EMERY WORLDWIDE AIRLINES

RESPONSE TO

EXHIBIT 17R

EWA'S RESPONSE TO LETTER DATED Opil 29, 1998



May 18, 1998

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letter dated April 29, 1998 regarding an enroute inspection on April 20, 1998 performed by Mr. Wilbert Robinson.

Item 1) of your letter was addressed April 28, 1998 (see attached log page).

Item 2) is correct in that this is no longer an EWA Line Station by Classification I, II or III.

The current revision 19, dated November 12, 1997 has been updated accordingly as EWA does not normally operate into this station.

Mr. Edward Jones will provide you the current EWA Line Station listing. As you know, this list changes frequently and is maintained by Quality Control.

Thank you for your report on these items.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

Attachments



Administration

San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

April 29, 1998

Emery Worldwide Airlines Mr. Thomas M. Wood Director, Quality Control 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

On April 20, 1998, an Enroute Inspection was conducted on an Emery Worldwide Airlines (EWA) DC-8 aircraft, N988CF, Flight 009, San Jose/Dayton, by Inspector Wilbert J. Robinson of this office. In addition, a Maintenance Facility Inspection was conducted by this same Inspector at the EWA New Orleans, Louisiana (MSY), maintenance facility. Accordingly, two issues have evolved as a result of these inspections:

- 1) It was noted by Inspector Robinson and verbally communicated to the flight deck crew that the left side aft most cockpit side window appeared crazed and distorted.
- 2) Inspection of the EWA New Orleans maintenance facility revealed that EWA maintenance personnel were not stationed at this facility. Rather, the EWA maintenance requirements were attended by Ryan International Airlines mechanics as per a contractual agreement.

As you are aware, Revision 19 to the EWA Maintenance Policy & Procedures Manual dated November 12, 1997, revised the status of this maintenance facility from a Class III facility, wherein contractual agreements between EWA and other maintenance providers are the norm; to that of a Class II facility, wherein EWA personnel are staffed. Therefore, a clarifying amendment to the Maintenance Policy & Procedures Manual, Chapter 2, is hereby requested regarding maintenance personnel staffing.

Should you have any questions regarding these issues. please call at your convenience.

Sincerely,

Joseph A. Abramski Principal Maintenance Inspector

cc: Richard Jacobson - EWA Richard Hickey - EWA W.J. Robinson - FAA

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EWA'S RESPONSE TO LETTER DATED

EMERY WORLDWIDE AIRLINES MAINTENANCE RELIABILITY PROGRAM DOCUMENT NO. EWA-51890

EMERY WORLDWIDE AIRLINES

MAINTENANCE

RELIABILITY

PROGRAM

DOCUMENT NO. EWA-51990

JANUARY 15, 2000 REVISION 8

Original Issue: October 1, 1990

EMERY WORLDWIDE AIRLINES MAINTENANCE RELIABILITY PROGRAM DOCUMENT NO. EWA-51990

LIST OF EFFECTIVE PAGES

Page No.		Revision	Date	Page	No.	Revision	Date
Cover RevList	i ii	Rev 8 Rev 8	01/15/00 01/15/00	Chpt 6 Chpt 6 Chpt 6	Pg 1 Pg 2 Pg 3 Pg 4	Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00
EffPg EffPg	1 2	Rev 8 Rev 8	01/15/00 01/15/00	Chpt 6 Chpt 7	Pg 1	Rev 8	01/15/00
TofC TofC	Pg 1 Pg 2	Rev 8 Rev 8	01/15/00 01/15/00	Chpt 7 Chpt 7	Pg 2 Pg 3	Rev 8 Rev 8	01/15/00 01/15/00
Chpt 1 Chpt 1 Chpt 1 Chpt 1 Chpt 1	Pg 1 Pg 2 Pg 3 Pg 4 Pg 5	Rev 8 Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00	Chpt 8 Chpt 8 Chpt 8 Chpt 8 Chpt 8	Pg 1 Pg 2 Pg 3 Pg 4 Pg 5	Rev 8 Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00
Chpt 1	Pg 6	Rev 8	01/15/00	Chpt 9 Chpt 9	Pg 1 Pg 2	Rev 8 Rev 8	01/15/00 01/15/00
Chpt 2 Chpt 2	Pg 1 Pg 2	Rev 8 Rev 8	01/15/00 01/15/00	Chpt 10 Chpt 10	Pg 1 Pg.2	Rev 8 Rev 8	01/15/00 01/15/00
Chpt 3	Pg 1 Pg 3 Pg 4 Pg 5 Pg 6	Rev 8 Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00	Chpt 10 Chpt 10 Chpt 10 Chpt 10 Chpt 10 Chpt 10 Chpt 10 Chpt 10	Pg 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00
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Chpt 5 Chpt 5 Chpt 5 Chpt 5 Chpt 5 Chpt 5	Pg 1 2 2 Pg 3 4 Pg 5 6	Rev 8 Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00	Chpt 10 Chpt 10	Pg. 19 Pg. 20 Pg. 21 Pg. 22 Pg. 23	Rev 8 Rev 8 Rev 8 Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00
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EMERY WORLDWIDE AIRLINES MAINTENANCE RELIABILITY PROGRAM DOCUMENT NO. EWA-51990

LIST OF EFFECTIVE PAGES

Page No	Revision	Date	Page No.	Revision	Date
Chpt 10 Pg. 25 Chpt 10 Pg. 26 Chpt 10 Pg. 27 Chpt 10 Pg. 28 Chpt 10 Pg. 30 Chpt 10 Pg. 31 Chpt 10 Pg. 32 Chpt 10 Pg. 33 Chpt 10 Pg. 33 Chpt 10 Pg. 35 Chpt 10 Pg. 36 Chpt 10 Pg. 36 Chpt 10 Pg. 37 Chpt 10 Pg. 37 Chpt 10 Pg. 38 Chpt 10 Pg. 39 Chpt 10 Pg. 39 Chpt 10 Pg. 39 Chpt 10 Pg. 40 Chpt 10 Pg. 41 Chpt 10 Pg. 42	Rev 8 Rev 8	01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00 01/15/00			

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January 15, 2000 Revision 8



Air Carrier Certificate

This certifies that

EMERY WORLDWIDE AIRLINES, INC.

One Emery Plaza

Vandalia, OH 45377

has met the requirements of the Federal Aviation Act of 1958, as amended, and the rules, regulations, and standards prescribed thereunder for the issuance of this certificate and is hereby authorized to operate as an air carrier and conduct common carriage operations in accordance with said Act and the rules, regulations, and standards prescribed thereunder and the terms, conditions, and limitations contained in the approved operations specifications.

This certificate is not transferable and, unless sooner surrendered, suspended, or revoked, shall continue in effect indefinitely.

By Direction of the Administrator

Certificate number: RRXA558B David E. Hanley (Signature)

Effective Date: June 19, 1987 Manager, Flight Standards Division (Title)

Issued at: GL05 Great Lakes Region (Region/Office)

Reissued: December: 17, 1999

EWA'S LETTER IN RESPONSE TO LETTER DATE



September 18, 1998

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to conversations this week during your spot inspection of aircraft records.

I understood per your advisement Monday, that the purpose of your inspection was to clean up some open issues in preparation for the possible certificate move.

Per your advisement to me on Wednesday, you stated you would provide any problems found at a debrief to me.

I was informed this morning by Edward Jones, that you had completed your inspection yesterday and would not provide an outbrief to me.

Please call me next week if I can be of further assistance

Sincerely,

Thomas M. Wood
Director Quality Control

TMW/re

cc:

Kent Scott René Visscher





Thomas M. Wood

DIRECTOR QUALITY CONTROL

303 CORPORATE CENTER DR. VANDALIA, OH 45377 FAX: (937) 898-2803 PHONE: (937) 454-3940

FACSIMILE TRANSMISSION COVER SHEET

DATE:
DELIVER IMMEDIATELY TO:
NAME: JOE ABAAMSKI TELEPHONE #:
COMPANY / DEPARTMENT: FAA SJC
This is page 1 of pages sent in transmission regarding the following principal subject(s):
goe Food Morning: I received a copy of your
fax to Bot- and Jum. Per their brief to
me I understand you requested the following
_ changes to the program.
1) Incorporate the MRB document zonal Inspection
Reference Information 2) AD numbers and 3) CPCP
numbers. I have instructed Bot to perform these
revious. Please call today and we can discuss
numbers. I have instructed Bob to perform these revisions. Please call today and we can discuss FACSIMILE MESSAGE FROM: To you and your office.
NAME: Thomas M. Wood



Fax Cover Sheet

FEDERAL AVIATION ADMINISTRATION Flight Standards District Office 1250 Aviation Avenue Suite 295 San Jose, California 95110-1130

To: Bob Peck / Jim Feisley - EWA

Phone:

Fax:

From: Joseph A. Abramski

Principal Maintenance Inspector

Phone:

Fax:

Date: November 4, 1998

Total Pages

(with cover): 2

Comments - Bob Peck / Jim Feisley:

This will confirm our telecon this day regarding the EWA DC-10 aircraft MSG3 maintenance program submitted to this office for our review. Present at this discussion were Inspectors Roger Sigg, Larry Moheit, and myself of this office; and you gentlemen.

Our discussion concerned the irregularities, inconsistencies, and incomplete EWA work cards and references thus far discovered by us in our review merely of the A1 and C1 inspection documents. As I stated to you in this telecon, in our opinion these program documents should not have been sent to this office for our review based upon their present order and format. We have expended an inordinate amount of time on a fragmented maintenance program that, as you are aware, was twelve (12) weeks late in arriving at this office for our review, according to the original schedule of events.

It is our understanding that EWA will finalize at the end of this workweek, a <u>complete</u> and <u>comprehensive</u> MSG3 inspection program to be forwarded to this office for our review. Please insure that these documents are indeed

<u>pristine</u>, as otherwise, undesireable delayed approval may be encountered which would be counterproductive to our mutual efforts. Should you require additional time to verify the contents of the program for completeness prior to submission, please advise us accordingly.

Regards,

Joe Abramski

CC: Rene Visscher

ENA'S RESPONSE TO LETTER DATEN.

November 13, 1998



November 20, 1998

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to our letters, faxes and telecon discussions of EWA's submitted DC-10 Inspection Program Manual and Time Limits Manual.

I would first like to thank John Howard, Roger Sigg, Larry Moheit and yourself for your efforts in providing support for the continuance of the inspection program review.

The Reliability and Quality Control Sections have completed their final audit and have revised the program to meet all regulatory requirements and in accordance with the MSG-3 MRB program requirements. This revision also includes changes as required by your previous letters, and will be overnighted to you today.

We request your expeditious review of these revisions, and as always will provide management and technical personnel to your office to present and answer questions you may have.

Thank you for your support on this issue.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

Attachment

cc: Rene Visscher

DC-10 Inspection Program Submittal Record of Correspondence/Telephone Discussions Summary

September 15, 1998	Bob Peck letter to Joseph Abramski, subject; advising FAA figures will not be included in EWA inspection program due to delay from Boeing.
October 26, 1998	Thomas Wood letter to Joseph Abramski, subject; EWA DC-10 Inspection Program/Time Limit Manual submittal for FAA review/approval.
November 2, 1998	Thomas Wood letter to Joseph Abramski, subject; EWA notification of submitted DC-10 Inspection program irregularities noted in a second Quality Control audit, scheduled to review/correct and provide changes by November 6, 1998.
November 3, 1998	Roger Sigg fax to Thomas Wood, subject; follow-up to telecon same day with Roger Sigg, Larry Moheit and John Howard, discussing Nov. 2, 1998 faxed letter to Joseph Abramski. FAA requested CPCP manual and stated they would continue to work the submitted inspection program until they received the revised program.
November 4, 1998	Joseph Abramski fax to Bob Peck and Jim Feisley, subject; follow-up to telecon same day with Larry Moheit, Roger Sigg and Joseph Abramski concerning inspection program irregularities. FAA agreed to additional time needed by EWA to complete, and required three additional items to be included.
November 5, 1998	Thomas Wood fax to Joseph Abramski, subject; follow-up to 11-4-98 telecon with Bob Peck and Jim Feisley, and provided statement of incorporation into EWA program 1) MRB zonal inspection 2) A.D. numbers and 3) CPCP numbers.
November 9, 1998	Joseph Abramski fax to Thomas Wood, subject; follow-up to telecon same day which Joseph Abramski provided concurrence of the FAA required three additions to the inspection program, and additional time to complete.

November 13, 1998

Joseph Abramski letter to Thomas Wood, subject; EWA's submitted inspection program "C" Check interval does not meet new operator criteria, therefore requiring adjustment and resubmittal.

EWA'S RESPONSE TO LETTER DATED Movember 30, 1998

F



November 25, 1998

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to our telephone call yesterday, November 24, 1998 concerning the conference call with Larry Moheit, Roger Sigg, Bob Peck, Jim Feisley and Steve Gillespie.

I understand from Bob Peck that Larry Moheit addressed some initial concerns and Bob agreed to address these issues immediately.

Per Larry Moheit's request to Bob Peck, and your agreement, I will send Bob Peck, Jim Feisley and Bob Crabtree to your SJC office on December 7, 1998 to provide technical support for the completion of the program review. The EWA staff is prepared to spend the week if it is needed. This joint review process, as you and I have experienced, provides excellent results.

Thank you for your support.

Sincerely,

Thomas M. Wood
Director Quality Control

TMW/re

cc:

Rene Visscher Bob Peck Mr Joseph Abramski FSDO-SJC 1250 Aviation Avenue, Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is to inform you of the contents of this package. The contents are as follows:

- 1. DC-10 Maintenance Figures Reference
- 2. DC-10 A-Check
- 3. DC-10 C-Check
- 4. DC-10 Road Map

The corrections for the Time Limits Manual and the front matter for the DC-10 Maintenance Figures Manual will be sent out Monday. If you have any questions please call.

Thanks for your support.

Sincerely,

Bob Peck

Manager of Reliability

cc:

Larry Moheit

Roger Sigg

Tom Wood

FAA TELECON WITH JOHN HOWARD 12-10-98 9:00 A.M.

- I advised John I was calling him in the absence of Joe Abramski.
- I went over the fax sent to me 12-9-98 by Joe concerning the DC-10 certification status.
 - 1. Requested Roger Sigg review status and call me. Mr. Howard assured me he would.
 - 2. Discussed Nick Pearson's PAI review not complete.
 - a) John stated they have augmented this with two additional PAI types, Mr. Hardy and Jim Carnathan. John stated Mr. Pearson by office priority of working enforcement items against EWA, has been covered up with several letters of investigation, and certification work would follow this priority.
 - 3. I requested from John to receive the completion of the IPM review with exception of the avionics, as soon as possible, and he stated Mr. Sigg would provide me with that review status information.
- John asked if the DC-10 delivery schedule had changed, and if so to send him a revised DC-10 Certification Plan schedule for the maintenance documents.
- I stated we had not received the scheduled output date change in writing, and upon receipt would provide the revised schedule to him.
- I discussed Mr. Abramski's objection to using the DC-10 Continental Maintenance Manual as EWA's manual, in place of the Douglas Master. John stated their concern is that EWA does not have equipment and or staffing to support these procedures.
- I informed John that I talked to the Manager of Technical Publications Mr. David Heggland of Continental (713-324-8147) and he advised me that they made all manual revisions themselves, making it necessary to use their manual. I also told John that Mr. Heggland used the carriers manual when they purchase aircraft and it is Industry Standard. John stated he would call Mr. Heggland and discuss this.
- I informed John that I felt it necessary to visit the office next week to meet with the EWA's Principals and bring closure by agreed solutions to any open issues.
- John stated to coordinate this visit through the principals, and I assured him I would.

FAA TELECON WITH ROGER SIGG, LARRY MOHEIT BOB PECK, JIM FEISLEY, ROBERT CRABTREE AND MYSELF 12-10-98 2:00 P.M.

- Roger stated he was working to complete his review of the "C" Check package. He
 identified by card number approximately ten cards with minor changes, and or he accepted
 the card based on our explanation.
- Larry Moheit stated he was working to complete his review of the "A" Check package. He identified by card number approximately five cards with minor changes, and or he accepted the card based on our explanation.

FAA Telecon's Page 2

• I summarized their minor concerns as not representing any significant problems, and requested that Jim and Bob come to SJC Monday, December 14, 1998 to work with them to bring closure to their review process. Roger Sigg understood this process and stated he would present it to John Howard.

FAA TELECON WITH JOHN HOWARD 12-10-98 4:00 P.M.

- John called and stated Roger had addressed our request to send two technical people to their office December 14, 1998 to complete the FAA review of the DC-10 IPM.
- John stated that Roger would be complete not later than Monday on the "C" Check and Larry was completing the "A" Check. He also stated he was not aware they were this far along.
- I requested from John this final review jointly as we have previously done on the B-727
 Maintenance Program, B-727 MEL, Cargo Loading Manual and the most recent DC-10
 MEL at their office, for final review and receive FAA approval after any agreed changes.
- I expressed to John this process would be like we were just a few blocks apart, to which we would be exchanging office visits to expedite the process, and he agreed.
- John and I agreed to send the two technical people to SJC on Monday, and this review process would be accomplished, with the EWA technical people making the changes to the program onsite, and therefore having a completely FAA audited program.
- John stated it may not be possible to finish the avionics review next week, but that the two
 additional avionics people are presently working this review.
- I told John that we had sent the avionics mod status to Nick Pearson, and we are reviewing the program again to make sure all items are included.
- I expressed my hope to John that during the review next week that the avionics review could be complete with the support of EWA's technical people onsite.
- John stated he wanted to make sure we were not going to pressure the inspectors review next week to complete the process, to bring home an IPM FAA approved form.
- I assured John that our intent is to promote the final review process, and to provide onsite
 technical support to our FAA Principals of their assigned tasks. I stated to John that since
 this collective review has been going on for seven weeks and is in the final state, it would
 benefit the FAA and EWA to complete this and move on to other required workload
 requirements.
- John stated he would contact Joe Abramski and inform him of this plan.
- I thanked John for his support on this subject and the other approved manuals we are already in receipt of on schedule.

Thomas in went



April 8, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your March 8, 1999, letter that we discussed during my March 23 and 24, 1999, meeting at your office.

It is my understanding from our meeting, and your meeting with Bruce Robbins and James Feisley on March 30 and 31, 1999, at your office, all items have been satisfied by your office.

I speak on behalf of the EWA senior management, in expressing our appreciation to your office's professional support and contributions to the success of the DC-10 certification plan.

We look forward to seeing Mr. Pearson and yourself, next week here in Dayton.

Sincerely,

Thomas M. Wood

Director, Quality Control

Attachments

TMW/csh

cc: Rene' Visscher
Tim Alman
Bruce Robbins
Edward Jones
Abraham Michael

Wood, Thomas M

From:

Wood, Thomas M

Sent:

Sunday, December 13, 1998 11:12 AM Scott, Kent T; Visscher, Rene P DC-10 IPM Final Review

To:

Subject:

Kent/Rene: I presented to Rene Friday morning our plan to send two Reliability Technical people to SJC Monday, Dec. 14,98, to conduct a joint review of all the A and C check cards with the the FAA to bring closure to this ongoing review process. I spoke to John Howard, the Office Supervisor and he agreed to this process. The faxed letter from Larry Moheit to me confirms this arrangement. We have addressed all known items of the FAA, and are preparing to address any thing else with them jointly this coming week. We will make changes on the spot, as we are taking a PC to perform this task.

> Kent, I will fax you the Telecon record that was performed to achieve this task.

As you will see from my notes with John Howard, he was supportive in this process. It is my intent and goal, to finish the tolal review/approval process of the program by end of next week. I believe as in the past, this possible stonewall will be resolved by a technical front as it is a FAA APPROVED PROGRAM.

Thomas

MEMORANDUM

TO:

Thomas Wood, Director, Quality Control

FROM:

Bob Peck, Manager of Reliability

SUBJECT:

DC-10 Inspection Program Final Review Schedule

Date:

21 December, 1998

Dear Sir,

The following DC-10 Inspection Program Final Review schedule is forwarded for your review and comments. This review schedule is based on having two dedicated individuals. The schedule is set up so that a continuous flow of DC-10 Workcards is going on at all times. The plan as discussed with Jim Feisley, Edward Jones, and Andrew Porter of Quality Control is as follows:

- 1. The A-Check will be completed through Reliability by 24 December, 1998. Cards will be forwarded to Quality Control for their review as sufficient quantities are completed by Reliability.
- 2. The Quality Control review should be completed by 31 December, 1998, with submittal to the FAA on that date.
- 3. The C-Check review will begin in Reliability on December 28, 1998. It is anticipated that this review/edit will require at least 2 weeks. Cards will be forwarded to Quality Control as sufficient quantities are ready for their review.
- 4. The Quality Control review will be completed on 0r before 15 January 1998. The C-Check cards will be forwarded to the FAA as sections have completed the Quality Control review.

The above schedule is contingent on each section having two individuals available for the review process. Mr. Feisley and Mr. Porter will work together on the initial review/edit with Ed Jones, Manager of Quality Control, doing the final Quality Control review.

Bob Peck

Manager of Reliability



January 18, 1999

176.

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letter dated December 21, 1998, in regard to the schedule of the final submittal of Emery Worldwide Airlines (EWA) DC-10 MSG-3 Maintenance Program, for your review and approval.

An immediate action plan was put in place as addressed in Robert Peck's Memorandum attached.

As agreed in our telecon, EWA Quality Control has audited and forwarded in part the DC-10 Aircraft Inspection Program to your office for your review. As of Friday, 1-15-99, your review of the Service Check and "A" Check had been completed, with changes made and faxed to you. We thank you for this review and your recommendations.

The "C" Check package is being overnighted to you today C- 1 thru C-3, and the remaining cards by Wednesday 1-20-99.

I would like to thank you and your staff for your continued support in the DC-10 Certification Plan.

Sincerely,

Hamos M. Wood

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc:

Rene' Visscher QC Managers

December 9, 1998



December 14, 1998

Mr. Joseph Abramski FSDO - SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to my letter sent to you September 23, 1998, which provided you the DC-10 Formal Training Program to support the DC-10 Certification Plan.

You stated to me your acceptance of this program per a telephone conversation following your receipt of this letter.

This letter provides you the completion of the initial scheduled training program.

DC-10 Avionics/Electrical Course - 8 completed course

DC-10 Systems Course - 34 completed course

DC-10 Executive Course - 10 completed course

The attached course certificates are provided to substantiate the completion.

In addition to this initial formal training given, three weeks (240 hours) of specialized DC-10 on-the-job training will be provided by AMT DC-10 Technically qualified instructors upon delivery of the aircraft with three weeks of direct daily maintenance support. One instructor will be positioned at Dayton, and another at SJC. Other trained DC-10 line stations are: LAX, PDX, and DFW.

Additional AMT DC-10 training has been scheduled for January and February 1999.

This will complete the training certification requirements for the DC-10 aircraft to be placed on EWA's operation specification. Please call if you have any questions.

Singerely,

Thomas M. Wood

Director Quality Control

TMW/re

Attachment

cc:

Rene' Visscher Bruce Robbins



THE RESERVE TO THE PROPERTY OF THE PROPERTY OF

WORLDWIDE FIRLINES A COMPANY

This is to certify that

James Coors

has satisfactorily completed a 40 hour course in DC-10 Avionics/Electrical

October 22, 1998

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October 22, 1998

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Daniel Maier

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October 15, 1998

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October 15, 1998

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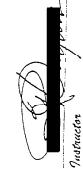
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October 8, 1998

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October 8, 1998

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October 15, 1998

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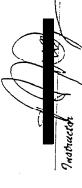
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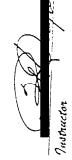
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October 8, 1998



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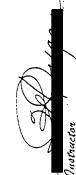
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October 1, 1998

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October 8, 1998





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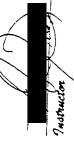
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October 8, 1998 Date



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October 28, 1998

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October 28, 1998

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October 31, 1998

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January 15, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to my letter sent to you December 14, 1998, which provided you the certificates of the DC-10 training program.

EWA has completed another DC-10 System Course that trained an additional eight (8) mechanics. The current total mechanics training are as follows:

DC-10 Avionics/Electrical Course -8 completed course

DC-10 System Course - 45 completed course

DC-10 Executive Course - 10 completed course.

Quality Control has reviewed the American Airlines 40 hour DC-10 Airframe, Powerplant and Electrical course provided to the four (4) EWA SJC Line Station mechanics, and find it acceptable. It is also important to note that the SJC line station has been providing maintenance support to the Gemini Air Cargo DC-10-10 for over three months. Please advise me of your acceptance of this training.

Sincerely,

Amusmadel

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc:

Rene' Visscher

Bruce Robbins Edward Jones WORLDWIDE FIRELINES

ACTF COMPANY

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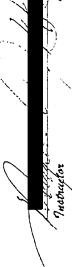
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For Satisfactory Completion of the

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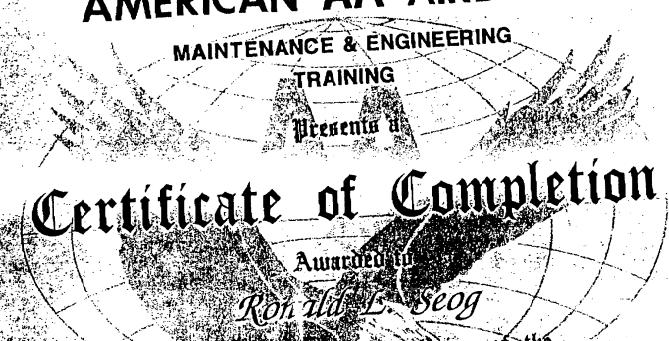
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February 11, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is a follow-up to my letter sent to you January 15, 1999, which provided you the certificates of the DC-10 training program.

EWA has completed another two (2) System Courses (40/120 Hours) that trained an additional twenty one (21) mechanics. The current total mechanics trained are as follows:

DC-10 Avionics/Electrical Course - 8 completed course

DC-10 System Course 40/120 Hours - 70 completed course

DC-10 Executive Course - 10 Completed course.

The DC-10 System Course numbers reflect (4) EWA SJC mechanics that received the 40 Hour American Airlines System Course.

Sincerely,

Thomas M. Wood Director Quality Control

TMW/re

Attachments

cc: Rene' Visscher

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Austin Neel

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James Schald

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DC-10 Systems

February 11. 1999

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February 11, 1999

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DC-10 Systems

February 11, 1999

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February 11, 1999

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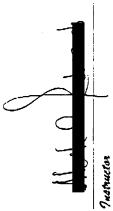
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February 11, 1999



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February 11, 1999

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DC-10 Systems

February 11, 1999

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February 11, 1999

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A COF COMPANY

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John Malls

has satisfactorily completed a 120 hour course in

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January 28, 1999

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January 28, 1999

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January 28, 1999

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Robert Hartley

has satisfactorily completed a 120 hour course in

DC-10 Systems

January 28, 1999

Instructor

HEVENEY! WORLDWIDE AIRLINES A COMPANY

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Datrick Fredrick

has satisfactorily completed a 120 hour course in

DC-10 Systems

WORLDWIDE FIRLINES
A COFCOMPANY

This is to certify that

Michael Benner

has satisfactorily completed a 120 hour course in

DC-10 Systems

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January 28, 1999 Date

Instructor

I EMERY WORLDWIDE FIRLINES

This is to certify that

Frank Exilen

kas satisfactorily completed a 120 hour course in

DC-10 Systems

ENAS RESPONSE TO LETTER DATED. December 15, 1998



January 11, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter constitutes Emery Worldwide Airlines, Inc. (EWA)'s formal response to your letter of investigation (991NP150023), addressed to EWA's President and Chief Operating Officer, dated December 15, 1998. (See Attachment 1).

EWA reported this damage to the Los Angeles Aircraft Certification office to Mr. Greg Delibero on December 17, 1998. (See Attachment 2).

As a result of an initial/indepth inspection of the subject cargo door and previous maintenance history, it was concluded that the door was damaged due to human factors.

EWA's Reliability Section reported the closed status of this subject which required a major repair to return the aircraft to service, by faxing the Operational Difficulty Report to you on January 3, 1999. (See Attachment 3).

A comprehensive review was performed by EWA Technical Service Management to determine why the SDR Report was not reported per the accepted EWA Maintenance Policy and Procedure Manual. The following is a summary of events and a Comprehensive Fix to prevent future occurrences.

1. On November 25, 1998, aircraft N10EWP was taken out of service and removed from the operation during a schedule inspection, to perform unscheduled maintenance to repair the cargo door. Maintenance Control did not process the EWA ME023 Mechanical Interruption and MRR Advisory form and submit to Reliability as it was removed from service. At this time Maintenance Control had no knowledge of the requirement of a major repair.

Mr. Joseph Abramski Page 2 January 11, 1999

2. The Manager of Reliability submitted the required MRR Report upon receipt of the repair package from the FAA DER. This report was delayed due to the Thanksgiving and Christmas Holiday period.

Comprehensive Fix:

- The Manager of Reliability will provide re-current training to all Staff members on the Maintenance Policy and Procedure Manual, Chapter 4, Section X, Mechanical Interruptions and MRR Reporting. He will also focus on the timeliness of reporting to prevent future delay reporting occurrences.
- The Manager of Maintenance Control will provide re-current training to all Staff members on the Maintenance Policy and Procedure Manual, Chapter 4, Section X, Mechanical Interruptions and MRR Reporting. A copy of this letter will be addressed in this training.
- 3. The Manager of Quality Control in addition to auditing the submitted ME028 forms to Reliability, will add to the daily audit of all log pages, the review of discrepancies sign-off indicating a major repair was performed. The Quality Control Inspection Rep. will contact the Manager of Reliability and ensure that a MRR has been reported.
- 4. The Manager of Line Maintenance will provide re-current training to all Line Station Supervisors/Mechanics on the Maintenance Policy and Procedure Manual Chapter 4, Section X, Mechanical Interruptions and MRR Reporting, specifically to the reporting responsibility of the Maintenance personnel. A copy of this letter will be addressed in this training.

