discrepan	cy:			Corrective Action:			
	LIGHT FOUND HOI INOPERATIVE.	RIZONTAL ST	AB MOTTON		H AT HORIZ STAB ELE PS CHECKS NORMAL.		
Tail# .	Aircraft Type DC8-73F	Flt Date 11/11/98	Fit # EB0035	Fit Leg KORD -to- KDAY	Delay Length 1 Hr. 35 Min.	ATA 8011	
Discrepand		11/11/70	20000	Corrective Action:	1 10.55 Will.	0011	
•	/ILL NOT START			REMOVED AND R	EPLACED #3 ENG STA REPLACEMENT STAR N P696C	•	
Tail # N796AL	Aircraft Type DC8-63	Flt Date 11/19/98	Flt # EB137	Fit Leg CYYZ -to- KDAY	Delay Length 0 Hr. 40 Min.	ATA 8011	
Discrepand	cy:			Corrective Action:			
	TART SWITCH BLI ATE DURING STAI		NOT	RELAMPED #3 ENG START SWITCH LIGHT. OPS CHECKS GOOD.			
Tail # N796AL	Aircraft Type DC8-63	Fit Date 11/20/98	Flt # EB123	Fit Leg KMSP -to- KDAY	Delay Length 2 Hr. 13 Min.	ATA 3236	
Discrepand	ey:			Corrective Action:			
GEAR WI GEAR BU RETURNE	TB DUE TO LOUD NEEL WHEN LDG. IT NOISE DID NOT ED TO KMSP WITH (OVERWEIGHT LA	RETRACTED, CEASE. AIRC LANDING WI	CYCLED RAFT	LOCK INTERMITT DOOR REMAINS O	LT NOSE GEAR DOOR ENT. CLEANED AND CLOSED. ACCOMPLISE NDING INSP. ALL CHI MANUAL CH 2.	LUBED STRUT ÆD	
Tail #	Aircraft Type	Flt Date	Fit #	Flt Leg	Delay Length	ATA	
N796AL	DC8-63	11/20/98	EB382	KDAY -to- KOAK	0 Hr. 00 Min.	7221	
Discrepano	ey:			Corrective Action:			
ON POST FLIGT WALKAROUND FOUND #4 ENGINE FAN BLADES NICKED.			PRESSED OUT BLADES IAW PRATT & WHITNEY MM 72-00. OK FOR SERVICE.				
Tail # N796FT	Aircraft Type DC8-73F	Flt Date 11/11/98	Fit # EB0025	Fit Leg KSEA -to- KDAY	Delay Length 0 Hr. 00 Min.	ATA 3233	
Discrepano	cy:			Corrective Action:			
UPON GE GEAR DO LATCHEI	AR RETRACTION, XXRS NOT LATCHE D LIGHT AND ALT GHTS TO REMAIN	ED LIGHT, GEA ERRNATE MA	AR NOT		EPLACED RT MLG BO RETRACTION OPS CH 1-62.		

DC8	FLEET		DELA	YSUMMARY	Nava	- h 1000
					14046	mber 1998
Tail # N796FT	Aircraft Type DC8-73F	Fit Date	Fit#	Flt Leg	Delay Length	ATA
		11/16/98	EB057	KFLL -to- KDAY	Cancelled	2811
Discrepan	•			Corrective Action:		
FUEL LE	AK LEFT WING AT	7#2 PYLON.		PARA 'C'. REINST ENTERED HEAT I DUE WITHIN 50 F 11-17-98 REMOVE	3584-558 TUBE IAW U. ALLED, NO LEAKS NO REAT PROCESS ON N LT HRS OR 7 DAYS. C ED AND REPLACED FU ON MEL 320 CLEAREI	OTED. ON MEL 320 LOSED KATL JEL TUBE NO
Tail #	Aircraft Type	Flt Date	Flt #	Flt Leg	Delay Length	ATA
N797AL	DC8-63	11/5/98	EB0115	KPHL -to- KDAY	3 Hr. 14 Min.	5755
Discrepano	:y:			Corrective Action:		
ON WALK	C AROUND RIGHT	INBOARD SL	OT OPEN.	REPLACED SLOT AS PER MM 27-80	DOOR FITTING, OPS C	HECK GOOD
Tail #	Aircraft Type	Fit Date	Fit#	Flt Leg	Delay Length	ATA
N797AL	DC8-63	11/13/98	EB0117	KEWR -to- KDAY	0 Hr. 20 Min.	5234
Discrepano	y:			Corrective Action:		
	UE TO CARGO DO PER NON-MEL DEI		MANUALLY		ED CARGO DOOR, VE RITY OF LATCH PINS,	
Tail #	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N797AL	DC8-63	11/29/98	EB305	KROC -to- KDAY	3 Hr. 41 Min.	2811
Discrepand	y:			Corrective Action:		
	LAROUND FOUND EDGE INBOARD (INBOARD OF #3 E?	O LEAKING GAMMA (NGINE. REPLACED SE K CHECK GOOD, NO I	ALS IAW DC-8
Tail #	Aircraft Type	Flt Date	Flt#	Fit Leg	Delay Length	ATA
N8076U	DC8-71F	11/8/98	EB0380	KDAY -to- KLAX	0 Hr. 24 Min.	3419
Discrepanc	y:			Corrective Action:		
	ARNING WENT OF D LAND 130 KTS.	F AT FL 35.0 3	00KTS ALSO	R+R STALL WARN	ING XDCR OPS CK GO	OOD.
Tail # N8076U	Aircraft Type DC8-71F	Fit Date 11/13/98	Flt # EB003 l	Flt Leg KFLL -to- KDAY	Delay Length 0 Hr. 27 Min.	ATA 2752
Discrepanc	y:			Corrective Action:		
STALL WA	ARNING CAME ON RAIGHT & LEVEL		N, AGAIN AT		DUND CONTROL REL. PER MM 27-80.	AY, R2-63. OPS

DC8 FLEET			DELAT	SUMMARY	November 1998		
Tail # N3084U	Aircraft Type DC3-71F	Fit Date 11/8/98	Fit ≠ EB0331	Fit Leg KOAK -to- KDAY	Delay Length Cancelled	ATA 3023	
Discrepanc	y:			Corrective Action:			
BLEED AI VALVE	R LEAK IN NOSE	SECTION, PO	SSIBLE SCOOP		SCOOP ANTI-ICE VA E. OPS NOW NORMAL		
Tail # . N8087U	Aircraft Type DC8-71F	Fit Date 11/26/98	Flt # EB233	Flt Leg KDFW -to- KDAY	Delay Length 3 Hr. 00 Min.	ATA 2752	
Discrepanc	y:			Corrective Action:			
FLAP WAF	NING HORN FAI	LED TEST ON	PREFLIGHT.	CLEANED FLAP FO	OLLOW UP SWITCH.	GROUND OPS	
Tail # N811AL	Aircraft Type DC8-71F	Fit Date 11/6/98	Fit # EB0013	Flt Leg KBOS -to- KDAY	Delay Length I Hr. 02 Min.	ATA 2721	
Discrepane	y:			Corrective Action:			
PREFLIGH WILL POW	7-21 6266251 077 T-FOUND THAT A ER RUDDER & E TH YAW DAMPER	NOV98 BOS C ACTUATING S XTINGUISH R	TAB TRIM		ÆD, RE-INSTALLED C R PACKAGE. OPERAT		
Tail # N811AL	Aircraft Type DC8-71F	Fit Date 11/6/98	Flt # EB0213	Fit Leg KBOS -to- KDAY	Delay Length 0 Hr. 33 Min.	ATA 2721	
Discrepancy	<i>y</i> :			Corrective Action:			
CAUSES R OFF, RUDE	ING STAB TRIM I UDDER KICK. WI DER KICK SEVER N LIGHT GOES O DN.	TH RUDDER F E AND RUDDE	POWER LEVER ER	BROKEN PART & I SEVERAL TIMES T KICKS NOTED WIT	ER & STABILIZER PAC LEAKS. OPERATED ST THROUGH FULL UP & ITH RUDDER PWR OFF JDDER IS NORMAL IA	ABILIZER DOWN, NO S SLIGHT	
Tail # N811AL	Aircraft Type DC8-71F	Fit Date 11/7/98	Flt # EB0308	Flt Leg KBOS -to- KDAY	Delay Length Cancelled	ATA 7721	
Discrepancy	<i>/</i> :			Corrective Action:	,		
AND HIGH	NG DOWN AT 053 EGT. RETARDED EAD 804 DEGRE	POWER LEV		REMOVED AND RI CHANGE CARD M	EPLACED #4 ENG IAW E091.	EMERY ENG	
Tail # N8177U	Aircraft Type DC8-71F	Fit Date 11/10/98	Flt # EB0011	Fit Leg KMCI -to- KDAY	Delay Length 2 Hr. 36 Min.	ATA 552	
Discrepancy	er.			Corrective Action:			
DECLARE	BY LIGHTINING A D. ALL SYSTEMS NT WORKED OK.				CIAL INSPECTION IAW TRIKE INSP OK FOR S		

			DELAY	'SUMMARY		
DC8	FLEET				mber 1998	
Tail # N3177U	Aircraft Type DC3-71F	Fit Date 11/17/93	F(t # EB043	Fit Leg KLAX -to- KDAY	Delay Length 0 Hr. 30 Min.	ATA 3463
Discrepano	īy:			Corrective Action:		3403
ALL CAPT ARE ALL	TS FLIGHT INSTR INOPERATIVE ON	UMENTS #1 F PREFLIGHT.	MS, #2 COMM	CYCLED SHIP EL BUSES CAME UP	ECT. POWER SEVERAL NO FAULTS NOTED.	TTMES, ALL
Tail # N832AL	Aircraft Type DC8-73F	Fit Date 11/1/98	Flt # EB0013	Fit Leg KBOS -to- KDAY	Delay Length 2 Hr. 23 Min.	ATA 5234
Discrepanc	λ:			Corrective Action:		
MAX CAB	IN PREE 8PSI BOT	TH PACKS MA	x	IAW MM CHAP 52	EPLACED MAIN CARC -36-2. SEAL , AND REI AL RETAINER IAW M	LACED
Tail # N832AL	Aircraft Type DC8-73F	Flt Date 11/5/98	Fit # EB0106	Fit Leg KDAY -to- KROC	Delay Length 0 Hr. 20 Min.	ATA 5234
Discrepance	y:			Corrective Action:		
	OOR SEAL RETAI 'D ON 11/01/98. ST			RETAINER, MM CI	EPLACED CARGO DOO H 52-26-8, GROUND PR HIS CLEARS NON-MEL	ESSURE
Tail # N832AL	Aircraft Type DC8-73F	Flt Date 11/6/98	Flt # EB0305	Fit Leg KROC -to- KDAY	Delay Length 0 Hr. 47 Min.	ATA 5234
Discrepancy	<i>y</i> :			Corrective Action:		
NON-MEL ATTEMPTS	ON MAIN CARGO S TO GET DOOR T	DOOR: TOOF	K SEVERAL NUALLY	LOCK PIN ENGAGI ATTEMPTSDOOR (MAINTENANCE AT	DOOR MANUALLY CO ED. AFTER SEVERAL CLOSED AND LOCKED I KDAY REPAIRED BR MIC SWITCH, SYS OPS IE.	OKEN WIRE
Tail # N832AL	Aircraft Type DC8-73F	Fit Date 11/24/98	Flt # EB013	Flt Leg KBOS -to- KDAY	Delay Length 0 Hr. 23 Min.	ATA 2811
Discrepancy	:			Corrective Action:		
ON PREFLIGHT FOUND FUEL LEAKING FROM #3 BOTTOM OF ENGINE COWLING.			FUEL LEAK FOUND WITHIN LIMIT. REF UNITED AIRLINE MM PG 203 (71-70-02) ON MAIN FUEL PUMP.			
Tail # N865F	Aircraft Type DC8-63F	Fit Date 11/18/98	Flt # EB331	Fit Leg KDAY -to- KFLL	Delay Length 3 Hr. 40 Min.	ATA 2912
Discrepancy	7.			Corrective Action:		
HYDRAULIC FLUID LEAKING FROM TAIL SKID.				REMOVED & REPLACED RUDDER HYDRAULIC POWER PACK IAW MM 27-21-6/27-20-6 ADJUSTMENT TEST, OPS CHECKS NORMAL.		

D	C8	FL	.E	E	Г

November 1998

Tail#	Aircraft Type	Fit Date	Flt#	Flt Leg	Delay Length	ATA
N370TV	DC8-73F	11/13/98	EB0118	KDAY -to- KEWR	0 Hr. 58 Min.	2724
Discrepan	ich:			Corrective Action:		
	CONTROL MANU. VATED W/GUST LO		Τ	REPAIRED BROK SWITCH. OPS CK	EN WIRE AT RUDDER GOOD.	REVERSION
Taii # N870TV	Aircraft Type DC8-73F	Flt Date 11/19/98	Flt # EB021	Fit Leg KDFW -to- KDAY	Delay Length Cancelled	ATA 5611
Discrepan	cy:			Corrective Action:		
CENTER	WINDSHIELD OUT	PANE SHATT	ERED FL 350.		DSHIELD & TEMP CON -04. CHECKS NORMA	
Tail # N870TV	Aircraft Type DC§-73F	Flt Date 11/20/98	Fit# EB004	Fit Leg KDAY -to- KCLT	Delay Length 0 Hr. 19 Min.	ATA 5271
Discrepan	cy:			Corrective Action:		
	RGO DOOR LACK OT INDICATED LOC				ERIFIED ALL LOCK PR D AND VENT DOOR LO	
Tail # N870TV	Aircraft Type DC8-73F	Fit Date 11/28/98	Fit# EB424	Fit Leg KDAY -to- CYMX	Delay Length 0 Hr. 20 Min.	ATA 8011
Discrepan	cy:			Corrective Action:		
RELEASE	TING ENGINE #1 W ED MANIFOLD PRE ALVE OPEN LIGHT	SSURE RECO	VERED BUT	08-98 PLACARD IN 98 CLEANED CAN	AW 80-1 CAT C DMI #8 ISTALLED. CLOSED # NON PLUG ON START CONTROL #811362 RE	MCO 12-01- VALVE THIS
Tail# N921R	Aircraft Type DC8-63F	Fit Date 11/10/98	Fit# EB0018	Fit Leg KSLC -to- KRNO	Delay Length 1 Hr. 27 Min.	ATA 2421
Discrepano	cy:			Corrective Action:		
T.O. INTI TAKEOFF PUMP, M CIRCUIT EACH TH IN.) UPC	ACK O HEAT LIGH ERMITTENT DURIN F, NUMBER ONE AL LAIN TANK BOOST BREAKERS POPPE GREE POPPED, MID ON GEAR EXTENSIO P TO 2,100 PSI. NU	NG CRUISE. A LTERNATE TA PUMP AND F ED.(TOP AND DLE BREAKE ON, SPOILER I	AFTER ANK BOOST EED PUMP BOTTOM OF R STAYED PRESSURE	TRANSFORMER FA OVERHEAT LIGHT #1 CURRENT TRAI BREAKERS ALL PI SPOILER PUMP CU CHECKS GOOD 3.0	ENT LIMITERS AT #1 C AN OPS CHECKS GOOD TREPLACED CURREN NSFORMER RESET BO UMPS OPS CHECK GOO JRRENT LIMITERS PUT 100 PSI SPOILERS OPS AN #1 ENGINE #1 EPR	D. NO I LIMITER AT OST PUMP DD.REPLACED MP OPS CHECK
Tail# N921R	Aircraft Type DC8-63F	Flt Date	Fit # EB025	Fit Leg KSEA -to- KDAY	Delay Length 0 Hr. 45 Min.	ATA 3421
		11/1//70	CD02J		0 111. 75 Willi.	J-761
RMI/FO'S	cy: IBACK #2 DIRECTION IN HIS) FLAGS CAME IS CARDS STOPPED	EIN VIEW AN		Corrective Action: R/R #2 DIRECTION DC-8 MM 34-21-1.	IAL GYRO. OPS CHEC	KS GOOD IAW

DC8 FLEET			DELA	Y SUMMARY	Nover	November 1998		
Tail # N921R	Aircraft Type DC8-63F	Fit Date 11/21/98	Flt # EB037	Flt Leg KATL -to- KDAY	Delay Length 1 Hr. 10 Min.	ATA 5282		
Discrepan	icy:			Corrective Action:				
LH OUTE	BOARD FLYTNG GE	AR DOOR LIN	K BROKEN.	REPLACED DOOR	LINK PER 32-10-0 MM			
Tail# N950R	Aircraft Type DC8-63	Fit Date 11/11/98	Flt # EB0123	Flt Leg KMSP -10- KDAY	Delay Length 1 Hr. 01 Min.	ATA 2421		
Discrepan	cy:			Corrective Action:				
	I BACK #1 GEN UN 'R FAIL. RETURNE			PER 24-1 CAT C DE PLACARD INSTAL GEN CONT PNL, G	ISCONNECTED #1 CSE UE DATE 11-22-98, DM LED. REMOVED AND IRD CHKS NORMAL ON I #811164 PLACARD RE KDFW)	I#811164 REPLACED#1 NENG RUN,		
Tail # N950R	Aircraft Type DC8-63	Flt Date 11/21/98	Flt# EB031	Fit Leg KCAE -to- KDAY	Delay Length 3 Hr. 22 Min.	ATA 2925		
Discrepan	cy:			Corrective Action:				
	ANDING SPOILERS TO PUMP CONTRO			LEAK CHECK GOO RETRACT NORMA	RAULIC PUMP. OPS C DD. SPOILERS EXTENI LLY NO DEFECTS NO ILER SYSTEM RESERV	D AND TED. MAIN		
Tail # N957R	Aircraft Type DC8-63F	Fit Date 11/21/98	Fit # EB018	Fit Leg KSLC -to- KRNO	Deiay Length 0 Hr. 00 Min.	ATA 7831		
Discrepand	cy:			Corrective Action:				
HYD LEA	.K. #2 ENGINE SUSF	PECT REVERS	ER.	DEACTIVATED IA PROCEDURES CAT INSTALLED. CLOS AND RETRACT HY LEAKS NOTED. RI	J REVERSER IAW MEL W MEO 87 AND C/W M I 'C' CONTROL #811293 SED KSEA 11-29-98 R/I 'D LINES TO SLOW/LA EACTIVATED REVERS THIS CLEARS MEL C '8-30-5.	AINTENANCE PLACARD REXTEND TCH CYL. NO ER, OPS		
Tail #	Aircraft Type	Fit Date	Flt#	Fit Leg	Delay Length	ATA		
N957R	DC8-63F	11/24/98	EB017	KRNO -to- KDAY	0 Hr. 50 Min.	0		
Discrepan	cy:			Corrective Action:				
LIGHT IL	N PUSH BACK RUD LUMINATED WITH & RUDDER POWE	GUST LOCK		OPERATED RUDDER HYD SYS AS PER DC8 MM CH 27 SYS PRESS GOOD. OPS CHECK OK. DETERMINED LIGHT TO BE CAUSED FROM CAUSED FROM HIGH WINDS ON THE FIELS.				

DC8	FLEET	-	OLLA!	SUMULARY	November 1998		
Tail # N959R	Airaraft Type DC8-63F	Fit Date 11/10/98	Fit # EB0115	Fit Leg KPHL -to- KDAY	Delay Length 0 Hr. 29 Min.	ATA 5273	
Discrepand	ıy:			Corrective Action:			
	OOR LIGHT ON DU CONT # 811149 Ca ED.			ADJUSTED STRIK OPS CK GOOD. TH REMOVED.	ER ON B-PIT DOOR SY NS CLEARS DMI #8111	vitch strike: 49 Placard	
Tail # · N961R	Aircraft Type DC8-73F	Flt Date 11/11/98	Fit # EB0233	Fit Leg KORD -to- KDAY	Delay Length Cancelled	ATA 5312	
Discrepano	;у:			Corrective Action:			
FOUND A LONG. 301	VC DAMAGED TO R	SHEETMETAI	L AT STA. 160,	REPAIRED DAMA I.A.W. DC8 SRM 53-2-1 CLASS	GED SKIN AT STA 160 THREE REPAIR	-LONG30R	
Tail # N961R	Aircraft Type DC8-73F	Fit Date 11/14/98	Fit # EB0174	Fit Leg KDAY -to- EBBR	Delay Length 0 Hr. 38 Min.	ATA 3423	
Discrepano	y:	:		Corrective Action:			
ARTIFICL VERTICA	AL HORIZON SHO L GYRO.	WS 8' CLIMB S	SUSPECT	R+R #1 VG CAPT A MM CH 34.	ADI OPS CHECKS NOR	MAL IAW DC-	
Tail # N990CF	Aircraft Type DC8-62	Flt Date 11/5/98	Flt # EB0522	Fit Leg KDAY -to- KDFW	Delay Length 1 Hr. 36 Min.	ATA 2821	
Discrepano	;y:			Corrective Action:			
ON POST	FLIGHT FOUND F	UEL LEAK #3 1	PYLON	R&R'D GAMMA SEALS			
Tail # N990CF	Aircraft Type DC8-62	Fit Date 11/26/98	Fit# EB107	Fit Leg KBSM -to- KDAY	Delay Length 1 Hr. 13 Min.	ATA 2741	
Discrepancy: H STAB TRIM YOKE SWITCH OPERATES IN UP POS ONLY - BOTH SIDES.				Corrective Action: R/R H STAB TRIM ACT IAW DC8 MM 27-41-10. OPS CHECKS NORMAL. NOTE: REINSTALLED BORROWE PART FROM UPS. SEE LOG PAGE 7278-20.			
Tail # N992CF	Aircraft Type DC8-54F	Flt Date 11/4/98	Fit # EB0039	Fit Leg KMEM -to- KDAY	Delay Length 0 Hr. 00 Min.	ATA 2911	
Discrepano	sy:			Corrective Action:			
ENG WEN	RT HYD QUANTIT VT TO 60 PSI AND D TO IDLE AND TE	ENG OIL TEM	P TO 143 DEG	ALLOWING FLUII SUPPLY. DRAINE R&R OIL FILTER (CONTAMINATION SYSTEM, R7R OIL PERFORMED ENC PERFORMED LEA SUPPLY, NONE N	YD PUMP INTERNAL S O TO FLOW INTO #2 EN ID, FLUSHED ENG OIL & ENG HYD PUMP. FO IN IN #2 OIL SUPPLY, FO FILTER, SAMPLE TAB FOWER RUN, NO DEI IX CHECKS OF ENG ON OTED, PERFORMED PO O DEFECTS NOTED.	NG OIL TANK 2 TIME DUND HYD LUSHED KEN, FECTS NOTED L & HYD	

DC8 FLEET			DELAY	SUMMARY	November 1998		
Tail # N992CF	Aircraft Type DC8-54F	Fit Date 11/24/98	Flt# EB039	Fit Leg KMEM -to- KDAY	Delay Length 0 Hr. 30 Min.	ATA 5282	
Discrepund	sy:			Corrective Action:			
	LG FLYTNG DOOR EQUIRED.	ATTACH BRA	\CKET, GE.\R	GEAR SWING ACC	OMPLISHED.		
Tail # N993CF	Aircraft Type DC8-62	Flt Date 11/6/98	Flt # EB0309	Fit Leg KDEN -to- KDAY	Delay Length 0 Hr. 00 Min.	ATA 5234	
Discrepano	;y:			Corrective Action:	,		
MAIN CAI CONTROI	RGO DOOR WILL	NOT LOCK US	ING HYD		ARGO DOOR RESERV POWER CONTROLS.	OIR. CLOSE	
Tail # N995CF	Aircraft Type DC8-62	Fit Date 11/4/98	Fit # EB0040	Fit Leg KMEM -to- KDAY	Delay Length 0 Hr. 57 Min.	ATA 3261	
Discrepano	y:			Corrective Action:			
DOORS N ARE ON, O HELP. AC	OACH NO GREEN OT LATCHED & G CYCLED THE GEA CCOMPLISHED QR ÆD EMERGENCY	EAR NOT LAT IR SEVERAL T IH PROCEDUR	CHED LIGHTS TMES NO	LIMIT SWITCH INC	OUND NOSE GEAR DO DP. REMOVED AND R ED TEST TO SYS OPS (0.	EPLACED	
Tail # N995CF	Aircraft Type DC8-62	Flt Date 11/5/98	Fit # EB0228	Flt Leg KCAE -to- KMCO	Delay Length 1 Hr. 28 Min.	ATA 7111	
Discrepano	:y:			Corrective Action:			
ON POST DOOR OP	FLIGHT FOUND #: EN.	ENGINE (INE	D) BLOWOUT	SECURED BLOWO	UT DOOR ON #3 ENG.		
Tail # N995CF	Aircraft Type DC8-62	Fit Date 11/5/98	Flt # EB0039	Flt Leg KMEM -to- KDAY	Delay Length 0 Hr. 38 Min.	ATA 7831	
Discrepano	:y:			Corrective Action:			
PRIOR TO DISPATCH REVERSER PRESS INDICATES SYSTEM PRESSURE AT ALL TIMES - T/R HYD SHUT- OFF VALVE ELECT ACTUATED OPEN			FOUND #1 T/R STOW LATCH SWITCH OUT OF ADJUSTMENT ADJUSTED #1 T/R STOW LATCH SWITCH IAW MM CH 78-30 OPS CK NORM				
Tail# N995CF	Aircraft Type DC8-62	Fit Date 11/6/98	Fit # EB0227	Fit Leg KMCO -to- KDAY	Delay Length Cancelled	ATA 3611	
Discrepand	cy:			Corrective Action:			
#1 PNEUMATIC BLEED (HIGH) INOP				R&R HP BLEED VALVE & INTERSTAGE BLEED VALV GRND RUN SYS. OP'S CKS NORMAL			

DC8 FLEET		DELAY	SUMMARY	November 1998			
Tail # N995CF	Aircraft Type DC8-62	Flt Date 11/24/98	Fit ≠ EB309	Fit Leg KDAY -to- KDEN	Delay Length 1 Hr. 30 Min.	ATA 7331	
Discrepan	cy:			Corrective Action:			
AND #2 F PWR SOU RESET B C/B OPER	INDICATOR C/B'S F CB RESET #3 & # JRCE TO RIGHT & 9 OTH C/B STAYED & NED AGAIN 1 & 2 S DPS WITH PWR SOL	4 WOULD NO C/B OPENED. APPROX 2 MIN TAYED CLOSI	T. SWAPPER ON LANDING IS AND 3 & 4 ED. #3 & #4		ED GROUND WIRE AT PS CHECKED OK.	*4 ENG F.F	
Tail #	Aircraft Type	Fit Date	Fit #	Fit Leg	Delay Length	ATA	
N997CF	DC8-62F	11/19/98	EB107	KBSM -to- KMCI	0 Hr. 17 Min.	7712	
Discrepan	cy:			Corrective Action:			
	OPERATIVE. NO IN THAN MINOR FLUX			SWAPPED #3 AND TROUBLESHOOTT HANDBOOK.	I N2 TACH FOR NG. OPS NORMAL PE	R EWWA RUN	
Tail #	Aircraft Type	Fit Date	Flt #	Fit Leg	Delay Length	ATA	
N997CF	DC8-62F	11/20/98	EB228	KDFW -to- KDAY	0 Hr. 00 Min.	2130	
Discrepan	cy:			Corrective Action:			
FROM DI WAS CON STABILLI 8.4 PSID. LEVER V FULL OP CABIN W 15,000' O	.0 A 4TH TC WAS T ESCENDING TOO F NTROLLED MANU. ZED. AT 300 FPM D ABOUT 45 SEC AF VAS RELEASED. O EN. COULD NOT C /AS AT APPROX 17 UTFLOW VALVE W IL LANDING.	AST. OUTFLO ALLY BY FE U DIFFERENTIAL TER OUTFLO UTFLOW VAL LOSE MANUA ,000', DESCEN	OW VALVE INTIL I PRESS WAS W VALVE VE WENT ALLY UNTIL DED TO	RELIEF VALVES, (ENG RUN PRESSU GROUND, NO LEA	PCTION OF OUTFLOW CARGO DOOR SEAL. I RIZED ACFT OPS CHE KS OR DEFECTS NOTI ED FOR SERVICE.	PERFORMED CKS GOOD ON	
T-:1 #	Airont Tra-	Fit Date	Flt#	· Fit Leg	Delay Length	ATA	
Tail # N998CF	Aircraft Type DC8-62F	11/5/98	EB0006	KDAY -to- KJFK	4 Hr. 40 Min.	2222	
Discrepan				Corrective Action:			
PTC EXTENDED & WILL NOT TEST.				REMOVED PTC ACT. REPLACED WITH SERV UNIT. OPS CHECKED GOOD.			
Tail # N998CF	Aircraft Type DC8-62F	Flt Date 11/13/98	Flt # EB0392	Fit Leg KMEM -to- KDAY	Delay Length Cancelled	ATA 7932	
Discrepar	ncy:			Corrective Action:			
•	YE OIL TEMP INDI	CATOR READS	S 40°C AND	R&R#2OILTEM	P SENSOR IAW MM CI	H 79-34-1 OPS	



January 18, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines Inc. (EWA)'s follow-up (initial letter sent 1-15-99 attached) formal response to your letter of investigation (99WP150025), addressed to EWA's President and Chief Operating Officer, dated January 7, 1999.

The Manager of Reliability has prepared a comprehensive fix and action plan to prevent future occurrence of this inadvertent isolated event (See Attachment).

It is equally important to consider and review the overall performance of EWA's Approved Maintenance Program over the past nine (9) years. I have provided a score card (See Attachment) that demonstrates the overall above average performance of EWA's Technical Services Department.

The very achievement of a 98% Mechanical Dispatch Reliability average for the past nine (9) years of an aging fleet reflects the overall achievement of an effective maintenance program under sincere management oversight and leadership.

EWA has maintained an average of two (2) pilot reports per flight hour since 1990. It is important to note that EWA increased its flight hours by 9% in 1998 and decreased the number of PIREP'S per flight hour by 25%.

This performance measurement also demonstrates the effectiveness of EWA's training program and manual system as the mechanics performance is a direct result of EWA's program administration.

Mr. Joe Abramski Page 2 January 18, 1999

I trust EWA's comprehensive fix is satisfactory. Should you wish to discuss this matter more fully, EWA's Senior Management and I would be more than willing to meet with you and address any concerns.

Sincerely,

Thomas M. Wood

Director Quality Control

TMW/re

Attachment

cc:

Kent Scott Rene' Visscher QC Managers

FAA TRAINING AWARDS

EMERY WORLDWIDE AIRLINES FOURTH ANNUAL FEDERAL AVIATION ADMINISTRATION TECHNICAL AWARDS PRESENTATION 1997

EWA Accomplishment Overview

Emery Worldwide Airlines (EWA) is pleased to receive for the fourth consecutive year, the Federal Aviation Administration (FAA) Technical Awards presented to the EWA Mechanics, Technical Service Management, Senior Director Technical Services, and Vice President and General Manager. A chronological history of the awards received to date is presented for your review.

1994 FAA Awards

The awards received during a ceremony on May 11, 1994 were as follows:

1. Mechanical Technical Awards

133 mechanics were presented these awards which represented 42% of the EWA mechanics.

This 42% or 133 mechanics actually represent 96% of EWA's full-time mechanics.

2. Organizational Awards

The highest award, the Diamond Certificate of Excellence was presented to Emery Worldwide Airlines.

S. <u>Macter Mechanic Award</u>

This prestigious aviation career accomplishment was presented to Mr. Roy Deeming. The requirement of selection for this award is fifty (50) years of serving as a certificate airframe and powerplant mechanic.

1995 FAA Awards

1. Mechanical Technical Awards

EWA employed 304 technicians/mechanics. Out of these, 228 or 75% have received awards. This was a 33% increase in training EWA personnel from the previous year.

This 75% or 228 mechanics actually represent 60% of EWA's full-time mechanics.