EWA submitted one MRR Report in 1998 that resulted in a major repair performed during line maintenance. All other major repairs MRR's were reported routinely through heavy maintenance visits, C and D checks. The isolation of this one line maintenance occurrence is representative in the isolated occurrence of the MRR not reported in a timely manner, even though it was only a few days.

Mr. Joseph Ambraski Page 3 January 11, 1999

I trust EWA's correction action taken is satisfactory to your concerns. Should you wish to discuss this matter more fully, EWA's Senior Management and I would be more than willing to meet with you and address your concerns.

Sincerely,

Thomas M. Wood Director Quality Control

- ...**-**

Attachments

cc: Kent Scott
Rene Visscher
Wayne Farnsworth
Jack Smith
Roger Rosher
Robert Peck
Edward Jones

DCQ:mlb



December 17, 1998

Mr. Greg Delibero
Federal Aviation Administration
Los Angeles Aircraft Certification Office
3960 Paramount Drive
Lakewood, CA 90712-4137

Dear Mr. Delibero:

This letter is a follow-up to our telephone conversation today regarding the DC-8 Rosenbalm STC SA1802SO Cargo Door repair performed on a Emery Worldwide Airlines (EWA) DC-8-73, Serial Number 46095, Production Number 497, N105WP at LAX.

The subject aircraft received a repair to the cargo door due to a crack found during preflight. The following details are pertinent to this aircraft.

- STC Door Installation
 - March 17, 1989
 - Installer, Zantop Macon, Georgia
 - Aircraft TAT 45,890 and TC 14,988
- II. Aircraft TAT/TC as of 11-27-98
 - TAT 65,843
 - TC 21,197
- III. TAT/TC on Cargo Door STC Installation
 - TAT 19,953
 - TC 6,209
 - 9 years and 8 months

Per our conversation, you informed me that the FAA considers this to be as safety concern and is preparing an Airworthiness Directives (AD) for a one-time inspection of the cargo door.

As I discussed with you, EWA issued a Fleet Campaign Directive No. 52-6 (attached) to inspect the Rosenbalm and Monarch Cargo Doors in this specific area.

Mr. Greg Delibero Page 2 December 17, 1998

Per our agreement, this letter is being copied to the DC-8 JTF members for their initial notification and opportunity to perform inspections on their aircraft. Please advise me if the EWA FCD inspection will comply with the proposed AD inspection.

I would like to thank you for your support in this matter.

Sincerely,

Thomas M. Wood/se
Thomas M. Wood
Director, Quality Control

Attachment

cc: Rene' Visscher

JTF Members

U. S. Department of Transportation

Operational Difficulty Report

Federal Aviation
Administration

OPERATOR CONTROL NUMBER

ATA CODE

Administration	
----------------	--

RRXA98959

5230

MAJOR EQUIPMENT IDENTITY

Enter pertinent data	MANUFACTURER	MODEL	SERIAL NO.	TOT. TIME	TOT. CYC.
AIRCRAFT	DOUG	DC373	46095	65,847	21,199
POWERPLANT					
PROPELLER					

PROBLEM DESCRIPTION

DATE	STATUS	OPER. DESIG.	OPER TYPE	A/C N NUMBER	PREC. PROCED.	NATURE	STAGE OF FLIGHT	STATION	FLIGHT #
981215	С	RRXA	01	105WP	к	J	IN	KLAX	835

Discrepancy/Corrective Action:

DURING FLIGHT FROM KALT TO KLAX CARGO DOOR (OPEN) LIGHT ILLUMINATED, NO CABIN PRESSURE CHANGE, LOCK BAR APPEAR TO BE AT FULL TRAVEL (CLOSED). FLIGHT CONTINUED TO KLAX WITHOUT INCIDENT ///// GROUND INSPESTION ///// FOUND DOOR LOCKING MECHANISMS OUT OF SEQUENCE, TORQUE TUB DAMAGED (PN: 23630352). DOOR HYDRAULIS SYSTEM, VALVE UNSERVICEABLE (PN:1629-3-24) AND CYLINDER ASSEMBLY (PN: 8520008009) MISSING BEARING. STRUCTURE DAMAGE, ALL 12 DOOR FRAMES CRACKED AT LOWER SECTION. ///// CORRECTIVE ACTION///// REMOVED AND REPLACED THE FOLLOWING COMPONENTS; TORQUE TUB (PN: 23630352), VALVE (PN:1629-3-24) AND CYLINDER ASSEMBLY (PN: 8520008009). STRUCTURE DAMAGE; REMOVED DAMAGES FROM 12 FRAMES, FASRCATED AND INSTALLED REPAIR DOUBLER IAW FAA DER COTNEY APPROVED ENGINEERING SKETCH F84-R01.

	SPECIF	IC PART CAUSING	PROBLE	М		
PART NAME MF	G. PART NUMBER	SERIAL #	!	PART CONDITION DAMAGED	PART/DEFECT LOC. FUSELAGE	
PART TOTAL TIME	PART TOTA	AL CYCLES	PA	RT TIME SINCE:	Overnau: Repair Inspection	
COMPONENT NAME COMPONENT MANUFACTURER		ANUFACTURER	COMPONENT PART #		COMPONENT SERIAL #	
COMPONENT TOTAL TIME	COMPONENT TO	OTAL CYCLES	COME	PONENT TIME SINCE:	Cvemaul Regair	
	:		į		Inscection	

SUBMITTED BY

NAME			
EMERY WORLDWIDE AIRLINES INC	SUE. CODE	DIST. OFF.	ALERT FILM
	A	WP:5	



U.S. Department of Transportation

Federal Aviation Administration

March 16, 1999

San Jose Flight Standards District Office

RECEIVED MAR 2 2 1999

KENT T. SCOTT

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

CERTIFIED-RETURN RECEIPT

Kent T. Scott, President and Chief Operating Officer Emery Worldwide Airlines, Inc. One Lagoon Drive Redwood City, CA 94065

Dear Mr. Scott:

FAA EIR Consolidation Notification

This letter is to inform you that the San Jose Certificate Holding Office has determined that administratively, it is advantageous to the Administrator to consolidate, into one Enforcement Investigation Report (EIR), the following several EIR's:

EIR 99WP150023 is incorporated into EIR 99WP150028 EIR 99WP150025 is incorporated into EIR 99WP150028 EIR 99WP150029 is incorporated into EIR 99WP150028 EIR 99WP150032 is incorporated into EIR 99WP150028 EIR 99WP150033 is incorporated into EIR 99WP150028

Any correspondence relating to any of the above EIR's listed should be referenced to EIR 99WP150028 as shown above. All Emery Worldwide Airlines, Inc. (EWA) responses received to date, have been incorporated into EIR 99WP150028.

If you have any questions, please advise.

Sincerely,

John R. Howard Asst. Manager, A/W





January 10, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letters dated December 15, 1998 and December 21, 1998 concerning aircraft N105WP experienced main cargo door damage, and Telegraphic Airworthiness Directive (TAD) T98-26-51. (See Attachment 1)

Emery Worldwide Airlines (EWA) reported this damage to the Los Angeles Aircraft Certification office to Mr. Greg Delibero on December 17, 1998. (See Attachment 2).

Based on the initial determination by EWA, in concert with the FAA DER performing the line maintenance repair it was determined that the door was damaged due to FOD. EWA elected to issue a Fleet Campaign Directive 52-6 to inspect the Rosenbalm and Monarch STC Cargo door aircraft in this effected area as a precaution. As a result of this inspection, no defects were noted on other aircraft.

Mr. Andrew Albright, EWA Reliability Structural Analyst, performed an in-depth review of the cause of the damage. The following is the results of an analysis on aircraft N105WP main cargo door damaged that occurred on November 26, 1998 and was removed from service and repaired on December 14, 1998. The main cargo door sustained damages to the door locking mechanism, hydraulic system and primary structures. The crew's first indication was maintaining pressurizations written on log pages (6913-22, 6913-24 & 6913-25). The aircraft log pages reported FOD found by EWA's maintenance personnel in the lower forward section of the door causing the following chain of events. The events as written by flight crew members, repairs made by maintenance personnel and damaged parts replaced during the repair will demonstrate a pattern to which the damage conclusion was determined.

Pilot Write-Ups Sequence of Events:

Prior to aircraft's main cargo door damage on November 26, 1998, aircraft departed KELP to KDAY, log page 6913-22 pilot reports; "Both packs have to be at maximum to maintain the cabin below 10,000 feet, only able to get 7.6 differential left pack appears weak". Maintenance personnel inspected aircraft at KDAY and performed corrective action; "Removed FOD from along bottom edge of main cargo door, aircraft pressurization operations checked good, both packs operations checked good, no defects noted".

On November 27, 1998, Aircraft N105WP, departed KATL to KLAX. During the flight, log page 6913-25, pilot reports; "needed both packs to maintain pressurization, cargo door lights illuminates during flight, no cabin pressure change—lock bar appears to be at full travel—closed, cargo door latches went to latch position before door was fully closed (out of sequence)." Corrective Action; EWA received on December 14, 1998, a repair for the main cargo door IAW FAA DER approved Cotney Engineering Sketch F84-R01.

Pilot Write-Ups Abbreviated (See Attachment 3 for actual log pages):

11-26-98

- FOD found in cargo door lower section
- Difficulties maintaining cabin pressurization
- Cargo door jerks while opening

11-27-98

- Difficulties maintaining cabin pressurization
- Cargo door open light illuminates
- Cargo door latches went to latch position before was fully closed

The FOD found by maintenance personnel at KDAY on November 26, appears to have affected the closure of the main cargo door at or before departing KELP. This appears to be true after reviewing prior log page (6913-21), did not indicate aircraft pressurization lost. Review of the remaining log pages 6913-23 and 6913-24 found discrepancy and write-up on pressurization. Aircraft log page 6913-24 pilot reports, "Air leak forward lower edge of cargo door, must run both packs at high flow to keep aircraft pressurized, and the cargo door jerks while opening". EWA maintenance personnel inspected aircraft at KLAX and performs corrective action by, "Straightened seal depressor at lower edge and serviced reservoir."

In summary, the log page revealed FOD in lower section of the main cargo door found by EWA maintenance personnel. Maintenance inspected the main cargo door lock mechanisms but were unable to find any damage and dispatched the aircraft for flight. The aircraft flew two additional legs that day without difficulties. The third flight indicated cabin pressurization problems and main cargo door jerked when opening. EWA has determined when the main cargo door closed at KELP, FOD

trapped in the locking mechanisms had shifted locking bar and locks to close unevenly, causing the locking system to become out of sequence.

Main Cargo Door Repairs:

The main cargo door damages repaired at KLAX by EWA maintenance personnel were recorded on EWA Non-Routine Maintenance Forms, on December 14, 1998. The main cargo door locking mechanisms and hydraulic system damages evaluated by EWA personnel. The locking device torque tub (PN: 23630352) damaged by excessive force created by hydraulic forces to shift locks combined with locks out of sequence.

EWA maintenance personnel evaluated the main cargo door hydraulic system damages. Their evaluation determined main cargo door jerking while opening caused by a damaged hydraulic valve (PN: 1629-3-24), removed and replaced. The Cylinder assembly (PN: 8520008009) found unserviceable, cylinder removed and replaced.

EWA maintenance personnel and FAA DER Mr. Derrick Seys, of the Cotney Company, inspected the internal door structure. Their findings were all 12 frames damaged, to which these frames run longitudinal each equipped with a locking hook. The Cotney Company analysis structural strengths lost which they can provide, they concluded repairs to all frames exceed limitations IAW DC-8 SRM type repairs. EWA requested and received a repair for all 12 frames from the Cotney Company (F84-R01), and repairs completed on December 14, 1998, IAW FAA DER approved Cotney Engineering Sketch F84-R01.

Summary:

The FOD found in the lower forward section of the main cargo door appears to have caused door locking system to become out of sequence induced by the torque tube. Evidence of FOD can be seen by the damage found to the lower forward door seal depressor. EWA has determined the damaged locks prevented the door from fully sealing, creating pressurization loss, as reported by crews. When the aircraft was in flight, pressurization engaged to the internal air pressure forcing the lower door edge out, creating just enough vibration from the loose door locks that illuminated the door open warning light. This constant vibration and air forces created enough stress on the forward door frames eventually cracked.

The request for information per your December 15, 1998 letter is enclosed (See Attachment 3).

Mr. Greg Delibero informed me on December 17, 1998 that a telegraphic airworthiness directive may be issued to inspect the Rosenbalm cargo doors in the area that the damage was found on aircraft N105WP.

On December 21, 1998, I coordinated the inspection requirements per T98-26-51 with Greg Dilibero and Rony Azzi of the Atlanta ACO, and of course yourself per the TAD. The inspection was performed on the nine (9) aircraft per the TAD item (B) and no defects were noted. (See Attachment 4).

I have addressed this TAD inspection requirement with other effected air carriers, and they also found no defects. I have also addressed this with Mike O'Neil and we will discuss this issue at the scheduled DC-8 Cargo Conversion Joint Task Force meeting next Tuesday, January 12, 1999 at Miami.

Please call if I can be of further assistance or provide additional information.

Sincerely,

Thomas M. Wood
Director Quality Control

attachments

cc: Rene Visscher QC Managers

TMW/amb

ATTACHMENTS

- 1. FAA PMI Letters to EWA Director Quality Control dated December 15, 1998 and December 21, 1998.
- 2. EWA Director Quality Control letter dated December 17, 1998 to FAA ACO Engineer Greg Delibero. (Copy of EWA FCD 52-6)
- 3. FAA PMI information requested per his December 15, 1998 letter.
- 4. EWA TAD notification letter to PMI dated December 21, 1998, and EWA response letter for TAD findings to ACO TAL, Rany Azzi, Copies of all MA's performing inspection, memo from EWA Manager Reliability for 30 day review per AD, AFM AD revision record.



December 17, 1998

Mr. Greg Delibero
Federal Aviation Administration
Los Angeles Aircraft Certification Office
3960 Paramount Drive
Lakewood, CA 90712-4137

Dear Mr. Delibero:

This letter is a follow-up to our telephone conversation today regarding the DC-8 Rosenbalm STC SA1802SO Cargo Door repair performed on a Emery Worldwide Airlines (EWA) DC-8-73, Serial Number 46095, Production Number 497, N105WP at LAX.

The subject aircraft received a repair to the cargo door due to a crack found during preflight. The following details are pertinent to this aircraft.

- I. STC Door Installation
 - March 17, 1989
 - Installer, Zantop Macon, Georgia
 - Aircraft TAT 45,890 and TC 14,988
- II. Aircraft TAT/TC as of 11-27-98
 - TAT 65,843
 - TC 21,197
- III. TAT/TC on Cargo Door STC Installation
 - TAT 19,953
 - TC 6,209
 - 9 years and 8 months

Per our conversation, you informed me that the FAA considers this to be as safety concern and is preparing an Airworthiness Directives (AD) for a one-time inspection of the cargo door.

As I discussed with you, EWA issued a Fleet Campaign Directive No. 52-6 (attached) to inspect the Rosenbalm and Monarch Cargo Doors in this specific area.

Mr. Greg Delibero Page 2 December 17, 1998

Per our agreement, this letter is being copied to the DC-8 JTF members for their initial notification and opportunity to perform inspections on their aircraft. Please advise me if the EWA FCD inspection will comply with the proposed AD inspection.

I would like to thank you for your support in this matter.

Sincerely,

Thomas M. Wood/re
Thomas M. Wood
Director, Quality Control

Attachment

cc: Rene' Visscher

JTF Members

EUR'S RESPONSE TO LETTER DATED February 10, 1999



February 18, 1999

Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letter dated February 10, 1999 regarding your request for the damaged material removed from the N105WP cargo door repair on November 27, 1998, to be forward to Mr. Michael O'Neil (See Attachment).

Mr. O'Neil had requested this of me in the January 12, 1999 DC-8 Cargo Conversion Joint Task Force (JTF) meeting; however, I did not know the availability status of the material at that time. I expressed no objection of his review and informed him I would let him know.

I spoke to Mr. O'Neil today by telephone and e-mailed him, regretfully informing him that these parts were routinely disposed of after Quality Control inspected these items last year (See Attachment).

I have previously provided you the comprehensive data package on the facts of the subject damaged door due to FOD in my letter dated January 10, 1999. I provided Mr. O'Neil a copy of this letter also on January 12, 1999.

I am sending you a copy of the JTF meeting to which this subject was addressed and responded to by Mr. O'Neil (See Attachment).

If you have any other questions, please don't hesitate to call me.

Sincerely,

Chance Me Ubest

Thomas M. Wood Director Quality Control

Attachments

cc: Kent Scott Rene Visscher



Administration

San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 298 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

February 10, 1999

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

This office has received a request for assistance from the FAA Los Angeles Aircraft Certification Office (LAX ACO) in retrieving, for FAA structural analysis, the fractured main cargo door frames on Emery Worldwide Airlines (EWA) DC-8 aircraft N105WP. As you are aware, the failure of these door frames was discovered on November 27, 1998, by EWA maintenance personnel in Los Angeles, California, and is the subject of an open investigation by the FAA into this matter.

Since it is our understanding that FAA LAX ACO engineering personnel have politely requested these door frames for analysis on a number of occasions without success, we would appreciate your cooperation in this matter.

Please forward the requested parts to: Mr. Michael E. O'Neil or Greg DiLibero

Aerospace Engineer

FAA Los Angeles Certification Office 3960 Paramount Blvd.

Lakewood, CA. 90712-4137

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector

cc: Rene Visscher - EWA Michael E O'Neil - FAA Greg DiLibero - FAA

Wood, Thomas M

From:

Wood, Thomas M

Sent:

Thursday, February 18, 1999 7:39 AM

To:

'Michael E O'Neil'

Cc:

Farnsworth, Wayne E; Tackett, Jerry B

Subject:

RE: FAA Airplane Review

Mike: EWA is pleased to make available a Rosenbalm STC aircraft at ATL on 2-22-99 per your request. The JTF Bidders Committee meet on 1-27-99 and 2-16-99 to discuss and implement a plan to move forward with Phase II. Both meetings with the STC Holders was very positive in taking a continued forward motion. I will send

you the Meeting Minutes.

With regard to your request of the damaged parts from EWA's aircraft N105WP that received FOD damage to the cargo door, and was repaired at LAX, I regretfully inform you that these parts were routinely disposed of after the Quality Control inspected these items last year. I provided you and my PMI a complete report of this subject. Wes Plattner has informed me that his Rosenbalm Vent Door STC has been taken off the fast track of being reviewed by the ATL ACO and does not know the completion date. AS EWA,s aircraft is soon to depart the scheduled Heavy Maintenance, this will not be installed as we had planned. Please call when you get time if you need more details of this message.

Thomas Wood

From: Sent:

Michael E O'Neil[■

To: Cc:

Wednesday, February 17, 1999 12:52 PM wood.thomas

Randy.Avera

Subject:

Tom:

On February 5, 1999, the FAA requested the JTF make available, for FAA review of the door annunciation system, a DC-8 modified by STC SA1802SO. At that time we suggested Monday, February 22, 1999 at the Emery facility at Hartsfield Int'l. Airport in Atlanta, GA.

Is this proposal acceptable? Please let me know when, where, and whom to contact to finalize plans for the visit.

Thanks,

MEO



JANUARY 12, 1999

MEETING MINUTES

- I. General Introduction and Welcome Members
 - The Chairman, Thomas Wood opened the meeting by welcoming all JTF members and guests.
 - Each attending person identified themselves and their relationship to the JTF effort.
 - Mike O'Neil introduced his guest from the FAA Seattle ACO, Mr. Mike Zielinski who is currently assigned to the B-727 Cargo Door STC reevaluation team.
 - Mike O'Neil stated that the FAA recognizes that the JTF is an ice breaking process resulting in a precedent setting group. He thanked the JTF for the progress made to date.
 - A total of nine (9) FAA ACO Engineers were present at this meeting.
 - The Chairman reviewed the sign-in sheet and confirmed that eight (8) of the eleven (11) JTF members were present, which constituted a quorum to perform business.
- II. Review/Accept November 3 & 4, 1998 Meeting Minutes.
 - The minutes were prepared to be read by the Chairman, and a motion by John Zappia of Fine Air was made to accept them as written, which received unanimous approval by the JTF membership.
 - The Chairman requested the membership to submit changes as applicable
 to update the "Operator Cargo Door STC Summary". Dan Johnson of Air
 Transport International (ATI) provided this information at the end of the
 meeting. All members agreed to review the listing and provide changes
 to the Chairman by end of January 1999.

Meeting Minutes - January 12, 1999 Page 2

III. JTF Funding Status

- The Chairman reviewed the "JTF Payment Status Report" provided to the membership in the meeting handout.
- The final cut off date for the JTF Joint Venture Agreement for membership entitlements to receive payment by December 15, 1998, was administered by the Chairman.
- Eleven (11) members out of fourteen (14) eligible members signed up and made the required payment which covered the cost of Phase I reevaluation of the five (5) STC data packages.

JTF Membership:

Fine Air Services, Inc./Agro Air STC Holder
(SA1862SO/ST00309AT)
Aeronautical Engineers Inc. - AEI STC Holder (SA1063SO)
Airborne Express/Airborne - STC Holder (SA2315SO)
American International Airways
Air Transport International
Aerolease International
Arrow Air, Inc.
Emery Worldwide Airlines
M.K. Airlines
Airmec
Aer Turas

NOTE: The JTF membership acts as the STC Holder for SA1802SO/SA421NW and SA1832SO.

The three airlines who chose not to join the JTF was Trans Continental Airlines (2 aircraft), Iberia Airlines (2 aircraft) and Florida West International Airways (1 aircraft).

 The Chairman confirmed that fifty percent payment was sent to all three contracted engineering companies performing the STC re-evaluation process.

Meeting Minutes - January 12, 1999 Page 3

- A special recognition and appreciation was given to Aerolease for providing the initial funds to cover the engineering cost of the STC reevaluation process until the membership could be established and receive funds. Without this exceptional sacrifice of a large amount of money, it would have been difficult to possibly not probable to complete Phase I by the end of 1998. <u>THANK YOU AEROLEASE</u>.
- The Chairman will request final invoices from Matt Creager and Wes Plattner, and request the CAA to return the balance of the funds to Aerolease.
- Aerolease agreed to make payment for aircraft N781AL previously operated by BAX Global, and the five aircraft that was not paid by the Airlines choosing not to join the JTF.
- The JTF payment shortage identified on the Payment Status report was accepted by Aerolease, to which will be considered a future credit to Aerolease in membership fees.

IV. Aircraft Conformity Inspection FAA Inspection Status

- The JTF members were provided the schedule and FAA proposal for the three aircraft inspections scheduled in January 1999.
- An Arrow Air DC-8-62 modified by STC SA1063SO was inspected at the Arrow Air facility. A Fine Air DC-8 modified by STC's SA1862SO and ST00309AT was inspected at the Fine Air facility. An Emery Worldwide Airlines DC-8-73 modified by STC SA1802SO was inspected at Timco facility.
- Mike O'Neil presented a brief review of the findings during each visit. Mr.
 O'Neil stated a copy of the reports would be provided to the JTF for distribution to the members.
- John Zappia of Fine Air requested what protocol was in place regarding
 the inspection results of these inspections. His concern was the request
 of STC data from his FAA PMI, upon completion of their inspection. Mr.
 O'Neil stated their findings were only observations and he would contact
 his PMI to address his concerns and explain the JTF review process.

Meeting Minutes - January 12, 1999 Page 4

- Mr. O'Neil stated that the airplane reviews the FAA performed on five aircraft revealed no immediately identifiable unsafe conditions with the STC installations. There were, however, discrepancies identified in the smoke detection/annunciation systems on two aircraft and noted that both operators initiated campaigns in their respective fleets to address the discrepancies.
- Mr. O'Neil requested that an EWA aircraft be made available at Atlanta on February 1, 1999 for the JTF FAA ACO team to inspect a Rosenbalm STC door aircraft (SA1802SO). The Chairman took this action item and will notify Randy Avera.
- Mr. O'Neil thanked the JTF for their support in setting up these inspections and providing support to enable the FAA to complete the airplane reviews in the time frame desired by the FAA.

V. STC Re-evaluation Status Phase I

- The Chairman provided the membership a copy of the letter sent to Mr. O'Neil on December 22, 1998 in their handout. This letter provided the STC re-evaluation status report that concluded the Phase I process.
- Mr. O'Neil asked the JTF if these reports reflected any known unsafe condition, and a collective response from the JTF was no.
- Mr. O'Neil stated that the submitted reports from the JTF probably address about 90% of the issues under review by the FAA in the reevaluation process; the remaining 10% would be addressed in their acceptance of the workscope for Phase II. He also stated that it made little sense to delay initiating Phase II until completion of FAA review of the Phase I results.
- The JTF requested of Mr. O'Neil to review and comment the submitted reports within 45 days, to promote the beginning of Phase II. Mr. O'Neil stated the JTF FAA team would provide a response in 45 days to the reports.
- Mr. O'Neil stated that the Safety Analysis for modifications may be FAA
 DER approved, and service bulletin/STC modifications will be FAA ACO
 approved as they may be the subject of FAA rule making.

Meeting Minutes - January 12, 1999 Page 5

> Marge Jones of Safety Analytical Technologies provided a statement of work and compliance checklist for the Smoke Detection re-evaluation process. A copy will be sent to Mr. O'Neil by the Chairman to complete his report.

VI. JTF Phase II Development

• The Chairman presented the status of the JTF Phase I, II & III program that was reviewed and accepted by Mike O'Neil.

Phase I

- STC Holders perform a technical review of the STC data package to determine compliance with CAR4b, utilizing the DC-8 Cargo Conversion Review Scope of Work.
- Upon completion of the STC Holder review, submit to the FAA for review and concurrence.
- After receiving FAA concurrence with the technical review, STC Holders will develop an inspection/conformity package.
- STC Holder/Operator will inspect the aircraft during Heavy Maintenance visit. FAA review/inspection may be performed in concert with this inspection.

Phase II

 Develop engineering as required to correct structural anomalies to meet CAR4b criteria by amending the original STC(s) or providing FAA 8110-3 approval per aircraft.

Phase III

- Modify aircraft to incorporate the FAA approved design changes developed in Phase II.
- Wes Plattner of National Aircraft Services, Inc. provided a status of the development of an STC program that will address re-evaluation results of the Cargo Door System Category #5 for the Rosenbalm STC SA1802SO. The STC proposes to install a vent door and improve the door locking and indication system. This STC data is under review by the Atlanta ACO and approval is expected in February 1999. An EWA aircraft DC-8-73, N105WP will have the first NAS Vent Door STC installed during heavy maintenance and will return to service in March 1999.

Meeting Minutes - January 12, 1999 Page 6

- Mike O'Neil asked Mr. Plattner and the JTF members if any other STC of this type was in work for the other effected door systems? Mr. Plattner responded in that none he knew of.
- Mr. O'Neil asked Mr. Plattner how many days it would take him to develop his STC for incorporation on the other doors? Mr. Plattner stated approximately thirty (30) days each to the remaining three (3) STC door installations.
- The Chairman concluded the NAS STC discussion by stating that a Phase II
 and III process would be complete by March 1999. All members agreed that
 this positive timely support as a group, would not have been possible without
 the JTF, FAA ACO Team support.
- The Chairman presented the history of the issuance of the Telegraphic Airworthiness Directive T98-26-51.

Emery Worldwide Airlines DC-8-73, N105WP received FOD damage to the main cargo door (STC SA1802SO) due to the results of human factors and was repaired. The total aircraft flight hours and flight cycles since installation of this STC was TAT 19,953, TC 6,209 and 9 years and 8 months.

The Chairman provided Greg DiLibero the aforementioned information prior to the issuance of the TAD. Mr. DiLibero gave the Chairman the opportunity to notify the JTF members of a possible issuance of this AD.

Thirty-two (32) effected aircraft were inspected per this TAD with no defects noted. A formal response letter on the FOD damaged caused to the EWA aircraft was provided by the Chairman to Mike O'Neil.

 Mr. O'Neil thanked the JTF for their support in this TAD inspection and stated future AD's concerning this review process would attempt to utilize the JTF to communicate with affected parties.

VII.Update of B727 NPRM

 Mike O'Neil introduced and requested that Mr. Mike Zielinski of the FAA Seattle ACO office assigned to the B-727 Cargo Conversion STC team address this issue.

DC-8 CARGO CONVERSION JOINT TASK FORCE

Meeting Minutes - January 12, 1999 Page 7

 Mr. Mike Zielinski advised that the 727 Cargo floor ADs had been issued and become effective February 16, 1999 under AD numbers 98-26-18, 98-26-19, 98-26-20 and 98-26-21.

As an example, Mr. Zielinski referenced the limitations of the AIE STC SA1368SO.

• The Chairman thanked Mr. Zielinski for the update.

VIII.Questions and Answers

 Mr. Rob Jackson, FAA LGB-AEG presented to the JTF the opportunity to include the DC-8 structural STC's into the ongoing DC-8 MSG III program per AC121-122A, as approved by the FAA working with Douglas and the DC-8 Air Carriers.

It was moved and unanimously approved that the JTF participate in this task. It was also noted several JTF air carriers are active members of this DC-8 MRB.

• The Chairman adjourned the meeting and advised the FAA, members and guests that the JTF membership would continue their meeting to discuss Phase II implementation.

IX. JTF Meeting

- The Chairman assigned Thierry Derrien of Airborne to chair this meeting.
- A motion was made to allow Wes Plattner to complete the re-evaluation for the Cargo Door System Category #5 on the other STC cargo doors, like was performed on the Rosenbalm with a Phase II solution.

A vote was taken which received unanimous approval by the JTF.

- A discussion was held regarding the Class E Cargo Compartment Smoke Detection Category #4. The JTF proposed the following solutions for further review;
 - 1. Look at the existing systems and modify as an option.
 - 2. Review new wireless installation as an option.

DC-8 CARGO CONVERSION JOINT TASK FORCE

Meeting Minutes - January 12, 1999 Page 8

3. Review the ATI system that was recently tested to the 1 minute rule as an option.

The JTF assigned Airborne the task of researching these options and report back to the JTF.

A discussion was held regarding the structural categories of 1,2 & 3. Matt
Creager of Structural Integrity Engineering proposed to the JTF to provide a
draft proposal that will define the Phase II requirements. He stated he would
fax this to the Bidders committee on or before 1-30-99.

A vote was taken which received unanimous approval by the JTF.

Thierry Derrien informed the JTF that he would present the minutes to the Chairman by telephone in the next few days.

A decision was made to establish the next meeting after Mr. Derrien spoke with the Chairman. A JTF Bidders meeting was proposed to be held at Airborne on 1-27-99.

The meeting was adjourned.

DC-8 CARGO CONVERSION JOINT TASK FORCE Meeting Minutes - January 12, 1999 Page 9

ATTENDEES

NAME	COMPANY	TELEPHONE	FAX
Mike O'Neil	FAA-LAACO		
Greg DiLibero	FAA-LAACO		
Thierry Derrien	Airborne Express		
George Mabuni	FAA-LAACO		
Albert Lam	FAA-LAACO		
Rob Jackson	FAA-LGB-AEG		
Carl Fountain	FAA		
Rany Azzi	FAA		
Mike Zielinski	FAA		
Tim Corley	Aerolease		
Wes Plattner	NASI		
Paul Hedding	NASI		
Richard Saltivan	AIA		
Bill Cotney	Cotney/Fine		
Dan Johnson	A.T.I.		
Charles Perry	AEI		
John Zappia	Fine Air		
Mary Arabi	Airborne Express		
Matt Creager	SIE		
Cawas Commissariat	Airborne Express		
Thomas Wood	EWA		
Nelson Gonzalez	Arrow Air		
Randy Avera	FAA-Atlanta		

EUR'S RESPONSE TO LETTER DATED. December 21, 1998





Thomas M. Wood

DIRECTOR QUALITY CONTROL

303 CORPORATE CENTER DR. VANDALIA, OH 45377 FAX: (937) 898-2803 PHONE: (937) 454-3940

FACSIMILE TRANSMISSION COVER SHEET

DATE: 1 15199	SEND TO FAX #:_
DELIVER IMMEDIATELY TO:	
NAME: JUE ABRAMSK	TELEPHONE #:
COMPANY / DEPARTMENT:	FAA SIC
This is page 1 of 1 pages sent in transi	mission regarding the following principal subject(s):
your status of the for	you all week by phone. I meed of lowing, I would like the oppurmite of phone, instead of letters, please. 1-7-99 perous agreement, to agree or outmitt to you.
2. Maint manual	MPP neuroion sent 1-7-99
3. Working with 7 4 'C' Check being sent FACSIMILE MESSAGE FROM:	Boling on training, will advise. Today Have a great weekend.
	NAME: Numas M. Wood

EUA'S RESPONSE TO LETTER DATED December 17, 1998



February 9, 1999

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Mr. Joe Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your December 18, 1998 letter requesting information (See Attachment). We have discussed many of these items over the phone in the past two months in preparing for the DC-10 certification. The following responses address your items by number per your letter.

- 1) EWA has provided you previously this information that is contained in the Continental Maintenance Manuals. Bruce Robbins has also provided Nick Pearson the Avionics Modification Listing. The Douglas Freighter Conversion Supplement will contain all of the specific differences not contained in the reference manuals.
- 2) Same as above.
- 3) See Attachment.
- 4) None at this time.
- 5) See Attachment.
- 6) EWA will utilize Timco, Greensboro, NC and Commodore Aviation, Miami, Florida.
- 7) Same as above.
- 8) EWA has reviewed the aircraft tooling, test and support equipment per the aircraft maintenance manual. In support of the line maintenance operations and other in-depth system trouble shooting, EWA is purchasing additional equipment as listed. Please contact Bruce Robbins if you have any questions.

Please call me upon receipt and your review if you have any questions. Thank you for your support.

Sincerely,

Thomas M. Wood
Director Quality Control

TMW/re

Attachments

cc: Rene' Visscher
Bruce Robbins

3C3 CORPORATE CENTER DRIVE, VANDALIA, OH 45377



U.S. Department of Transportation Federal Aviation Administration San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7581 FAX: (408) 279-5448

December 18, 1998

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

In conjunction with the Emery Worldwide Airlines (EWA) DC-10 certification, please inform this office in writing of the following information:

- 1) A listing of all applicable major aircraft components by part number and serial number.
- 2) A listing of all applicable aircraft sub-components by part number and serial number.
- 3) A copy of applicable aircraft lease agreements.
- 4) A copy of applicable parts pooling and borrowing agreements.
- 5) A copy of applicable aircraft major component lease and/or interchange agreements.
- 6) A copy of the applicable aircraft substantial maintenance provider agreements.
- 7) A copy of any other applicable aircraft maintenance provider/vendor agreements.
- 8) A listing of all applicable aircraft tooling, test, and support equipment.

Please forward the requested information as soon as possible.

Sincerely,

Íoseph A. Abramski

Principal Maintenance Inspector

cc: Rene Visscher - EWA

AIRCRAFT LEASE AGREEMENT N68041

dated as of

May 18, 1998

between

FIRST SECURITY BANK, NATIONAL ASSOCIATION, not in its individual capacity, except as expressly set forth herein, but solely as Owner Trustee or its assigns,

as Lessor,

and

EMERY WORLDWIDE AIRLINES, INC.,

as Lessee

ONE MCDONNELL DOUGLAS DC-10-10 AIRCRAFT
U.S. REGISTRATION N68041
MANUFACTURER'S SERIAL NUMBER 46900
as equipped with three (3) General Electric CF6-6D engines

COUNTERPART NO. ___ OF 6 SERIALLY NUMBERED, MANUALLY EXECUTED COUNTERPARTS. TO THE EXTENT THAT THIS AIRCRAFT LEASE AGREEMENT CONSTITUTES CHATTEL PAPER UNDER THE UNIFORM COMMERCIAL CODE IN THE U.S. OR ANY CORRESPONDING LAW IN ANY FOREIGN JURISDICTION, NO SECURITY INTEREST IN THIS AIRCRAFT LEASE AGREEMENT MAY BE CREATED THROUGH THE TRANSFER OR POSSESSION OF ANY COUNTERPART HERETO OTHER THAN COUNTERPART NO. 1.

INDEX

	11027	
SECTION	HEADING	<u>PAGE</u>
1. 2. 3. 4. 5. 6. 7. 8. 9. 10. 11. 12. 13. 14. 15. 16. 17. 18. 19. 20.	Preamble Definitions Lease and Conditions Delivery and Acceptance; Effective Date; Term Rent Representations and Warranties Possession and Use Information and Inspection Covenants of Lessee Replacement of Parts; Alterations, Modifications and Additions General Tax Indemnity Casualty Occurrences Insurance Indemnification Liens Perfection of Title and Further Assurances Return of Aircraft and Records Events of Default Remedies Alienation Miscellaneous	1 1 9 11 16 18 21 28 29 31 33 37 39 42 43 44 45 48 50 53 54
EXHIBIT A EXHIBIT C EXHIBIT C EXHIBIT C EXHIBIT C EXHIBIT F EXHIBIT F EXHIBIT H EXHIBIT I EXHIBIT I EXHIBIT I EXHIBIT I EXHIBIT I EXHIBIT M	Aircraft Specifications List of Manuals, Documents and Records Definitions and Values Lease Supplement No. 1 Delivery Conditions Return Conditions Maintenance Recording Requirements Part 1 — Calculation of Complete Aircraft Multiplier Part 2 — Mandatory Extension of Possession Term Form of Lessee's Guaranty Form of Lessor's Guaranty Form of Legal Opinion from Lessor Guarantor's In-House Counsel Form of Legal Opinion from Lessee's In-House Counsel Schedule of Permitted Foreign Sublessees	I II IV VIII XI XXIII XXIII XXV XXIX XXIX XXXIII XXXVIII LXII

ROTABLES LEASE AGREEMENT

between

AVIATION SALES LEASING COMPANY

and

EMERY WORLDWIDE

Contract Number:

COUNTERPART NO. OF SEQUENTIALLY NUMBERED, MANUALLY EXECUTED COUNTERPARTS. TO THE EXTENT, IF ANY, THAT THIS LEASE CONSTITUTES CHATTEL PAPER UNDER THE UNIFORM COMMERCIAL CODE, NO SECURITY INTEREST IN THIS LEASE MAY BE CREATED THROUGH THE TRANSFER AND POSSESSION OF ANY COUNTERPART OTHER THAN COUNTERPART NO. 1.

ROTABLES LEASE AGREEMENT

between

AVIATION SALES LEASING COMPANY

and

EMERY WORLDWIDE

,	•	Contract Number:
office	s at 69	es Lease Agreement ("Lease") is made and entered into as of this day,, by and between Aviation Sales Leasing Company ("Lessor") with 05 N.W. 25 th Street, Miami, FL 33122-1898 and Emery Worldwide with offices at 303 Corporate Center Drive, Vandalia, Ohio 45377.
In cor Lesse	nsiderat e hereb	tion of the premises and mutual promises herein contained, Lessor and y agree as follows:
1.	TERM	M OF THIS LEASE
	a. :	Subject to the terms and conditions of this Lease, Lessor agrees to lease to Lessee and Lessee agrees to lease from Lessor the components and rotable parts described in Exhibit A and such other components and rotable parts as may become subject to this Lease pursuant to the terms hereof (the "Rotables"), and all records relating to the Rotables in the possession of Lessor which are requested by Lessee and required by the Federal Aviation Administration ("FAA") for Lessee's operation, and all records generated by Lessee during the Lease Term (as defined in Section 1.b) relating to the Rotables (the Rotables, such Rotable records and related items are collectively called the "Equipment").
	b.	The term of this Lease ("Lease Term") will be for the period specified in Exhibit A, including such additional time as may be required for return of the Equipment in accordance with the terms and provisions of Section 16.
	c.	Lessee will forthwith redeliver the Equipment, at the Return Location specified in Exhibit A, to Lessor upon the expiration or earlier termination of this Lease in accordance with the terms and conditions of Section 16.

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reference.

Conditions of the terms of this Lease are set forth in the Addendum,

attached hereto as Exhibit A, which is incorporated herein by this

2. <u>DELIVERY, INSPECTION, ACCEPTANCE AND INSTALLATION</u>

- a. Lessor will deliver the Equipment to Lessee on the Delivery Date specified in Exhibit A at the Delivery Location specified in Exhibit A (the "Delivery"). At time of Delivery, the Equipment will have valid FAA approved return to service and maintenance release tags affixed to them.
- b. Lessee will deliver to Lessor a receipt for the Equipment substantially in the form of Exhibit B, duly executed by an authorized representative of Lessee upon Delivery of the Equipment.
- c. Delivery by Lessor is subject to the following conditions precedent:
 - i. Receipt by Lessor of the Initial Payment described in Section 3.b.
 - ii. Receipt by Lessor of the insurance certificates described in Section 12.
 - iii. Evidence satisfactory to Lessor that this Lease or documents or other instruments have been duly filed in such jurisdictions as shall be specified by Lessor.
- d. Lessee may, at its expense, conduct an inspection of the Rotables within the inspection period stated in Exhibit A to determine whether the Rotables are acceptable to Lessee. Lessee will notify Lessor in writing of any discrepancy, within the inspection period, at which time Lessor may substitute other Rotables or remove unacceptable Rotables from Exhibit A.

CHARGES AND PAYMENT

- a. Lessee will pay to Lessor:
 - i. The Monthly Rent specified in Exhibit A in advance for each month or fraction thereof during the Lease Term, commencing with the Delivery Date specified in Exhibit A and on the first day of each month thereafter continuing until the return of the Equipment in accordance with the terms and conditions of Section 16.
- b. Prior to Delivery of the Equipment, Lessee will pay Lessor an initial payment ("Initial Payment") consisting of (i) an amount equal to Monthly Rent for one (1) month in advance applied against the first month's Monthly Rent and (ii) a security deposit in an amount as defined on

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Exhibit A (the "Security Deposit"). The Security Deposit shall be held by Lessor as security for the timely and faithful performance by Lessee of all of Lessee's obligations under this Lease, and Lessee hereby grants Lessor a security interest therein. If Lessee fails to pay Monthly Rent hereunder or to pay any other sums due or to perform any of the other terms and provisions of this Lease or is otherwise in Default (as defined in Section 17) hereunder, in addition to all other rights Lessor shall have under the Uniform Commercial Code as a secured party, Lessor may use, apply or retain all or any portion of the Security Deposit in partial payment for sums due to Lessor by Lessee, to compensate Lessor for any sums it may in its discretion advance as a result of a Default by Lessee, or to apply toward losses or expenses Lessor may suffer or incur as a result of Lessee's Default hereunder. If Lessor uses or applies all or any portion of such Security Deposit, such application shall not be deemed a cure of any Defaults, and Lessee shall within five (5) days after written demand therefor deposit with Lessor in cash an amount sufficient to restore the Security Deposit to amount equal to Monthly Rent for two (2) months. Provided Lessee is not in Default under this Lease, upon the expiration of the Lease Term and the return of the Equipment in accordance with Section 16, any undisbursed portion of the Security Deposit shall be returned to Lessee. Lessor may commingle with its other funds all funds paid by Lessee to Lessor under this Section 3.b and shall not be required to segregate such funds. Lessee (i) shall not be entitled to receive any interest on the Security Deposit, and (ii) shall not assign, hypothecate or otherwise transfer its residual interest, if any, in such funds.

- c. Lessee will pay Lessor in advance on the first day of each month during the period of the Lease Term the Monthly Rent for such month without demand or notice from Lessor.
- d. All payments hereunder shall be made by wire transfer of immediately available funds, in U.S. Dollars, to the following account of Lessor:

Aviation Sales Leasing Company Aviation Sales Company Citibank New York 399 Park Avenue New York, NY 10043 ABA # 021000089 Account # 40659379

e. Any payment due on a day, which is not a Business Day, shall be due on the next Business Day. "Business Day" shall mean any day other than a Saturday, Sunday or other day on which banking institutions in New York are authorized or required by law to be closed.



Brian McCarthy . Director of Marketing

Triad International
Maintenance Corporation
623 Radar Road
Greensboro, NC 27410
Phone (336) 668-4410
Fax (336) 665-9011

July 9, 1998

Mr. Richard Hickey Emery Worldwide 303 Corporate Center Drive Vandalia, Ohio 45377

Dear Richard:

This letter, when accepted in the manner setforth below, will constitute an agreement between TIMCO and Emery Worldwide for the technical support of DC-10 aircraft operated by Emery Worldwide Airlines.

TIMCO agrees to perform heavy maintenance, repairs, and modification (as required) on Emery Worldwide's DC-10 aircraft. The work will be accomplished in accordance with the rates, charges, terms, and conditions contained in the 1998 Pricing Proposal dated December 26, 1997 (Rev 1). In September 1998, we will submit a new 1999 Pricing Agreement which will reflect both DC-8 and DC-10 understandings.

Emery Worldwide agrees to pay TIMCO for all DC-10 services performed in accordance with the following payment terms.

Payment Terms

"C" Check DC-10

- \$75,000 on input
- 100% of total estimated invoice (less input payment) payable prior to departure
- Remaining balance due net 30

"D" Check DC-10

- \$250,000 on input
- \$450,000 payable 25 days after input date
- 100% of total estimated invoice (less input payment) payable prior to departure
- Remaining balance due net 30

Drop-in and Field Trip Work

Total invoice due net 30

For DC-8 and DC-10 field trip repairs authorized and accomplished away from TIMCO's facility, Emery Worldwide agrees to pay the overtime rate for labor expended by the appropriate skill.

All other pricing, rates, charges, terms, conditions and understandings contained in the 1998 Pricing Proposal will remain in full force and effect.

If this meets with your approval, please execute this letter in the space provided and return one original to me.

Brian McCarthy
Accepted and Agreed this of July, 1998.
Emery Worldwide TIMCO



P.O. BOX 681078 • MIAMI, FLORIDA 33266-1078 4900 N.W. 36TH STREET • MIAMI, FLORIDA 33122 E-Mail: comaviation @luno.com

MARKETING/SALES (305) 869-1900 EXT 202 / 256 FAX: (305) 869-1952 • SITA: MIAICTX

AIRCRAFT SERVICES AGREEMENT

This Aircraft Services Agreement is made and entered this 1st day of February 1999, by and between EMERY WORLDWIDE AIRLINES (hereinafter "EMERY") duly organized under the laws of the State of Ohio, with offices in Vandalia, Ohio and COMMODORE AVIATION, INC. a corporation duly organized under the laws of the State of Delaware and having its principal place of business at Miami International Airport (hereinafter "COMMODORE").

WHEREAS, EMERY desires that COMMODORE provide maintenance, repair and/or overhaul and related services with respect to EMERY's DC-10-10Fseries aircraft and engines (collectively, the "Aircraft"), and COMMODORE agrees to perform said services in accordance with the terms and conditions of this Agreement.

NOW, THEREFORE, the parties agree as follows:

ARTICLE 1 SCOPE OF SERVICES

- 1.1. EMERY hereby agrees to purchase, and COMMODORE hereby agrees to perform, at COMMODORE's, premises, Miami International Airport (hereinafter the "Site"), the heavy maintenance services set forth in Exhibit "A" as per the specific Registration Number attached hereto and Exhibit B for "On-Call" Flight Line Maintenance Services/Field Team Maintenance Services (hereinafter the "Services"), all in accordance with the terms and conditions of this Agreement.
- 1.2 COMMODORE shall accept and perform the Services on Aircraft requested by EMERY in accordance with the schedule set forth in Schedule 1 or, as amended, for Aircraft not specifically identified therein, on any other Aircraft, provided EMERY shall have given COMMODORE at least ten (10) days notice of the date such Aircraft will be delivered to COMMODORE

ARTICLE 2 <u>DELIVERY</u>, RE-DELIVERY

- 2.1. EMERY shall deliver the Aircraft to COMMODORE's facility in Miami, Florida per the respective "Exhibit A" of each Aircraft.
- 2.2. AIRCRAFT DOWNTIME: Based on the workscope defined above, the downtime of the Aircraft is stated in the respective "Exhibit A" of each Aircraft.

- 2.3 EMERY will, seven (7) days prior to delivery of the Aircraft, provide COMMODORE with one (1) full work package for each Aircraft to be delivered.
- 2.4 Re-delivery schedule may increase or decrease in time depending on:
 - 2.4.1 Addition or reduction in work scope.
 - 2.4.2 Conditions under Article 7.

ARTICLE 3 TEST FLIGHTS:

- Upon completion of all work contemplated under the Agreement, EMERY may conduct one or more test flights and COMMODORE shall correct those items identified by such Test Flights which EMERY requires/authorizes correction prior to redelivery of the Aircraft. Should any item that EMERY requires/authorizes COMMODORE to be corrected/repaired not being a defect arising from services or additional services performed by COMMODORE, EMERY shall be charged and invoiced for said items. Said items to be mutually agreed to.
- 3.2 All test flights shall be conducted by EMERY crews unless otherwise agreed, and any liability arising therefrom shall be at EMERY risk. EMERY shall bear the cost and expense of test flights.

ARTICLE 4 ENTIRE AGREEMENT, JURISDICTION

This Agreement together with the Exhibits or other documents referred to herein or attached hereto shall constitute the entire agreement between the parties hereto and supersedes all previous communications, representations or agreements, either oral or written, heretofore made between the parties regarding the subject matter hereof. This Agreement shall not be varied other than in writing and signed by the duly authorized representatives of both parties. Otherwise, any provision appearing in any document with respect to the obligations or terms and conditions contemplated herein shall, in so far as said provisions change, add to, conflict with and/or differ from the terms and conditions herein contained be deemed inapplicable and this Agreement shall control.

In the event that the parties hereto shall, in addition to this Agreement execute an agreement with respect to any non-routine items to be performed hereunder or any additional services not contemplated herein, the parties shall be bound by the terms and conditions appearing in such subsequent agreement.

This Agreement and any agreement referred to in this Article herein (unless specifically stipulated to the contrary in such agreement) shall be construed and enforced in accordance with the laws of the State of Florida, both substantive and remedial.

In the event that any proceeding, suit or action is brought by or against either party and in any appeal(s) thereof, the prevailing party shall be entitled to recover costs and expenses including reasonable attorneys fees.

ARTICLE 5 MISCELLANEOUS

5.1. All notices and other communications sent by the parties shall be by first class mail or by telex or telefax or come to the following addresses (any notices sent by first class mail will be established based on the postmark date):

TO COMMODORE:

COMMODORE AVIATION, INC.

4900 N.W. 36th Street Miami, Florida 33122 P.O. Box 661078 (Mailing) Miami, FL 33266-1078

Attention: R.E. Weltmann, Director

Marketing/Contracts

Fax No:

(305)869-1952

Telephone No:

(305)869-1900 ext 256/232

TO EMERY

EMERY WORLDWIDE AIRLINES

303 Corporate Center Drive Vandalia, Ohio 45377

Attention: R. Hickey, Director

Heavy Maintenance

Telephone No.:

(937)264-2721

Fax No.:

(937)890-6346

5.2. Assignment

This Agreement is personal to EMERY and to COMMODORE and shall not be assigned or transferred, in whole or in part, by either party without the other party's prior written consent.

5.3. Non-Waiver

The failure of a party to enforce at any time any of the provisions of this Agreement, or to require at any time the performance by the other party of any of the provisions hereof, shall in no way be construed to be a waiver of such provision, or in any way affect the validity of this Agreement or any part thereof, or the right of the first party to enforce each and every such provision at some later date.

5.4. Captions

The caption headings of the Articles appearing in this Agreement are for convenience of reference only and shall not be construed as in any way limiting or extending the language of the provisions to which the captions refer.



P.O. BOX 651078 • MIAMI, FLORIDA 33266-1078 4900 N.W. 36TH STREET . MIAMI, FLORIDA 33122 E-Mail: comaviation@junc.com

MARKETING/SALES (305) 869-1900 EXT 232 / 256 FAX: (305) 869-1952 • SITA: MIAICTX

EXHIBIT "B"

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ON-CALL FLIGHT LINE MAINTENANCE AND FIELD TEAM SERVICE AGREEMENT BETWEEN COMMODORE AVIATION, INC. AND EMERY WORLDWIDE AIRLINES DATE:

On behalf of COMMODORE AVIATION, INC. (hereinafter "COMMODORE") we are pleased to submit the following ON-CALL FLIGHT LINE MAINTENANCE AGREEMENT ("AGREEMENT") to provide Aircraft Maintenance Services to EMERY WORLDWIDE AIRLINES, INC. (hereinafter "CUSTOMER").

This Agreement, between COMMODORE, located at the Miami International Airport, Miami, Florida, and CUSTOMER with principal offices in Vandalia, Ohio, will be effective upon the execution hereof by authorized representatives of the parties.

This Agreement entails that COMMODORE will provide the following ON CALL flight line maintenance services for CUSTOMER's DC-10 series aircraft (hereinafter, "Aircraft") at the Miami International Airport, as requested/authorized by CUSTOMER on the following terms and

1. SCOPE OF WORK:

- COMMODORE will perform aircrast maintenance work on CUSTOMER's Aircraft as requested and authorized by CUSTOMER's Technical Representative (hereinafter "Tech. Rep.") at the Miami International Airport ("Site") or "Off-Base" at location other than Miami International Airport.
- COMMODORE shall perform ON-CALL maintenance services (hereinafter, "On-1.2 Call Flight Line / Field Team Maintenance Services") as authorized by the CUSTOMER'S Tech. Rep. including the following types of work:
 - 1.2.1 Correction and repair of log book reported defects
 - 1.2.2 Customer request items
 - 1.2.3 Discrepancy item correction/repair
 - 1.2.4 Engine Change's and Component's changes
 - 1.2.5 Trouble shooting on components, Avionics
 - 1.2.6 Battery Charging Servicing ("Site only")
 - 1.2.7 Non-Destructive Testing Work (N.D.T.)

grinese are to be charge a against the original M-14 for

IEM.	DC 10 A/C) MPN	CPN	моим	QTY	UI	IIT PRICE	TOTAL PRICE	PO#	DUE DATE	STATUS	CODE2
1	NAS133A2624D	1160013	NCSE WHEEL STEERING	6	\$	48.00	288.00	1160013001	11/27/98	closed 1ea 11-25	С
2	1758***	D918050	ADAPTER, TIRE SERVICING	1	\$	1.92	1.92	D918050001	2/4/99	closed 1-11	С
3	7350-0540-0007	D918060	Y ADAPTER	1	\$	1,160.00	1,160.00	D91806001	1/19/99	closed 1-18	С
4	2471F	T300193	MEGO METER	1	\$	1,350 00 3	1,350.00	T300193001	12/12/98	closed 1ea rec 12-20	C
5	TEE62-2		VSV DYNAMIC TRACKING	1	\$	58,750 00	58,750.00	AWAIT BRUCE	ADDED PER 1-12-98MTG	Await Demo Test on A/C per Bruce Robbins	W
6	BH112JD-901	1300188	TEST SET, ENGINE	1	\$	47,139.00		ROBBINS T300188001	CANCELLED per 1-12-99mtg	2/8 PER 1-12-99 MTG	сх
7	33410LH-90-4	D918051	ADAPTER, STATIC PORT	1	\$	265.00	265.00	D918051001	11/30/98	closed 1ea 11-20	С
8	33410LH-90-4	[)918051	ADAPTER, STATIC PORT	1	\$	265.00	265.00	D918051002	12/8/98	closed 1ea 12-4	С
9	33410LH-125-4	T300175	ADAPTER, STATIC PORT	1	\$	283.00	283.00	T300175002	12/15/98	closed 1ea rec 11-30, 12-24	С
10	MILH-5593-3-3600	D918052	HOSE, PITOT, 30 FT	1	\$	250.00	250.00	D918052001	11/30/98	clcsed 11-20	С
11	MILH-5593-3-3600	D918052	HOSE, PITOT, 30 FT	1	\$	250.00	250.00	D918052002	12/8/98	clcsed 12-28	С
12	MILH -5593-4-7200	D918052	HOSE, STATIC, 60 FT	1	\$	494.00	494.00	D918053001	11/30/98	closed 11-20	С
13	MILH-5593-4-7200	D918053	HOSE, STATIC, 60 FT	1	s	494.00	494.00	D918053002	12/8/98	closed 12-28	С
14	P52287M3	D918054	ADAPTER, PITOT TUBE	1	\$	545.00	545.00	D918054001	11/30/98	closed 11-20	С
15	_{,н} Р52287М3	D918054	ADAPTER, PITOT TUBE	1	\$	545.00	545.00	D918054002	12/8/98	closed 12-28	С
16	22D10-1093	T300194	LVDT TESTER	1	\$	3,680.00	3,680.00	T300194001	2/12/99	will ship on or before 2/12/99	cs
17	CTS-700	T300193	SELCALIATSCALL RAMP TEST SET	1	\$	3,395.00 \$	3,395.00	T300183001	1/19/99	closed 1-15	С
18	G02D9	L)918055	KIT, ELECTRICAL CONNECTOR TOOLS	1	\$	7,995 00	· .	D918055001	3/25/99	cancelled per 1-12-98,mtg	СХ
19	102078-1	1160014	DC10 TOW BAR	1	\$	4,500.00 \$	4,500.00	1160014001	2/11/99	closed 1-11	С
20	H394-115	1300137	TESTER, THERMOCOUPLE	1	\$	7,346 00	\$ -	T300187001	5/4/1999 - HOLD ON ORDER	Cancel per Bruce Robbins CX'D 2/8/99	СХ
21	RRX-367	1000937	REMOVAL TOOL, SWITCH SOCKET	1	\$	95.00	95.00	T000987001	2/3/99	closed 1-15	С
22	RRX-367	T000937	REMOVAL TOOL, SWITCH SOCKET	10	\$	95.00 \$	950.00	T000987002	1/25/99	Closed 1/20/99	С
23	STC32225-1	1000933	SPECIAL TOOL	1	\$	638.00 \$	638.00	T000983001	3/17/99	closed 11-19	С
24	STC90572-1	D918046	HOIST ADAPTER MLG/CLG	1	\$	184.80 \$	184.80	D918046001	3/5/99	closed 1/22	С
25	STC90650-1	1000984	EXTRACTOR WRENCH	1	\$	168.40 \$	168.40	T000984001	11/18/98	closed 11-19	С
26	STC90650-1	1000984	EXTRACTOR WRENCH	2	\$	168.40 \$	336.80	t000984002	4/17/99	PER 1-12-99 MTG	cs
27	833834-1	T300185	TEST SET	1	\$	42,900.00 \$	42,900.00	T300185001	2/21/99		cs
28	DLTFRPSKT3009	1160002	PERMASWAGE TUBE REPAIR KIT	1	\$	49,814.00 \$	49,814.00	1160002001	2/3/99	closed 12-3	С
29	A1540AA0R2	C901736	HYDRAULIC SERVICE LADDER	1	\$	515.00 \$	515.00	C901786001	12/31/98	closed 12-22	С
30	AML-08	C901784	8' SOLID LADDERS	1	\$	476.00 \$	476.00	C901784001	12/31/98	closed 12-28	С
31	F900H-06	T000989	€' FOLDING LADDER	1	\$	258.00 \$	258.00	T000989001	12/31/98	closed 11-30-98	С
32	JL-7411-D-R2	C901735	WHEEL WELL LADDER	1	\$	1,184.00 \$	1,184.00	C901785001	12/31/98	closed 12-14-98	С
33	1N588-7	D918114	BATTERY 120V BLACK&DECKER	1	\$	50.25 \$	50.25	D918114002	12/12/98	closed 11-16	С
34	4X673-9	D918113	CHUCK STRAIGHT TIRE SERVICING	1	\$	1.75 \$	1.75	D918113001	11/19/98	closed 11-16	С
35	4Z122	D918093	SOLVENT SPRAYER	1	\$	78.45 \$	78.45	D918093001	11/19/98	closed 11-16	С
36	7W086	T000990	WORK BENCH	1	\$	154.28 \$	154.28	T000990001	11/19/98	closed 11-17	С
37	; 175S-2	T000986	DOLLY, WHL & BRK CHANGE	1	\$	4,467.00 \$	4,467.00	T000986001	1/15/98	closed	С
38	1755-2	T000986	DOLLY, WHL & BRK CHANGE	6	\$	4,467.00 \$	26,802.00	T000986002	3/16/99	PER 1-12-99 MTG	cs
39	60P10	T000173	WHEEL JACK	1	\$	6,880 00 \$	6,880.00	T000173003	12/26/98	closed 12-31	С
40	BC-400	T000988	TIRE DOLLY	1	\$	428.00 \$		T000988001	12/11/98	closed 1-7-99	C