2. Organizational Awards

For the second consecutive year, required training percentage achieved by EWA surpasses the requirement stated in the Advisory Circular. The Diamond Certificate of Excellence requires 25% of eligible employees to be trained. Therefore in view of the great achievement of training rendered to its employees, EWA qualified itself to receive again the Diamond Certificate of Excellence.

1996 FAA Awards

1. Mechanical Technical Awards

EWA employment 320 technicians/mechanics. Out of these, 264 or 83% received awards. This is a 14% increase in training EWA personnel from the previous year.

2. Organizational Awards

For the third consecutive year, the required training percentage achieved by EWA surpassed the requirement stated in the FAA Advisory Circular. Therefore, in view of the great achievement of training rendered to its employees, EWA qualified and received the Diamond Certificate of Excellence Award.

1997 FAA Awards

1. Mechanical Technical Awards

EWA employed 338 technicians/mechanics. Out of these, 181 or 54% received awards. This is a 49% decrease in training EWA personnel from the previous year. This decrease reflects the previously accomplished extensive training provided in the previous seven years.

2. Organizational Awards

For the Fourth consecutive year, the required training percentage achieved by EWA surpassed the requirements stated in the FAA Advisory Circular. Therefore, in view of the great achievement of training rendered to its employees, EWA qualified for and received again the Diamond Certificate of Excellence Award.

Awards Summary:

This training is a direct contribution to the continued success of EWA. We have experienced for the past nine years an average of 98% Mechanical Dispatch Reliability performance, a standard desired by many Air Carriers.

These FAA awards exemplify EWA's professional approach to lead its employees to produce the highest level of safety possible and the most cost effective process to provide the customer the best product.

C. EWA'S Maintenance Program Continues to Produce Successful Results

Emery Worldwide Airlines Maintenance Program is tested by other means than it's Mechanical Dispatch Reliability that has maintained 98% average over the past nine years. EWA has gone through several very in-depth FAA/DOD/Outside Firms inspections over the past nine (9) years. The successful results of these inspections continued to reveal EWA's ratings to be higher than the Industry performance of the 121 Air Carriers and average to excellent ratings from the Department of Defense (DOD).

In 1992, EWA went through a very in-depth FAA NASIP Inspection to which EWA rated 64% higher than the Industry performance of the 121 Air Carriers. EWA received honorable recognition for this achievement from the San Jose FAA Certificating Holding Office Manager.

In 1995, EWA received a specific FAA inspection that was administered by FAA Washington, DC to be accomplished on all 121 Air Carriers in 1995. This inspection was titled a Regional Aviation Safety Inspection Program (RASIP). This inspection lasted ten days and covered the Operations/Maintenance Departments. On June 22, 1985, the FAA RASIP team provided EWA Senior Management a debrief of their findings. The team reported that their inspection did not reveal any major discrepancies and overall EWA was above average in performance.

In 1997, EWA received a comprehensive Internal Evaluation performed by the SH&E International Air Transport Consultancy. This evaluation was performed based on the FAA NASIP items to ensure EWA has adequate systems and controls in place to support the growth of the airline. A report was provided to EWA Senior Management from the SH&E team that reflected an excellent rating of the Technical Services Organization. Their report specifically reflected that all aspect of the necessary systems and controls were in place and performing excellent ratings.

EWA Technical Services Department has gone through four Department of Defense (DOD) inspections in the past nine years. We received average to excellent ratings on all inspections.

EWA's Maintenance Program success is a direct result of true team effort promoting synergy.

Another indicator for EWA's performence is reflected by the law number of Fag. Enforcement Actions received. The falls wing data provides an ensignous summary of this performance.

EMERY WORLDWIDE AIRLINES MAINTENANCE PERFORMANCE BASED ON FAA SAFETY INSPECTION/ENFORCEMENT HISTORY

YEAR	#ADMIN ENFORCEMENT'S	FAA <u>NPTRS</u>	FLEET SIZE	FLT <u>HOURS</u>	CYCLES	PILOT REPORTS
1990	4	Ref. Total	7	11,070	4,732	3,679
1991	3	Ref. Total	20	28,095	12,565	10,512
1992	3	Ref. Total	29	40,606	20,559	17,196
1993	2	Ref. Total	29	42,473	20,718	15,443
1994	1	Ref. Total	37	52,465	23,704	16,667
1995	2	Ref. Total	37	55,178	25,169	16,280
1996	1 '	Ref. Total	39	57,994	23,960	15,284
1997	0	Ref. Total	43	62,405	28,127	14,760
1998	1	Ref. Total	43	68,140	32,561	22,061
TOTALS	17	4,944		418,426	192,095	131,882

EWA PERFORMANCE FACTORS SUMMARY - 1/90 THROUGH 12/98

- During the nine (9) year period of Air Carrier Operations, EWA Technical Department experienced the following:
 - FAA Administrative Enforcement's compared to # of Safety Inspections = .3%
 - FAA Administrative Enforcement's compared to # of Flight Hours = .004%
 - FAA Administrative Enforcement's compared to # of Flight Cycles = .008%
 - FAA Administrative Enforcement's compared to # of Pilot Reports = .01%
- EWA's FAA Administrative Enforcement's are minor in numbers as represented during 1990 thru 1998.
- EWA increased its fleet size by 22% in 1094 and decreased its number of PIREP's per flight hour by 5%.
- EWA increased its fleet size by 6% in 1996 and decreased its number of PIREP's per flight hour by 3%.
- EWA increased its fleet size by 10% in 1997 and decreased its number of PIREP's per flight hour by 10%.
- EWA increased its flight hours by 9% in 1998 and decreased its number of PIREP'S per flight by 25%.

FAA/SPOT RAMP INSPECTION RESULTS 1998

EWA incorporated an airline industry standard "FAA Spot/Ramp Inspection Procedures" into our Maintenance Policy and Procedure Manual (MPP) in 1995.

The purpose of this program was to enhance EWA's Continuing Analysis and Surveillance System (FAR 121.373) for the continuing analysis and surveillance of the performance and effectiveness of its inspection program and the program covering other maintenance, preventative maintenance, and alterations and for the correction of any deficiency in those programs.

It also provides direct support to FAR 119.59 to assure that EWA properly handles FAA Inspector contacts, and expedites the handling of any FAA request for information.

In 1998, 78 FAA Station Inspections of the EWA's 43 line stations were reported. A total of 173 minor findings was noted and corrected. This number of findings reflected 70% of the inspections resulted in an average of 2 write-ups per visit, and 30% no findings.

This audit performance continues to reflect EWA's compliance of FAA regulations and company policies and procedures.



February 22, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines Inc., (EWA)'s formal response to your letters of investigation (99WP150028) dated February 8, 1999, and (99WP150033) dated February 8,1999 addressed to EWA's President and Chief Operating Officer (See Attachment #1).

At the outset, I would like to assure you that your letters have merited EWA's immediate and undivided attention. EWA, as a certificated air carrier, its management and employees are fully appreciative of their responsibilities arising under pertinent laws and the Federal Aviation Regulations (FAR's) and strive to fulfill these responsibilities in a professional and conscientious manner.

These two additional letters of investigation 99WP150028 and 99WP150033 dated February 8, 1999 pertain to the same subject matter of your previous letters 99WP150028 dated January 27, 1999 and 99WP150025 dated January 7, 1999. EWA appreciates the FAA's voicing of its possible concerns, but now questions why four separate letters of investigation and three separate files, each bearing a different File Number, were required or even appropriate.

As the initial response letter to 99WP150025 addressed all issues of your subsequent referenced letters, EWA does not understand these actions of enforcement and questions if they are reasonable or prudent. Attached is a chronological order of events that represent possible redundancy of enforcement action taken.

FAA Enforcement Action Sequence of Events

<u>Date</u>	<u>Subject</u>
January 7, 1999	FAA LOI 99WP150025 dated 1-7-99 sent by EWA PMI, regarding not receiving MISR reports for September, October and November 1998. (See Attachment #6)
January 15, 1999	EWA Manager Reliability contacted the EWA PMI by telephone to acknowledge receipt of the LOI 99WP150025, and explained the details of the delay and of the inadvertent failure to send to him the Monthly Reliability Reports which contained the MISR reports, and the comprehensive fix that was in place. (See Attachment #5)

Mr. Joseph Abramski Page 2 February 22, 1999

<u>Date</u> January 15, 1999	Subject EWA Director Quality Control's initial response letter to PMI LOI 99WP150025, that included the completed MISR reports for September, October and November 1998. The letter provided initial notice of the delay and of the inadventent failure to send the Monthly Reliability Reports which included the MISR reports. (See Attachment #8)
January 18, 1999	EWA Director Quality Control's follow-up letter to the initial response to PMI LOI 99WP150025, which included a formal response from the Manager Reliability, that detailed the delayed Reliability Reports, a comprehensive fix to the Reliability Reports and the Merit Program delays. (See Attachment #7)
January 27, 1999	FAA LOI 99WP150028 dated 1-27-99 sent by EWA PMI, regarding Reliability meetings for September, October and November 1998 not having been conducted. (See Attachment #4)
February 2, 1999	EWA Manager Reliability initial response letter to PMI LOI 99WP150028, dated 1-17-99, including reiterating their telephone conversation on 1-15-99 in which they discussed the delayed Reliability Reports, program meetings, MISR and the problems experienced with the MERIT data base. This letter provided the scheduled meeting dates for the Reliability meetings; 2-12-99 review of September, October and November 1998 to which this meeting was held; and 2-26-99 review December and January data. (See Attachment #5)
February 8, 1999	EWA Director Quality Control formal response letter to PMI LOI 99WP150028, providing details that support this LOI was addressed by the previously issued LOI 99WP150025. (See Attachment #4)
February 8, 1999	FAA LOI 99WP150028 dated 2-8-99 (continuation of LOI 99WP150028 dated 1-27-99) regarding Reliability meeting for December 1998 not having been conducted. (See Anachment #1)
February 8, 1999	FAA LOI 99WP150033 dated 2-8-99 regarding activation of the MERIT database system of maintenance data collection. (See Attachment #1)

Mr. Joseph Abramski Page 3 February 22, 1999

<u>Date</u>	Subject
February 8, 1999	EWA Manager Reliability's letter to PMI to provide January 1999 MISR report, and an update of the scheduled Reliability meetings; 2-12-99 review September, October and November data; 2-26-99 review December and January data. (See Attachment #3) The completed December 1998 Reliability Report is attached.
February 11, 1999	EWA Director Quality Control's letter to PMI follow-up to LOI 99WP150028 informing him of Reliability Meeting dates and completed copies of the September, October and November Reliability Reports, to be addressed in the scheduled 2-12-99 meeting. (See Attachment #2)
February 12, 1999	EWA Director Quality Control received LOI 99WP150028 continuance and 99WP150033 both dated 2-8-99. (See Attachment #1)

EWA's continuous airworthiness maintenance program has received over nine years of internal evaluation programs that continually monitor company policies and procedures and have proven compliant to ensure that the highest level of safety and airworthiness have and are being maintained. (See Attachment #7). During the past nine (9) years of EWA's Air Carrier Operation, EWA Technical Services Department has only received seventeen (17) FAA Enforcement actions, which is an average of less than two a year.

EWA's FAA Approved Maintenance Reliability Program has successfully managed the Continuous Airworthiness Maintenance Program for over nine (9) years maintaining a consistent 98% mechanical dispatch reliability. An example of this performance is represented by the consistent decline of Reliability ATA Alert Levels (See Attachment #9).

EWA's DC-8 fleet arrival performance was compared favorably with Southwest, American Airlines and United Airlines from December 1997 to July 1998. EWA's DC-8 thirty (30) years plus aged aircraft fleet performance was comparable to the referenced major carriers with newer fleets. (See Attachment #10).

The FAA enforcement program has always been considered a means to promote compliance with the FAA's regulations, not an end in themselves. In addition to the deterrence achieved by the appropriate use of enforcement action, the public interest is also served by positive incentives to promote and achieve compliance. The Air Carrier Industry understands that the FAA believes that aviation safety is well served by incentives for certificate holders regarding compliance to identify and correct their own instances of noncompliance and to invest more resources in efforts to preclude their recurrence. EWA believes this process may not have been achieved by the aforementioned redundancy of LOI's.

Mr. Joseph Abramski Page 4 February 22, 1999

In the subject alleged non-compliance issues, EWA has proven that the alleged violations were inadvertent, did not indicate a lack of qualifications, immediate corrective action was in place and now completed by the demonstration of the comprehensive fix.

With specific regard to LOI 99WP150033, EWA elected to upgrade the EWA01 computer software system in early 1996. The Maxi-Merlin software program was purchased from U.S. Airways to accommodate the growth of EWA's fleet, and provide a Major Carrier control system that has many years of proven effectiveness. EWA has invested over 2.5 Million Dollars in this program enhancement.

I have enclosed the correspondence to your office dating back to October 11, 1996 for your review (See Attachment #11). As you can see from this correspondence, EWA has provided extensive details regarding the implementation plan of the MERIT software.

The August 8, 1997 letter to you provided the transition plan regarding the log page data being entered into the MERIT software. Revisions to EWA's Maintenance Policy and Procedures Manual regarding the MERIT system in the appropriate Chapters have also been accepted by you in 1998. I have enclosed a transition letter for the log page discrepancies and corrective action, in continuance of notification of the Merit transition process.

Summary of LOI Responses

- 1. This isolated occurrence of delayed Maintenance Reliability Reports/meetings is the first since the FAA approval of the Reliability Program in 1990.
- 2. The failure to hold the monthly Maintenance Reliability Meeting has not hindered the EWA Reliability program, as can be evidenced by the improving trend in the Mechanical Dispatch Rate for the last four months of 1998. The rate for the four months was 96.3%, 97.0%, 97.4% and 97 for an average 97% for 1998.
- 3. EWA has maintained an average of two (2) pilot reports per flight hour since 1990. It is important to note that EWA increased its flight hours by 9% in 1998 and decreased the number of PIREP'S per flight hour by 25%.
- 4. All aspects of the Continuous Airworthiness Maintenance Program was performed on a daily basis as is referenced in EWA's Maintenance Reliability Program Document No. EWA-51990. In addition daily/weekly scheduled meetings are held with the Technical Services Management (MRB members) to address all aspects of EWA's continuous maintenance program including daily corrective action. Based on the day to day interactive support system, no information was relayed that would or did cause non-compliance to the Continuous Airworthiness Program.

Mr. Joseph Abramski Page 5 February 22, 1999

- 5. EWA elected to go to the U.S. Airways Maxi Merlin (MERIT) data system. This data system ties all aspects of airline operation together. With the implementation of a new system, there can be unforeseen problems. The problem experienced with the September data is a prime example of the type of problems that can occur. The problem was aggressively attached, and was put back on track in a timely manner.
- 6. A comprehensive corrective action plan was in place since October 1998 to prevent future non-compliance of our program. Aircraft log page discrepancies/corrective action was entered into the Merit software commencing in September 1998. A monthly audit by Reliability in October revealed data entry discrepancies and report errors. Reliability and Aircraft Records worked in concert to audit and verify the data entered into the MERIT Program was correct from the original log page. The Information Management section worked to correct these problems of processing the data. These program changes were complete and verified correct by audits in November.

I trust that this has been responsive to your letters. Should you wish to discuss this matter more fully, EWA's Senior Management an I would be more than willing to meet with you and address your concerns. If this is the case, please contact me to arrange the meeting.

Sincerely,

Morrow 11 US

Thomas M. Wood
Director Quality Control

TMW/re

Aπachments

cc: Kent Scott

Rene' Visscher Robert Conlon Michael Dworkin



U.S. Department of Transportation

Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

MAR 2 2 1999
KENT T. SCOTT

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028 EIR 99WP150025 is incorporated into EIR 99WP150028 EIR 99WP150029 is incorporated into EIR 99WP150028 EIR 99WP150032 is incorporated into EIR 99WP150028 EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W



Emery Worldwide Airlines

Technical Services Department

Response letter to

Mr. Jay Howard, SJC FSDO

Office Manager, dated April 2, 1999

Subject: Letter of Investigation

File No. 99WP150028 amendment of

EWA's Operations Specification

D74 and D76.



April 2, 1999

Mr. Jay Howard Officer Manager FSDO SJC 1250 Aviation Ave. Suite 295 San Jose, CA 95110

Dear Mr. Howard:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA's) formal response to your March 18, 1999, letters regarding File No. 99WP150028 addressed to Mr. Kent Scott, President and Chief Operating Officer. (Reference Attachment 1.)

I would like to assure you that your letters have merited EWA's immediate and undivided attention. EWA, as a certified air carrier, it's management and employees are fully appreciative of their responsibilities arising under pertinent laws, and under the Federal Aviation Regulations (FAR's), and strive to fulfill these responsibilities in a professional and conscientious manner.

In demonstration of this professional compliance attitude, a meeting was held with Mr. Scott and the Director, Quality Control on March 19, 1999, and an immediate proactive plan was established. This plan was discussed with you and the FAA Principals during a conference call with the Director, Quality Control the same day, and was agreed to by your office. (Reference Attachment 2.)

EWA's Director, Quality Control arranged for an immediate meeting to discuss the concerns of the FAA and provide immediate solutions to these concerns. Per Mr. Joseph Abramski, an agenda was sent representing items to be discussed with the Director of Quality Control, during his scheduled meeting at the SJC office March 23 and 24, 1999. (Reference Attachment 3.)

This letter will provide substantiation that EWA did not compromise safety in air commerce, and/or negate the public's interest, to constitute the requirement of amending EWA's Operation Specification D74 and D76. EWA requests that you thoroughly review the contents of this data package, and provide EWA the opportunity to discuss any other concerns that you may have, if you are not satisfied this constitutes our compliance position. It is our utmost desire to continue to resolve all open issues, with the first opportunity face-to-face.

Your letter cites eight (8) reasons for considering the modification of EWA's Operational Specifications. All eight (8) reasons will be addressed in this letter in the order listed in your referenced letter. Supportive data is provided as attachments in these responses.

I. FAA REASON/EWA RESPONSE

FAA Reason:

 EWA failed to conduct monthly Reliability Program meetings to review and analyze aircraft maintenance data as required in their Maintenance Reliability Program Document No. EWA 51990, for the months of October, November, December 1998, and again in January 1999.

EWA Response:

- The monthly meetings are for review of the reports only. As stated in EWA 51990, Chapter 3, Page 8; C; 9; b) ".....reviews by the Manager of Reliability pertaining to the previous month's fleet performance and reliability highlights provided in the monthly Fleet Reliability Report." Further this section of EWA 51990 clearly indicates that the analysis of the data has been performed in order to present to the meeting attendees; proposed amendments to the CAMP, and other special interest subjects. (Reference Attachment 4.)
- The duration indicated in your letter appears to be incorrect. Data collection for the September Reliability report began in the second week of October. At this time the Reliability Section discovered a possible problem with the September log book data stored in MERIT. Immediate corrective action taken by EWA resulted in an eight (8) week delay in the reporting process. The distribution of the January Reliability report placed the monthly Reliability report process back on schedule.
- The Reliability data was continually analyzed as required and actions were taken, if necessary. The meetings that were scheduled had to be canceled, not due to a lack of data, but rather due to the fact that the data required for the report was incomplete in the MERIT data base. This was explained in the EWA response letter to LOI 99WP150025. (Reference Attachment 5).
- The EWA response to the LOI number 99WP150025 included supportive information such as; 1) telephone conversations with the PMI concerning this issue had taken place with the Manager, Reliability; 2) this isolated occurrence was the first over the past nine (9) years; 3) EWA's training performance and FAR compliance record during this nine (9) year period has been exceptionally high in demonstrating results of a professional attitude as a company, to FAR compliance. This achievement by EWA is only enforced and promoted by the surveillance and Principal oversight management of your office, to which we are very grateful. (Reference Attachment 5.)

FAA Reason:

2. EWA voluntarily implemented an aircraft maintenance electronic data collection system; specifically known as Maintenance/Material, Engineering, Reliability, Information Technology (MERIT), which was not accepted by the administrator. Further, that the MERIT system was faulty and contributed to the absence of accurate fleet airworthiness data for the months of September, October, November and December 1998. This deficiency of data, and the subsequent recovery and verification of such by EWA, continued from the months of October 1998 through February 1999.

EWA Response:

- The MERIT program, much like the EWA01 system, is not listed in EWA's
 Operational Specifications. Data is collected and input into the MERIT
 system similar to the process used for EWA01 for the past nine (9) years.
 Previous revisions to the MPP reflect EWA's use of MERIT, and have been
 accepted by the Administrator.
- The MERIT system has functioned as designed, and was not "faulty" as indicated in your letter. A change in the data input process resulted in the data in MERIT to be incomplete. Because of the thoroughness in analyzing the data by the Reliability Section, this problem was discovered and immediately acted upon by EWA's Management. The data was checked against the log pages to ensure that all inputs were correct. This process took approximately five (5) weeks.
- EWA established an "EWA Computer Based Maintenance Program Manual" dated 1/20/92 with Mr. John Howard, EWA's previous PMI from January 1990 to December 1997. This procedure provided guidelines for the administration of this program, which EWA has followed to date. (Reference Attachment 6.)
- Since the implementation of the new Maxi-Merlin Computer System, EWA
 has provided correspondence to the Principal Maintenance Inspectors
 beginning in 1996 through February, 1999, and at all times received FAA
 acceptance. (Reference Attachment 7.)
- EWA elected to upgrade the EWA01 computer software system in early 1996. The Maxi-Merlin software program was purchased from U.S. Airways to accommodate the growth of EWA's fleet, and provide a major carrier control system that has many years of proven effectiveness. EWA has invested over \$2.5M in this program enhancement.

This new software provides growth options to EWA's FAA approved Reliability Program, which is under implementation consideration currently.

FAA Reason:

3. EWA failed to reasonably and prudently notify the Administrator of encountered deficiencies with respect to their MERIT system; which subsequently compromised their Maintenance Reliability Program, resulting in ineffective and inadequate program functions. Further, EWA failed to provide acceptable documentation and testing of their parallel EWAO1 and MERIT systems.

EWA Response:

- EWA responded immediately, internally, to resolve the issue of incomplete data contained within the MERIT system. At no time did the lack of information in MERIT compromise EWA's Maintenance Reliability Program. The effect on the Maintenance Reliability Program was simply a delay in the publishing and distribution of the monthly Reliability Report.
- Extensive comparison testing was performed between MERIT and EWAO1. Several months of testing all aspects of both systems was accomplished. The MERIT database was incomplete, not because of a system hardware or software problem, but due to a data input process change. In addition EWAO1 and MERIT ran parallel for nearly one (1) year to ensure that the process would be intact.
- EWA01 and MERIT operated parallel for approximately one (1) year. In that year over 20,000 man-hours were devoted to ensuring that the system functions would not have an impact once EWA converted to the MERIT database. Once the data collection issue was identified, EWA spent five (5) weeks reviewing the data and correcting errors.
- EWA provided the Merit Data Audits of the Time and Cycle Reports in August, 1997, and the log page in February, 1999. This submittal of a parallel program comparison is in compliance with the procedures. (Reference Attachment 7.)

FAA Reason:

4. EWA failed to maintain adequate Reliability Program personnel to consistently analyze, evaluate and address acquired maintenance data, relative to the complexity and composition of the EWA's forty-one (41) DC-8 aircraft and their continuing analysis and surveillance system.

EWA Response:

- EWA aircraft performance data is continually collected, monitored and analyzed by the Reliability Section. During the month of September, the Reliability Section experienced three separate events that contributed to the delay of producing the monthly Reliability Report. These events are as follows:
 - MERIT data was found to be incomplete during a routine download.
 - One of the Technical Analysts under went emergency surgery and was out of the office for 3 weeks.
 - The ongoing review of the DC-10 Maintenance Program review.
 - EWA aircraft performance and reliability statistics indicate that the Reliability Section has been, and is, adequately staffed.
 - EWA's Maintenance Program has demonstrated a proactive approach to identifying and correcting potential problems. This is done by setting the alert levels at one standard deviation for the aircraft systems, and investigating those systems that exceed their alert level. Alert levels set at one standard deviation, identifies emerging problems, and in most cases, the potential problem is related to only a small percentage of EWA's fleet. These tight alert levels allow EWA to correct potential problems before they become a fleet problem. (Reference Attachment 8.)
 - An EWA presentation was provided March 14, 1995, to the FAA SJC office personnel by the Director, Quality Control, regarding the support of Mr. John Howard's decision to approve the D74 Operations Specification, Maintenance Policy and Procedures Manual, and the Reliability Manual revisions.

The results of this meeting were very positive in resolving the misunderstandings which seemed to have existed regarding the Reliability Program. Additionally, and more importantly, the EWA and FAA Principals were able to solidify a very good working, professional relationship for the ongoing interaction between the respective members.

Based on this presentation, and substantiation provided to Mr. John Howard, March 30, 1995, EWA received Revision #5 to the Maintenance Reliability Program, Document EWA-51990, that provided control of the Reliability Program to the EWA MRB, without the requirement of being FAA approved.

It is important to note, however, that EWA has received FAA approval by the Principals for all Inspection/Maintenance Program changes from 1989 to date. The Director, Quality Control, as a courtesy to Mr. Joseph Abramski in 1997 upon assignment as the PMI, has and continues to send all changes for his approval, in order to give him time to acclimate to our program.

EWA continues to believe that the professional relationship and ability to interact with our Principals remains present, as it was demonstrated in 1995, to jointly resolve any and all FAA concerns with this substantiated data, and EWA's willingness to provide solutions real time. (Reference Attachment 9.)

- The EWA MRB organization has grown in technical, qualified staffing since 1995. The most recent addition and reorganization was the development of the Engineering Department. (Reference Attachment 10.)
- In November, 1996, the EWA Finance and Administration Division implemented a new "project oriented" organization known as Systems and Controls. This organization was established to create, improve, and support our various airline business functions through development and implementation of computer and non-computer related projects, and to maintain the integrity of the systems and information produced.

This dedicated, skilled, professional computer group was added to provide direct support to the Merit Program. (Reference Attachment 11.)

 EWA's Material Department is now under the direction of Tracy Chaplin, Director Material Management.

This department has expanded in several areas, but one important establishment was the addition of seven (7) Inventory Controllers, staffed 24 hours a day/7 days a week. This 24/7 now enables EWA Merit system to operate real time. (Reference Attachment 12.)

FAA Reason:

5. EWA failed to evaluate, analyze, and submit to the Administrator, regulatory required Mechanical Interruption Summary Report, or Mechanical Reliability Reports for the months of September, October, November and December 1998 and again in January 1999. These reports were eventually received by the CHDO on February 12, 1999 and March 1, 1999 respectively.

EWA Response:

- During the months in question, EWA's Reliability Section continued to evaluate and analyze the Mechanical Interruption Summaries. No action items were generated, as none were required.
- During the months in question EWA's Reliability section continued to evaluate, analyze and submit Mechanical Reliability Reports as required per FAR 121.703. A comprehensive fix for this reporting requirement was provided in the response letter to the LOI 99WP150023, dated January 11, 1999, from the Director, Quality Control. (Reference Attachment 17.)
- There is no specific interval for submitting a Mechanical Interruption
 Summary Report per FAR 121.705, nor does EWA 51990 state a specific
 interval for submitting this report. However, EWA's Maintenance Polices and
 Procedures Manual (MPP) does provide a procedure for sending monthly
 reports.
- The Mechanical Interruption Summary Report is part of the EWA monthly Reliability Report and as a result was not submitted during the indicated months.

FAA Reason:

 EWA failed to maintain their continuing analysis and surveillance system in a manner which identifies and corrects deficiencies, as reflected in their untimely submitted Mechanical Interruption Summary Reports for the months of September, October, November and December 1998.

EWA Response:

- The EWA Reliability Section continually collects, monitors and analyzes
 aircraft performance data. The conclusion that "EWA failed to maintain their
 CASS" based solely upon the lack of a report being submitted to the
 administrator cannot be substantiated. EWA's fleet reliability continued to
 show positive results throughout the months in question.
- It should be noted that between the months of September and November the fleet mechanical dispatch reliability increased by 1.1%.
- EWA increased it's flight hours by 9% in 1998, due to the PMPC operation, and decreased it's number of PIREP's per flight hour by 25%.
- The Mechanical Interruption Summary Report is part of the EWA Reliability Report, and therefore, was not submitted for the indicated months due to the facts provided in items 4 and 5 above.

- During this time EWA had formed a Tiger Team to analyze, evaluate and correct, repeat and chronic problems identified through the daily log pages, Merit Data System, EWA01, Reliability, and Maintenance Control. Representatives from the Tiger Team attended the daily 7:15 morning meeting chaired by the Manager of Maintenance Control to gather information concerning identified chronic and repeat problems. (Reference Attachment 13.)
- EWA's Maintenance Review Board (MRB) formally advised Mr. Joseph Abramski by letter, dated September 17, 1998, that EWA was an active member on the DC-8 MSG-3 Steering Group, and would implement the new Douglas DC-8 MSG-3 Maintenance Inspection Program upon FAA approval, (expected to be complete in August, 1999.)

In addition to this program improvement, EWA senior management contracted Avitech to develop a new Maintenance Program and Maintenance Task Cards for the DC-8 aircraft. This project is underway and projected to be complete and sent to your office for review and approval, by May 1999. EWA has invested \$125,000 to this program enhancement.

These proactive measures of the EWA MRB demonstrate the effectiveness of the CASS Program, and specifically the Reliability Program. (Reference Attachment 14.)

FAA Reason:

7. EWA failed to submit to the Administrator required Major Alteration Reports which directly impacts EWA's data collection system for the Reliability Program.

EWA Response:

- EWA provided notification to the Principal concerning the Stage III Husk Kit (major alteration) installation STC SA5455NM, initially by a copy of the referenced STC in 1998 prior to installation.
- The Aircraft Maintenance Manual was revised October 23, 1998, and received FAA approval from the Principal providing notification. (Reference Attachment 15.)
- A revision to the MPP, Chapter 4, Page 135, dated October 22, 1998, was sent to the Principals, to which the requirement of 121.707 was added to the MA procedures. This revision is currently at your office under review by the Principals. (Reference Attachment 16.)

- Based on the Director, Quality Control's discussion on March 23 and 24, 1999, at your office, I understand there was a misunderstanding of the new FAR 121.707 procedure added to the MPP revision. This procedure provides them a copy of the MA for the aircraft to be issued. Your understanding was that EWA would send each completed MA to them, for each aircraft. An action item was taken by the Director, Quality Control to revise the procedures accordingly.
- EWA's Reliability Program receives data relative to these modifications based on it's continuous airworthiness maintenance program. In no case did this adversely affect the data collection system for the Reliability Program.

FAA Reason:

8. Contrary to EWA procedures outline in the EWA Maintenance Policy and Procedures Manual and EWA's Maintenance Reliability Program Document EWA-51990, EWA escalated five (5) DC-8 aircraft "C Check" Inspection intervals without benefit of adequate Reliability Program analysis or evaluation.

EWA Response:

 As previously stated, EWA had all information available for review and evaluation. EWA only failed to submit monthly reports per our own manual procedures, which we believe falls under the category of an FAA Administrative Enforcement Action, to which a letter of correction would serve both the FAA and EWA, to promote procedural controls, as corrective action has already been accomplished.

II. SUMMARY OF SUBSTANTIATION:

EWA at no time had a lack of data available to support its fleet of DC-8's. The Technical Services staffing is adequate to manage and produce the amount of work, and the detail level of analysis required to evaluate this data. Since January, 1999, two (2) DC-8 aircraft have been removed from the Operations Specifications, and seven aircraft are scheduled to be parked this year. This reduction in the fleet will reduce workload and fleet types to monitor, therefore, improving man-hour availability.

As outlined above, a data input process change resulted in an approximate eight (8) week delay in the publishing/distribution of the EWA Reliability Report. As a result of incomplete data within the MERIT database, EWA decided to withhold publishing/distributing the September report until the end of December 1998. This was a delay of two (2) months or eight (8) weeks.