HEM	MP11	CPN	NOUN	QTY	U	NIT PRICE		TOTAL PRICE	PO#	DUE DATE	STATUS	CODE2
41	PF53467-2PWS	D918083	SKYDROL SERV	1	\$	2,461.00	\$	2,461.00	D918083001	12/11/98	closed 12-28	Ċ
42 -	PF53467-2PWS	D918083	SKYDROL SERV	6	\$	2,461 00	\$	14,766.00	D918083002	2/25/99	PER 1-12-99 MTG	cs
43	PF55451-2	D918084	OIL SERVICE	1	\$	2,620.00	\$	2,620.00	D918084001	12/11/98	closed 12-21	С
44	PF55451-2	D918084	OIL SERVICE	6	\$	2,620.00	\$	15,720.00	D918084002	3/12/99	PÉR 1-12-99 MTG	cs
45	PF55543	1160026	DISPENSER, FAILREVERSER	1	\$	2,301.00	\$	2,301.00	1160026001	3/1/99	closed	С
46	PF55688-3	1160027	ACTUATOR MTC ACTUATOR, STATOR VANE	1	\$	6,713.00	\$	6,713.00	1160027001	3/16/99		cs
47	TC250/230/DE9P	1160017	TRICKLE CHARGE RIS	1	\$	175 00		-,	1160017001	3/16/99	CANCEL PER BRUCE/TIM	CX
48	TC250A	1160015	TRICKLE CHARGER FOR EMERG BAT.	1	\$	130.00		130.00	1160015001	11/18/98	closed 11-13-98	C
49	TG250A .M83723	FC00995	TRICKLE CHARGER MD11 EMERG BAT	1	\$	300.00	\$	300.00	T000995001	12/11/98	closed 1-7-99	С
50	CONNECTOR 912948-1-1	D918085	ADAPTER B	1	\$	996.00	ς.	996 00	D918085001			
5‡	AV1027	F300186							D916063001	1/22/98	Waiting on Allied to call back with shipping info 2/2	CS
52	LMC2570BDC	DS18086	TESTER, PNEUMATIC SYSTEM	1	S	22 500 00		22,500 00		2 WEEKS ARO	WAITING APPROVAL FROM R. VISSCHER	w.,
53	LMC2570BDC		COVER, ENGINE INLET	1	\$	489.00		489.00	D918086001	11/20/98	closed 11-19-98	С
54	AM-1005	D918086 1160028	COVER, ENGINE INLET	14	\$	489.00		6,846.00	D918086002	3/5/99	PER 1-12-99 MTG	cs
55	AM-1460		SLING, QEC HANDLING	1	\$	7,920.00		7,920.00	1160028001	1/11/99	closed 1-6-99	С
56	. AM-2552	1160029	SUPPORT, ENGINE COWL DRS	1	\$	25,935.00		25,935.00	1160029001	2/8/99		CS
57	9769	U918087	ENGINE CHANGE KIT	1	\$	75,695.00		75,695.00	D918087001	2/22/99		cs
58	P-159	D918127	HOSE	1	\$	77.00		77.00	D918127001	11/27/98	closed 11-30-98	С
59		1150030	FAN COWL PUMP	1	\$	362.00	\$	362.00	1160030001	11/27/98	closed 12-1-98	С
60	MSE42 MSE43	T030998	SPANNER WRENCH	1	\$	1,498 00			T000998001	2/11/99	cancelled per b. robbins	CX
61		T000998	SPANNER WRENCH	1	\$	1,256.00		1,256.00	T000998002	3/30/99	closed 1/26	С
	MSE-51A **	T000999	TORQUE MULTIPLIER	1	\$	1,659.00		1,659.00	T000999001	4/28/99		cs
62	T&E 20-00-07-5	1)918088	RING PIN KIT	1			S	-		TOOL&EQUIP "MANUAL MAY BE MFR'D BY A/L OR	order manual's - ANGELA BRUNNER	w
63	T&E 22:00 01	D918119	BREAKOUT BOX & CABLE ASSY	1			\$	-		DESIGNEE "	order manual's - ANGELA BRUNNER	w
- 64	T&E 22-10-02	D918120	BUSS SPLITTER SUMULATOR	1			\$	-		11	order manual's - ANGELA BRUNNER	w
95	T&E 22 10 03	D918089	TEST ADAPTER	1			\$	-		44	order manual's - ANGELA BRUNNER	w
56	T&E 22-13-01	D918121	BREAKOUT BOX, SERVO-ACTUATOR	1			\$	-		4	order manual's - ANGELA BRUNNER	w
67	T&E 22-22-01	D918122	BREAKOUT BOX, AUTO PITCH	1			\$	-			order manual's - ANGELA BRUNNER	w
68	T&E 22-31-01	T001000	WRENCH, AUTO-THROTTLE ADJ	1			S			· ·	order manual's - ANGELA BRUNNER	w
69	TEE 24A	1300184	TESTER, APU	1	S	21,450 (X)	\$	21,450 00	AWAIT BRUCE	RMS	Waiting Decision from R V	w
70	191-191H	1160001	TOWBAR, HYD HEIGHT CONTROL	2	\$	4,308.00	s	8,616.00	ROBBINS check with Rob/Quote			w••
71	214030-1	1160001	TOWBAR, HYD HEIGHT CONTROL	1	5	5,000 00					Duplicate? See Item 70 PER 1-12-99 MTG	СХ
72	DZZ7465-1	1160023	LINE TESTER	1	\$	13,790.00	\$	13,790.00	1160023001	5/9/99	,	cs
73	Y2004MG	D918059	BRAKE, BLEED VALVE FITTING	1	3	32.50	\$	32.50	D918059002	11/21/98	closed 11-19	С
74	Y2004MG	D918059	BRAKE, BLEED VALVE FITTING	20	\$	32 50	\$	650.00	D918059003	3/5/99	closed 1/29	c
75	600A	1300192	ATC TEST BOX	1	\$	7,645.00	\$	7,645.00	T300192001	11/18/98	closed 11-16	С
76	NAY402AP	7300191	NAV 402 AP	1	\$	11,795.00	s	11,795 00	T300191001	11/18/98	closed 11-16	С
77	01-0235-06	T300190	NAV1 TESTER	1	\$	3,025.00	\$	3,025.00	T300190002	2/10/99	3,3332 7.1.75	cs
78	6701A41	D918092	CHUCK DUAL HEAD	1	8	6.07		6.07	D918092001	11/18/98	closed 11-12	C
79	CJ-93B	T000539	PULLER SET	1	\$	187.25		187.25	T000539002	11/18/98	closed 11-20	С
80	LDH302	D9 8056	SOCKET, ENGINE MOUNT BOLTS	1	\$	24.10		24.10	D918056001	11/18/98	closed 11-25	C
81	SES-301	D9:8057	SOCKET, ENGINE MOUNT BOLTS	1	\$	24.20		24.20	D918057001	11/18/98	closed 11-25	C
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82	72-500-05	D918058	ADAPTER CABLE	1	\$	350.00	\$ 350.0	0 D918058001	11/18/98	closed	С
83 •	72-500B	1300182	TESTER, ANTIOSKID VALVE	1	\$	11,000.00	\$ 11,000.0	0 T300182001	11/18/98	closed 11-17	С
84	V-800	7000985	TOOL, SUMP DRAIN	1	\$	495.00	\$ 495 0	0 T000985001	11/21/98	closed 11-19	С
85	V-800	7000985	TOOL, SUMP DRAIN	5	\$	495.00	\$ 2,475.0	0 T000985002	2/4/99	closed 2/4/99	С
86	2436	D918090	COUPLING FEMALE PIPE TREAD	1	\$	3.89	\$ 3.8	9 D918090001	11/20/98	closed 1/12/99	С
87	2726	[)918091	COUPLING MALE PIPE TREAD	1	\$	3.89	\$ 3.8	9 D918091001	11/20/98	closed 11-16	С
88	DZZ7554 1	D918082	ACCESSORY KIT	1	\$	38,500 00				PER 1-12-99 MTG	CX
89	DZZ7625-503	T 300189	TEST SET, LWR CARGO DR RIGGING	1	\$	4,900.00	\$ 4,900.0	T300189001	5/25/99		cs
90	286-11	1160016	TRICKLE CHARGE MAIN BATT.	1	\$	725.00	\$ 725.0	1160016001	2/9/99		CS
91	3NA	D918095	SAFETY HARNESS WAIST				\$	 SEE NOTES 		PER 1-12-99 MTG	CX
92	2C6068G03	D918047	LOCKING STRAP DC10 2 ENG FERRY	1	\$	1,728.00	\$ 1,728.6	D918047001	4/6/99	closed 2/1	С
93	2C6068G03	D918047	KIT LOCKING STRAP DC10 2 ENG FERRY KIT	4	\$	1,728.00	\$ 6,912.0	D918047002	4/15/99		cs
94	2C6081G05 **	T000396	WRENCH, FUEL MANIFOLD	1	\$	942.00	\$ 942.0	00 T000996001	11/22/98	closed 12-1-98	С
95	2C6081G05 **	T000996	WRENCH, FUEL MANIFOLD	5	\$	942.00	\$ 4,710.0	00 T000996002	5/25/99		cs
96	2C6352G02 **	T000997	WRENCH SET- SPANNER	1	\$	2,346.00	\$ 2,346.0	00 T000997001	11/22/98	closed 12-1	С
97	2C6367G01	D918048	FIXTURE, R/I	1	\$	221.40	\$ 221.	D918048001	11/22/98	closed 12-1	С
98	2C6373G01	D918049	ADAPTER	1	\$	488.85	\$ 488.	35 D918049001	3/29/99		cs
99 1	2C6373G01	D918049	ADAPTER	4	\$	488.85	\$ 1,955.	10 D918049002	5/15/99	PER 1-12-99 MTG	cs
100	2C6955G01	1160018	FIXTURE, LIFT	1	\$	1,525.00	\$ 1,525.	00 1160018001	4/6/99		cs
101	2C6983P06 **	1160019	PUMP, COWL DOOR ACTUATOR	7			\$	-		Pricing	w
102	4771879-1	T000240	TORQUE ADAPTER	1	\$	332.90	\$ 332.	0 T000240002	2/9/99	closed 12-31	С
103	4916759-505	D918061	TORQUE ADAPTER	1	\$	352.65	\$ 352.	55 D918061001	11/22/98	closed 12-1	С
104	AXG7000-501	D918062	PIN ASSY	1	\$	38.80	\$ 38.	0 D91806200	11/22/98	closed 12-1	C
105	AXG7000-501	D918062	PIN ASSY	2	\$	38.80	\$ 77.	50 D918062002	1/4/99	closed 1-6-99	С
106	AXG7012-501	D918063	LOCK ASSY	1	\$	127.79	\$ 127.	79 D918063001	11/22/98	closed 12-1	С
107	5489N-6 WIHARNESS PIN 201R-8	D918097	LANYARD 6'				\$	- SEE NOTES			СХ
108	PF32003	T000993	BRAKE LIFT TOOL	2	\$	290.00			4/6/99		cs
109	3ZL78	D918094	SAFETY HARNESS	4	\$	70,38	\$ 281.5	52 D918094001	1/28/99	CLOSED	C .
110	344L0LH9045	D918045	ADAPTER, STATIC PORT	2			\$	-	dup of 7 & 8	NAVAID	CX
111	DZZ-7048-501	D918111	DC10-30 THREAD PROTECTOR	1	\$	345.00	\$ 345.	00 D918111001	11/18/98	closed 12-1	С
112	DZZ0006-509	D918115	FLAP LOCK, GROUND SAFETY	1	S	-	\$	- D918115001	4/12/99	CANCEL PER TIM ALMAN	CX
113	DZZ0012-503	D918064	ADAPTER, JACKING, WING	1	\$	620.89	\$ 620.	39 D918064001	3/4/99		cs
114	DZZ0012-503	D918064	ADAPTER, JACKING WING	3	\$	620.89	\$ 1,862.	37 D918064002	3/15/99	PER 1-12-99 MTG	cs
115	DZZ7023-1	D918065	HANDLING ADAPTER	1	\$	348.00	\$ 348.	D918065001	3/4/99		CS
116	DZZ7023-1	D918065	HANDLING ADAPTER	1	\$	348.00	\$ 348.6	00 D918065092	4/15/99		CS
117	DZZ7026-1	D918066	HOISTING ADAPTER SET	1	\$	2,207.00	\$ 2,207.	00 D918066001	11/17/98	closed 12-1	С
118	DZZ7027-1	D918098	SLING, WINDSHIELD	1	\$	430.60			3/4/99		cs
119	DZZ7027-1	D918098	SLING, WINDSHIELD	1	\$	430.60	\$ 430	50 D918098002	4/15/99	PEF 1-12-99 MTG	cs
120	DZZ7030-1	D918099	SLING ASSY, MLG ACTUATOR	1	\$	154.80			11/17/98	closed 12-1	С
121	DZZ7034-1	D918067	HOISTING BRACKET	1	\$	198.30			11/17/98	closed 12-1	С
122	DZZ7040-501	D918068	BRACE, DETACHABLE ELEV	1	\$	1,320.00	\$ 1,320.		3/4/99		cs
123	1D846	D918096	LANYARD 10' (IS COMPATIBLE WIHARNESS #3ZL78)	5	\$	33,36	\$ 166	D918096001	1/29/99	CLOSED.	С

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124	DZZ7041 1	D918:00	GUIDE MLG BRAKE BUSHING	1	\$	378.00	\$ 378.00	D918100001	11/4/98	closed 11-4-98	C
125*	DZZ7041-1	D918:00	GUIDE MLG BRAKE BUSHING	5	\$	378.00	\$ 1,890.00	D918100002	1/29/99	closed 2/3/99	С
126	DZZ7042-1	D918069	TORQUE ADAPTER, MLG AXLE NUT	1	\$	253.00	\$ 253.00	D918069001	11/17/98	closed 12-1	С
127	DZZ7042-1	D918069	TORQUE ADAPTER, MLG AXLE NUT	5	\$	253.00	\$ 1,265.00	D918069002	1/29/99	closed 2/3/99	С
128	DZZ7043-1	D918070	ADAPTER, NLG AXLE NUT	4	\$	193.60	\$ 774.40	D918070001	11/17/98	closed 12-1	С
129	DZZ7043-1	D918070	ADAPTER, NLG AXLE NUT	. 2	\$	193.60	\$ 387.20	D918070002	1/29/99	closed 2/3/99	С
130	DZZ044-1	D918 01	SAFETY LOCK, MLG DOOR	1	\$	464.00	\$ 464.00	D918101001	11/17/98	cosed 12-1	С
131	DZZ7044-1	D918:01	SAFETY LOCK, MLG DOOR	1	\$	464.00	\$ 464.00	D918101002	1/4/99	closed 1-6-99	С
132	DZZ7044-1	D918101	SAFETY LOCK, MLG DOOR	12	\$	464.00	\$ 5,568.00	D918101003	1/29/99	closed 2/3/99	С
133	DZZ7048 1	D918102	PROTECTOR, MLG/CLG AXLE	1	\$	345.00	\$ 345.00	D918102001	11/17/98	cosed 12-1	С
134	DZZ7048-1	D918102	THREADS PROTECTOR, MLG/CLG AXLE THREADS	6	\$	345.00	\$ 2,070.00	D918102002	1/29/99	closed 2/3/99	С
135	DZZ7053-1	D918071	ADAPTER, JACKING, FUSALAGE	1	\$	398.53	\$ 398 53	D918071001	3/19/99		CS
136	DZZ7048-1	D918071	ADAPTER, JACKING , FUSALAGE	1	\$	398.53	\$ 398.53	D918071002	5/15/99	PER 1-12-99 MTG	cs
137	DZZ7055-1	1000991	SPANNER WRENCH, NLG STRUT	1	\$	425.00	\$ 425.00	T000991001	11/17/98	closed 12-1	С
138	DZZ7055-1	1000991	SPANNER WRENCH, NLG STRUT	1	\$	425.00	\$ 425.00	T000991002	1/29/99	closed 2/3/99	С
139	DZZ7056-1	T000992	SPANNER WRENCH, MLG STRUT	1	\$	311.50	\$ 311.50	T000992001	11/17/98	closed 12-1	C
140	DZZ7056-1	T000992	SPANNER WRENCH, MLG STRUT	1	\$	311.50	\$ 311.50	T000992002	1/29/99	closed 2/3/99	С
141.14	DZZ7073-1	D918103	SLING, MLG/CLG WHEELS	1	\$	948.00		D918103001	3/18/99		СХ
142	DZZ7074-1	D918104	GUIDE, STRUT BEARING	1	\$	183.65	\$ 183.65	D918104001	11/17/98	closed 12-1	С
143	DZZ7074-1	D918104	GUIDE, STRUT BEARING	1	\$	183.65	\$ 183.65	D918104002	1/29/99	closed 2/3/99	С
144	DZZ7077-1	D918072	TORQUE ADAPTER	1	\$	378.95	\$. 378.95	D918072001	11/17/98	closed 12-1	С
145	DZZ7077-1	D918072	TORQUE ADAPTER	1	\$	378.95	\$ 378.95	D918072002	5/25/99	SCHED SHIP 5/25/99, PER 1-12-99 MTG	CS
146	DZZ7078-1	D918073	TORQUE ADAPTER	1	\$	719.00	\$ 719.00	D918073001	11/17/98	closed 12-1	С
147	DZZ7078-1	D918073	TORQUE ADAPTER	1	\$	719.00	\$ 719.00	D918073002	2/5/99	PER 1-12-99 MTG	CS
148	DZZ7087-1	1160003	LOCK, UPPER RUDDER POSITIONING	1	\$	3,666.00	\$ 3,666.00	1160003001	11/17/98	closed 12-1	С
149	DZZ7087-1	1160003	LOCK, UPPER RUDDER POSITIONING	1	\$	3,666.00	\$ 3,666.00	1160003002	1/29/99	closed 2/3/99	С
150	DZZ7088-1	1160004	LOCK LOWER RUDDER POSITIONING	1	\$	2,890.00	\$ 2,890.00	1160004001	4/1/99		cs
151	DZZ7088-1	1160004	LOCK LOWER RUDDER POSITIONING	1	\$	2,890.00	\$ 2,890.00	1160004002	6/4/99	PER 1-12-99 MTG	cs
152	DZZ7089-1	D918105	SLING, AILERON ACTUATORS	1	\$	260.70	\$ 260.70	D918105001	11/17/98	closed 12-1	С
153	DZZ7089-1	D918105	SLING, AILERON ACTUATORS	1	\$	260.70	\$ 260.70	D918105002	1/29/99	closed 2/3/99	С
154	DZZ7100-501	1160005	STOP ASSY	1	\$	753.00	\$ 753.00	1160005001	3/4/99		cs
155	DZZ7100-501	116005	STOP ASSY	1	\$	753.00	\$ 753.00	1160005002	5/5/99	PER 1-12-99 MTG	cs
156	DZZ7101-1	1160006	SLING, RADOME	1	\$	4,379.00	\$ 4,379.00	1160006001	3/19/99		cs
157	DZZ7104-503	1160007	LOCK, TAIL CONE HOLD OPEN	1	\$	1,839.00	\$ 1,839.00	1160007001	11/17/98	c'osed 12-1	С
158	DIT-78-500	1160011	FAN COWL STUT LOCK							possible dup same as 182 - lisa research	w-•
159	DZZ7109-1	D918074	SAFETY BRACE	1	\$	1,125.00	\$ 1,125.00	D918074002	2/29/99		cs
160	DZZ7126-1	D918075	PULLER KIT	1	\$	9,767 00		D918075001	4/29/99	cancelled per1-12-98mtg	CX
161	DZZ7130-1	D918106	GUIDE, NLG PISTON LWR BRG	2	\$	147.50	\$ 295.00	D918106001	11/17/98	cosed 12-1	С
162	DZZ7138-1	D918107	GUIDE, PISTON UPPER BRG	2	\$	399.00	\$ 798.00	D918107001	11/17/98	cosed 12-1	С
163	DZZ7149-1	D918076	TORQUE ADAPTER	1	\$	481,00	\$ 481.00	D918076001	11/17/98	closed 12-1	С
164	DZZ7150-1	D918077	TORQUE ADAPTER	1	\$	445.20	\$ 445.20	D918077001	11/18/98	closed 12-1	С
165	DZZ7151-1	D918078	ADAPTER	1	\$	370.45	\$ 370 45	D918078001 ;	11/17/98	closed 12-1	С
166	DZZ7151-1	D918078	ADAPTER	1	\$	370.45	\$ 370.45	D918078002	1/29/99	closed 2/3/99	С

HEW	Whit	CHI	HOUH	ЦIY	Ųį	III PRICE	TOTAL PRICE	PU#	DOC DATE	SIAIOS	UUULZ
167	DZZ7162-1	1160008	DC10-30 NOSE WHEEL WRENCH	1	\$	173.50	173.50	1160008001	3/4/99	closed 1-4-99	С
168 *	D227171-1	D913116	PULLER SET, LANDING GEAR PINS	1	\$	6,250.00	6,250.00	D918116001	4/1/99		cs
169	DZZ7172-1	D913108	SLING, NOSE COWL	1	\$	928 00	928.00	D918108001	2/10/99		cs
170	DZZ7174 1	D913117	SPINNER, ANTI-SKID TRANDUCER	1	\$	589.00	589.00	D918117001	11/17/98	closed 12-1	С
171	DZZ7287-1	D913118	SUPPORT, ENGINE COWL DRS	1	\$	51,325 30	ş .	D918118001	210 day LT	CANCELLED, NO FEES.	CX
172	D2Z7296-1	D913079	TEMPLATE SET	1	\$	1,864.00	1,864.00	D918079001	3/18/99		cs
173	DZZ7317-501	()913080	RIG PIN KIT	1	\$	4,115.00	4,115.00	D918080001	4/1/99	closed 12-2-98	С
174	DZZ7317-501	D913080	RIG PIN KIT	1	\$	4,115.00	4,115.00	D918080002	6/4/99	PER 1-12-99 MTG	cs
175	DZZ7343-1	D913109	BRACE, SPOILER MTC	1	\$	118.00	118.00	D918109001	11/17/98	closed 12-1	С
176	DZZ7343-1	D913109	BRACE, SPOILER MTC	17	\$	118.00	2,006.00	D918109002	4/27/99	PER 1-12-99 MTG Shipped 4 on 1/29 Waiting	cs
177	DZZ7351-1	D913110	SLING, MLG/CLG BRAKE	1	\$	338.25	338.25	D918110001	11/17/98	on AWB# closed 12-1	С
178	DZZ7371-1	D913081	ADAPTER, TORQUE	1	\$	328.00	328.00	D918081001	3/4/99		cs
179	DZZ7375-1	1160020	SLING, CF6 ENGINE SUPPORT	1	s	2,650.00	2,650.00	1160020001	11/17/98	closed 12-1	С
180	DZZ7390-1	1160021	PULLER, WHEEL, MLG	1	\$	3,461.00	3,461.00	1160021001	4/18/99		cs.
181	DZZ7390-501	1160022	PULLER, BRAKE, MLG	1	s	6,633.00	6,633.00	1160022001	11/17/98	closed 12-1	С
182	DIT-78-501	1160012	FAN COWL STUT LOCK				s -			same as 158 lisa research	w
183	DZ-7154		DC10-30				S -			Cancelled	СХ
184 1	d 110-0460-102	T300137	LRRA	1	\$	12,995.00	12,995.00	T300137003	2/9/99	closed 2/5/99	С
185	DZZ7550-501	1160024	CONTROL BOX	1	\$	49,000.00			14WKS	PER 1-12-99 MTG	СХ
186	DZZ7709-1	1160025	TEST BOX	1	\$	2,485.00	2,485.00	1160025001	4/5/99		cs
187	DZZ7709-3	T001001	TEST SET W/O BOX	1	\$	1,393.00	1,393.00	T001001001	3/22/99		cs
188	TC8-3350-7-4302	T000994	CHARGER TRICKLE CHARGER EMERG							PER 1-12-99 MTG	СХ
189	3SF2828-501		TIRE PRESSURE/FILLER TESTER	6		?		7	?	lisa research wilh Spec Tool for new number	s
190	AMZ68		GROVE MANLIFT	1	\$	85,173.00	85,173.00	33163	2/9/99	will dc-8 work 215#	cs
191			WHOLE BODY SAFETY HARNESS	4			s -			Cancelled Duplicateof D918095 or Item	СХ
192	125137-35	t001005	WING JACKS	4	\$	23,995.00	95,980.00	t00100600I	4/19/99	#109??	cs
193	4582-35	t001006	NOSE JACKS	2	\$	17,000.00	-	t001005001	4/19/99		cs
194	856A1221G01		PIN SET, LOCKING VSV RIG PIN	6	5	606.00	•			WAITING DECISION FROM R.V.	w
195	856A1336G02		TORQUE ADAPTER SET-MEC		\$	779 00				WAITING DECISION FROM R.V.	w
			FASTENERS WRENCH SET-MEC/FUEL PUMP	2	5	2,117 00				WAITING DECISION FROM R.V	w
196	856A1469G02		CRATEX POLISHING BLOCK, EXTRA	2	•		\$ 4,254.00 \$ -	CRATEX MFG CO		NEED MORE INFO TO SOURCE WAITING	11
197	E-mail Part number		FINE GRADE					CRATEX WITG 50		DECISION FROM R V	
198	856A2664G01		YOKE, SETTING - VBV DOORS	2	\$	1,848.00				WAITING DECISION FROM R.V.	W
199	856A3408G02		GAGE, MASTER VBV	2	\$	651 00				WAITING DECISION FROM R.V.	w
200	TMRX10		CROWFOOT WRENCH-GEAR MOTOR ASSEMBLY	2	\$	27 25	54 50	SNAP ON		WAITING DECISION FROM R.V.	w
201	856A1331G02		GAGE SET, STAGE IGV, 1, 2 AND 3 VSV	2			s -	-		Awaiting Bruce Robbins	Н
202	AN8508-14B		. KIT CROWFOOT WRENCH, FLARE NUT	2	s	20 45	40 90	SNAP ON		WAITING DECISION FROM R.V.	w
303	856A3317P02		TEST LEAD	2	5	583 OC	1,166 00			WAITING DECISION FROM R V.	w
204	856A33A8P02		TEST LEAD	2	\$	530 00	1,060 00			WAITING DECISION FROM R.V.	w
205	856A9507G01		LIFTING BEAM	1	5	18,944 00	\$ -			Cancel per Bruce Robbins	СХ
206	856A9004G01		MOUNT PIN, INSTALL/REMOVE	t	\$	418 00	418 00			WAITING DECISION FROM R V	w
⊋07	856A9009G01		HYDRAULIC ACTUATION TOOL CONNECTOR, JUMPER-LOCKING	1	\$	1,219 00	1,219 00	: .		WAITING DECISION FROM R V	w

Ht:M	MPH	CPN	NOUN	YIP	U	NIT PRICE	1	TOTAL PRICE	PO#	DUE DATE	STATUS	CODE
208	856A9032G01		ADAPTER, PULLER-HYDRAULIC ACTUATOR TRUDNION PINS	1	\$	428 00	S	428 00			WAITING DECISION FROM R.V.	W
209	TEE.61		PMC-TESTER	1	\$	56,930 00	\$	56,930 00		10-12 WK ARO	VENDOR-RMS WAITING DECISION FROM	w
210	A000111		EGT TESTER BARFIELD	2			\$		404-761-4321		R V LEFT VOICE MSG 1/28 BARFIELD WAITING	s
211	TEE62-2 RMS		VSV DYNAMIC TRACKING	1	\$	60,000 00	s				DECISION FROM R V SAME AS #5??? WAITING DECISION FROM	СХ
212	2C6983P09		PUMP, HYDRAULIC OPEN FAN	4	\$	800 00	s	3,200 00	LOCKHEED/MARTIN		R V WAITING DECISION FROM R.V	w
213	2C6639G03		REVERSER INLET FILTER SCREEN REMOVAL	2	\$	3,090 00	s	6,180 00	LOCKHEED/MAFTIN		WAITING DECISION FROM R V.	w
∠14	2C14244G01		TOOL WRENCH SET, MEC	2	\$	4,059 00	s	8,118 00			WAITING DECISION FROM R.V.	w
215	2C6276G03		WRENCH SET, FUEL CONTROL PUMP	2	\$	1,100 00	s	2,200 00			WAITING DECISION FROM R V	w
216	2C6613G01		GAGE, IMMERSION TEST	1	\$	1,050.00	s	1,050.00			APPROVED BY R.V. AWAITING BRUCE	w
217	PBS-4100		VIBRATION/TRIM BALANCE ANALYSER	1	\$	65,191 00	s	-	\$69366 INCL TRNG		ROBINS Caricel per Bruce Robbins	cx
218	856A1353G01		FIXTURE TEST TCC VALUE, HPTACC	1	\$	3,085 00	\$	-			Cancel per Bruce Robbins	сх
219	REF NO Q13875		5TH & 9TH STAGE ENGINE CHANGE HARDWARE	1	\$	21,000.00	\$	21,000.00			Bruce to advise if the same as 55 & 56	w
220	856A1084G02		ACTUATOR PORTABLE HYDRAULIC HAND OPERATED SYSTEM	3	\$	5,488 00	S	16,464 00			WAITING DECISION FROM R.V.	W
221	QC1R50 0:50		TORQUE WRENCH	3	\$	179 95	\$	539 85	SNAP ON		WAITING DECISION FROM R.V.	w
.?22	QC2R200 40-200 INC/LI3		TORQUE WRENCH	3	\$	189 95	\$	569 85	SNAP ON		WAITING DECISION FROM R.V.	w
13,	CR1000 200-1000 IN/LBS		TORQUE WRENCH	3	\$	189 95	\$	569 85	SNAP ON		WAITING DECISION FROM R.V.	w
224	0-100 IN/HG		PRESSURE GAGE PT7 PART POWER TRIM BEARING BREATHER PRESSURE	2	\$	500.00	\$	1,000 00			WAITING DECISION FROM R.V.	w
.225			TEST OSCILLOSCOPE				\$	-			NEED MORE INFO TO SOURCE WAITING	н
2 2 6			SPECTRUM AMALYZER				\$	-	HEWLET PACKARD-		DECISION FROM R V NEED MORE INFO TO SOURCE WAITING	н
727	·		DIGITAL MULTIMETER				\$	-	MFG FLUKE-MFG		DECISION FROM R.V NEED MODEL # WAITING DECISION FROM	н
. 28			DATA BUS ANALYZER (ARINC & RS				\$	-			R.V NEED MORE INFO TO SOURCE, WAITING	Н
.50			232) FREQUENCY COUNTER				\$	-			DECISION FROM R V NEED MORE INFO TO SOURCE WAITING	н
230	12 602 9		RF SIGNAL GENERATOR		J.	49,495 00	s	49,495.00	Dac Intil	90 Days	DECISION FROM R V WAITING DECISION FROM R V	н
731	TCAS 201		TCAS TEST SET	1	\$	14,695 00	\$	14,695.00		STK	ORDER GTY? VENDOR-DAC INTL. TCAS REPLY GENERATOR-RAMP, WAITING	w
732	TCAS 201 COUPLER MODEL B (AG-210B)		TCAS ANTENNA COUPLER TESTER	1	\$	2.675 00	\$	2,675.00		1-2 WK	DECISION FROM R.V. ORDER QTY? VENDOR-DAC INTL. BENDIX COUPLER FOR TCAS201 PART #2018.	w
33			WATIMETER				\$	-			WAITING DECISION FROM R V NEED MORE INFO TO SOURCE WAITING	н
134			WEATHER RADAR PRIMUS 90				\$	-			DECISION FROM R V MED MORE INFO TO SOURCE WAITING	11
-35			WEATHER RADAR TESTER (BENDIX)				\$	-			DECISION FROM R.V. NEED MORE INFO TO SOURCE WAITING	H
136			WEATHER RADAR TESTER (COLLINS)				\$	-			DECISION FROM R V NEED MORE INFO TO SOURCE WAITING	н
:3 <i>i</i>			RADOME TESTER				\$	-			OF USION FROM R V NELD MORE THEO TO SOURCE WALLING	H
-38	PSD30-2 TyPL		FUEL QUANTITY TEST SET				\$		JCAIR-MFG		ÐFCISION FROM R.V. FAX REG 1726/99-WAITING RESPONSE.	REQ
-39	PSD 60 1AF		AC CAPACITANCE FIELD CALIB UNIT	1	3	6 025 00	\$	6,025.00		90 DAY	WAITING DECISION FROM R.V. ORDER GLY? VENDOR-BEG JCAIR, PH# 01-	w
÷40	78°r T 18°F		MAINTENANCE KITS				\$	-,	JCAIR-MEG		0605-00 WAITING DECISION FROM R V CAX RCG 1/20/99 WAITING RESPONSE	REQ
.41	GSM 100		ANALOG GYRO				\$		JUAIR MEG		WALTING DECISION FROM R V EAX REG 1/20/95-WALTING RESPONSE	REQ
-42	T322967		RADIO ALTIMETER TEST SET				3 S		JICAIR MEG		WALTING DECISION FROM R V FAX REQ 1/26/99 WAITING RESPONSE	REQ
	I OI E.CH		det a martin (Ed) QL				•	-	on or the Co		WALLIE DE JSJOH ROM S V	IVI (3

ILM	МРИ	GPN NOUN	QTY	·	NIT PRICE	TOTAL PRICE	PO#	DUE DATE	STATUS	CODE2
243	A1G-600A • •	DME/XPO:IDER	1	s	7,495 00	7,495.00		STK	ORDER GTY? VENDOR-DAC INTL ATC/DML RAMP TEST SET WAITING	w
244	C1S-700	VHF/HF/SELCAL/ATSCALL TESTER	₹			s -	COLTECH-MFG		DIFCISION FROM R V DUPL OF #17 OR CPN T300183 CX PER	СХ
45		ADFITESTER				s -			BRUCE NEED MORE INFO TO SOURCE WAITING	H
246	402AP	MKR/HAVICOMM RAMP & BENCH TE SET	ST 1	\$	11,695 00	11,695.00		sīk	DECISION FROM RIV ORDER OT 12 VENDOR-DAWINTL NAVICOM RAMPITEST SET IPN# NAV-	w
247	HAV-750C	ILS/VOR/MB/ADF/SELCAL/&COMM TESTER	1	\$	14,595 00	14,595.00		STK	402AP WAITING DECISION FROM R V ORDER GTY VENDOR-DAC INTL NAVICOM FEST SET WAITING DECISION	w
248	2945A	AVIONICS COMMUNICATIONS	1	\$	16 750 00	16,750.00	Dac Int'l	6 Weeks	FROM R V WAITING DECISION FROM R V.	w
.349		MONITOR AIR DATA TESTER		\$	52 000 00	s -	PENNYSGILES		WALLING DECISION FROM R V	w
.350		GPWC TEST SET				\$ -			WAITING DECISION FROM R.V.	w
251	7000-4133	PBS-4100 PORTABLE BALANCING SYSTEM	1	\$	35,995.00	35,995.00			BRUCE ROBBINS TO REVIEW	w
252	7000-4141	PORTABLE PRINTER	1	\$	695 00	695.00			BRUCE ROBBINS TO REVIEW	w
253	1400-4118	SHIPPING/STORAGE CASE	1	\$	375.00	375.00			BRUCE ROBBINS TO REVIEW	w
254	1400-4119	(DAU/COMPUTER) SHIPPING/STORAGE CASE	1	\$	325.00	325.00			BRUCE ROBBINS TO REVIEW	w
255	2053-4801	(PRINTER&CABLES) ENGINE DATA DISK SET;JT8D-1,-5,-	7 1	\$	-	\$ -			BRUCE ROBBINS TO REVIEW	w
256	2053-4809	ENGINE DATA DISK SET; JT8D-9,-11	i ₋ 1	\$	180.00	180.00			BRUCE ROBBINS TO REVIEW	w
257	8900-5056	15,-17 ENGINE INTERFACE CABLE SET (INC	CL 1	\$	1,995.00	1,995.00			BRUCE ROBBINS TO REVIEW	w
258	8900-5037	8900-5034, -5035, -5036) VELOCITY PICKUP JUMPER CABLE	1	\$	635.00	635.00			BRUCE ROBBINS TO REVIEW	w
259	8213-6680	JT8-D ENGINE JUMPER CABLE SET	1 1	s	1,835.00	1,835.00			BRUCE ROBBINS TO REVIEW	w
260	8213-5048	(INCL 8213-5042, -5043, -5044) N1/N2 BREAKOUT CALE (B707/B727/B737 COCKPIT TO N1)	1	\$	630.00	630.00			BRUCE ROBBINS TO REVIEW	w
261	8213-6563	N1/N2 BREAKOUT CABLE (DC-9 AND		\$	450.00	450.00			BRUCE ROBBINS TO REVIEW	w
262	4900-4005	MD-80 COCKPIT) 1/REV PULSE GENERATOR	1	\$	2,875.00	2,875.00			BRUCE ROBBINS TO REVIEW	w
263	4900-4002	PULSE GENERATOR GEAR BOX	1	5	3,475.00	3,475 00			BRUCE RO3BINS TO REVIEW	w
264	5500-4000	VELOCITY PICKUP (2 REQD ON JT80	D) 1	\$	1,100 00	1,100.00			BRUCE ROBBINS TO REVIEW	w
265	2053-4300	ENGINE DATA DISK SET; JT3D-3B	1	\$	180.00	180.00			BRUCE RO3BINS TO REVIEW	w
266	2053-430?	ENGINE DATA DISK SET; JT3D-7	1	\$	180.00	180 00			BRUCE RO3BINS TO REVIEW	w
267	4900-400?	PULSE GENERATOR GEAR BOX-GEA RATIO FOR JT3D	AR 1	s	3,475.00	3,475.00			BRUCE RO3BINS TO REVIEW	w
268	8213-5049	N1/N2 BREAKOUT CABLE (DC-8	1	\$	1,895.00	1,895.00			BRUCE ROBBINS TO REVIEW	w
269	2053-3200	COCKPIT TO N1 INDICATOR) ENGINE DATA DISK SET; CFM56-2	1	s	180.00	180.00			BRUCE ROBBINS TO REVIEW	w
270	8001-5024	CFM56-2 JUMPER CABLE SET (INCL	L 1	\$	2,536.00	2,536.00			BRUCE ROBBINS TO REVIEW	w
271	8000-4225	8001-6512, -6513, -6511) CHARGE AMPLIFIER	1	\$	2,395.00	2,395.00			BRUCE ROBBINS TO REVIEW	w
272	1400-4150	CHARGE AMPLIFIER MOUNTING	1	\$	500.00	500.00			BRUCE ROBBINS TO REVIEW	w
273	2053-1606	BRACKET ENGINE DATA DISK SET; CF6-6	1	\$	180.00	180.00	•		BRUCE ROBBINS TO REVIEW	w
274	8114-6430	CF6-50 ENGINE CONNECTION CABL	E 1	\$	2,110 00	2,110.00			BRUCE ROBBINS TO REVIEW	w
275	8113-5047	SET (INCL 8113-5164, -5027, -5038) BLIPPER CONDITIONER CABLE	1	\$	995.00	995.00			BRUCE ROBBINS TO REVIEW	w
276	4100-4102	TRAINING	1	\$	4,175 00	4,175 00			BRUCE ROBBINS TO REVIEW	w
277	TF1068-613	HYPERTRONICS OUTLET TEST FIXTURE	2	\$	350.00	700 00	T001007001	2/3/99	Closed 2/4/99	С
278	DC-1	DIGITAL CONDUCTIVITY INSTRUMEN	√T 1	\$	3,300.00	3,300 00	T300197001	2/10/99		cs
:79	BORESCOPE	100977 VIDEOSCOPE	1	3	27.239.00	37,239 00			WALLES OF ISSUITED MIRV	w

HEM	MPN	CPN	иоин	QTY	U	NIT PRICE		IOIAL PRICE	PO#	DUE DATE	STATUS	CODES
240	DC10BORESCOPE	1001003	FIBERSCOPE	1	s	71,444 00	\$	71,444.00			WAITING DECISION FROM R.V.	w
781 [™]	GS 2668 RT 4X4		NEW GENIE SCISSORS LIFT MODEL	2	s	27,475 00	\$	54,950.00			WAITING DECISION FROM R V	w
282			(BODE FINE) SMOKE GENERATOR (TOOLING				\$	•	TIM ALMAN TO			s
283			STROBE LIGHT CHECKER/TO COUNT				5		TIM ARYUSE TO			s
284	HALF AE95075N?		FLASHES PER MIN (TOOLING REQDO HYD PRESS HOSE COUP (TOOLING				s	-	ADVISE CANCEL PER TIM			СХ
285	HALF AE94186P?		REQD) HYD PRES HOSE COUP (TOOLING				\$		ALMAN CANCEL PER TIM			СХ
286			REQD) HYD MULE 3000 PSI 50 GPM?				s		ALMAN CANCEL PER TIM			СХ
287	2651-133?		(TOOLING REQD) LAV SERVICE DRAIN HOSE (TOOLING				s	-	ALMAN TIM ALMAN			сх
288			REQD) MLG DOOR LOCKS (D2270441)	2			\$		TIM ALMAN		Already purchase qty of 12 on po#	СХ
289	DPS350		(TOOLING REQD) AIR DATA TESTER -MODEL 127-1N	2	\$	9,995.00	s	19,990.00	BARFIELD	16 WKS	E918101003 LISA RESEARCH WITH BARFIELD RVSM	w••
290	34410LH90-5	D918386	(MILHARD ENGRG)(TOOLING REQD) STATIC PORT ADAPTER (CANADIAN		\$	265.00	\$	265.00	D918386001	2/18/99	CCMPLIANCE?	cs
291	P52287M3		AEROINST. CO)(TOOLING REQD) PILOTHEAD ADAPTER (MILHARD	1	\$	545.00	\$	545.00	D918054003	2/11/99		cs
292			ENGRG)(TOOLING REQD) IGNITER PLUG PLIERS (TOOLING				s	-	TIM ALMAN		CANCEL PER TIM	СХ
293	2C6613G01		REQD0 DEPTH GAUGE (TOOLING REQD)	1	\$	1,045 00	s	1,045 00	TIM ALMAN	1 WK	SPEC TOOL - AWAITING APPROVAL TO	w••
294	DZZ0006-511		FLAP ACCT SAFETY STRUTS	4	\$	6,965.00	\$	_	TIM ALMAN	20 WKS	PURCHASE 562-945-3351 CANCEL PER TIM ALMAN-DUPL OF CPN	СХ
295	\$		(TOOLING REQD) TIRE PRESSURE GAUGE TO 215PSIG				s	•	TIM ALMAN		D918115 OR #112. CANCEL PER TIM	СХ
296	MILH-5593-3-3600	D918052	(TOOLING REQD) HOSE	1	\$	250.00	\$	250.00	D918052003	2/11/99	ORDER PER TIM ALMAN	cs
297	MILH-5593-4-7200	D918053	HOSE	1	\$	494.00	\$	494.00	D918053003	2/11/99	ORDER PER TIM ALMAN	CS
298			DANIEL KIT	1	\$	10,000.00	s	10,000.00			BRUCE WRIGHT TO ADVISE PART #	w
299			WIRE LABEL	1	\$	1,100.00	\$	1,100.00		•	BRUCE WRIGHT TO ADVISE PART #	w
							\$	1,360,170.18				

EWA'S RESPONSE TO LETTER DATED [Secendor 21, 1996]



January 10, 1999

Mr. Joseph Abramski FSDO-SJC 1250 Aviation Ave., Suite 295 San Jose, CA 95110

Dear Mr. Abramski:

This letter is in response to your letters dated December 15, 1998 and December 21, 1998 concerning aircraft N105WP experienced main cargo door damage, and Telegraphic Airworthiness Directive (TAD) T98-26-51. (See Attachment 1)

Emery Worldwide Airlines (EWA) reported this damage to the Los Angeles Aircraft Certification office to Mr. Greg Delibero on December 17, 1998. (See Attachment 2).

Based on the initial determination by EWA, in concert with the FAA DER performing the line maintenance repair it was determined that the door was damaged due to FOD. EWA elected to issue a Fleet Campaign Directive 52-6 to inspect the Rosenbalm and Monarch STC Cargo door aircraft in this effected area as a precaution. As a result of this inspection, no defects were noted on other aircraft.

Mr. Andrew Albright, EWA Reliability Structural Analyst, performed an in-depth review of the cause of the damage. The following is the results of an analysis on aircraft N105WP main cargo door damaged that occurred on November 26, 1998 and was removed from service and repaired on December 14, 1998. The main cargo door sustained damages to the door locking mechanism, hydraulic system and primary structures. The crew's first indication was maintaining pressurizations written on log pages (6913-22, 6913-24 & 6913-25). The aircraft log pages reported FOD found by EWA's maintenance personnel in the lower forward section of the door causing the following chain of events. The events as written by flight crew members, repairs made by maintenance personnel and damaged parts replaced during the repair will demonstrate a pattern to which the damage conclusion was determined.

Pilot Write-Ups Sequence of Events:

Prior to aircraft's main cargo door damage on November 26, 1998, aircraft departed KELP to KDAY, log page 6913-22 pilot reports; "Both packs have to be at maximum to maintain the cabin below 10,000 feet, only able to get 7.6 differential left pack appears weak". Maintenance personnel inspected aircraft at KDAY and performed corrective action; "Removed FOD from along bottom edge of main cargo door, aircraft pressurization operations checked good, both packs operations checked good, no defects noted".

On November 27, 1998, Aircraft N105WP, departed KATL to KLAX. During the flight, log page 6913-25, pilot reports; "needed both packs to maintain pressurization, cargo door lights illuminates during flight, no cabin pressure change—lock bar appears to be at full travel—closed, cargo door latches went to latch position before door was fully closed (out of sequence)." Corrective Action; EWA received on December 14, 1998, a repair for the main cargo door IAW FAA DER approved Cotney Engineering Sketch F84-R01.

Pilot Write-Ups Abbreviated (See Attachment 3 for actual log pages):

11-26-98

- FOD found in cargo door lower section
- Difficulties maintaining cabin pressurization
- · Cargo door jerks while opening

11-27-98

- Difficulties maintaining cabin pressurization
- Cargo door open light illuminates
- Cargo door latches went to latch position before was fully closed

The FOD found by maintenance personnel at KDAY on November 26, appears to have affected the closure of the main cargo door at or before departing KELP. This appears to be true after reviewing prior log page (6913-21), did not indicate aircraft pressurization lost. Review of the remaining log pages 6913-23 and 6913-24 found discrepancy and write-up on pressurization. Aircraft log page 6913-24 pilot reports, "Air leak forward lower edge of cargo door, must run both packs at high flow to keep aircraft pressurized, and the cargo door jerks while opening". EWA maintenance personnel inspected aircraft at KLAX and performs corrective action by, "Straightened seal depressor at lower edge and serviced reservoir."

In summary, the log page revealed FOD in lower section of the main cargo door found by EWA maintenance personnel. Maintenance inspected the main cargo door lock mechanisms but were unable to find any damage and dispatched the aircraft for flight. The aircraft flew two additional legs that day without difficulties. The third flight indicated cabin pressurization problems and main cargo door jerked when opening. EWA has determined when the main cargo door closed at KELP, FOD

trapped in the locking mechanisms had shifted locking bar and locks to close unevenly, causing the locking system to become out of sequence.

Main Cargo Door Repairs:

The main cargo door damages repaired at KLAX by EWA maintenance personnel were recorded on EWA Non-Routine Maintenance Forms, on December 14, 1998. The main cargo door locking mechanisms and hydraulic system damages evaluated by EWA personnel. The locking device torque tub (PN: 23630352) damaged by excessive force created by hydraulic forces to shift locks combined with locks out of sequence.

EWA maintenance personnel evaluated the main cargo door hydraulic system damages. Their evaluation determined main cargo door jerking while opening caused by a damaged hydraulic valve (PN: 1629-3-24), removed and replaced. The Cylinder assembly (PN: 8520008009) found unserviceable, cylinder removed and replaced.

EWA maintenance personnel and FAA DER Mr. Derrick Seys, of the Cotney Company, inspected the internal door structure. Their findings were all 12 frames damaged, to which these frames run longitudinal each equipped with a locking hook. The Cotney Company analysis structural strengths lost which they can provide, they concluded repairs to all frames exceed limitations IAW DC-8 SRM type repairs. EWA requested and received a repair for all 12 frames from the Cotney Company (F84-R01), and repairs completed on December 14, 1998, IAW FAA DER approved Cotney Engineering Sketch F84-R01.

Summary:

The FOD found in the lower forward section of the main cargo door appears to have caused door locking system to become out of sequence induced by the torque tube. Evidence of FOD can be seen by the damage found to the lower forward door seal depressor. EWA has determined the damaged locks prevented the door from fully sealing, creating pressurization loss, as reported by crews. When the aircraft was in flight, pressurization engaged to the internal air pressure forcing the lower door edge out, creating just enough vibration from the loose door locks that illuminated the door open warning light. This constant vibration and air forces created enough stress on the forward door frames eventually cracked.

The request for information per your December 15, 1998 letter is enclosed (See Attachment 3).

Mr. Greg Delibero informed me on December 17, 1998 that a telegraphic airworthiness directive may be issued to inspect the Rosenbalm cargo doors in the area that the damage was found on aircraft N105WP.

On December 21, 1998, I coordinated the inspection requirements per T98-26-51 with Greg Dilibero and Rony Azzi of the Atlanta ACO, and of course yourself per the TAD. The inspection was performed on the nine (2) strongft per the TAD item (B) and no decesses were noted. Note Attachment 4,.

I have addressed this TAD inspection requirement with other effected air carriers, and they also found no defects. I have also addressed this with Mike O'Neil and we will discuss this issue at the scheduled DC-8 Cargo Conversion Joint Task Force meeting next Tuesday, January 12, 1999 at Miami.

Please call if I can be of further assistance or provide additional information.

Sincerely,

Junes Mittles

Thomas M. Wood Director Quality Control

attachments

cc: Rene Visscher QC Managers

TMW/amb

ATTACHMENTS

- 1. FAA PMI Letters to EWA Director Quality Control dated December 15, 1998 and December 21, 1998.
- EWA Director Quality Control letter dated December 17, 1998 to FAA ACO Engineer Greg Delibero. (Copy of EWA FCD 52-6)
- 3. FAA PMI information requested per his December 15, 1993 letter.
- 4. EWA TAD notification letter to PMI dated December 21, 1998, and EWA response letter for TAD findings to ACO TAL, Rany Azzi, Copies of all MA's performing inspection, memo from EWA Manager Reliability for 30 day review per AD, AFM AD revision record.

ATTACHMENT

NO.1



San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

December 15, 1998

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

Request For Information

On November 27, 1998, at Los Angeles, California, a structural failure within the main cargo door was discovered by Emery Worldwide Airlines (EWA) maintenance personnel on an EWA DC-8-73 series aircraft, N105WP.

Accordingly, the following information is hereby requested :

- 1) The total aircraft time in service since the STC'd main cargo door installation.
- 2) A copy of the aircraft log book pages 6913-12; 6913-13; 6913-17; 6913-22; 6913-24; 6913-25; 6913-26; 6913-27.
- 3) Copies of all Non Routine Maintenance Work Forms associated with the repair to the main cargo door and any other maintenance performed during the aircraft out of service time in Los Angeles.
- 4) A copy of the last maintenance "A" Check work forms and all Non Routine Work Forms associated with that check.
- 5) A copy of FCD 52-6.
- 6) An aircraft listing of FCD 52-6 accomplishment and findings for each aircraft.
- 7) A copy of AI-5233-04:07.
- 8) An aircraft listing of AI-5233-04:07 accomplishment and findings for each aircraft.
- 9) A copy of the previous log book pages that state cabin pressurization problems; aircraft hull or main cargo door leaks; main cargo door malfunctions; or main cargo door indication malfunctions.

Please provide the requested information to the undersigned within ten (10) working days after receipt of this letter. Should you have any questions or require clarification regarding this matter, please call at your convenience.

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector

cc: Kent Scott - EWA Rene Visscher - EWA Michael O'Neil - FAA



Administration

San Jose Flight Standards District Office

San Jose International Airport 1250 Aviation Avenue, Suite 295 San Jose, CA 95110-1130 Phone: (408) 291-7681 FAX: (408) 279-5448

December 21, 1998

Mr. Thomas M. Wood Director, Quality Control Emery Worldwide Airlines, Inc. 303 Corporate Center Drive Vandalia, OH 45377

Dear Mr. Wood:

This letter will confirm our telecon of this day regarding Emery Worldwide Airlines (EWA) compliance with Telegraphic Airworthiness Directive (TAD) T98-26-51 as applied to EWA's fleet of DC-8 aircraft.

It is understood that EWA will issue Maintenance Authorization AI-5234-04:00 which will address the specific inspection guidance as contained in paragraph (B) of the aforementioned TAD; and EWA plans completion of the inspection on or about December 25, 1998. In addition, EWA will inform this office and the Atlanta Aircraft Certification Office (ACO) of its inspection findings.

Thank you for your cooperation in this matter.

Sincerely,

Joseph A. Abramski

Principal Maintenance Inspector

cc: Rene Visscher - EWA

ATTACHMENT

NO.2



December 17, 1998

Mr. Greg Delibero
Federal Aviation Administration
Los Angeles Aircraft Certification Office
3960 Paramount Drive
Lakewood, CA 90712-4137

Dear Mr. Delibero:

This letter is a follow-up to our telephone conversation today regarding the DC-8 Rosenbalm STC SA1802SO Cargo Door repair performed on a Emery Worldwide Airlines (EWA) DC-8-73, Serial Number 46095, Production Number 497, N105WP at LAX.

The subject aircraft received a repair to the cargo door due to a crack found during preflight. The following details are pertinent to this aircraft.

- I. STC Door Installation
 - March 17, 1989
 - Installer, Zantop Macon, Georgia
 - Aircraft TAT 45,890 and TC 14,988
- II. Aircraft TAT/TC as of 11-27-98
 - TAT 65.843
 - TC 21,197
- III. TAT/TC on Cargo Door STC Installation
 - TAT 19,953
 - TC 6.209
 - 9 years and 8 months

Per our conversation, you informed me that the FAA considers this to be as safety concern and is preparing an Airworthiness Directives (AD) for a one-time inspection of the cargo door.

As I discussed with you, EWA issued a Fleet Campaign Directive No. 52-6 (attached) to inspect the Rosenbalm and Monarch Cargo Doors in this specific area.

Mr. Greg Delibero Page 2 December 17, 1998

Per our agreement, this letter is being copied to the DC-8 JTF members for their initial notification and opportunity to perform inspections on their aircraft. Please advise me if the EWA FCD inspection will comply with the proposed AD inspection.

I would like to thank you for your support in this matter.

Sincerely,

Thomas M. Wood/re Thomas M. Wood Director, Quality Control

Attachment

Rene' Visscher

JTF Members

EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

		No. 52-6
		issue Date: 12/08/98
		Rev. <u>Criginal</u>
		Task Code: 852346
<u>Title:</u> Rosenbalm and Monarch Main Cargo Door Inspection	Reference:	N/A
Compliance Requested: As scheduled by Maintenance Planning	Approvéd by	<u>.</u>
Manpower: 3 hrs/aircraft		.ffected: Aircraft equipped alm and Monarch Main
Priority: N/A	Cargo Doors	
Publications Affected: None		
Weight Change: N/A		
GENERAL:		
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Upon identification of a crack, contact Mx. Control immediately for

Note:

further disposition.

EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

Page 2 of 2 No. 52-6 Rev. <u>Original</u>

ACCOMPLISHMENT INSTRUCTIONS:

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3.	Make a lo complete	M			
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303 CORPORATE CENTER DR. VANDALIA, OH 45377

To: Thierry Derrien

Company: Airborne Express

At: 1937383-4336

From: THOMAS M. WOOD

Fax: 937-898-2803

Company: EMERY WORLDWITE A'RUNES

Date: 12/17/98

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Time: 12:26PM

Pages Including Cover: 5

FAX MESSAGE

Notes: SEE ATTACHED.

ATTACHMENT

NO.3

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☐ HYDROSTATICALLY TESTED
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BENCH CHECKED

weight CHECKED Ш ERVICEABL 0 10: DATE 76 ASSEMBLY OVERHAULED
CALIBRATED
SERVICEABLE SERIAL NUMBER OFF N+23 REV. 5-94 . SS STATION Macon, Georgia hoqilA lanolgeA International Airlines, Inc. Georgia Місһідап

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r <u>15)</u>	0008	WORLD	RUNES	Garage	
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13	and till 2 21
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ME 034 (REV. 1	ATTACH HARD COPY TO UNIT.