The EWA Reliability Program is based on Pilot Report (PIREP) data. This data is collected from the aircraft log pages which are submitted to Aircraft Records (hard copy), and are always available for all departments use, including the Reliability Section. To assume that an airline cannot run effectively without the aid of data management by computer is incorrect. The task of analyzing data is much easier using a computer, but it is not a requirement for operation or regulatory control by an FAR.

EWA's use of MERIT merely aids in the tracking of parts for accounting purposes and assists maintenance by generating reports quickly. Thousands of man-hours were expended making sure that the functionality of MERIT was as good as, or better than, those functions found in the EWA01 system. Every effort was taken to ensure a smooth transition to MERIT from EWA01. When it was noted that there was a problem in the data that populated the MERIT data base, immediate corrective action was taken by EWA. As stated before, this demonstrates EWA's Reliability Program performed, and continues to perform, as designed and approved by the FAA.

If EWA's Reliability Section had failed to evaluate or analyze data properly, or did not have the staff or resources to effectively perform its tasks, the problem found in the MERIT system would have taken much longer to be discovered. Only by the evaluation and analysis by the Reliability Section in preparation of the September Monthly Report, was this discrepancy found.

EWA has operated since 1989 with a growing fleet of DC-8 aircraft. EWA has always maintained a high degree of reliability at a reasonable cost. A key factor in EWA's ability to compete with other operators, is the proficiency of the Reliability section. EWA's proven 98% Mechanical Dispatch Reliability average of it's DC-8 fleet in the past nine (9) years, is relative to a successfully managed program.

I have also enclosed the FAA EIR Consolidation Notification letter received from Mr. John Howard, March 16, 1999, and a copy of the letters and EWA responses, for your information. (Reference Attachment 17.)

III. EWA PROPOSED MAINTENANCE PROGRAM PLAN

 During the FAA meeting March 23 and 24, 1999, at the SJF FSDO office, the Director, Quality Control presented a proactive plan to address the FAA's concern regarding this letter. This discussion was held in part with Mr. John Howard, and in detail with Joseph Abramski, Nick Pearson and Shawn Skaggs.

- 2. This plan consisted of some of the following items:
 - A. Provide an updated EWA DC-8 C/D Check Inspection Program that is under contract development with Avitech. A discussion was held reflecting this program implementation with a C Check interval at 24 months, with a phase in plan to be provided for FAA approval.
 - B. A previous letter of commitment was sent to Mr. Abramski, to implement the DC-8 MSG-3 Program upon approval this year, expected in August, 1999.
 - C. The development of the EWA Engineering Department provided to Joe on March 12, 1999.
 - D. Develop FAA/EWA communication policy, and incorporate into the Maintenance Policy and Procedures Manual (M.P.P.).
 - E. Revise the Reliability Program Manual to include the Director of Engineering as an MRB Judiciary Member.
 - F. Increase data to be sent to the Principals, i.e., MA's, FCD's, MSL, OEM correspondence, etc..
 - G. Upgrade the Reliability Program to monitor non-routines and other data analysis processes, as EWA is equipped to manage, or increase staffing to support upgrades agreed upon by the FAA/EWA.
 - H. EWA is maintaining the contract training support of AMT, to facilitate in the overall reliability improvement focus. In addition, EWA has purchased an interactive Computer Based Training (CBT) program to be used for EWA Line Station maintenance recurrent training.
 - Manager of Maintenance Training, Manager Quality Assurance, Director Line Maintenance and Director Base Maintenance positions will be filled in thirty (30) days.
 - J. Forward the MPP, Chapter 6 revisions currently being revised by Abraham Michael regarding new log page procedures.
 - K. Based on the proposed change to the Inspection Program to address the CPCP Level II findings found in 1998, a management decision was made to report the more conservative CPCP findings to Douglas and the FAA. These items were changed and reported on March 23, 1999.

- Revise the Inspection Program Manual, CPCP section, with additional guidelines (logic chart) for determining corrosion levels, based on written notification from Douglas concerning the DC-8 and DC-10 level assignment.
- EWA's new DC-8 Inspection Program will incorporate additional corrosion inspections to maintain Level I corrosion between scheduled inspection visits.
- The Principal's stated objection of the use of previous operators manual 3. (UAL) as EWA's manual. This was discussed and continued to agree that the solution is an EWA customized manual system. Mr. Abramski requested a time be provided for this solution. EWA has committed to a December 1999, completion date.

Per our conversation on March 30 and 31, 1999, at your office, EWA will commit to revising the M.P.P. to reflect Maintenance Manual usage based on your acceptance.

Based on the provided written substantiation, excellent compliance history of EWA, previous working professional relationship with your office, it is confirmed that no safety was comprised, and EWA should continue to hold D74 and D76 Operations Specification, as issued.

I trust this has been responsive to your letter. Should you wish to discuss this matter more fully, EWA's senior management would be more than willing to meet with you and address your concerns. If this is the case, please contact me to arrange this meeting.

Sincerely,

Bruce A. Robbins

Director of Engineering

Attachments

BR/csh

cc: Kent Scott Rene' Visscher Thomas Wood Robert Conlon Ted Ellett Michael Dworkin

, -



U.S. Department of Transportation Federal Aviation Administration San Jose Flight Standards District Office

San Jose International Aircont 1250 Aviation Avenue, Surte 295 San Jose, CA 95110-1130 Phone: (408) 291-7581 FAX: (408) 279-5448

RECEIVED

KENT T. SCOTT

March 18, 1999

Mr. Kent T. Scott
President and Chief Operating Officer
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton-International Airport
Vandalia, OH 45377

Mr. Scott:

The attached is the letter Mr. David Gilliom said would be forthcoming from the San Jose Flight Standards District Office (FSDO) regarding Emery's reliability and short term escalation programs. Basically, the attached letter proposes to amend Emery's Operations Specifications by rescinding Operations Specifications D74 - Maintenance Reliability Program; and D76 - Short Term Escalation for the reasons specified in that letter. However, the letter also gives Emery the opportunity to provide, to the San Jose FSDO, justification in the form of written information, views and arguments as to why Emery should continue to hold D74 and D76 Operations Specifications.

As stipulated in the attached letter, Emery has 10 days to respond to the proposal to amend Emery's Operations Specifications Time and D76. In the interim, no action will be taken in regard to the subject Operations Specifications until we have reviewed all material submitted by Emery. Should Emery choose not to reply within the specified time, we will continue the amendment process as outlined in the attached letter.

Sincerely,

Jáy P. Howard

Manager, San Jose FSDO

1 Enclosure

Letter: File No 99VP150028 to Kent Scott

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U.S. Department of Transportation
Federal Aviation
Administration

San Jose Flight Standards District Office

San Jose Internationa: Althori 1250 Awation Avanue Suite 255 San Jose, CA 95110-1130 Phone: (408) 251-17581 FAX: (408) 279-5448

March 18, 1999

REGISTERED RETURN-RECEIPT

Mr. Kent Scott President & Chief Operating Officer Emery Worldwide Airlines, Inc. One Emery Plaza Dayton International Airport Vandalia, OH 45377

Dear Mr. Scott:

FILE NO.: 99WP150028

This letter is to inform Emery Worldwide Airlines, Inc. (EWA), holder of Air Carrier Confifeate Number RRXA558B, that in accordance with the provisions of 14 CFR §119.51, the San Jose Certificate Holding District Office finds that safety in air commerce and the public interest requires the proposed amendment of EWA's Operations Specifications. EWA is hereby notified that their operations specifications may be amended by rescinding D74 - Maintenance Reliability Program Authorization; and D76 - Short Term Escalation Authorization. EWA may within ten (10) days after receipt of this notice, submit to our office written information, views, and arguments regarding this proposed amendment to EWA's operations specifications specified above.

After considering all material presented by EWA, the San Jose Certificate Holding District Office will notify EWA of:

- i) The adoption of the proposed amendment;
- ii) The partially adoption of the proposed amendment; or
- iii) The withdrawal of the proposed amendment.

If the San Jose Certificate Holding District Office issues an amondment to the EWA's Operations Specifications, it becomes effective not less than 30 days after EWA receives notice. At that time, EWA may petition for reconsideration under to the provisions of 14 CFR §119.51(d).

This action is necessary due to the following reasons:

- EWA failed to conduct monthly Maintenance Reliability Program meetings to review and analyze aircraft maintenance data as required in their Maintenance Reliability Program Document No. EWA-51990, for the months of October, November, December, 1998; and again in January, 1999.
- 2) EWA voluntarily implemented an aircraft maintenance electronic data collection system; specifically known as Maintenance/Material, Engineering, Reliability, Information Tochnology (MERIT), which was not accepted by the Administrator. Further, that the MERIT system was faulty and contributed to the absence of accurate fleet airworthiness data for the months of September, October, November, and December, 1998. This deficiency of data, and the subsequent recovery and verification of such by EWA, continued from the months of October, 1998 through February, 1999.

3) EWA failed to reasonably and prudently notify the Administrator of encountered deficiencies with respect to their MERIT system; which subsequently compromised their Maintenance Reliability Program, resulting in ineffective and inadequate program functions. Further, EWA failed to provide acceptable documentation and testing of their parallel EWA1 and MERIT systems.

4) EWA failed to maintain adequate Reliability Program personnel to consistently analyze, evaluate, and address acquired maintenance data relative to the complexity and composition of the EWA's forty-one

(41) DC 8 aircraft and their continuing analysis and surveillance system.

5) EWA failed to evaluate, analyze, and submit to the Administrator regulatory required Maintenance Interruption Summary Reports, or Mechanical Reliability Reports for the months of September. October, November, and December, 1998; and again in January, 1999. These reports were eventually received by the CHDO on February 12, 1999 and March 1, 1999; respectively.

6) EWA failed to maintain their continuing analysis and surveillance system in a manner which identifies and corrects deficiencies, as reflected in their untimely submitted Maintenance Interruption Summary

Reports for the months of September, October, November, and December, 1998.

7) EWA failed to submit to the Administrator required Major Alteration Reports which directly impacts

EWA's data collection system for the Reliability Program.

8) Contrary to EWA procedures outlined in the EWA Maintenance Policy and Procedures Manual, and EWA's Maintenance Reliability Program Document EWA-51990, EWA escalated five (5) DC-8 aircraft "C Check" Inspection intervals without benefit of adequate Reliability Program analysis or evaluation.

If the San Jose Certificate Holding District Office finds that adoption of this amendment as proposed, EWA will be notified, in accordance with 14 CFR §121.373(b), that your continuing analysis and surveillance program and your program covering other maintenance, preventive maintenance, and alterations, and for the correction of any deficiency in those programs, does not contain adequate procedures and standards to meet the requirements of 14 CFR Part 121. EWA may have to make changes in those programs that are necessary to meet those requirements; including the following:

- Within ten (10) after receipt of FAA notification, EWA may have to revise and obtain FAA approval
 for their Time Limits Manual reflecting original Hard Times (HT); On-Condition (OC) and Condition
 Monitoring (CM) items, cutlined in the currently revised McDonnell Douglas DC-8 Maintenance
 Planning Document (MPD); and
- Within thirty (30) days after obtaining FAA approval for their revised Time Limits Manual as outlined above. EWA may have to immediately conform their fleet of forty-one (41) DC-8 aircraft to the time limits and maintenance processes contained therein.

Sincerely,

Ly P. Howard Manager

cc: Rene P. Visscher - EWA Thomas M. Wood - EWA EWA'S RESPONSE TO LETER DATED.

JOHN WORLD 29, 1999



February 8, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines Inc., (EWA) formal response to your letter of investigation (99WP150029) addressed to EWA's President and Chief Operating Officer, dated January 29, 1999 (See Attachment).

At the outset, I would like to assure you that your letter has merited EWA's immediate and undivided attention. EWA, as a certificated air carrier, and its management and employees are fully appreciative of their responsibility arising under pertinent laws and under the Federal Aviation Regulations (FAR's) and strive to fulfill these responsibilities in a professional and conscientious manner.

During October through November 1998 EWA added three additional C & D Maintenance Checks and Modifications vendors to the Operation Specifications D91. The heavy maintenance vendors were Commodore Aviation Inc., Miami Modification Center and Tennessee Technical Services. (See Attachment). The Technical Service Department organized a team made up of Heavy Maintenance Reps, Quality Control Inspection Reps and Reliability Technical Analysis to perform training to these new vendors on our Inspection Procedures, Corrosion Prevention and Control Program and to include direct oversight of these new vendors to ensure compliance of EWA's programs. Several manhours and additional cost was incurred to aggressively eliminate this unforseen or controllable circumstance of our two heavy maintenance vendors experiencing manpower shortages, parts shortages and the manufacturers delay in providing repairs which were beyond EWA's control.

It was noted that 1998 was reported a record year for shortage of A & P mechanics in the United States. It was estimated to be 35% less than 1997. This mechanic shortage impacted EWA's Heavy Maintenance facilities specifically and resulted in longer down days to perform the check.

EWA escalated seven (7) aircraft in 1998 primarily due to this shortage of manpower, material, Douglas engineering support and utilizing the maximum capacity of EWA's two Heavy Maintenance facilities. This written information and several telephone calls with you on this subject, was provided during the year 1998. (See Attachments).

EWA escalated the aircraft in 1998 with your concurrence and under Technical Services controlled conditions that did not effect airworthiness or safety. The escalation utilized was in compliance with the Maintenance Policy and Procedure Manual Chapter 4, Section XXI. Short Term Escalation A. General items 1,2,3 and 4. (See Attachment).

Mr. Joseph Abramski Page 2 February 8, 1999

The referenced escalation's time used beyond the approved check interval must be subtracted from the time interval of the next regularly scheduled inspection, therefore there is no loss inspection time to the continuous airworthiness maintenance program providing a conservatory approach to ensure airworthiness and safety.

At no time in the past nine (9) years of EWA history has there been repetitive abuse for cause for revocation of the Short Term Escalation Privilege, and as substantiated in this letter, continues to be in full compliance.

I trust this has been responsive to your letter. Should you wish to discuss this matter more fully, EWA's Senior Management and I would be more than willing to meet with you and address your concerns. If this is the case, please contact me to arrange this meeting.

Sincerely,

Ou - mules

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc.

Kent Scott Rene' Visscher Robert Conlon Michael Dworkin

EWA "C" Check Short Term Escalation History

Previous escalation history for indication of abuse.

EWA has only used the escalation procedure a very few times in the past nine years. The last escalation was in November of 1996 for a component to align with a "C" Check inspection interval. This is no history of abuse with this program.

DATE	AIRCRAFT	REASON
February 9, 1998	N811AL	Heavy Maintenance Schedule Adjustment to meet FAR compliance of the RVSM Modification Program and schedule aircraft for FAA STC Cargo door inspection.
March 16, 1998	N603AL N964R	Heavy Maintenance escalation for the purpose of scheduling the "C" Check to be performed in conjunction with the "D" Check that are written reasonable proximity to one another.
April 20, 1998	N990CF	Heavy Maintenance delay of four aircraft due to the shortage of manpower, material, Douglas Engineering support, and utilizing the maximum capacity of EWA's two Heavy Maintenance facilities.
June 15, 1998 -	N602AL	This escalation is required to the grounding of aircraft N995CF that is being placed into "C" Check early and the current Heavy Maintenance schedule delay of four aircraft due to the shortage of manpower, material, Douglas engineering support, and utilizing the maximum capacity of EWA's two Heavy Maintenance facilities.
November 6, 1998	N8085U N797AL N997CF N996CF N959R	Heavy Maintenance Schedule delay of four aircraft due to the shortage of manpower, material, Douglas Engineering support, and utilizing the maximum capacity of EWA's two Heavy Maintenance facilities.



U.S. Department of Transportation

Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

RECEIVED MAR 2 2 1999

KENT T. SCOTT

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681

FAX: (408) 279-54-18

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028

EIR 99WP150025 is incorporated into EIR 99WP150028

EIR 99WP150029 is incorporated into EIR 99WP150028

EIR 99WP150032 is incorporated into EIR 99WP150028

EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W ENAS RESPONSE TO LETTER DATED
February 2, 1999



February 9, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letters dated February 2, 1999 (See Attachments) regarding your status request of the 1998 quarterly aircraft corrosion reports.