ME 034 (REV. 1) (2/8/93)

TATEMENT OF COM	U.B. DEPARTMENT OF TR	ANSPORTATION WINISTRATION EEDED AT AVI	ATION REGULA	TIONS	December 14, 1998
TATEMENT OF COM	AIRCRAFT OR AIR	PEDEROC ATT	ONENT LOENTIFE		
	AIRCRAFT OR AIR	TYPE (LUMP	. Radio, Mail-sophur,	NAME OF	APPLICANT
AKE	MODEL NO.	814. /)	Em	very Worldwide Airlines Vandalia, Ohio
DOUGLAS	DC-8-73	1	PLANE	l	Valuatia, Otto
DOOGLAS		LIST OF DA	TITLE		
IDENTIFICATION					
WWA Dwg: 84-R01 (evision: I.R. Dated: 12/6/98	CRACKS.		- END	on Dal	riglas DC-8-73, s/n 46095, not constitute installation
	gpprov				
PURPOSE OF DATA	compliance with the reg	ulations listed t	sclow and docume	nt the rep	pair on Douglas DC-8-73, s/n
4 CADE 1) -	MINSWP				
. CAR 4b.		4b.202(a)(b)(c 4b.306			302, 4b 303, 4b.304(a),
AS ABOVE requirements of the Feder	nder the entherity vested by an 183 of the Federal Avia have been examined ral Aviation Regulations. I Recommend approval of t	i in accordance t	Administrator and in a dota based above an dota based above and with established proc	iccordance d on attach colores ac	with the conditions and internation and sheets new berod
i [(Same) Fligratore			DESIGNATION NUMBE	36/6/ T	CLASS/FICATION(S)
SENATUREIST OF DESIGN	MATED ENGINEERING REPRES	ENTATIVE(S)			
1//			SO-969		-STRUCTURES-
Derrick P.	Seys		l		
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1					.3684

FAA Form 8110-3 (11-70) SUPERSEDES PREVIOUS EDITION

- 1. THIS DRAWING APPLIES TO AIRCRAFT MODIFIED PER STC SA1802SO ONLY.
- 2. BREAK ALL SHARP EDGES AND DEBURR ALL HOLES.
- 3. HOLE PREPARATION AND CONDITION TO BE PER DC-8 SRM CHAPTER 51 UNLESS OTHERWISE NOTED.
- 4. ALL MACHINED SURFACES TO HAVE SURFACE FINISH OF 125 MICRO 'NCH OR BETTER
- 5. ALODINE AND PRIME ALL BARE SURFACES.
- 6 REFER TO ROSNEBALM DRAWING 39250 AND DOOR RIGGING PROCEDURES FOR PROPER ALIGNMENT OF DOOR LATCH AND LOCKS
- 7. FORM ALL PARTS IN THE "O" CONDITION AND HEAT TREAT TO ~16 PER MIL-H-6088 OR EQUIVALENT.
- B. ACCEPTABLE TO USE NEXT OVERSIZE IF REQUIRED FOR PROPER HOLE CONDITION IF A MINIMUM OF 2 FASTENER DIA EDGE DISTANCE EXISTS.
- 9. DETERMINE FINAL SIZE OF REPAIR HARDWARE UPON INSTALLATION TO ACCOMODATE FASTENERS SHOWN AND NOTED.
- PARTS MAY BE PURCHASED FROM ZANTOP INTERNATIONAL AIRLINES OR MANUFACTURE REPLACEMENTS.

		1		Z40583-2	FRAME RING	OR EQUIVALENT	
			1	Z40583-1	FRAME RING	OR EQUIVALENT	
1				Z30583-2	SIUB FRAME	OR EQUIVALENT	
	1			Z30583-1	STUB FRAME	OR EQUIVALENT	7
		1	1	-11	CHANNEL	7075-0 Q0-A-250/12	
						10/0_0 44 // 200/12	-
				-4	REPAIR, STUB	FRAME	
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				-1	REPAIR		
QTY	QTY	QTY	OTY	PART NUMBER	DESCRIPTION	MATL	NOTES
10LERANCES EMER					EMERY W	ORLWIDE AIRLI	NES

Preliminary

EFFECTIVITY

AIRCRAFT: SERIAL NO.

MTE 12/6/98 DEC 0 8 1998 REPAIR: CARGO DOOR FRAMES AT LOCK ATTACHMENT DUE TO CRACKS

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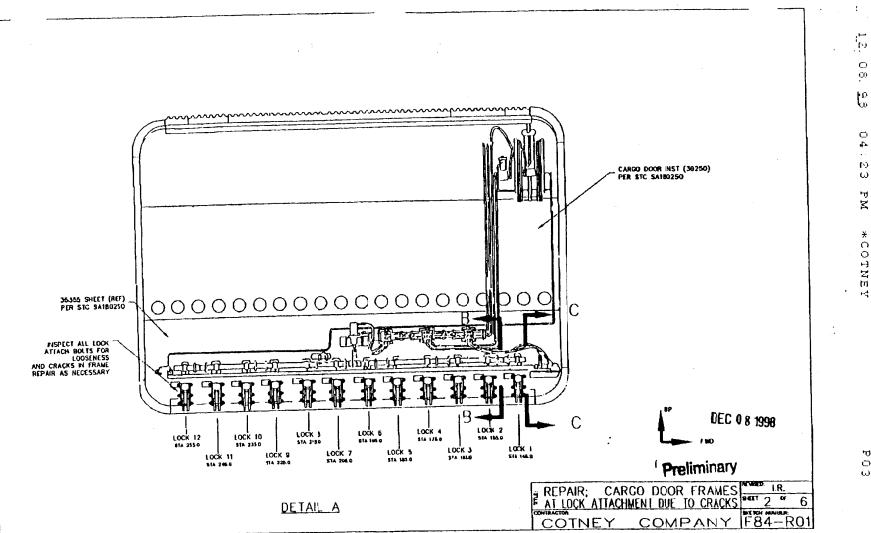
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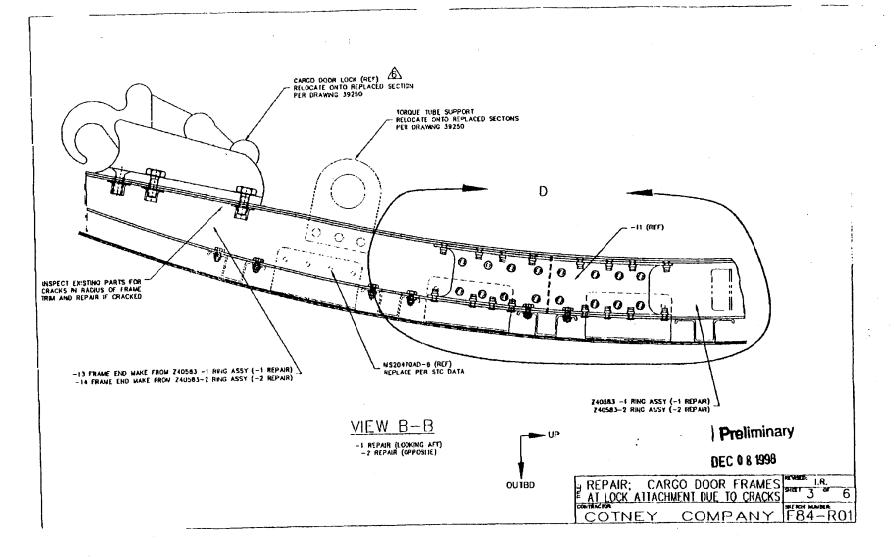
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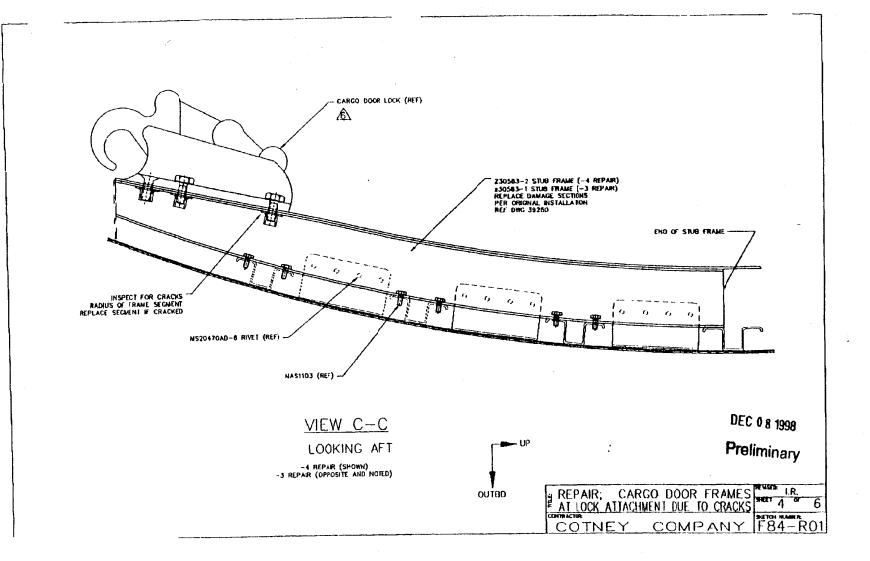
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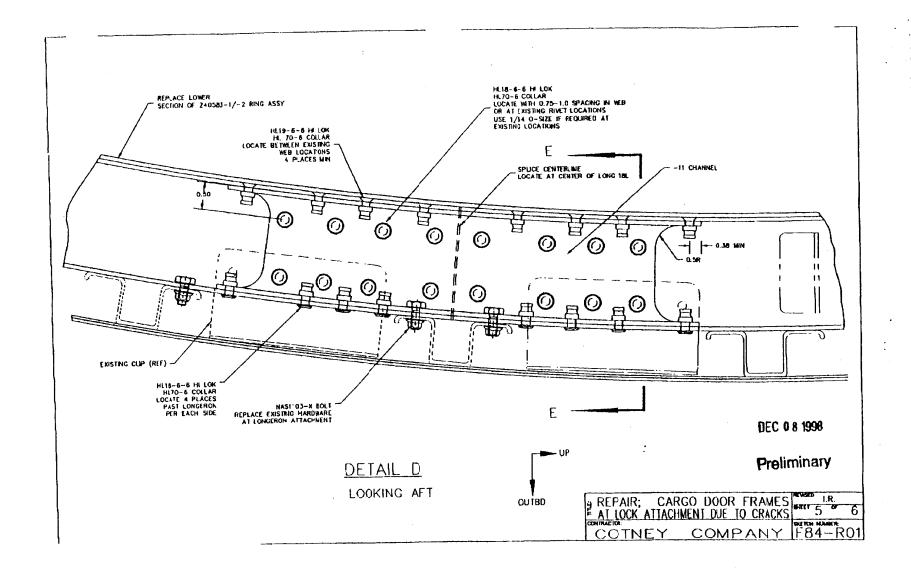
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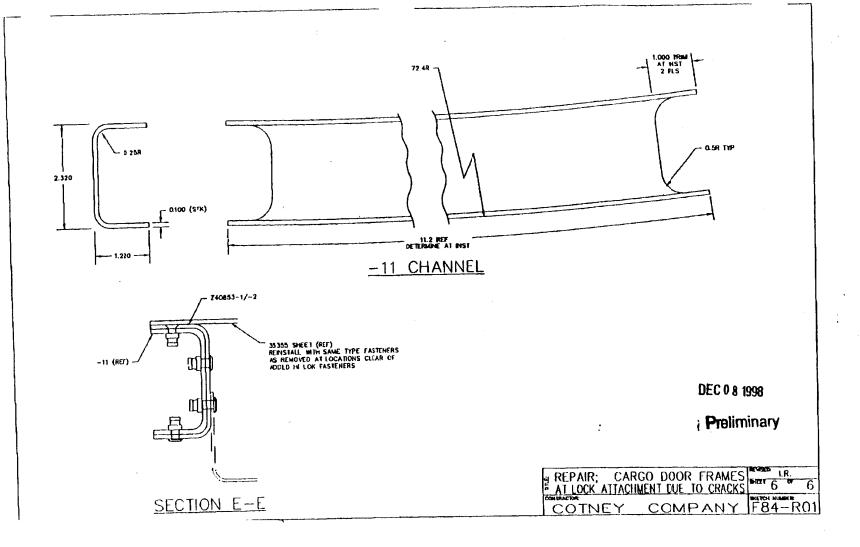
DC-8-73 46095 N105WP











CARGO DOOR - LUBRICATION 99.40 σ_{r} TO FHEAT LENST THE A3 LATCH ON THE DOOR. THE SProchATCH WALKING BEAM ON FUSEIASE 13 FINE AS FAR AS WE DOOD IS PAILING AWAY FRONITHE STANDANTHIS CAUSING CAM TELL, WHEN AIRCRAFT IS PRESSURED THE GREAT LARGE GAP/MISMATCH. BETWEEN CRACK RUMS FROM FWD OF #1 (14) CHARSER REV. 08 .015

	2-01						
AIRCRA /05 V	AFT NO.	DATE 12-12-12-12-12-12-12-12-12-12-12-12-12-1	STATION	TYPE CHECK	PAGE OF		
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<i>L</i>		N	FOLL	. 	INSP		
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ITEM #3	D	ISCREPANCY	CORRE	CORRECTIVE ACTION			
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Act. a		ckets ^L	Actuator	Removed torque tube Actuator and Brackets K			
			FGAN	FEAN			
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ITEM #9	DISCREPANCY	CORF	MECH		
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		' -			
ITEM #B	SCREPANCY		COR	RECTIVE ACTION	MECHOL
#5 FRAI	HEAT LECK FEIT	10,0	REPAIRES	- 13-06-4	
towns DAMAG	EŲ.		F84-R0	4/17	
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100000000000000000000000000000000000000	<u>, </u>		F 54-12CI	REUISICIU TIR.	INSP
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AIRCRAFT NO.	DATE		TATION	TYPE CHECK	PAGE OF A
	DISCREPANCY		CORF	RECTIVE ACTION	, MECH :
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FIC FR	ALME AT LCCK FEST	1010	KENAIRED	Par GWWA Dwy. Ravisiau I. R.	t-19-
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ITEM #K	DISCREPANCY		CORR	'₩ĔĊĤ [₽]	
#11 FRA FELIND DANNIKIN	ME AT LOCK POSITI	10%	1-84-RCI	Finder	
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<u> </u>	DISCREPANCY		CORR	MECHE	
FIX FR	TIME AT LOCK POUT	16,70	FEY-ROI	Tradi-	
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AIRCRAFT NO. DATE ITEM #2 DISCREPANCY #13 FRAINCE AT LOCK FOUT TOWN DANNINGED PIN OFF SIN OFF	K	HEDAIRED FSY-ROI	ECTIVE ACTION FOR EWAN IXIE. KEUWA IX.R.	PAGE OF & 6
FIN OFF SIN OFF	16,0	HEDAIRED FSY-ROI	per Ewwa 1216.	
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ITEM #23 DISCREPANCY		P/N ON	S/N ON	POS
		CORRE	ECTIVE ACTION	"WECH
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pin actuater		,		INSP
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80-58042-11 2963	18	30-58042	1308	enly
TEM #33 DISCREPANCY		CORRE	MECH	
INSTAll 12 FA latch	165	Install		
		latches.	Kenin B Curnel	
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TEM #24 DISCREPANCY		CORRE	CTIVE ACTION	MECH
	Fナ		d Fwd and	MECHE
LOZQUE LUBES.		AFT topo	ue tubes	Levin O. Cranal
		IAW STO		INSP
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EMERY WORLDWIDE AIRLINES

AIRCRAFT NO. DATE STATION TYPE CHECK PAGE 7 OF E ITEM #25 DISCREPANCY CORRECTIVE ACTION MECH INStall to reque tube Installed to reque tube Actual to R and Brackets Archator (same) and knumi Brackets Inw INSP STC \$AIBO250 PIN OFF SIN OFF PIN ON SIN ON POS ITEM #26 DISCREPANCY CORRECTIVE ACTION MECH Reinstall gang bar Reinstalled gang bar and Lock Pin. Required Ith 5TC 5A/60250 INSP
ITEM #25 DISCREPANCY CORRECTIVE ACTION MECH TINSTALL to Eque tube Installed to Eque tube Actualtor and Brackets Artualor (same) and keyind Brackets Inw INSP 5TC \$A180250 PIN OFF SIN OFF PIN ON SIN ON POS ITEM #26 DISCREPANCY CORRECTIVE ACTION MECH Reinstall gang for Reinstalled gang for and Lock Pins as swings
ITEM #25 DISCREPANCY CORRECTIVE ACTION MECH INSTALL to reque tube Installed to reque tube Actual to r and Brackets Artuator (same) and keyind Brackets Inw INSP STC \$A180250 PIN OFF SIN OFF PIN ON SIN ON POS ITEM #26 DISCREPANCY CORRECTIVE ACTION MECH Reinstall gang for Reinstalled gang for and Lock Pins as swince
ITEM #25 DISCREPANCY CORRECTIVE ACTION MECH Tinstall to eque tube Installed to eque tube Actual to R and Brackets Actual to R gue tube Brackets Inw INSP 5TC \$A180250 PIN OFF SIN OFF PIN ON SIN ON POS ITEM #26 DISCREPANCY CORRECTIVE ACTION MECH Reinstall gang to R Reinstalled gang to R and Lock Pin. and lock Pins as sewing.
Install torque tube Actuator and Brackets Brackets Inw INSP STC \$A180250 PINOFF SINOFF PINON SINON POS ITEM #26 DISCREPANCY CORRECTIVE ACTION Reinstall gang bar Reinstalled gang bar and lock pins as while of and lock pins as while of and lock pins and lock pins as while of and lock pins are likely the pins and lock pins and lock pins are likely the pins are likely to the pins are li
Install to eque tube Installed to eque tube Actuator and Brackets Actuator (same) and tound: Brackets Inw INSP STC \$A180250 PIN OFF SIN OFF PIN ON SIN ON POS TEM #26 DISCREPANCY CORRECTIVE ACTION MECH Reinstall gang bar Reinstalled gang bar and Lock Pin. and lock pins as seminal
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5A 18025'C
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TEM #28 DISCREPANCY CORRECTIVE ACTION MECH. Reinstall Sequence Valous Reinstalled Sequence
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DISCREPANCY CORRECTIVE ACTION MECH Reinstall sequence valours Reinstalled sequence and brackets SEA and HYd lines values and brackets towns insp
TEM #28 DISCREPANCY CORRECTIVE ACTION MECH Reinstall sequence valours Reinstalled sequence and brackets SEA and HYd lines values and brackets town Com

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EMERY WORLDWIDE AIRLINES "A" CHECK TALLY SHEET

Rev. Date: 3/20/97

Rev. No. 17

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A01	1 of 31	15	06/11/96	A01	17 of 31	14	11/15/95
A01	2 of 31	17	05/08/97	A01	18 of 31	17	05/08/97
A01	3 of 31	17	05/08/97	A01	19 of 31	17	05/08/97
A01	4 of 31	17	05/08/97	A01	20 of 31	17	05/08/97
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A01	13 of 31	14	11/15/95	A01	29 of 31	17	03/20/97
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MAINTENANCE:

		Y WOR	RLDWIDE ES	REV. DATE 06/11/96	REV. NO. 15	PAGE NO.	INSPEC. CK	CARD NO
		DC-8			•	ACFT. NO.	STATION	DATE
6	OG PAG 9/3	=1.3		INSTRU	JCTION		SIGN- MECHAN	OFF
1.	UPO	N ARR	IIVAL	"A" CH	IECK			
	a.	Mair Insta	ntenance Manual landing gear adding points. E: Verify the	ual Chapter 10 locking pins. El	"Parking Ma lectrically groun unit voltage re	ing to the Dou intenance Practic id aircraft to appro ads 110 to 120 V	es". oved	21075
	b.	Posit	tion switches as	ency is 380-420 s follows:	nertz. Connec	t ground power.	 b. [<	XIVI
		1. 2. 3. 4. 5.	Emergency Air Condition Battery-Exte	Light Not Armed ning rnal Power	EXT	OFFILLUMINATEDOFF ERNAL POWEROFF		
	C.	Oper	the following o	circuit breakers:			c. S	71075
		1. 2. 3. 4. 5. 6. 7. 8. 9.	Thrust Reverse Thrust Longitudinal Blow away Johann Standby Attit	Control Fach Power Sup rse Emergency S ust	Stow			
		WAR		ADVERTENT ERSERS COUL SONNEL WORK		RIOUS INJURY		
	d.	Revie action	ew log book an n on Open disc	d discrepancies repancies.	with flight crew	and take correct	ive d.	1075
	e.	Revie dates	ew open deferre with Mmainter	ed maintenance lance Control an	items and verif d take necessa	y time limits and d ry action.	lue e.	31075
	f.			echanism. Close			f.	31025
	WAR	RNING:		RCRAFT OF (LOWERING FLA		AND PERSONN	EL	

E	MERY WO	RLDWIDE INES	REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 2 OF 31	INSPEC. CK A	CARD NO. A01	
	DC	-8			ACFT. NO.	STATION	DATE 1/2/-92	
LOG	S PAGE /3-/2		INSTR	UCTION		<u>SIGN</u> - MECHAN		
	g. Ex	tend wing flaps a	and ground spoil	ers.		g. [81075	
					al open control va gear downlock pir		94503	
2.	ENGINE (OIL SERVICING						
	a. Pe	rform ENGINE 1	oil service as fo	llows:				
	WARNING: WAIT AT LEAST 5 MINUTES AFTER ENGINE SHUT-DOWN BEFORE REMOVING OIL TANK CAP, TO ALLOW TANK PRESSURE TO BLEED OFF. HOT OIL FROM TANK COULD CAUSE SEVERE BURNS. Note: Check engine oil tank level within 15 minutes after engine shutdown. If check has not been made within specified time, engine must be operated at approximately 75 percent N2 RPM for 1 minute immediately before checking tank oil level.							
	1)				e engine as requed on the log sheet		24505	
	2)	Check engined, se		" ring for ser	viceability, replace	e if 2)	04503	
**	3)		in gear box chip stall chip detector		eck, clean, replace	80- 3)	HA	
.•	4)		spect and reins	stall supply and	d common scave	nge		
		a) Rem	ove chip detecto	ors and magneti	c plugs (4 each).	a) c	1674	
		(1)	Clean area ar before remov		ctor or magnetic p	olug		
	•	CAUTION:	CHIP DETE	CTORS AND	MAGNETIC PLU	IGS		

SHOULD BE REMOVED AND INSTALLED

Remove chip detectors or magnetic plugs by depressing and turning one-quarter turn counterclockwise (CCW).

ONE AT A TIME.

(2)

EMERY WO			REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 3 OF 31	INSPEC. CK A	CARD NO. A01
DC	-8				ACFT. NO.	STATION	DATE 1/-2/-98
LOG PAGE 6913-13.			INSTRUCTION		SIGN- MECHANI	OFF	
		(3)		exact sump a	ne. Mark each nd engine from w		
	b)	Inspe	ect chip detector	s and magnetic	plugs (AD89-23-0	06). b)	24674
		(1) Examine each unit very closely, taking care not to displace any retained particles.					
		(2)			nagnetic plug sho s, notify Foreman.		
5) Perform starter magnetic plug check and service.							
	a)	a) Remove pneumatic starter magnetic plug and check for abnormal metal particles, chips, or fuzz. Notify Maintenance Control for required action if findings are abnormal.					
	b)	Clean and s		nagnetic plug w	rith serviceable O-ı	ring b)	0467y
	c)	Check	k oil level in pn	eumatic starte	aircraft and 105V s. Service to full Y. Secure servic	as 🗠	24674
	Note:	eq:	uipped with ass	isted wet splin servicing is red	CEPT 105WP, e cavity starters f quired at any inter		
6)	Perforr	n CSD	magnetic plug	check and serv	rice.		
		for al	bnormal metal enance Control	particles, chi	netic plug and che ps or fuzz. No action if findings	fity	14674
		Clean and sa		agnetic plug w	ith serviceable O-r	ing b) 2	4674
			and service Co		ssary. Service slov 2380 oil ONLY.	wly c)	4674

			•							
EMERY Al	WORL IRLINE:		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 4 OF 31	INSPEC. CK A	CARD NO. A01			
	DC-8				ACFT. NO.	STATION	DATE V2/9x			
LOG PAGE	3.		INSTRU	JCTION		SIGN- MECHAN	OFF			
b.		m ENGINE 2	and inspect fo	llows:		7) [24674			
	WARNING: WAIT AT LEAST 5 MINUTES AFTER ENGINE SHUT-DOWN BEFORE REMOVING OIL TANK CAP, TO ALLOW TANK PRESSURE TO BLEED OFF. HOT OIL FROM TANK COULD CAUSE SEVERE BURNS.									
Note	Note: Check engine oil tank level within 15 minutes after engine shutdown. If check has not been made within specified time, engine must be operated at approximately 75 percent N2 RPM for 1 minute immediately before checking tank oil level.									
	1)				e engine as requed on the log sheet		04503			
	2)	Check engin required, sec	•	' ring for sen	viceability, replace	e if 2) <u>(</u>	14503			
	3)		n gear box chip all chip detector		eck, clean, replace	0- 3)	HA			
	4)	Remove, ins		tall supply and	d common scave	nge				
-		a) Remo	ve chip detecto	rs and magneti	c plugs (4 each).	a) [1674			
··· .		(1)	Clean area ar before remova		ctor or magnetic p	iug				
		CAUTION:		E REMOVED	MAGNETIC PLU AND INSTALL					
		(2)		and turning	magnetic plugs one-quarter t	by urn				

Remove units one at a time. Mark each unit, indicating the exact sump and engine from which

Inspect chip detectors and magnetic plugs (AD89-23-06).

(3)

b)

it was removed.

EMERY WORLDWIDE AIRLINES DC-8 LOG PAGE (0913 13		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 5 OF 31	INSPEC. CK	CARD NO. A01
				ACFT. NO.	STATION SEA	DATE 11-21-98
		INSTRUCTION			SIGN-OFF MECHANIC ONLY	
	(1)		h unit very clos	sely, taking care n les.	ot to	

- (2) If any chip detector or magnetic plug shows abnormal or suspect deposits, notify Foreman.
- 5) Perform starter magnetic plug check and service.
 - a) Remove pneumatic starter magnetic plug and check for abnormal metal particles, chips, or fuzz. Maintenance Control for required action if findings are abnormal.
- b) Clean and re-install magnetic plug with serviceable O-ring and safety.
- C) Starter servicing - 50/60 series aircraft and 105WP: Check oil level in pneumatic starters. Service to full as required with Exxon 2380 oil ONLY. Secure servicing plug.

All EWA DC8-73 aircraft, EXCEPT 105WP, are Note: equipped with assisted wet spline cavity starters P/N 3505154-22. No servicing is required at any interval other than at installation.

- 6) Perform CSD magnetic plug check and service.
 - a) Remove constant speed drive magnetic plug and check for abnormal metal particles, chips or fuzz. Maintenance Control for required action if findings are abnormal.
- b) Clean and re-install magnetic plug with serviceable O-ring and safety.
- c) Check and service CSD oil as necessary. Service slowly to avoid over servicing. Use Exxon 2380 oil ONLY.

7) Motor engine and inspect for leaks.

1	EMERY WORLDWIDE AIRLINES		REV. NO. 17	PAGE NO. 6 OF 31	INSPEC. CK	CARD NO. A01
DC-8				ACFT. NO.	STATION KS Ex	DATE 11-21-98
LOG PAGE		INSTRI	UCTION		<u>SIGN-</u> MECHANI	

c. Perform ENGINE 3 oil service as follows:

required, secure cap.

2)

WARNING: WAIT AT LEAST 5 MINUTES AFTER ENGINE SHUT-DOWN BEFORE REMOVING OIL TANK CAP, TO ALLOW TANK PRESSURE TO BLEED OFF. HOT OIL FROM TANK COULD CAUSE SEVERE BURNS.	
Note: Check engine oil tank level within 15 minutes after engine shutdown. If check has not been made within specified time, engine must be operated at approximately 75 percent N2 RPM for 1 minute immediately before checking tank oil level.	,
Check engine oil for proper level. Service engine as required with Exxon 2380 oil ONLY. Record oil added on the log sheet.	1) OFD

Check engine oil cap "O" ring for serviceability, replace if

- 3) Remove main gear box chip detector. Check, clean, replace O- 3) Ring. Reinstall chip detector. (JT3D)
- 4) Remove, inspect and reinstall supply and common scavenge chip detectors (CFM56-2).
 - a) Remove chip detectors and magnetic plugs (4 each).
 - (1) Clean area around chip detector or magnetic plug before removal.

CAUTION: CHIP DETECTORS AND MAGNETIC PLUGS SHOULD BE REMOVED AND INSTALLED ONE AT A TIME.

- (2) Remove chip detectors or magnetic plugs by depressing and turning one-quarter turn counterclockwise (CCW).
- (3) Remove units one at a time. Mark each unit, indicating the exact sump and engine from which it was removed.
- b) Inspect chip detectors and magnetic plugs (AD89-23-06).
 - (1) Examine each unit very closely, taking care not to displace any retained particles.

1	ORLDWIDE LINES	REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 7 OF 31	INSPEC. CK	CARD NO.	
	C-8			ACFT. NO.	STATION	DATE	
LOG PAGE (913-13		INSTR	UCTION	· · · · · · · · · · · · · · · · · · ·	SIGN- MECHANI	OFF	
•	(2)	• •	detector or r suspect deposit	magnetic plug sh ts, notify Foreman.	ows		
5)	Perform s	tarter magnetic plu	ug check and se	ervice.			
	a) Remove pneumatic starter magnetic plug and check for abnormal metal particles, chips, or fuzz. Notify Maintenance Control for required action if findings are abnormal.						
	b) Cle and	an and re-install n i safety.	nagnetic plug w	rith serviceable O-	ring b)	4674	
	Che	eck oil level in pn uired with Exxon	eumatic starter	aircraft and 105Vrs. Service to full Y. Secure service	as	24674	
	3	equipped with ass	isted wet splin servicing is rec	CEPT 105WP, e cavity starters f juired at any inter	P/N		
6)	Perform CS	SD magnetic plug	check and serv	ice.			
••••••••••••••••••••••••••••••••••••••	for Mai	abnormal metal	particles, chip	netic plug and che os or fuzz. No action if findings a	fity 💆	4674	
		an and re-install m safety.	agnetic plug wi	th serviceable O-ri	ing b)	4674	
	c) Che to a	ck and service CS void over servicing	SD oil as neces g. Use Exxon 2	sary. Service slov 380 oil ONLY.	viy c) 🗾	4674	
7)	Motor engin	e and inspect for	leaks.		7) 2	4674	
d. Per	form ENGINE 4	oil service as follo	ows:			(-()	

WAIT AT LEAST 5 MINUTES AFTER ENGINE SHUT-DOWN

BEFORE REMOVING OIL TANK CAP, TO ALLOW TANK PRESSURE TO BLEED OFF. HOT OIL FROM TANK COULD CAUSE SEVERE BURNS.

WARNING:

EMERY WO		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 8 OF 31	INSPEC. CK	CARD NO. A01	
DC	-8			ACFT. NO. NIOSWP	STATION	DATE 4-21-95	
LOG PAGE 6913-13		INSTRU	JCTION		SIGN-OFF MECHANIC ONLY		
Note: Check engine oil tank level within 15 minutes after engine shutdown. If check has not been made within specified time, engine must be operated at approximately 75 percent N2 RPM for 1 minute immediately before checking tank oil level.							
1)	Check engine oil for proper level. Service engine as required 1) with Exxon 2380 oil ONLY. Record oil added on the log sheet.						
2)	Check engine oil cap "O" ring for serviceability, replace if 2)						
3) ု	Remove mai Ring. Reinst	n gear box chip all chip detector	detector. Che . (JT3D)	ck, clean, replace	0- 3)	A/A	
4)	Remove, ins	pect and reinst s (CFM56-2).	tall supply and	common scaver	nge		
	a) Remo	ve chip detector	s and magnetic	plugs (4 each).	a) 5	4674	
	(1)	Clean area ard before remova	ound chip detection.	ctor or magnetic p	lug -		
	CAUTION:	CHIP DETEC SHOULD BE ONE AT A TH	REMOVED	MAGNETIC PLUC AND INSTALLI	GS ED		

- (2) Remove chip detectors or magnetic plugs by depressing and turning one-quarter turn counterclockwise (CCW).
- (3) Remove units one at a time. Mark each unit, indicating the exact sump and engine from which it was removed.
- b) Inspect chip detectors and magnetic plugs (AD89-23-06).
 - (1) Examine each unit very closely, taking care not to displace any retained particles.
 - (2) If any chip detector or magnetic plug shows abnormal or suspect deposits, notify Foreman.
- 5) Perform starter magnetic plug check and service.

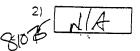
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		IRLIN	_		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 9 OF 31	INSPEC. CK	CARD NO. A01
	·	DC-8	.				ACFT. NO.	STATION	DATE
69	DG PAG	5.			INSTR	JCTION			OFF Jung
			a)	abno Maint	rmal metal p	articles, chips	ic plug and check , or fuzz. Naction if findings	otifv +	1674
			b)	Clear and s	n and re-install r afety.	nagnetic plug w	rith serviceable O-	ring b)	04674
			VP: c) (as cing	24674					
			Note:	eq. 350	uipped with ass	isted wet spline servicing is req	CEPT 105WP, e cavity starters f uired at any inter	2/N	
		6)	Perforn	n CSD	magnetic plug	check and servi	ce.		
				for at	onormal metal enance Control	particles, chip	netic plug and che os or fuzz. No ction if findings a	fity	14674
•••			p)	Clean and sa	and re-install m ifety.	agnetic plug wit	th serviceable O-ri	ing b)	4674
	·		c)	Check to avoi	and service CS d over servicing	SD oil as neces: g. Use Exxon 2	sary. Service slov 380 oil ONLY.	vly c)	4674
		7)	Motor e	ngine	and inspect for	leaks.		7) 5	417A
3.	POWE	R PLA	NT INSP	ECTIC	N				(61)
	a.	50 and	d 60 Serie	es					
		1)	Perform	No. 1	power plant ins	pection as follo	ws:	12 [NA
			a) į	nspect	t inlet and fan bl	lades for visible	damage.	81075	1 [[]

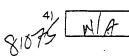
Check engine cowling, hinges, latches, and surrounding areas for general condition, leakage and security.

b)

EMERY WORLDWIDE AIRLINES	REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 10 OF 31	INSPEC. CK	CARD NO. A01
DC-8				STATION	DATE
LOG PAGE (69(3-13	INSTR	INSTRUCTION		SIGN-OFF MECHANIC ONLY	

- c) Visually inspect components, lines, hoses and drains for general condition, leakage and security.
- Visually inspect exhaust and reverser areas for leakage, damage and security.
- 2) Perform No. 2 power plant inspection as follows:
 - a) Inspect inlet and fan blades for visible damage.
 - b) Check engine cowling, hinges, latches and surrounding areas for general condition, damage, corrosion and security.
 - c) Visually inspect components, lines, hoses and drains for general condition, leakage and security.
 - d) Visually inspect exhaust and reverser areas for leakage, damage and security.
- 3) Perform No. 3 power plant inspection as follows:
 - a) Inspect inlet and fan blades for visible damage.
 - b) Check engine cowling, hinges, latches, and surrounding areas for general condition, damage, corrosion and security.
 - Visually inspect component, lines, hoses, and drains for general condition, leakage and security.
 - Visually inspect exhaust and reverser areas for leakage, damage and security.
- 4) Perform No. 4 power plant inspection as follows:
 - a) Inspect inlet and fan blades for visible damage.
 - b) Check engine cowling, hinges, latches, and surrounding areas for general condition, damage, corrosion and security.
 - Visually inspect components, lines, hoses and drains for general condition, leakage and security.





. AIRLINI	EMERY WORLDWIDE AIRLINES		REV. NO. 14	PAGE NO. 11 OF 31	INSPEC. CK	CARD NO. A01
DC-8				ACFT. NO.	STATION	DATE
LOG PAGE 10913-13			JCTION	(SIGN- MECHANI	
					CONTAIN	CONLY

 Visually inspect exhaust and reverser areas for leakage, damage and security.

b. 70 Series

Perform #1 power plant inspection as follows:

1) 24674

a) ENGINE INLET

Check nose cowl inlet area T-12 probe, acoustical panel, fan blades, spinner and fan shroud for unsatisfactory condition.

b) ENGINE EXHAUST

Visually check reverser exterior, turbine from strut, exhaust nozzle and ;lug, flame arrestor screen for security and L.P.T. stage 4 blades for obvious unserviceable condition.

2) Perform No. 2 power plant inspection as follows:

21 24674

a) ENGINE INLET

Check nose cowl inlet area T-12 probe, acoustical panel, fan blades, spinner and fan shroud for unsatisfactory condition.

b) ENGINE EXHAUST

Visually check reverser exterior, turbine from strut, exhaust nozzle and plug, flame arrestor screen for security and L.P.T. stage 4 blades for obvious unserviceable condition.

3) Perform No. 3 power plant inspection as follows:

31 24674

a) ENGINE INLET

Check nose cowl inlet area T-12 probe, acoustical panel, fan blades, spinner and fan shroud for unsatisfactory condition.

	EMERY WORLDWIDE AIRLINES DC-8		REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 12 OF 31	INSPEC. CK	CARD NO. A01
ĺ					ACFT. NO.	STATION	U-21-98
LOG PAGE 1		INSTRU	JCTION		<u>SIGN-</u> MECHANI		

b) ENGINE EXHAUST

Visually check reverser exterior, turbine from strut, exhaust nozzle and plug, flame arrestor screen for security and L.P.T. stage 4 blades for obvious unserviceable condition.

4) Perform No. 4 power plant inspection as follows:

4) 24674

a) ENGINE INLET

Check nose cowl inlet area T-12 probe, acoustical panel, fan blades, spinner and fan shroud for unsatisfactory condition.

b) ENGINE EXHAUST

Visually check reverser exterior, turbine from strut, exhaust nozzle and plug, flame arrestor screen for security and L.P.T. stage 4 blades for obvious unserviceable condition.

4. ENGINE PYLON EXTERNAL AREA INSPECTION

Perform No. 1 engine pylon external area inspection as follows:

a. 04503

- Inspect pylon exterior for visible damage, corrosion, deformation, fluid leakage, loose or missing fasteners, and security of all panels.
- 2) Inspect pylon to engine and wing-to-pylon fairings for general condition and security.
- b. Perform No. 2 engine pylon external area inspection as follows:

b. 04503

- Inspect pylon exterior for visible damage, corrosion, deformation, fluid leakage, loose or missing fasteners, and security of all panels..
- 2) Inspect pylon to engine and wing-to-pylon fairings for general condition and security.

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		AIRLIN	IES	REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 13 OF 31	INSPEC. CK	CARD NO. A01
		DC-8	3			ACFT. NO.	STATION KS EX	DATE
69	OG PA	3E / 3		INSTRU	JCTION		SIGN- MECHANI	
	c.	Perf	orm No. 3 eng	ine pylon exterr	nal area inspec	tion as follows:	c. [04703
		1)	Inspect py deformation security of a	i, fluid leakage	for visible , loose or mi	damage, corros ssing fasteners,	ion, and	2 / 3 - 2
		2)	Inspect pylocondition an	on to engine and did security.	d wing-to-pylo	n fairings for gen	eral	
	d.	Perf	orm No. 4 engi	ne pylon extern	al area inspect	ion as follows:	d	2
		1)	Inspect py deformation, security of a	lon exterior , fluid leakage, Ill panels.	for visible of loose or mis	damage, corrosi ssing fasteners,	ion, and	04307
		2)	Inspect pylo condition and	n to engine and d security.	l wing-to-pylor	n fairings for gene	erai	
5.	FUSE	LAGE	NOSE SECTION	NEXTERIOR INS	SPECTION			
	a.	Perfo	orm fuselage no	se section exte	rior inspection	as follows:	_	
		1)	inspect nose general cond	e radome area ition, and secur	for visible da ity.	mage, deformatio	on, 1)	14503
***		2)	Inspect nose security.	scoops for dar	nage, deforma	ition, condition, a	and 2)	7503
	٠	3)	Inspect pitot	tubes for obstr	uctions, damag	ge and security.	3)	04503
		4)	Inspect statio	ports for dama	ege and obstru	ctions.	4)	4503
	door(s) exter			nd pneumatic su ior for signs of tion and securit	damage, corr	d ground cooling osion, deformatio	air 5) <u>C</u>	4503
		6)	Inspect all Air	Data Temp pro	obes for damag	ge and security.	6) 0	4503
		7)	Inspect externation,	nal power fuse damage, and s	s and utility o ecurity of insta	utlet breaker par allations (RH only)	nel 7) 🕝	4503
		8)	Inspect wate damage, and	er and waste security. (Insur	panels for le doughnut ins	eakage, corrosio stalled).	n, 8)	4503

	EMERY WO	RLDWIDE	REV. DATE	REV. NO.	DAGENG		
	AIRLI	NES	3/20/97	17	PAGE NO. 14 OF 31	INSPEC. CK A	CARD NO A01
.	DC	-8			ACFT. NO.	STATION	DATE
L	LOG PAGE (9(3-13)		INSTRU	JCTION	7,4,1,0,1	SIGN: MECHAN	OFF IC ONLY
	9)	Inspect grou	and conditioned of installations (air inlet panel fo (RH only).	or damage, corros	ion, 9)	04503
	10)	Inspect LH signs of consecutive	and RH heat ex rosion, damage,	changers FWD deformation, b	and AFT outlets and inding, condition,	for 10) and	0503
	11)	Check radio	rack exhaust co	oling air outflow	·.	11)	m
	b. Gro		iC	,743 US			
	1)		1) (29503			
	2)	Compressor	belt wear and te	nsion		2)	04503
	c. Perf	orm NLG assem	bly inspection as	follows:			
	1)	Check for dar	maged or loose i	nstallations.		1)	4503
	2)	riacard for p	roper extension	raulic leakage and improper oleo extension (soper extension). Wipe down landing gear standing moistened with MIL-H-5606.			4674
		Check accum	ulators for prope	r pre-charge.		3)	X4503
	4)	Nose tires for 155 PSI; Serie	wear and propes 62 - 165 PSI).	er inflation (Se	ries 54/63/ 71 <i>/</i> 73	- 4) 2	4674
6.	TURBO CO	MPRESSOR CO	MPARTMENT II	NSPECTION			
	a. Inspe	ect turbo compres	ssor compartmen	nt as follows:			
	1)	Inspect turbo damage, signs of lines, and in	or corrosion, air	mpartment inte r and fluid leaks	rior for componer s, general conditio	nt 1) 🕝	4503
	2)	Inspect turbo of and security (5	compressors (fo 0 & 60 Series).	ur) for oil and	air leaks, conditio	2)	IA
						81013	

			1					
	EMERY WORLDWIDE AIRLINES		REV. DATE 10/18/96	REV. NO. 16	PAGE NO. 15 OF 31	INSPEC. CK A	CARD NO. A01	
ļ		C-8			ACFT. NO.	STATION SEA	DATE	
	LOG PAGE 6913-13.		INSTRI	JCTION	<i>70 \$7 \$2 </i>	SIGN- MECHANI		
	3)	heating reci	rculating valves	, pneumatic cr	st ducts, compres ossfeed valve, so general condition	000	04503	
	4)	4) Inspect turbo compressor magnetic plugs for contamination Refer to Maintenance Manual for contamination limits. Repany contamination to Maintenance Control. Install magnetic pwith new "O" ring and torque to 210 (± 10) inch pounds. Saf plug with lockwire (50 & 60 Series).						
	5)	Check oil level of left a check oil for routine to dra Exxon 2380 a	ally on-	RIGHT				
	. 6)	usually devel	k chin scoop ca lops near the ca lining clip (70 Se	p 6 o'clock pos	n. Failure of the o sition in the area j	cap 6) (0503	
	/. AIR CONI	DITIONING/FORV ION	VARD ACCESS	ORY COMPAR	TMENT			
	a. Pe	rform air condition ows:	ning/forward acc	cessory compa	rtment inspection	as		
	1)	Inspect interiorsecurity of in kind.	or of access do stalled compon	oor for signs of ents, signs of	corrosion, damag fluid leakage of a	ge, 1) (74,707	
	2)	Inspect radio	cooling exhaust	fan for conditio	n and security.	2)	2003	
8	3. FORWAR	D LOWER CARG	O COMPARTM	ENT		<u>ر د</u>	77,07	
	a. Per	rform cargo compa	artment inspecti	on as follows:				
	1)	Inspect cargo corrosion, cor	compartment indition, and secu	nterior for dama urity.	ige, cracks, signs	of 1) 🙋	74503	
	2)	seals, latches	s, rollers, spools	and attach br	T-components (i. rackets, hooks, and general condition	d	4503	

	EMERY WORLDWIDE AIRLINES			REV. DATE 11/15/95	REV. NO.	PAGE NO.	INSPEC. CK	CARD NO.	
		DC-8	3			16 OF 31	STATION	A01	
_ [OG PAC	3E -13		INSTRU	JCTION	MASWY	SIGN-OFF MECHANIC ONLY		
		3)				that are inoperativ		04503	
9.	cen a.			RNAL AREA IN					
Toblago died inspection as follows.									
		1)	deformation	iter fuselage ar i, corrosion, flu ral condition, ar	uid leakage o	or signs of dama f any kind, miss	ige, 1) (24503	
		2)	Inspect all corrosion, coscurity.	access panels leformation, flu	and service id leakage, ge	panels for dama eneral condition,	ge, 2)	०५०%	
10.	WING	GENE	RAL INSPECT	ION					
	а.	Perfo	orm center win	g external surfa	ce inspection a	as follows:			
		1)	Inspect cen deformation	ter wing exterr , fluid leaks, cor	nal surface fo ndition, and se	r signs of dama	ge, 1) 🙋	2407	
		2)	Inspect wing leakage, con	fillets for dam dition, and secu	nage, corrosion Irity.	n, deformation, fl	uid 2)	0503	
	b.	Perfo	rm RH wing ge	eneral inspection	n as follows:				
		1)	Inspect exterior leakage, go components	eneral conditio		eformation, signs curity of install		04503	
		2)	Inspect flap security.	wells and all	components	for condition a	nd 2) 🧷	24503	
		3}	Check fuel d and fluid leal	ump indicator fo tage.	or extension a	nd signs of damag	ge, 3) <u>C</u>	1450 3	
		4)	Inspect win condition, an	g mounted lar d security.	nding light fo	r damage, gene	ral 4)	74503	
		5)	Check press (open access	ure refueling ad door 555A).	apters for cor	ndition and securi	ty 5)	4503	
		6)	Inspect venobstructions,	ts, scoops, ar leakage, genera	nd exhaust pal condition, ar	ports for damag	e, 6) (C	4503	

EMERY WORL		514055			Ĭ					
ſ	AIRLINE		REV. DATE 11/15/95	REV. NO.	PAGE NO. 17 OF 31	INSPEC. CK A	CARD NO. A01			
<u> </u>	DC-8	i			ACFT. NO.	STATION SEA	DATE 1/-21-9			
LOG PAG 6913-	E 13		INSTRU	JCTION		SIGN-OFF MECHANIC ONLY				
	7)	Sump all tai	nks and check f	for contaminati	on.	7)	24674			
	8)	Inspect win	g mounted lift	transducer for	r damage, condit	ion, 8) [04503			
	9)	Inspect flig	ht control su d security.	mage, deformati	on, 9) [04503				
c.	 c. Perform LH wing general inspection as follows: 1) Inspect external surfaces for damage, deformation, signs of leakage, general condition, and security of installed components. 									
							0403			
	2)	Inspect flap wells and all components for condition and 2) security.								
	3)	Check fuel d and fluid leal	ump indicator f kage.	or extension a	nd signs of dama	ge, 3) [24103			
	4)	inspect win condition, an	g mounted land description of the security.	nding light fo	r damage, gene	ral 4) [04503			
** .	5) .	Check press (open access	ity 5)	04503						
	6)	Inspect ven obstructions,	ge, 6)	7503						
	7)	Sump all tani	ks and check fo	r contaminatio	n.	7)	24674			
	8)	Inspect fligh	it control surf I security.	faces for dan	nage, deformatio	on, 8)	04503			
d.	with d	ate the left ar ry film lubrica lowing proced	int - Felpro P/N	ing trailing edg 1 C200, CGF 1	e flap hinge cove 123., or equivaler	rs It.	·			
	Note:	Lube flap s	sliding covers.	Four (4) covers	per wing.					
	1)	With flaps FL	ILL DOWN.			1) [04503			

,									
		Y WOR		DE	REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 18 OF 31	INSPEC. CK	CARD NO. A01
		DC-8					ACFT. NO.	STATION	, DATE
LC	OG PAG /3 -	E / 3		·i	INSTRUCTION		1/1/05-001	SIGN. MECHAN	
		2)	COVE	er track v	side (top) spray (where exposed.	In areas on co	s applicable, into e over, depress cove es.	each 2) [04503
		3)	brea	n front ker poir h as pos	it). Use extens	e sliding cove sion and libera	r leading edges lly spray this area	(ice 3) [as g	0503
		4)	Wipe	off exc	ess lube from bo	oth sides of flap).	4) [24503
11.	MAIN WING	LANE	DING (GEAR (ECTION	(MLG) ASSEMI	BLY, WHEEL	WELL, DOOR, A	ND	
	a.	RH M	lain Lai	nding G	ear (MLG)				
		1)	Perfo	rm RH i	MLG wheelwell	and door inspe	ction as follows:		
			a)	Inspe and secur	installed compo	vell area for sign conents for ge	gns of fluid leaka neral condition a	ge, a) and	4503
			b)	inspe opera	ct wheelwell ser tion.	vice light for d	amage, security, a	and b)	04503
			c)	Inspe dama	ct wheelwell inn ge, deformation,	er and outer do condition, and	oors hinges, jams security	for c)	74503
			d)	Inspec condit	ct accumulators ion, and security	s for leaks, : /.	air charge, gene	eral d)	0403
			e)	Chevr	k and service sp on Hyjet IV-A. ved on DAC DM	The intermix	with Skydrol 500B x of these fluids	or c)	0503
		2)	Perfo	rm RH N	ALG assembly in	nspection as fol	lows:		
			a)		for damaged or			a) [4503
			b)	(see F	Placard for prop	er extension).	roper oleo extensi Wipe down landi vith MIL-H-5606.	on b) 2	4674
			c)	Check and s	for signs of corecurity of attach	Tosion, bogie tr ed installations	rim cylinder leakag	ge, c)	450 7
• *			d)	Check	brake wear indi	cator.		d) 🕜	4503

EMERY WORLDWIDE AIRLINES		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 19 OF 31	INSPEC. CK	CARD NO. A01
DC-	DC-8			ACFT. NO. P	STATION	DATE 11-21-98
LOG PAGE (0913-13		INSTRUCTION			SIGN-OFF MECHANIC ONLY	

e) Check tires for proper inflation pressure.

e) 24674

MLG TIRE/BRAKE SERVICE LIMITS

Aircraft Effectivity	Tire Size	Tire Pressure	Max Brake Pin Depth AD 94-06-10
DC8-50 Series Light Gear	44 X 16	170 PSI	11/16" P/N 154252-2
DC8-62/71 Light Gear	44 X 16	190 PSI	11/16" P/N 154252-2
DC8-62/63/73 Heavy Gear	44.5 X 16.5	195 PSI	3/4" P/N 2601412-1
			1/2" *P/N 2601412-2
*NOTE:	BRAKE ASSY IDENTI	FIED BY A PERMANE	NT YELLOW DOT

f) Brake lockouts/deboosters-apply brakes. Check lockouts/deboosters leakage and proper heights. Fill if below limits. Release brakes on completion of check. WA CAST?

g) Check tires and wheels for damage, general condition, and security.

9) 0403

f)

h) Lubricate bogie swivel, unlock cylinder and unlock linkage.

h) NA

- 3) Perform RH wing root inspection as follows:
 - a) Inspect all system components and lines for damage, leakage, general condition, and security of installations.

a) 04503

b) Check emergency air brake bottle for signs of damage, general condition, security of installation, and proper change. Service if required. (60 & 70 Series Only)

b) 04503

- b. LH Main Landing Gear (MLG)
 - 1) Perform LH MLG wheelwell and door inspection as follows:
 - a) Inspect MLG wheelwell area for signs of fluid leakage, and installed components for general condition and security.

a) 04503

EMERY WO AIRLII		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 20 OF 31	INSPEC. CK	CARD NO. A01
DC-	·8 				STATION	DATE 1/2/-98
LOG PAGE 69/3-/3		INSTR	UCTION			-OFF
	b) Insp oper	ect wheelwell seration.	ervice light for d	amage, security, a	and b) [04503
	c) Insp defo	ect wheelwell in rmation, condition	nner and outer on, and security	doors for dama	ge, c) [04503
	conc	ect accumulato lition, and securi nsion.	rs for leaks, ty. Flap lockou	air charge, gene t cylinders for pro	eral d) [per	04503
2)	Perform LH	MLG assembly i	inspection as fo	llows:		
	a) Ched	k for damaged or loose installations.			a) [04503
	exter dowr	nsion (see Plac	for hydraulic leakage and improper oleosion (see Placard for proper extension). Wipe landing gear strut piston with rag moistened with -5606.			24674
	c) Chec and s	k for signs of co security of attach	k for signs of corrosion, bogie trim cylinder leakage, ecurity of attached installations.		ge, c)	0503
	d) Check		k brake wear indicator.		d)	04503
	e) Chec	k tires for proper	inflation pressu	ıre.	e) [24674
	MIC	TIDE/DDAVE C	COMOC LIMIT	_	<u> </u>	-

MLG TIRE/BRAKE SERVICE LIMITS

			.~
Aircraft Effectivity	Tire Size	Tire Pressure	Max Brake Pin Depth AD 94-06-10
DC8-50 Series Light Gear	44 X 16	170 PSI	11/16" P/N 154252-2
DC8-62/71 Light Gear	44 X 16	190 PSI	11/16" P/N 154252-2
DC8-62/63/73 Heavy Gear	44.5 X 16.5	195 PSI	. 3/4" P/N 2601412-1
			1/2" *P/N 2601412-2
*NOTE:	BRAKE ASSY IDENTI	FIED BY A PERMANE	ENT YELLOW DOT

EMERY WO	RLDWIDE	REV. DATE	BEV NO			T	
AIRLII	NES	3/20/97	REV. NO. 17	PAGE NO. 21 OF 31	INSPEC. CK	CARD NO. A01	
DC-	8			ACET. NO.	STATION	DATE	
LOG PAGE 6913-13.		INSTRU	JCTION	7-74-5-6-7	SIGN- MECHAN		
	outs	e lockouts/debo deboosters leak . Release brake	age and prope	r heights. Fill if h	lock- n r	NA 04503	
	g) Chec and s	ck tires and wh security	eels for dama	ge, general cond	ition, g)	7503	
	h) Lubricate bogie swivel, unlock cylinder and unlock linkage. h)						
3)	Perform LH v	wing root inspect	tion as follows:		81075 L	MA	
	fluid a) C	2503					
·	b) Service aircraft hydraulic fluid, if necessary with Skydrol b) 500B, or Chevron Hyjet IV-A. This intermix of these fluids is approved on Douglas DMS/WPL-Z014.						
	gener	c emergency air al condition, se e. Service if req	ecurity of inst	for signs of dama allation, and pro is only)	ge, c) per	1/A 57503	
12. AFT LOWER	R CARGO/ACCI	ESSORY COMP	ARTMENT INS	SPECTION			
a. Perfo	rm cargo compa	artment inspectio	n as follows:				
1)	Inspect cargo security of ins	compartment i tallations.	nterior for dan	nage, condition, a	and 1)	4503	
2)	seais, latches,	compartment or rollers, spools, anage, corrosion,	attach brackets	T) components (i , hooks, and supp andition.	i.e. 2) \bigcirc	4503	
3) Inspect interior of AFT accessory compartment through panel 645 3) for general condition, signs of fluid leakage, corrosion, and security of installed components.						503	
4)	Inspect AFT abinding, condit	ccessory compa ion, and security	rtment, door for	damage, corrosic	on, 4) 🙋	503	
5)	Check all lights	s. Replace lights	that are inope	rative.	5) 🙋	4503	

									
	EMER'	Y WOF VIRLIN	RLDWIDE	REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 22 OF 31	INSPEC. CK	CARD NO	
ł		DC-8	3			ACFT. NO.	STATION	DATE	
L	OG PAG	13.		INSTRU	ICTION	NIOS WP	SIGN-OFF MECHANIC ONLY		
13.	AFT	FUSE	LAGE AND EM	PENNAGE ARE	A INSPECTIO	VI			
	a.		orm AFT fusel			e area inspection	n as		
		1)	Inspect exte	rior surfaces fo kind, deformation	r signs of dar n, general cond	mage, corrosion, dition, and security	fluid 1)	24503	
		age, 2)	04503						
		3)	Inspect cabir obstruction, a	n air outflow bu and fluid leakage	tterfly valve fo of any kind.	r damage, corros	sion, 3)	04503	
		4)	Inspect cabir damage, and	air outflow noz fluid leakage of	zle valve for o any kind.	bstruction, corros	sion, 4) <u>(</u>	24503	
		5)	Inspect tail sk	id for missing re	d paint, and ev	idence of damage	5)	7503	
		6)	and trailing e	al stabilizer surfa dges, damage tubes and static	and deformati	I condition of lead ion, security of fl	ding 6) (24503	
		7)	Inspect tail st	trobe navigatio	n light for dan	nage, condition, a	and 7) (74503	
14	HORIZ			LOWER SURFA					
	a.			Il lower surface in					
		1)	damage, defo	surface of RH rmation, fluid le neral condition,	akage of any	abilizer for signs kind, missing sta	of 1) c	4503	
	b.	Perfor	m LH horizontal	lower surface in	spection as fol	lows:			
		1)	damage, deto	suface of LH rmation, fluid le eneral condition,	akage of anv	bilizer for signs kind, missing sta	of 1)	4503	
5.	FLIGHT	COM	PARTMENT IN	SPECTION					
	Note:	Veri Mar	ify that all circunual/Maintenanc	uit breakers are e Manual specifi	set in accordations.	ance with the Flig	ght		
	a.	Open I	Radio Rack Pan	el access panel:)	a. 🕥		
	b.	Perforr	n flight compart	ment area inspe	ction as follows	:	<u>\\ \</u>	075	

EMERY WO	RI DWIDE	DEV DATE						
AIRLIN	NES	3/20/97	REV. NO. 17	PAGE NO. 23 OF 31	INSPEC. CK	CARD NO. A01		
DC-	8	·		NIGSWY	STATION	DATE		
LOG PAGE 69 13-13		INSTRU	JCTION	1014501	SIGN-OFF MECHANIC ONLY			
1)	Inspect all ir damage, co	nstrument panels andition, and sect	s, instrument, s urity.	witches and gauge		81075		
2)	Inspect fligh leakage, and	nt compartment I visible defects.	windshields f	or security, corro	sion, 2)	81075		
3)	Inspect fligh condition and	Inspect flight compartment seats, cushions and seat belts for condition and security.						
4)	Inspect mag correction ch	Inspect magnetic compass for condition and security. Check 4) Correction chart for legibility.						
5)	Check opera	Check operation of fire warning system.						
6)	Check flight OUT.	data recorder.	Turn switch	ON, lights should	go 6)	31075		
7)	Inspect flight security, and	compartment d other discrepand	emand oxygen cies.	masks for condit	ion, 7)	\$1075		
8)	escape tapes	compartment and First Aid and broken sea	Kit) for genera	equipment (includ	ling 8)	\$1075		
	Note: If Fi	rst Aid Kit seal is	s broken compl	y with FCD 25-15.				
9)		compartment lig		for condition, secu		1075		
10)	Inspect landir	ig gear lever d	own limit swite	ch for condition a	nd 10) 🕡	100		
11)	security. Inspect cockpi	t floor panel hea	iter for conditio	n and security.	11) 🛛	1075		
12)	cushions, rad	nd second obs io panels and lity and signs of	oxygen equic	s, seats, seat bel ment for condition cies.	ts, 12) 8	1075		
13)	Inspect radio rack interior for signs of fluid leakage, corrosion, damage, general condition, and security of all installed components.							
14)	ioi darriage,	ency portable o general condi security of stowa	tion, proper	regulator, and ma pressure, leakag	sk 14) (1075		