As you know from your visit in October 1998 here at Dayton, and several other written references sent to you in recent correspondence, EWA performed C or D Checks on over 50% of our fleet in the year 1998.

Also you are aware due to manpower shortages, material and Douglas Repair support to our two Heavy Maintenance Vendors Timco and Aerocorp, EWA was required to add three additional Heavy Maintenance facilities to address this shortage. The Reliability Section Staff was required to make several visits to these new vendors to perform CPCP inspection training requirements, which included follow-up inspections to assure compliance of EWA's CPCP Program.

I have reviewed the completion and submittal status of the referenced CPCP Level II Reports with the Manager of Reliability and received assurance from him that they will be complete and sent by February 26, 1999.

I. PACKETS COMPLETED AND SENT 1998

Aircraft	Date Packets Completed/Sent	Type of Check
N865F	June 98	С
N8177U	April 98	С
N961R	March 98	С
N2674U	February 98	С
N796AL	June 98	С
N796FT	May 98	С
N990CF	July 98	С

II. PACKETS COMPLETED AND SENT 1999

<u>Aircraft</u>	Date Packet Completed/Sent	Type of Check	
N995CF	February 99	C*	
N811AL	February 99	C*	

Mr. Joseph Abramski Page 2 February 9, 1999

III. PACKETS NEED COMPLETED

<u>Aircraft</u>	Date Packet to be Completed/Sent	Type of Check
N870TV	February	C*
N8076U	February	C*
N792FT	February	C*
N602AL	February	С
N605AL	February	С
N8084U	February	С
N795FT	February	C*
N998CF	February	С
N2674U	February	Ċ
N603AL	February	С
N500MH	February	С
N801GP	February	C

^{*}Note: Compiled with CPCP six year initial requirements.

Sincerely,

Dans M. Day

Thomas M. Wood Director Quality Control

TMW/re

Attachment

cc:

Rene Visscher

Robert Peck



San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue; Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7631 FAX: (408) 279-5443

February 2, 1999

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

This letter is a follow-up of your response dated January 4, 1999, to our request of that same day, in which you provided a listing of Emery Worldwide Airlines (EWA) DC-8 fleet aircraft which had heavy maintenance (C&D) checks performed in the 1998 calendar year.

Of those twenty one (21) aircraft, this office has received only six (6) of the required inspection reports to date, which are:

N8177U	C-Check completed on 2/3/98	Received SJC FSDO on 9/9/98
N961R	C-Check completed on 2/25/98	Received SJC FSDO on 3/30/98
N796AL	C-Check completed on 4/21/98	Received SJC FSDO on 6/23/98
N796FT	C-Check completed on 5/1/98	Received SJC FSDO on 8/4/98
N2674U	C-Check completed on 2/19/98	Received SJC FSDO on 10/5/98
N990CF	C-Check completed on 7/2/98	Received SJC FSDO on 1/11/99

Therefore, please be advised that we are requesting Emery Worldwide Airlines to provide an explanation for the absence of the required report submissions. Please respond in writing to this office no later than ten (10) working days after receipt of this letter.

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector

cc: Rene P. Visscher - EWA

February 8, 1900



February 25, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA)'s initial formal response to your letter of investigation (99WP150032) addressed to EWA's President and Chief Operating Officer, dated February 8, 1999, and your letter (Request for Information) addressed to me dated February 10, 1999.

I have enclosed a copy of the Maintenance Authorization AM-7112-01:01 that installed the STC SA5455NM on the subject aircraft. I will overnight to you Monday, 3-1-99 a signed copy for aircraft N998CF. The other two requested aircraft N996CF and N964R paperwork has not yet been received from the heavy maintenance checks.

I have performed a review of the referenced STC accomplishment, to which was audited by the Quality Control Section, and we are not aware of any violation of Federal Aviation Regulations, pertaining to this subject.

Please contact me by telephone upon receipt of this letter to provide EWA the opportunity of the knowledge of your concerns. I can assure you that we will immediately address your concerns.

Sincerely,

Tnomas M. Wood

Director Quality Control

TM(W/te

Attachment

cc:

Kent Scott Rene' Visscher Edward Jones



March 4, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA)'s second follow up letter to the initial formal response to your letter of investigation (99WP150032) addressed to EWA's President and Chief Operating Officer, dated February 8, 1999, your letter (Request for Information) addressed to me dated February, and my initial letter to you dated February 25, 1999.

I am forwarding you a copy of the Maintenance Authorization, AM-7112-01:00 that installed the STC SA5455NM on Aircraft N998CF. Again, as I previously informed you, I will forward Aircraft N996CF and N964R paperwork, upon receipt.

I have again reviewed EWA manual revisions that have been previously accepted/approved by you, and still am not aware of any violation of the Federal Aviation Regulations pertaining to this subject.

I would hope that you would contact me by telephone, upon receipt of this letter, to provide EWA the opportunity of acknowledging your concerns. As EWA continually demonstrates, I can assure you that we will take immediate action in addressing your concerns.

Sincerely,

Thomas M. Wood

Director Quality Control

TMW/csh

Attachments

cc: Kent Scott Rene' Visscher Edward Jones



U.S. Department of Transportation Federal Aviation Administration

San Jose Flight Standards Distnet Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (403) 291-7531 FAX: (405) 279-5443

February 10, 1999

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

Request for Information

This correspondence requests a copy of the completed work orders installing STC SA5455NM on Emery Worldwide Airlines (EWA) DC-8 aircraft N996CF, N998CF, and N964CF.

Please provide the requested information within ten working days after receipt of this letter. Should you require clarification on this matter, please call at your convenience.

Sincerely,

Joseph A. Ābramski

Principal Maintenance Inspector

cc: Rene Visscher - EWA



U.S. Department of Transportation

Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

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MAR 2 2 1999
KENT T. SCOTT

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028 EIR 99WP150025 is incorporated into EIR 99WP150028 EIR 99WP150029 is incorporated into EIR 99WP150028 EIR 99WP150032 is incorporated into EIR 99WP150028 EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W EWAS RESPONSE TO LETTER DATED February 8, 1999

See Johnson 27, 1999,
Some letter



U.S. Department of Transportation

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Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

RECEIVED MAR 22 1999 San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028 EIR 99WP150025 is incorporated into EIR 99WP150028 EIR 99WP150029 is incorporated into EIR 99WP150028 EIR 99WP150032 is incorporated into EIR 99WP150028 EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W April 22 ; 1999

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7581 FAX: (408) 279-5448

April 8, 1999

File Number: 99WP150045

CERTIFIED MAIL - RETURN RECEIPT REQUESTED

Mr. Kent Scott
President & Chief Operating Officer
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Scott:

On March 24, 1999, an inspection of Emery Worldwide Airlines (EWA) DC-8 aircraft records was conducted at your Vandalia, Ohio, facility by Aviation Safety Inspectors. Specifically, seven aircraft heavy maintenance check records were reviewed for the reporting compliance requirements of Airworthiness Directive (AD) 92-22-07, Corrosion Prevention and Control Program, applicable to DC-8 series aircraft.

As a result, the inspection of those records revealed the downgrading of maintenance inspection corrosion level determinations, which may be contrary to the provisions of Federal Aviation Regulations.

This letter is to inform you that Emery Worldwide Airlines, Inc., the holder of Air Carrier Certificate Number RRXA558B, may be in violation of Federal Aviation Regulations, and that this matter is under investigation by the Federal Aviation Administration. We offer you the opportunity to submit a written statement to this office regarding this matter, which should be accomplished within ten (10) working days following receipt of this letter. Your response should contain all pertinent facts and extenuating or mitigating circumstances that you believe may have a bearing on this matter. Should you elect not to respond within the specified time, our report will be processed without the benefit of your statement.

Sincerely,

Joseph A. Abramski Principal Maintenance Inspector

cc: Rene P. Visscher - EWA Thomas M. Wood - EWA



April 13, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA's) initial formal response to your letters File Number 99WP150045 and 99WP150044, addressed to EWA's President and Chief Operating Officer, and myself, dated April 8, 1999. (Reference Attachment 1.)

As the Director of Quality Control for EWA, I am disappointed in the receipt of these letters for three primary reasons that were previously communicated to you;

- 1. EWA's proactive demonstration of performing the Corrosion Prevention and Corrosion Control (CPCP) program since 1990, three years prior to the requirement of the Airworthiness Directive (AD) 92-22-07.
- 2. This subject was discussed, with you, during my visit to your office March 23, 1999, when I informed you of the misapplied corrosion finding levels by the Quality Control Representatives, and that I had authorized them to be changed to the original level assignment that day, (March 23, 1999), which was accomplished by the Manager, Quality Control and they were reported per the AD requirements to Douglas and yourself, by fax on that day. (Reference Attachment 2.) This was also provided to you in writing, in meeting minutes, Item Number Nine (9). (Reference Attachment 3.)
- This same subject was again addressed to Mr. Jay Howard, in Mr. Bruce Robbins, Director of Engineering's letter dated April 2, 1999, Page 11, Item K. (See Attachment 4.)

As you are aware, EWA received FAA approval to provide information regarding applicability, methods, and procedures for performing corrosion prevention and control, in accordance with EWA's equivalent program, per Airworthiness Directive 92-22-07, as represented in EWA's Inspection Program Manual, Volume III.

I can assure you that EWA has maintained compliance of their program regarding all aspects, including the reporting requirements of the subject AD.

I trust I will have the opportunity to discuss this during your visit to our office this week, and to provide you this initial information.

Mr. Joseph Abramski Page Two April 13, 1999

I will compile the historical data of EWA's equivalent program, that will substantiate compliance of AD 92-22-07, and submit to your office, no later than April 26, 1999, (within ten (10) working days of receipt of your letters).

As always, I look forward to resolving your concerns in an expeditious manner, and promoting an open line of communication with the first opportunity face-to-face.

Sincerely,

Dime Willer

Thomas M. Wood Director, Quality Control

Attachments

TMW/csh

cc: Kent Scott
Rene' Visscher
Jay Howard
Bruce Robbins
Edward Jones
Robert Conlon
Michael Dworkin
Ted Ellet

ENR'S RESPONSE TO LETTER DATED

TIME LIMITS MANUAL DC-8-62/63, DC-8-71, DC-8-73

I. DC-8 CHECK INTERVALS

Transit Check

The Transit Check will be accomplished upon arrival and prior to aircraft departure:

1) On aircraft with less than six (6) hours of ground time unless a higher maintenance check is accomplished.

Note: Check is only required if EWA Maintenance personnel are assigned.

Terminating Check

The Terminating Check will be accomplished upon arrival and prior to aircraft departure:

1) On aircraft with six (6) or more hours of ground time if within the preceding twenty-four (24) hours a higher check has not been accomplished.

Note: Check is only required if EWA Maintenance personnel are assigned.

Service Check

The Service Check will be accomplished prior to aircraft departure:

1) On aircraft with twenty-four (24) or more hours of ground time and a higher check has not been accomplished.

Note: Check is only required if EWA Maintenance personnel are assigned.

B Check

The B Check will be accomplished prior to aircraft departure:

1) On aircraft in sequential segmented checks each one hundred and thirty-six (136) flight hours <u>unless</u> a "C" Check is accomplished.

Note: Check is only required if EWA Maintenance personnel are assigned.

Note: The "B" Check Inspections are incorporated in multiple segments as "B1", "B2", "B3" and "B4".

EMERY WORLDWIDE AIRLINES TIME LIMITS MANUAL DC-8-62/63, DC-8-71, DC-8-73

C Check

The C Check shall be accomplished each twenty-four (24)

calendar months.

D Check

The D Check shall be accomplished each twelve (12) calendar

years. A "C" check will always be accomplished when

performing a D Check.

Corrosion Program

See the Inspection Program Manual Volume III for a complete

detail of the Policy and Procedures for monitoring, planning, and compliance of EWA's FAA approved equivalent DC-8

Corrosion Prevention and Control Program.

SID

See the Inspection Program Manual for a complete detail of

the Policy and Procedures for monitoring, planning, and

compliance of the Douglas SID Program.

July 21, 2000 Revision 63 Chapter 4 Page 2 EUR'S RESPONSETTI LETTLE DATED

JUNE 16, 1999



May 5, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is an update to my March 12, 1999, letter of formal announcement of the development of the Engineering Department. (See attachment.)

Mr. Bruce Robbins, Director of Engineering, has assigned Mr. Jim Feisley as the Acting Manager of Technical Publications.

Please join me in welcoming Jim to this assignment. Mr. Feisley has several years of experience in technical publication development and management.

Sincerely,

Thomas M. Wood Director Quality Control

Attachment

TMW/csh

cc: Rene' Visscher Bruce Robbins Jim Feisley Nick Pearson



March 12, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

On behalf of the Emery Worldwide Airlines, Inc.'s (EWA) Senior Management, it gives me great pleasure to announce the development of the Engineering Department. This addition and reorganization of the Quality Control Department responsibilities will promote an overall increased effectiveness of the Technical Services Department.

Interviews are being scheduled to hire the replacement Manager of Maintenance Training, and filled within thirty (30) days.

I am sure you share with us the excitement of this department expansion, and can appreciate the true economical contributions made by the Company.

I trust from your previously voiced concerns of EWA's lack of an Engineering Department, this development meets with your concurrence. I will formally submit the Maintenance Policy and Procedures Manual Revision to you that will incorporate these changes.

Thank you for your support and help in promoting the growth of the EWA Technical Services Department.

Please call if you have any questions.

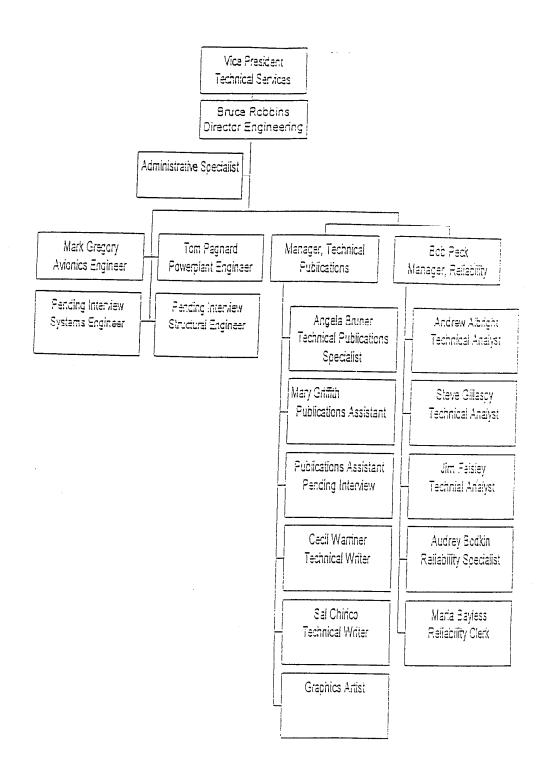
Sincerely,

Thomas M. Wood
Director Quality Control

Attachments

TMW/csh

cc: Rene' Visscher Jay Howard John Howard Bruce Robbins



EWA'S RESPONSE TO LETTER DATED August 9, 1999

EWA did provide a transition plan.
Could not locate at this time.

EWA'S RESPONSE TO LETTER DATED September 16, 1999



Air Carrier Certificate

This certifies that EMERY WORLDWIDE AIRLINES, INC. One Emery Plaza Vandalia, OH 45377

has met the requirements of the Federal Aviation Act of 1958, as amended, and the rules, regulations, and standards prescribed thereunder for the issuance of this certificate and is hereby authorized to operate as an air carrier and conduct common carriage operations in accordance with said Act and the rules, regulations, and standards prescribed thereunder and the terms, conditions, and limitations contained in the approved operations specifications.

This certificate is not transferable and, unless sooner surrendered, suspended, or revoked, shall continue in effect indefinitely.

	110
	10000 C 1 1000
Certificate number: RRXA558B	David E. Hanley
	(Signature)
Effective Date: June 19, 1987	Manager, Flight Standards Division
	(Title)
Issued at: GL05	Great Lakes Region
	्रिक् _{रिक्र} (Region/Office)
Reissued: December 17, 1999	
FAA Form 8430-18 (6-87) AFS Electronic Forms System - JetForm FormFlow - 12/	수가에 가장 그리고 있었다면서 가지 않는 그가 하는 것이 없는 것이 없는 것이 없다.

EMERY WORLDWIDE AIRLINES MANUAL REVISION SUBMITTAL -- FORM ME059

ro. IMI. Harold Campen					
The attached manual revision is submitted for your review and acceptance or approval a required. We request that you review the subject revision at your earliest opportunity an return completed form to Emery Worldwide Airlines within ten (10) working days after dat of submission. Should you have questions or comments concerning this revision, please d not hesitate to contact this office.					
Manual: Inspection Program Manual (IF	PM) Volume III				
Revision Number: 21	Revision D	ate: <u>April 2, 1999</u>			
Purpose of Revision:					
Revise IPM Volume III to include DC-10 chart. Refer to Revision Highlights for) information a additional info	nd CPCP level determination flow rmation on the "Purpose of Revision".			
	·				
Submitted by: <u>Jim Feisley</u>		Date: <u>02/29/00</u>			
FAA					
() Accepted	W	Approved			
() Not-Accepted	()	Disapproved			
Signature:		Date: / - 7 - 0 0			
Grounds for disapproval:					

EMERY WORLDWIDE AIRLINES MAINTENANCE REVIEW BOARD RECOMMENDATION

No. 99002

ATTENTION:	EWA MRB JUDICIAL MEMBERS Vice President of Technical Services, Director of Quality Control, Director of Operations
1.	Change maintenance inspection program interval, maintenance process, and/or work content specification.
2.	Modifications
3.	Fleet inspections
X4.	Revised maintenance practices
5.	Improved troubleshooting methods
6.	Maintenance and/or operations training
7.	Operating procedures
8.	No-specific action
Subject:	EWA Maintenance Inspection Program Manual Volume III
changes. Required Act	
	evised as recommended. Refer to attached revision 21 highlights for a
MRB Approv	(Check One) YES NO (Director of Quality Control) (Director of Operations)



of Transportation Federal Aviation Administration San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (403) 279-5448

October 20, 1999

Mr. Bruce Robbins
Director, Aircraft Engineering
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Robbins:

This correspondence will address your response to my letter of disapproval to the submitted Emery Worldwide Airlines, Inc. (EWA), Inspection Procedures Manual, Volume III, Revision 21. Again, the submission, has been disapproved for the following reasons as conjunctive to your letter's items:

Chapter 2

- 1) Item 1 Page 5a, Corrosion Level Determination, missing from Chapter 2 Table of Contents Page i. Suggest you review Chapter 1 of the EWA Maintenance Policies and Procedures Manual regarding manual revisions. Style and format exclusions are not contained therein. This Inspection Program Manual is an FAA approved document which is not exclusively operator discretionary nor selective in content.
- 2) Item 3 Page 2, Paragraph 8 does not accurately reflect Douglas Report MDC-K4608, Section 2, paragraph 13 instructions as alleged. The MDC-K4608 paragraph allows operators to convert calendar time to flight hours, landings, or cycles for ease of scheduling. Moreover, the cited MDC-K4608 paragraph and its associated section contents, does not contain any verbiage regarding the maintenance of Level 1 corrosion. I have included a copy of the MDC-K4608 document page as a part of this response for your review and edification.
- 3) Item 5 Page 3, Paragraph 2. Your response elicits our curiosity regarding EWA's established CPCP program per the MDC-K4608 document, Section 3.2, which addresses an Alternate Means Of Compliance (AMOC) to AD 92-22-07. Neither this office nor the Los Angeles Aircraft Certification Office has an archived FAA approved AMOC for EWA on this issue. Therefore, we request that EWA provide a copy of the AMOC for our review.
- 4) Item 6 Page 5, Paragraph 5. We believe your reasoning in this matter is erroneous, and therefore, unacceptable. As was stated in our previous letter dated September 16, 1999, EWA's Maintenance Reliability Program in concert with the Continuing

Analysis & Surveillance System as required by FAR Part 121.373 is the evaluative tool. As you are aware, the effectiveness and objectivity of the EWA Reliability Program has been and remains, the expressed subject of concern to this office.

In conclusion, please resubmit Chapter 2 of this revision for our review and approval. Should you have any questions regarding this matter, please call at your convenience.

Sincerely,

ORIGINAL SIGNED BY

Joseph A. Abramski
Principal Maintenance Inspector

cc: Rene P. Visscher - EWA
Jim Feisley - EWA
Thomas M. Wood - EWA
C. Robert Peck - EWA

EMERY WORLDWIDE AIRLINES MANUAL REVISION SUBMITTAL -- FORM ME059

The attached manual revision is submitted for your review and acceptance or approval as required. We request that you review the subject revision at your earliest opportunity and return completed form to Emery Worldwide Airlines within ten (10) working days after date of submission. Should you have questions or comments concerning this revision, please do not hesitate to contact this office.				
Manual: Inspection Program Manual (IPM) Volume III				
Revision Number: 21 Revision Date: April 2, 1999				
Purpose of Revision:				
Revise IPM Volume III to include DC-10 information and CPCP level determination flow chart. Refer to Revision Highlights for additional information on the "Purpose of Revision".				
Submitted by: <u>Jim Feisley</u>	Date: <u>09/29/99</u>			
FAA				
() Accepted ()	Approved			
() Not-Accepted	Disapproved			
Signature:	Date: 10/30/99			
Grounds for disapproval: DEASE SEE ATTACHED LETTER!				

To: Mr. William J. Dime

3 .



Mr. William J. Dime FSDO - SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110 - 1130

Dear Mr. Dime

This letter is in response to Mr. Joe Abramski's correspondence (copy attached) to Mr. Jim Feisley on Emery Worldwide Airlines (EWA) Inspection Procedures Manual (IPM) Volume III, dated September 16, 1999. The following listed items are in response to the nine (9) reasons for disapproval presented under Chapter 2 of this letter.

- Item 1. EWA's IPM style and format does not require the Corrosion Level Determination Flow Chart to be listed in the Table of Contents for Chapter 2.
- Item 2. EWA's IPM Volume III will be revised to reflect MDC-K4608 Revision 3, dated February 1995.
- Item 3. This paragraph reflects Douglas Report MDC-K4608, Section 2 (Program Application) paragraph 13 instructions and EWA elects to retain this in its' Corrosion Prevention and Control Program (CPCP).
- Item 4. Reporting requirements and procedures for FAR 121.703 (a) (15) are contained in the EWA's Maintenance Policy & Procedures Manual.
- Item 5. EWA's DC-8 Corrosion Prevention and Control Program (CPCP) was established per MDC-K4608 report Section 2.1, Paragraph 7 and Section 3.2. The DC-8 Maintenance Program calendar intervals were established with consideration to MDC-K4608 report Section 2, Paragraph 13 and therefore; Section 2.1, Paragraph 7 and Section 3.2 are still applicable and page 3, paragraph 2 remains in effect for the DC-8 aircraft CPCP. However; this paragraph has been revised (see attached page) to reflect that the CPCP repetitive intervals for the DC-10 aircraft were established per the MDC-K4607 report, and will be controlled under the EWA Reliability Program, once eligible.

- Item 6. MDC-K4608 Paragraph 2, (page 1-1-2) states, "Should inspections reveal that corrosion exceeds Level 1 for a given task on an operator's fleet of aircraft, the inspection interval should be reduced and/or action taken to minimize reoccurrence." AD 92-22-07 paragraph 6e states, "a means approved by the FAA must be implemented to reduce future finding of corrosion in that area to Level 1 or better. The word "evaluated" is appropriate as an evaluation is required to determine "action" necessary to "minimize/reduce" reoccurrence of future findings of corrosion in that area. This evaluation "may" result in a recommendation to reduce the inspection interval, but the referenced paragraphs do not state that it is a requirement.
- Item 7. The word "compliance" will be added.
- Item 8. The word "expedition" will be changed to read "expeditious".
- Item 9. "DC-10" will be added as necessary to clarify EWA's CPCP methods and reporting as applicable to the DC-8 and DC-10 aircraft.

Please find attached IPM Volume III pages which contain changes made in response to Mr. Abramski's letter. A MEO59 for IPM Volume III is also included in this package.

Sincerely,

Bruce Robbins

Director of Aircraft Engineering

cc: R. Visscher

T. Wood

J. Feisley



U.S. Department of Transportation Federal Aviation Administration San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

September 16, 1999

Mr. Jim Feisley
Manager, Technical Publications
& Maintenance Programs
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Feisley:

The Emery Worldwide Airlines, Inc. (EWA), Inspection Procedures Manual, Volume III, Revision 21 as submitted, has been disapproved for the following reasons:

Chapter 2

1) Page 5a, Corrosion Level Determination, missing from Chapter 2 Table of Contents Page i.

Page 2, Paragraph 6, incorrectly references MDC K-4608, Revision 1, dated December, 1990. The current revision of the document is Revision 3, dated February, 1995.

3) Page 2, Paragraph 8 is inappropriate in this manual. It would however, be appropriate for inclusion to the EWA Reliability Program Manual. Please delete this paragraph.

Page 2, Paragraph 10 is incomplete. Corrosion Level 2 an 3 findings are reportable to the FAA via Operational Difficulty Reports as required by FAR 121.703 (a) (15). Please amend and include as appropriate.

5) Page 3, Paragraph 2, second sentence, second half, which states "however, they do not have to be exactly as listed if corrosion is controlled to Level 1 or better", is objectionable and requires deletion for the following reasons: (a) EWA's DC-8 Maintenance Program is presently converting from Flight Hours to Calendar Times. Therefore, the CPCP inspection task intervals as defined in the MDC K-4608, Section 4, is applicable and appropriate. (b) A corrosion level determination has not been established for the EWA DC-10 aircraft fleet. The DC-10 maintenance and inspection intervals are also a Calendar Time based program. In addition, the DC-10 aircraft is not eligible at this time for inclusion to the EWA Reliability Program.

6) Page 5, Paragraph 5, seventh sentence, the word "evaluated" is inappropriate and inconsistent with paragraph 2, page 1-1-2, of the MDC K-4608 document; and paragraph (e) of AD 92-22-07, which require "reduced" inspection intervals should

corrosion task levels exceed Level 1. The "evaluated" factor should be ongoing as part of an Air Carrier's Maintenance Reliability Program in concert with the Continuing Analysis & Surveillance System. Therefore, since this paragraph paraphrases the previously cited MDC K-4608 document, replace the word "evaluated" with the word "reduced".

- 7) Page 5b, Note 1, is absent the word "compliance". This note paraphrases the MDC K-4608 document Note 1 located on page C-2-4. Please insert accordingly.
- 8) Page 5b, Note 3, fifth sentence, the word "expedition" should instead state the word "expeditious".
- 9) As we collectively discussed this day on a telecon in the presence of Bob Peck, Reliability Program Manager, Chapter 2 of the Inspection Program requires more definitive consistent references to the DC-10 CPCP methods and reporting, in conjunction with the DC-8 references contained therein.

In conclusion, please resubmit Chapter 2 of this revision for our review and approval. Should you have any questions regarding this matter, please call at your convenience.

Sincerely,

Joseph A. Abramski
Principal Maintenance Inspector

cc: Rene P. Visscher - EWA Bruce Robbins - EWA Thomas M. Wood - EWA C. Robert Peck - EWA

Wood, Thomas M

From: Wood, Thomas M

Sent: Monday, October 25, 1999 5:01 PM

To: Visscher, Rene P

Cc: Robbins, Bruce A.; Feisley, James L; Peck, Charles R; Moody, Ronald E; Jones, Edward

В

Subject: FW: IPM Volume III, Rev. 21

Rene: Thank you.

From: Visscher, Rene P

Sent: Monday, October 25, 1999 4:54 PM

To: Wood, Thomas M

Subject: RE: IPM Volume III, Rev. 21

Keep it on hold

Rene Visscher

Vice President Technical Services

EWA - HDY

From:

Wood, Thomas M

Sent:

Monday, October 25, 1999 3:28 PM

To:

Visscher, Rene P

Cc:

Robbins, Bruce A.; Feisley, James L; Peck, Charles R; Moody, Ronald E; Jones, Edward B

Subject:

IPM Volume III, Rev. 21

Rene: I assume that the letter from our PMI received today, dated October 20,1999, dissapproving the subject revision, is his position letter for this task in regards to the certificate move. With your permission, I would like to hold this, and not futher respond. We will address this with our new PMI.

Thomas Wood

EUA'S RESPONSE TO VETTER DATED

October 20 1999

Induded in Sph. 16, 1999 1988ponse

Carac Regionse to Letter Daned POOL C regions of the



U.S. Department of Transportation Federal Aviation Administration

San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

November 15, 1999

Mr. James Feisley
Manager
Maintenance Programs & Publications
Emery Worldwide Airlines, Inc.
One Emery Plaza
Dayton International Airport
Vandalia, OH 45377

Dear Mr. Feisley:

This office has reviewed the Emery Worldwide Airlines, Inc. (EWA) Inspection Program Manual Volume II, Revision 23 submission, and has disapproved such for the following reasons:

Chapter 1, page 1, paragraph I.B., second sentence: Change the verbiage "the inspector blocks can be", to "the Inspector blocks will be completed". This verbiage will then be consistent with Volume I, Chapter 1, page 1, paragraph I.C., and the note in Volume II, Chapter 1, page 2, paragraph B.c.

√2) Chapter 1, page 5, paragraph 3.f.: Therein is stated in two places "the EWA Station Supervisor". What role does this individual contribute in the EWA "C & D" heavy maintenance check process? Where would we find his job description since it is not listed in the EWA Maintenance Policy & Procedures Manual?

Chapter 1, page 5, Paragraph D.5: Change the verbiage for the sake of clarity to read "The series of aircraft to which a work card is applicable".

4) Chapter 2, page 33, DC-8 C-Check work card PRE10, 1 of 5, items 1 and 5: In accordance with what conformity documents?

5) Chapter 2, page 34, DC-8 C-Check work card PRE10, 2 of 5, items 11 through 14; These system functional checks and tests are conducted in accord with what maintenance manual references?

6) Chapter 2, page 35, DC-8 C-Check work card PRE10, 3 of 5, items 16 through 20: These system functional and self tests are conducted in accord with what maintenance manual references? In addition, item 17 refers to which system (normal or standby;

7) Chapter 2, page 36, DC-8 C-Check work card PRE10, 4 of 5, items 21 through 29, functional checks/tests: The EWA DC-8 fleet has several different configurations of avionics. Which procedures are applicable to which aircraft; and where (in what manual) are the FAA approved/accepted maintenance procedures located?

8) Chapter 2, page 37, DC-8 C-Check work card PRE10, 5 of 5, item c., MINIMUM ACCEPTABLE VOLTAGE: Are these stated voltages applicable to <u>all</u> makes and models of ULBB for both FDR and CVR?

Should you desire to discuss this matter, please contact either the undersigned or Principal Avionics Inspector Shawn Skaags at your convenience.

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector

cc: Rene Visscher - EWA Thomas M. Wood - EWA



Air Carrier Certificate

This certifies that EMERY WORLDWIDE AIRLINES, INC. One Emery Plaza Vandalia, OH 45377

has met the requirements of the Federal Aviation Act of 1958, as amended, and the rules, regulations, and standards prescribed thereunder for the issuance of this certificate and is hereby authorized to operate as an air carrier and conduct common carriage operations in accordance with said Act and the rules, regulations, and standards prescribed thereunder and the terms, conditions, and limitations contained in the approved operations specifications.

This certificate is not transferable and, unless sooner surrendered, suspended, or revoked shall continue in effect indefinitely.