AIR	VORLDWID LINES	E REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 24 OF 31	INSPEC. CK	CARD NO
<u> </u>	C-8			ACFT. NO.	STATION	DATE
LOG PAGE 6913-13		INSTR	UCTION	WISSUP	SEA SIGN.	1/-2/-98 -OFF
1		ect the PBE for the ate a damaged or exact ation characteristics. Dents, cracks holes	xpended unit,	tampering or los	s of	81035
	b)	Dents, cracks, holes The sealed bag sh ensure seal. Loss becomes soft and package or if humid from a blue color to	ould be perion of vacuum is no longer con ity indicator inspink.	odically inspected indicated if the forms tightly to side bag has chan	i to bag the ged	
	CAU	REPLACEME	, CARE MUST	OM CASE DURI BE TAKEN DURI TO CASE TO AVO	NG	
16		t circuit breaker pand curity of installation.	els for damage	, general conditi	on, 16)	51075
17,	. •	t cockpit door and lo on, and security of in:	stallation.	•	[8	1075
18)		aircraft airworthines cense for condition, le	s certificate, a gibility and cur	aircraft registration	on, 18) S	10-75
19)	Perfrm engines	Operational Check (A	ural Check) of	both Igniters for	all	
	, , t	Push continuous ignificate for only the enging witch for engine became to position "A" to override position and belect to position "B neard.	ne being check ing checked i and place igni d RH igniter bli	ked. Select ignit n Pilot's overheater ter override switch	er <u>K</u> ad :h	108
	~	urn ignition override eack to "Both" and reaker on AC bus.	switch off, pl pull continuo	ace igniter switc ous igniter circu	h b) 8 /	1075
20)	Perform	courier compartment	inspections as	follows:		

Inspect courier seats, seat belts and cushions for

security, damage, and visible defects.

a)

:							
EME	RY WOI	RLDWIDE IES	REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 25 OF 31	INSPEC. CK	CARD NO.
	DC-8	3			ACFT. NO. NIGSWP	STATION KSEA	DATE
LOG PA 6913-			INSTRU	JCTION	170143601	SIGN MECHAN	
		b) inspe	ect courier der rity, and other	mand oxygen discrepancies.	masks for cond	ition, b) [81075
		c) Inspe secu	ect evacuation rity.	rope for g	eneral condition	and c) [8108
•	21)	Check oxyg	en cylinders for	proper pressu	res.	21)	24674
		a) Crew	/ fixed Cylinder:	1500-1800 F	PSI at 21°C (70°F)	- 00
		b) One (70°F	crew portable c	ylinders: 1800	0 ÷ 0-20 PSI at 2	21°C .	
		c) Couri 1800	er and lavator + 0-20 PSI at	y portable cy 21°C (70°F)	rlinders (if instal	led):	
	22)	Close Radio	Rack access pa	nel.		22)	81075
	23)	Verify aircra	ft battery voltag	ge, minimum 2	2 volts.	23)	31035
	24)	compartmen	pection of all t for general (&P Ch. 1 - Man	condition and	nuals in the fl presence on bo	ight 24)	4674
16. CAB	IN ARE	A INSPECTION					
 a.	Perfo	rm cabin area i	nspection as fo	llows:	· ·		
	1)	Inspect forw seal, and cle and signs of	earview windov	door latches, v for damage,	door jambs, hand corrosion, bindi	dle, 1) (8108
	2)	Check condit	ion and security	of forward se	rvice door.	2)	1035
	3)	Check galley security of in	equipment for stallation.	damage, corro	esion, condition, a	and 3)	1075
	4)	Check forward servicing, evi	d lavatories flu dence of leakag	shing system t e. (Ensure dou	for proper operations in the propertion of the properties of the p	on, 4)	1075
`.	5)	Check all w damage, prop	aste container er latching oper	access and ration and prop	disposal doors er sealing.	for 5)	1075

EMERY WO	VES	REV. DATE 11/15/95	REV. NO. 14	PAGE NO. 26 OF 31	INSPEC. CK	CARD NO.		
DC-	8			ACFT. NO. NIGS WP	STATION KSEA	DATE		
LOG PAGE 6913-13.		INSTRU	JCTION	10195 201	SIGN- MECHAN			
6)	Check cond	ition and securi	ty of lavatory	doors.	8108	NA		
7.)	Check ceilir defective re	ig and side pan pairs, etc), co	els for damage andition, and s	e (i.e. for holes, t ecurity of installa	ears, 7) [81075		
8)	Check cond	ition of all place	8) [3	81075				
. 9)	Check condi rollers, ballm	ition, operation nats and restrain	ocks, 9)	81075				
10)	compartmen door seal a	Inspect cargo door area and hinge from inside of cargo compartment for damage and proper lubrication. Check cargo door seal and seal depressor for any damage, tears, or irregularities.						
11)	Inspect carg	o door pins on. Check for s	Assure that alsecure attachm	I are present and ent to door.	d in 11)	1075		
12)	inspect cargo general cond	o door, vent do ition, security a	or (if installed) and operation.	and components	for 12)	31075		
13)	Check overw	ing exit doors (2LH/2RH) for s	security.	13)	1075		
14)	Check condit	ion and security	of AFT service	e door.	14)	51075		
15)	Check all cinoperative.	argo comparti	ment lights.	Replace lights	if 15)	1075		
b. Test	External Lightin	g						
1)	Test operation	n of the anti-co	llision lights.		1) 🛭	1075		
2)	Test operation of the position and strobe lights with POS/STROBE LTS/OFF/POS/BOTH switch in both position.					la		
3)	Test operation	n of the wing na	avigation lights	•	3)	1075		
4)	Check operati taxi lights.	on of fixed and	retractable lan	ding lights and N	LG 4) 🛭	1075.		

	EMERY WORLDWIDE AIRLINES	REV. DATE 10/18/96	REV. NO. 16	PAGE NO. 27 OF 31	INSPEC. CK	CARD NO.
	DC-8			ACFT. NO.	STATION	DATE
-	LOG PAGE	INSTRI	JCTION	NIGERAP	SIGN- MECHANI	1/-2/-9 OFF
	5) Check on 17. VISUAL INSPECTION (eration of logo ligh			5)	B1075
	Perform a visual inspect date, hydrostatic test date, hydrostatic test date than ninety (90) days tire unit. If any items are number (S/N), and due EWA Emergency Equiporation The EWA Emergency In completed. If	ate, or date of ma ne is remaining, re replaced, enter the date on this inspe nent Tag (MEO98)	nufacture, and replace new units par ection verification the installed tag (MEO98)	record below. If ace with a service t number (P/N), so form and install serviceable unit.	less eable serial II an	
	all information contacted if air	this tag is not prese completed. EWA de is needed in con aul Period = 18 Mo	Aircraft Record	S Department ma	y be	NA
	S/N	Due Da	mergency Equi	pment Tag	2467d	
	P/N_DK1669	verhaul Period = M EWA E GYB Due Da	mergency Equip	pment Tag	b. J	4674
	. с. С.V.R. Batt Оve P/N <u>DK/Ф</u> S/N <u>DJ3</u> (B EWA E	rcury 24 Months mergency Equip te34	oment Tag	c. 2	4674
	d. Halon Fire extinguing Cockpit P/N 354 S/N 5324 Galley/Courier	EWA Er ΣφιDue Da	mergency Equip te/スース。	oment Tag ダダダ	d. 2	4674
	P/N <u>369</u> S/ND-5550	EWA Er 23øDue Dat	nergency Equip te	•	_	

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EMERY WO		REV. DATE 10/18/96	REV. NO. 16	PAGE NO. 28 OF 31	INSPEC. CK A	CARD NO. A01
DC.	-8		·	ACFT. NO.	STATION	DATE 11-21-78
LOG PAGE		INSTRI	UCTION		SIGN- MECHANI	OFF
					WEGNAN	CONLY
	T Overhaul P v 607690	<i>*</i> ~		Equipment Tag	e. [24674
S/N	150C-24	160 Di	ue Date <u>/-6</u> -	-99		
f. P.E	3.E Overhaul P	eriod = 10 Year	S	•	f. 7	21157
(i) P/N	1 <u>802360-</u>	<u>-14</u> Ev	VA Emergency	Equipment Tag	<u> </u>	176 74
S/N	1_NSN	Du	ie Date 6-	99		
g. Life	raft(s) (if installe	ed) - Overhaul P	eriod = 36 Mon	ths	<u></u> а. Г	2111-716
<u>No</u>		each for overw				14674
1.		4/65RBEV				
	SIN TP03	7	ie Date <u>ಎ</u> –ವಿ			
2.	P/N R B/B2	4105 RBEV	VA Emergency	Equipment Tag		
			e Date シーン			
h. Life	Vests - Overhau	l Period = 36 M	onths		—- h. [-]	11/2/
Cod	kpit/Courier					7617
) P/N	3505-10	<u>5</u> EWA E	Emergency Equi	ipment Tag	-	
S/N	E93843	9Due Da	ate 1-8-2	<u> </u> భత్త		
2 P/N	\$-11150-21	/	Emergency Equi	pment Tag		
S/N	85-3295	YDue Da	ate <u>3-/3-</u>	2\$01		
3 P/N	3545-105	EWA E	mergency Equi	pment Tag		
S/N	E096280		ate //-/4-9			
4 P/N	5-11156-24	Ø EWA E	imergency Equi	pment Tag		
S/N	85-3366		ate <u>3-/3-</u>			
	3545-145		mergency Equi			
S/N	E494316		ate 1/-18-	_		
Ç P/N	3-11156-2		mergency Equi	,		
S/N	85-3316	_	ite <u>4-/3</u> -			
7 P/N_	35\$5-105		mergency Equi		_	
S/N_	E\$37913		ate 1/-6-9			
8 P/N_	3545-105	-	mergency Equip			
S/N_	E+3868	_	ite 11-18-	_		
f.)@	8 \$23 \$4.					
-	NSN	٠ 2-	-8-2 BBY			

	EMERY WO	DEL DIVIDE	DEV DATE		T		
	AIRLI	NES	3/20/97	REV. NO. 17	PAGE NO. 29 OF 31	INSPEC. CK	CARD NO A01
	,	-8			ACFT. NO.	STATION	DATE
	LOG PAGE 6913-13.		INSTRU	JCTION		SIGN-	
	i. Ox	ygen Cylinders				i F	
	Loc	Galley PINZ SIN_ Gralley PINZ SIN_ SIN_ SING CKPC PINZ	yinders. 8\$13\$7-\$ 562579 8\$13\$7-\$6 8\$13\$7-6 212\$71 7\$\$\$-\$1A	EWA Emergence Due Date Due Date EWA Emergence Due Date Due Date BE234 EWA Emergence	ency Equipment 77-99 ency Equipment 7 - 2000 ency Equipment T 0-1-99 ency Equipment T	ag ag	296 79
	•	S/N _j				ag	
		DC-8 DC-8 ACFT. No. STATION DATE					
	1. 2. 3.	te: If no batte Remove the a Remove the the aircraft. Install the batte	aft perform the fery is installed in ircraft (in-use) becattery installed ery from step 1 i	ollowing: the SPK, N/A tattery. in the SPK posento the SPK pos	this step. sition and install it	24674	NA
5	MAIN CARC	SPK position.					•

Verify that the warning lights on the flight engineer lower left instrument panel are operational (use press-to-test).

Door Warning Light Functional Check:

1)

	EMERY WORLDWIDE AIRLINES		REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 30 OF 31	INSPEC. CK	CARD NO. A01
	DC-8	3			ACFT. NO.	STATION	DATE
LOG PA	GE		INSTRU	JCTION	NIPSUP	SIGN- MECHANI	
	2)	Open main switches.	cargo door e	enough to gai	n access to wa	rning 2) [NA
	3)	Check secur	ity of switch elec	ctrical connecto	ors.	3) F	- A
	4)	Close all dod are out.	ors except main	cargo door. Cl	neck that all door I	ights 4) [K(A
	5)	Manually ope light remains	erate cargo door on. If light goes	jamb switch. s out, replace th	Verify that the war ne door latch switc	nina 5)	HA
	6)	Manually ope light remains	erate door latch on. If light goes	tube switch. \s out, replace th	erify that the war	ning 6) (h. 81076	HA
	7)	Manually ope warning light	erate both switc goes out.	hes simultaned	ously. Verify that		NA
b.	With inspe	access door to	cargo door cont	roi vaive open,	perform the follow	ving 8/017	,
	1)	Verify the instantial handle to corbe replaced.	stallation of clip atrol valve. If cli	, P/N S12052 p is missing, b	247, used to stow ent or broken, it n	T- 1) nust 8(0)	N/A
	2)	secured, the	access door, :	such that wher ity above the	d in a position on n door is closed control valve sh	and 1	
c .	Verify agree	that the door f	function for door ating instructions	up and door of located in the	down positions are control valve box.	e in c.	(A
d.	Close	and latch the c	argo door.			d. [Ila
e.	laten	iock pins are e	With hydraulic engaged 3/16 ir on its shoulder.	nch minimum :	rized, verify that and that the No red.	the e 6 8 10 75	MA
f.	Check links.	the condition Repaint as requ	of the alignmen uired.	t paint stripes	on the latch supp	oort f.	NA
g.	positio	that the externa in stripe and the in arrow.	al lock pin indica e stripe on the la	tor handle is ali tch drive fitting	gned with the lock s aligns with the lo	ted g.	NA

	••								
	E		IRLI	ORLDWIDE NES	REV. DATE 3/20/97	REV. NO. 17	PAGE NO. 31 OF 31	INSPEC. CK	CARD NO. A01
İ	DC-8		-8			ACFT. NO.	STATION	DATE //-21-9	
. [69	G PAG (3-1			INSTRU	JCTION		<u>SIGN-</u> MECHANI	
	20.	FINA	LINS	SPECTION - GEN	NERAL WALKA	ROUND			
		a.	En ins	sure all exterior talled.	cowlings, pane	eis, doors and	fairings are sec	urely a.	24674
		. <u>.</u>	Ē.	Bure ell interior po	aneis and doors	are sacurely in:	stailed.	b.	14/76
	21.	PAPE	RW	ORK COMPLETE	•			اعط	7617
		a.	Eng	gine oil added do	cumented in log	book.		a. 🔇	04503
		b.		nove gear pins, c				b. [14674
		C.	Con	inspection pape opleted and accor	erwork, non-roi unted for.	utines, DMIs,	parts tags prop	perly c.	4674

Sign Airworthiness Release in log book.

d.

EMERY WORLDWIDE AIRLINES NON-ROUTINE MAINTENANCE FORM

LOG PAGE NO. 69/3-13

		· ·		· · · · · · · · · · · · · · · · · · ·		
	AFT NO	DATE	s	TATION	TYPE CHECK	PAGE
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ivere	Tly:			IIZ mais	broke IAW	70425
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ITEM #	[DISCREPANCY		CORR	ECTIVE ACTION	MECH
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	<u> </u>	145	<u> </u>		·	
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. //.	0	3/N OFF		P/M UN	3/1/10/1	POS
ITEM #		NECREDANCY	·			T
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					-	1 ,03

CRAFT MAINTENANCE LOG	WORLD FIDE	Q.C. 10 RRXA	6913-13	ACFT. NO. ACFT.	TYPE TO
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NO. SOURCE DISCREPANCY 1 (P) M RADAR UNSTABLE IN PAINTS GRA, 2 IN Ref. WRIF B2935	1 3 0/1 5. SM 0/2 R.C. 0/3 J.T.O. 1 TURNS 1. D. W.C. 2. W.C. W.C.	1782 1824 151, 1816UZZ 713	CORRECTIVE ACTION WX FAME F	DATE STA 11-2518 GC	EMPN MECH SIOTS 14.
	11214, F.O.S 3. here MA MA MA 11-6-99. 4. H-11	Kerolace t controller 130-41-2 Transferra 099	F.C.s windshie Yreset CBs This clears A of to NoME	COPS Chek's war NELHSULLY Place	24674 -d Remar 24674
NO. PARTHOMENCLATURE Window heat Control 1- 1	PART NO. OFF	SER. NO. OFF 7821371	PART NO. ON 1339-1	SER. NO. ON 3 881 B 1411	Pos.
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			EMER'	Y WORLD	WIDE AIRLI	NES IVE		, : : N
				-		No	52-6	
			*** **********************************					<u> </u>
_						Issue Date:	12/08/98	:
						Rev <u>0</u>	riginal	
						Task Code	e: <u>852346</u>	
	Title: Roser	nbalm and N	lonerch Mair	ı d	Reference	: N/A		
		a Door Insp						
	Compliance							,
	Requested:	As schedul	ed by Mainte	nance	Approved	by: (5 (/	3	<u>}</u> . (
		Planning				3		
	in die de			: :			• (
	Manpower:	3 hrs/aircra	ft			Affected: Air		i
	•				with Roser Cargo Doo	ibalm and Moi rs	narch Main	. •
	Priority: N/A					· -		
					4.4			
	Publications	Affected: N	one	€ •€ -25			•	
	Weight Chan	ge: N/A						.:
						•		
4								1
	GENERAL:			•		:	i	**
	This F	CD visualiv	inspects the	main car	ro deor frame	s in the area	of the unner	four
			nount bolts			.s in this disa	or are apper	
	Noo	11				Mari Canada	· · ·	<u> </u>
	Note:	further	disposition.	or a cre	ick, contact	Mx. Control	immediately	for
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	ME012 (Rev. 4	01/15/96) 2					Page 1 of	2
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AIP TAFT MA	AINTENANCE LOG	E/	57	Q.C.					(多的	9
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100000	MAIN CARGO PU	OR INSPECTION	1 10	056	£675	NOTE	3	برم کمیا کرون درسه سرسا	12/5/8		2065
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CHECK C/W: STATION:				AIRCRAFT TIME / CYCLES		-			T -	NS READOL	JT .
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GMT TIME:	CERT. NO.:	LANDINGS 2119		IIS PAGE	-[]	LANDINGS	211	99	1_1()		
CHATTANE;	AUTH SIG.:		7.20 FL	T. Hns.	\mathcal{N} :	TOTAL A/C FLT. HRS.	1 - 0	15.	<u> }{</u>		
DISC OR MAINT AC	TION CARRIED FWD TO:	FLT, IIRS. 6581	\cup	IS PAGE	1)	FLT. HRS.	6584	17:39	1		
	TION CARRIED FWD TO:	DOOK CHANGED NEW	LOG PAGE	ENO:		CAPTAIN'S	SIGNATUE		I	·	

EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

No	52-6		
issue Date:	12/08/98	,	
Rev	Original		

Task Code: 852346

Title: Rosenbalm and Monarch Main

Cargo Door Inspection

Reference:

N/A

Compliance

Requested: As scheduled by Maintenance

Planning

Approved by:

Manoower: 3 hrs/aircraft

Equipment Affected: Aircraft equipped with Rosenbalm and Monarch Main

Cargo Doors

Priority: N/A

Publications Affected: None

Weight Change: N/A

GENERAL:

This FCD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Note: Upon identification of a crack, contact Mx. Control immediately for further disposition.

EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

Page 2 of 2 No. 52-6 Rev. <u>Original</u>

ACCOMPLISHMENT INSTRUCTIONS:

Note:	To gain access to some Rosenbalm cargo door locks center a require removal. To gain access to Monarch cargo door lock require removal.	
Note:	Clean frames as required to facilitate inspection.	
1.	Using a mirror and flashlight inspect main cargo door frames in the area of the upper four latch attach bolts for cracks and corrosion at each latch location.	86746
	Note: Pay special attention to frame radiuses located at latch mount bolt nutplates.	
	Note: Inspection mirror should be inserted into access area directly above latch.	
	Cracks Noted (Circle) YES NO Location of cracks:	
2.	Report all defects to Mx. Control and re-install lock covers (if removed).	M 57809.
3.	Make a log book entry indicating compliance with this FCD and complete the section below.	M 56746
	Acft. 4904 Date: 12/12/58 Sta.: 1084 Emp. # 86746 Log Page # 7226-13	

AIRCR IAINTENANCE LOG	MATHEMAN ENE			χ.	34 (0)
AIR-0092 (10 o U.S.A.	WORLDWIL A DE CHIMIY	. 7		CFT, NO.	. TYPE
	GMT BLOCK GMT		FUEL DATA		-8.62 GO DATA
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3		-6 P-17-1			
4	Mainta	16 DN W			
L DEPT. DELAY TRAIN. FLTS. DELAY CODE LDGS STATION 1	OIL ADD A (D	CREW EMP#	T.O. LDG A/P	CREW	5110 #
1 : SOUR EDGS STATION 1	2 3 1 APU A/P			CHEW	EMP#
2 :	- × 52				
3 :					
4 :					
	EPANCY		ECTIVE ACTION	DATE STA	MECH
1 Sompi V WITH F	CD \$2-6 Bosy/	Bozy/Monarch Tima BII AM	With FCID 52	1-6 12/12/88/TOAY	570,0°
Monarch Door Inspt		Bory Monarch	Door Ins. No	o Pafects Nota	+415
2. P(M) Completel MAA	A1-C1 22 - Q11 ' 67	7 cha 311 gh	Wab-2232 825	188746	(5)400
Main Cargo Door Insp	M/_3/\33_04\01	2. Comploilui	th MA-111-	5.737415/11/24/KDWF	73.4776
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4. P/M					
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PATE 10 10 C	LANDINGS 1430-7	LANDINGS O	TOTAL LANDINGS /ロシ/		3-0/31.
DATE: 12-12-48 CERT. NO.			LANDINGS 1430	//	
GMT TIME: AUTUSIG: 7	PHEV. A/C TT +. HRS. SO/120	FLT. HRS.	TOTAL A/C FLT. HRS. 50/20	33 8-	
10x(1/2	1.7.10		1 1000000		
DISC, OF MAINT, ACTION CARRIED FWD TO:	BOOK CHANGED MEM FO	GPAGENO.	CAPTAIN'S SIGNATURE		

	No. 52-6
	Issue Date: <u>12/08/98</u>
	Rev. <u>Original</u>
	Taak Oode: 852346
<u>Title:</u> Rosenbalm and Monarch Main Cargo Door Inspection	Reference: N/A
Compliance Requested: As scheduled by Maintenance Planning	Approved by:
Manpower: 3 hrs/aircraft	Equipment Affected: Aircraft equipped with Rosenbalm and Monarch Main
Priority: N'A	Cargo Doors
Publications Affected: None	
Weight Change: N/A	
•	

This FCD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Note: Upon identification of a crack, contact Mx. Control immediately for further disposition.

GENERAL:

Page 2 of 2 No. 52-6 Rev. <u>Original</u>

ACCOMPLISHMENT INSTRUCTIONS:

Note:	To gain access to some Rosenbalm cargo door locks center require removal. To gain access to Monarch cargo door lock require removal.	•
Note:	Clean frames as required to facilitate inspection.	
14010	Olean maries as required to recilitate inspection.	
1.	Using a mirror and flashlight inspect main cargo door frames in the area of the upper four latch attach bolts for cracks and corrosion at each latch location.	153473
	Note: Pay special attention to frame radiuses located at latch mount bolt nutplates.	
	Note: Inspection mirror should be inserted into access area directly above latch.	
	Cracks Noted (Circle) YES NO	
	Location of cracks:	
2.	Report all defects to Mx. Control and re-install lock covers (if removed).	M5 3443
3.	Make a log book entry indicating compliance with this FCD and complete the section below.	M 53443
	Acft. N993 CF Date: 12-15-78 Sta.: KISH Emp. #_ 57473 Log Page #_5753-07	

AIRC MAINTENANCE LOG = EME Empsay o U.S.A. WORLDWI. 5763-09 ACFT, NO. AIRUNE. N 993 STATION OM TO FLT DATE BLOCK GMT FLT. FROM FUEL DATA OUT Liouns 2,441 OFF 12/12/98/XIND/XIAH ON Houns UPLIFT(USG) 5335 DEPART (LBS) ARRIVAL (LBS) 225 0146 GAL'S CARGO 34031 2315 0128 2413 MAIL 58.2 27,2 DEPT. DELAY
DELAY COD TRAIN, FLTS, OIL ADD LDGS A/P STATION CREW EMP # APU T.O. LDG A/P CREW 3:12 Or, EMP# οЛ J. MYERS 59849 012 R. SPEAR 78910 013 M. ALOR POE 00363 NO. SOURCE DISCREPANCY NO. CORRECTIVE ACTION P (M) SCRUZEd DATE STA MEGH SPUILE 02 ee P / M 4. P/M P / M G NO. PART NOMENCLATURE PART NO. OFF SER. NO. OFF PART NO. ON SER, NO. ON POS. クブラウ AIRWORTHINESS RELEASE AMCRAFT TIME / CYCLES INS READOUT STATION: PREVIOUS LANDINGS 1-DIST. 2-DIST. TOTAL 3-DIST. **LANDINGS** HISPAGE CERT, NO.: 2239/ LANDINGS 22392 GIAT THAE: AUTH SIG.: PREV. A/C FLT. HIDS. TOTAL A/C FLT, HAS. FLT.HRS. 60858 53 DISC. OF MAINT, ACTION CATTRIED FWD TO: BOOK CHANGED NEW LOG PAGE NO: CAPTAIN'S SIGNATURE

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EMERY WORLDWIDE AIRLINES FLEET CAMPAIGN DIRECTIVE

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	Title:	Rosenb	alm and	Monerch	Main		Refere	ncë:	N/A	. •	:
		Cargo	Door Ins	pection			1.33-97	. i			
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	4	RAL: This FC cargo d	O visua loor late	h mount	bolts at	each.la	tch locatio	n.			
	GENE!	RAL: This FC cargo d	O visua loor late	h mount	bolts at	each.la	tch locatio	n.			
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	Note: To gain access to some Rosenbalm cargo door locks center actuator, cover may	¬ -
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	Note: Clean frames as required to facilitate inspection.	丁
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	1. Using a mirror and flashlight inspect main cargo door frames in 1 6363	
	the area of the upper rout later attach poits for cracks and	15
	corrosion at each latch location.	
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	Note: Pay special attention to frame radiuses located at	
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	latch mount bolt nutplates.	
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	directly:above latch.	
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	2. Report all defects to Mx. Control and re-install lock covers (if M/ R/2)	**
	removed). MAX Control and re-install lock covers (if M6362)	
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	Make a log book entry indicating compliance with this FCD and M63632	
	complete the section below:	
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	Rev. <u>Origi</u>	naı	•
	Task Code:_	852346	
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Reference:	N/A	,	
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Approved by:			,

Equipment Affacted: Aircraft equipped

with Rosenbalm and Monarch Main

Cargo Doors

<u>Title:</u> Rosenbalm and Monarch Main Cargo Door Inspection

Compliance >

Requested: As scheduled by Maintenance Planning

Manpowert 3 hrs/aircraft

Priority: N/A

Publications Affected: None -

Weight Change: N/A

GENERAL:

This FCD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Note: Upon identification of a crack, contact Mx. Control immediately for further disposition.



Page 2 of 2 No. 52-6 Rev. <u>Original</u>

ACCOMPLISHMENT INSTRUCTIONS:

Note:	To gain access to some Rosenbalm cargo door locks center a	
	require removal. To gain access to Monarch cargo door lock require removal.	s, lock cover will
Ē: ;		
Note:	Clean frames as required to facilitate inspection.	÷
1.	Using a mirror and flashlight inspect main cargo door frames in the area of the upper four latch attach bolts for cracks and corrosion at each latch location.	16365
•	Note: Pay special attention to frame radiuses located at latch mount bolt nutplates. Note: Inspection mirror should be inserted into access area	
	directly above latch. Cracks Noted (Circle) YES NO Location of cracks:	
2	Report all defects to Mx. Control and re-install lock covers (if removed). Make a log book entry indicating compliance with this FCD and complete the section below.	M/6365
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	No. 52-6
	Issue Date: 12/08/98
	Rev. <u>Original</u>
	Task Code: 852346
Reference:	N/A
Approved by:	Constant of the second
Equipment Af with Rosenba Cargo Doors	fected: Aircraft equipped Im and Monarch Main

Title: Rosenbalm and Monarch Main

Cargo Door Inspection

Compliance

Requested: As scheduled by Maintenance

Planning

Manpower: 3 hrs/aircraft

Priority: N/A

Publications Affected: None

Weight Change: N/A

GENERAL:

This FCD visually inspects the main cargo door frames in the area of the upper four cargo door latch mount bolts at each latch location.

Upon identification of a crack, contact Mx. Control immediately for Note: further disposition.

Page 2 of 2 No. 52-6 Rev. <u>Original</u>

ACCOMPLISHMENT INSTRUCTIONS:

Note:	require removal. To gain access to Monarch cargo door locks require removal.	
Note:	Clean frames as required to facilitate inspection.	
1.	Using a mirror and flashlight inspect main cargo door frames in the area of the upper four latch attach bolts for cracks and corrosion at each latch location.	26275
	Note: Pay special attention to frame radiuses located at latch mount bolt nutplates.	
	Note: Inspection mirror should be inserted into access area directly above latch.	
	Cracks Noted (Circle) YES NO Location of cracks:	
2.	Report all defects to Mx. Control and re-install lock covers (if removed).	M26275
3.	Make a log book entry indicating compliance with this FCD and complete the section below.	M 26275
	Acft. \$3244 Date: 12-12-48 Sta.: PDX Emp. # 64504 Log Page # 6423-11	