By Direction of the Administrator

Certificate number: RRXA558B David E. Hanley (Signature)

Effective Date: June 19, 1987 Manager, Flight Standards Division (Title)

Issued at: GL05 Great Lakes Region (Region/Office)

Reissued: December 17, 1999

FAA Form 8430-18 (6-87)

AFS Electronic Forms System - JetForm FormFlow - 12/1998

EMERY WORLDWIDE AIRLINES MANUAL REVISION SUBMITTAL -- FORM ME059

io: Mr. Joe Abramski		
The attached manual revision is submitted required. We request that you review the return completed form to Emery Worldwide of submission. Should you have questions not hesitate to contact this office.	Subject re Airlines v	vision at your earliest opportunity an
Manual: Inspection Program Manual Volume	11	
Revision Number: 23	evision Da	ite: <u>April 27, 1999</u>
Purpose of Revision:		
Revised manual to clarify information. Referinformation on the "Purpose of Revision".	to Revisio	on Highlights for additional
Submitted by: <u>Jim Feisley</u>	,	Date: 09/16/99
= A A		
<u> </u>	,	
() Accepted	121	Approved
() Not-Accepted	()	Disapproved
Signature:		Date: 3-2-00
Grounds for disapproval:		

MEDER DO JOSJONIO

EWA'S RESPONSE TO LETTER DATED November 27, 1999



Air Carrier Certificate

This certifies that EMERY WORLDWIDE AIRLINES, INC. One Emery Plaza Vandalia, OH 45377

has met the requirements of the Federal Aviation Act of 1958, as amended, and the rules, regulations, and standards prescribed thereunder for the issuance of this certificate and is hereby authorized to operate as an air carrier and conduct common carriage operations in accordance with said Act and the rules, regulations, and standards prescribed thereunder and the terms, conditions, and limitations contained in the approved operations specifications.

This certificate is not transferable and, unless sooner surrendered, suspended, or revoked shall continue in effect indefinitely.

By Direction of the Administrator

The second secon	
Certificate number: RRXA558B	David E. Hanley
	(Signature)
Effective Date: Junc 19, 1987	
Effective Date: June 19, 1987	Manager, Flight Standards Division (Title)
Issued at: GL05	Great Lakes Region
	(Region/Office)
Reissued: December 17, 19	999

EMERY WORLDWIDE AIRLINES MANUAL REVISION SUBMITTAL -- FORM ME059

To: Harold Camden (FAA PM	<u>/II)</u>		
required. We request that v	you review the subjec nery Worldwide Airlind have questions or con	t revis es with	iew and acceptance or approval as ion at your earliest opportunity and nin ten (10) working days after date concerning this revision, please do
Manual: EWA Maintenance	Policy and Procedures	s Manu	al
Revision Number: 21	Revisio	n Date	: <u>January 15, 2000</u>
Purpose of Revision:			
have been incorporated into	this revision. All chap requirements, and job	oters ha descri	pered, temporary revisions 6 and 7 have been reviewed to add current ptions. Obsolete procedures and drision highlights for additional
Quality Control Review:	Thomas M. C	صحك	Date <u>01-24-00</u>
<u>FAA</u>			
(×) Accepted	()	Approved
() Not-Accept	ed ()	Disapproved

Grounds for disapproval:

Signature:_

*			
CHAPTER	PAGE	ITEM	PURPOSE
1-6	All		This is a complete re-issue, all pages have been re- numbered, Temporary Revisions 6 & 7 have been incorporated into this revision.
T of C	1-3		Revision, addition and/or deletion as shown in the Revision Highlights.
LEP	All		Added pages, changed all to current revision and date
FAR Index	1	•	Added FAR 43,13°, 43.16, 91.203 and Titles, Chapter and Section
	2		Added FAR 119.65 (a), 121.133, 121.135 and Titles, Chapter and Section.
Form Index	All		Sorted all forms by number, added MEO148 and MEO155, Removed MEO24, combined MEO20 from Chapter 4 & 5 to one line.
1 T of C	1-2		Revised to indicate additions, and deletions as shown in the highlights.
1	1-6	1	Re-numbered pages, removed a & b from pages, incorporated temporary revision 6 & 7
		I.A	Added FAR Compliance and FAR 43.13°, 43.16. Removed "andmaintenance" after CAMP. Added paragraph "The responsibility", Changed Quality Control to Maintenance Programs & Publications.
		II.B.	Changed flow of sentence, added Line and Heavy to Maintenance, added Engineering Department.
		I.C.	Added FAR statement., added DC-10 and CF6 manuals.
		I.D.	Changed "three" to "Five" added service checks. Added DC-10 to Vol III, Added Vol IV & V.
		I.E.	Changed paragraph to remove reference to DC-8, changed master to workcard. Added controlled access requirements. Added Note. Added Manager Programs and Publications, removed QC.

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CHAPTER	PAGE	ITEM	PURPOSE
1 .	1-6	I.FS.	Re-numbered paragraphs, Added EWA to paragraph H. Added DC-10 to paragraphs L & M Added paragraphs Q, R, S on DC-10.
	7-12	11	Re-numbered pages, incorporated temporary revision 6 & 7
		II.A	Added DC-10
		II.B.	Added DC-8 to Title, removed 50 series. Added DC-10
		II.C.	Added sentence to paragraph " Any changesManuals". Removed 50 series Added DC-10
		II.D.	Removed Blank page. Re-numbered pages, removed 50 series. Added DC-8 to title, added DC-10. Added CF6-6 MM, IPC, and Overhaul Manuals.
	13-19	111	Revised section to change Quality Control to Maintenance Programs and Publications.
		III.A.	Removed Quality Assurance Section. Removed last sentence from paragraph A.1. Removed Chapter 9 from paragraph A. Removed requirement to assigned a person to place manuals on the aircraft.
		III.B.	Removed requirement to designated an individual to the aircraft.
		III.C.	Changed Quality Control to Engineering, changed QC to Maintenance Programs and Publications. Revised MEO 51.
		III.D.	Changed QC to Engineering in paragraph 1. Changed note. Changed QC to Maintenance Programs and Publications in paragraph 3d.
		III.E.	Changed Quality Control to Maintenance Programs and Publications.
	20-27	IV	Re-accomplished Section to show who maintains EWA manuals. Added Engineering Director and Department.

CHAPTER	PAGE	ITEM	PURPOSE
1	20-27	IV.A.	Added Engineering, added Inspection Manuals Volumes IV and V. Added Maintenance Programs and Publications. Removed Quality Control. Added DC-10 manuals container, added Maintenance Publication Library. Added Engineering Orders (EO)
		IV.B.	Remove requirement for FAA approval stamp.
		IV.D.	Revised MEO54 and MEO 59.
	28	V.	Changed Quality Control to Maintenance Programs and Publications.
2 T of C	1-2		Revised to indicate additions, and deletions as shown in the highlights.
2	2-3	11.	Changed address and phone numbers removed aircraft from director of aircraft engineering.
	5-12	111	Updated organizational charts, added Engineering and Safety Departments.
	13-117	IV.BAX	Re-numbered pages and re-lettered paragraphs. Added Director of Engineering, Added Manager of Programs and Publications, Added Engineers positions, Added Quality Control Inspector positions, removed Records Coordinator position. Added qualifications to positions. Removed Technical Administration Position. Changed Airframe to heavy Maintenance. Changed Senior Director to Vice President. Added Manager Programs and Publications to Organization relationships on positions. Changed Foreman to Supervisor.
	118-123	V.	Changed address and phone number. Added statement in Class I and Class II line stations. Changed address, point of contact, and phone number for Class of stations. Deleted Line Stations.
3 T of C	1-5		Revised to indicate additions, and deletions as shown in the highlights.
3	1-2	I.A	Changed Senior Director to Vice President.
		I.B.	Added FAR 43 to title, Changed "any" to "all" in paragraph B1. Quoted FAR 65.81(a) in paragraph B3. Added FAR 65.81(b) to paragraph B4. Quoted FAR 65.83 in paragraph B5.

CHAPTER	PAGE	ITEM	PURPOSE
3	3-8	II.A.	Changed paragraph into two sentences.
-		II.B.	Changed Foreman to Supervisor. Added Chapter 3 in paragraph B3. Added Chapter 4 in paragraph 4. Added FAA statement to B4. Added Chapter 4 to B5.
		II.C.2.	Changed the address for the FAA in paragraph C.2.a(6)Revised MEO108.
	9-16	IV.A.	Changed Senior Director to vice president. Changed maintenance scheduling to production control.
		IV.B.	Changed Senior Director to Vice President., Added Director of Engineering and Manager of Material Procurement. Changed paragraph B2b. Added Planning to Production in paragraph B2d. Removed reliability and added engineering Added DC-10 and CF-6. Revised MEO81.
	19-34	VI.A.	Added "Line and/or" to paragraph A4. Changed MEL to DMI's in paragraph A8. Added two section to paragraph A8. Added "line and/or Heavy" to paragraph A9. Added "to all" in front of maintenance in paragraph A13.
		IV.D.	Removed minor from paragraph D1. Added "Minor Primary/secondary" to paragraph D3b. Changed should to shall in Note. Changed Quality Control to Engineering in paragraph D3c. Changed Quality Control to Engineering in Note. Added (Such as, partial installation) in D3d.
		IV.E.	Added Non-MEL to E2. Changed "as soon as possible" to immediately in E5. Added Note 3.
		IV.F.	Added to paragraph F5.
	35-50	VII.	Added FAR References to title.
3	35-50	VII.B.	Changed Foreman to Supervisor in B8.
		VII.C.	Change item 18.
	51-52	VIII.A.	Added "trained" to paragraph. Changed Foreman to Supervisor, Deleted Senior in A5.

CHAPTER	PAGE	ITEM	PURPOSE
3	51-52	VIII.C.	Added "trained" to paragraph. Changed paragraph C1 & 2. Deleted last part of sentence in paragraph C2. Changed section A to section VIII and added (Aircraft Airworthiness Release) in C1c. Spelled out (airworthiness release) in C2 Added training statement in C3d.
	55-64	IX.A.	Added Engineering to A5. Deleted last sentence from note.
		IX.B.	Deleted last part of sentence paragraph B3, end with Maintenance Log.
		IX.D.	Changed MERIT to EWA Computer System Added disposition of MEO09 after item 16.
		IX.F.	Changed EWA1 to EWA Computer System.
	65-66	X.A.	Changed Foreman to Supervisor Added to help in front of ensure, added all in front of maintenance for A1.
		X.B.	Paragraph was re-worked. Deleted Senior in front of maintenance representative.
	67-76	XÍ.B.	After part/component, add and placed in stock. In B1c.
			Changed last sentence in B4b to read EWA computer System. Changed MA to EO in B4e.
		XI.C.	Remove Senior Director Technical Services from paragraph C2a. Changed Material Management to Material Control. Changed Material Planning to Inventory Planning in C2b. Changed Material Management to Material Control in C3a,b,c and 4a. Changed The computer program to EWA Computer System. Revised MEO 121 Added page 75 & 76 as blank pages.
	80-87	XIII.A.B.	Removed MERIT.
3	80-87	XIII.C.	Changed EWA1 to EWA Computer System
		XIII.D.	Changed paragraph D1 to explain what 121, 145, 129 and 135 stands for.
	98-108	XVII.B.2	Changed Foreman to Supervisor.

CHAPTER	PAGE	ITEM	PURPOSE
3	116-120	XXI.B.	Changed EWA 1 to EWA Computer System in item 7. Paragraph in item 17 changed, EWA1 removed.
	121-122	XXII.A.	Removed "and" added (,) after teardown.
	127-136	XXVI.E.	Removed 50 series from paragraph E3b.
	139-141	MEO134	Changed to show new address and phone numbers.
	174-189	XXXIB.	In paragraph B1 last line changed tome to time. Changed Senior Director to Vice President.
		XXXI.B	Changed "eight" to "nine" in block 12 of paragraph B9.
		XXXII	Deleted the entire section.
4 T of C	1-5		Revised to indicate additions, and deletions as shown in the highlights.
4	1-6	1	Removed Inspection from title
		I.A.	Added (A&P) behind Power Plant, Changed Senior Director to Vice President, Added paragraph for Chief Inspector.
		I.B.	Quoted FAR 65 sub-part D.
		I.C.	Changed Director to Manager in C1b. Removed "Inspection branch from paragraph 1 & 2. Added "Contractor" to C2e. Changed Foreman to Supervisor and removed Senior in C2h.
		I.D.	Added B checks to D1a. Removed "on the spot" and inserted "one time" in D3.
		I.E.	Removed "The" and "branch" from first sentence. Changed "MA" to "EO", changed Writes and/or coordinates to Review & approves, added engineering department. Removed paragraph f,g &, h re-numbered paragraph i & j. Added plank page.
4	7-31	II.A.7.	Changed Quality Control to Engineering.

CHAPTER	PAGE	ITEM	PURPOSE
4	7-31	II.C.	Changed around sentence structure to read smoother in C2a & b. Added Engineering to paragraph C3 a & b. Remove "action(s) to the field" from C6d. removed paragraph C6e and re-numbered remaining paragraphs. Added removed paragraph C6e to C8e and renumbered remaining paragraph.
		II.D	Changed Foreman to Supervisor.
		II.E.	Changed Foreman to Supervisor and deleted paragraph E6 and re-numbered E7 to E6. Changed last sentence in paragraph E6. Revised MEO 63. Added MEO148.Revised MEO65.
		11.1.	Removed "nebulous".
		II.J.	Added retention statement to end of paragraph.
		II.K.	Added () to enclose i.e.:etc. Revised MEO71.
	32-67	III.B.5.	Added the Directors of Engineering and Material Management and removed senior director of technical service. Removed paragraph B6 and re-numbered B7 to B6. Revised MEO94.
		III.C.3.	Changed Senior Director to Director of Quality Control and changed Director to Manager.
		111.1.	Revised MEO94a.
	68-70	IV.A.1	Changed EWA1 to EWA Computer System.
		IV.B.1	Added () around i.e. in paragraph B1d.
	71-72	V.B.1	Changed Air to Aviation in paragraph B1b.
	73-78	VI.	Added EWA to title.
	79-82	VII.A.3	Added Quality Control will again perform an audit, after requested,
		VII.C.	Added Deicing after respective.
	83-97	VIII.B.1	Moved last sentence and made into a note.
	83-97	VIII,C.3	Added class I or II in front of EWA in paragraph C3c. Changed nozzle pressures in paragraph C4a(7).

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CHAPTER	PAGE	<u>ITEM</u>	PURPOSE
4	98-99	IX.A	Added Engineering to paragraph.
		IX.B.4	Changed MA to EO.
		IX.B.6	Changed Quality Control to Quality Assurance.
	100-113	X.A.	Added Director of Engineering.
		X.B.	Paragraph renumbered correctly. Changed Quality Control to the Reliability Section of the Engineering Department.
		X.F.6.	Added FAR Par 43, Appendix A, added DC-10 and removed Previously Approved from SFAR 36 repairs.
	114-125	XI.B.	Changed "all fleet aircraft' to " DC-8 and DC-10.
		XI.C	Added and not directly to Note.
		XI.D.	Changed "Requiredon the spot" to RII.
		XI.E.	Added RII and removed inspection in paragraph E2a. Added "RII trained and" in front of.
		XI.G.	Added note about RII stamps.
	126-129	XII.	Revised MEO103.
	130-141	XIII	Added entire section on Request for Services. Added MEO 154. Renumber remaining sections.
	143-154	XiV	Changed Maintenance Authorization (MA) to Engineering Order (EO) for entire Section Added MEO 155, Engineering Orders.
	161-169	XVIII.B.4	Removed "and/or Designated" from B4a.
	170-176	XIV.A.	Added and DC-10 to first sentence.
		XIV.B.6	Changed "The Captain " to "All Flight Crew members".
		XIV.C.	Changed "airworthy" to safe in Note. Added MEO06 and MEO30.
	177-182	XX.A.	Added "through its Operations Specifications" in front of authority. Added FAR Part 91.611 to paragraph A3.

CHAPTER	PAGE	ITEM	PURPOSE
5	30-33	VII.B.	Revised MEO26.
	34-37	VIII.B.1	Changed Technical Publications to Maintenance Programs and Publications.
,	49-51	X.B	Changed Senior Director to Vice President.
	54-61	XII.D.	Changed the address of the FAA in Figure 4.