## \*\* CDC Document Control Field Repair/Alteration Released \*\* \*\* Release To: Cirrus Approved - Revision: A.3 \*\* \*\* Release Date: 2016-06-29 11:33:29 CDT \*\*

## FIELD REPAIR / ALTERATION

<b>CIRRUS DI</b>	ESIGN CORP	ORATION
------------------	------------	---------

AUTHOR:	David L Johnson		CHANGE CLASS:	MINOR	REPAIR#	CM00007163
Name: TAIL STRIKE		REASONS AND REMARKS:  TAIL STRIKE DURING LANDING CAUSING DAMAGE TO THE  AFT VERTICAL SPAR AND FUSELAGE SKINS.		Models Affected	ASN#' Impacted	
Description:				SR20	1432	
TAIL STRIKE				n/a		
DOCUMENT	IT TYPE: FIELD REPAIR		REFERENCE FSR 2726		n/a	
Items Affected by Repair/Alteration		REVISION	DESCRIPTION/TITLE		n/a	
					n/a	

## NOTICE:

THE REPAIRS HEREIN ARE PROVIDED SPECIFIC TO DAMAGE REPORTED TO CIRRUS AIRCRAFT. OTHER DAMAGE MAY BE PRESENT AND IT IS THE RESPONSIBILITY OF THE MAINTENANCE PERSONNEL OR ORGANIZATION TO ASSESS ANY ADDITIONAL DAMAGE AND REPLACE DISCREPANT PART(S). KNOWN DAMAGED PARTS WHICH CAN BE REPLACED IN ACCORDANCE WITH APPROVED METHODS SHOULD BE ADDRESSED AND MAY NOT BE ADDRESSED SPECIFICALLY IN THIS REPAIR.

## **REPAIR INSTRUCTIONS: SUMMARY**

- 1. GENERAL REQUIREMENTS
  - A. >>> CAUTION: READ INSTRUCTIONS COMPLETELY AND THOROUGHLY BEFORE ATTEMPTING ACCOMPLISHMENT OF THIS REPAIR.
  - B. FOLLOW PRACTICES DEFINED IN DOCUMENT 12137-001, "SR20 AIRPLANE MAINTENANCE MANUAL" (AMM), CHAPTER 51.
  - C. AFFECTED EMM SHALL BE REMOVED PRIOR TO ACCOMPLISHMENT OF THIS REPAIR AND REPLACED AFTERWARDS PER AMM 51-20.
  - D. TAKE EXTREME CARE TO PREVENT ADDITIONAL DAMAGE TO THE AIRCRAFT STRUCTURE OR SYSTEMS.
- 2. PROCURE FROM CIRRUS AIRCRAFT:
  - A. 29142-012, REPAIR, COMPOSITE, FUSELAGE SECTION, RH QUANTITY 1.
  - B. 29143-012, REPAIR, COMPOSITE, FUSELAGE SECTION, LH QUANTITY 1.
  - C. 20083-003 OR 20083-004, SPAR, VERTICAL, AFT QUANTITY 1.
    - I. REQUIRED SECTION IS LOWER ~30" OF SPAR.
    - II. REQUIRED SECTION CANNOT HAVE THE RDM CUTOUT.
- 3. REMOVE DAMAGED FUSELAGE SKINS AND AFT VERTICAL SPAR.
- 4. AFT VERTICAL SPAR REPLACEMENT.
- 5. FUSELAGE SKIN REPLACEMENT.
- EXTERIOR WETLAY REPAIR.
- 7. TAIL TIE DOWN FINISHING.
- 8. BODYWORK AND PAINT.

Cirrus Design Approval – This document has been approved in accordance with FAA approved procedure meeting the requirements defined in 14 CFR Part 21. This document was processed through an electronic release system. All approval signatures are stored electronically in the Cirrus Product Data Management (PDM) System. The approval state appearing in the watermark at the top of this document is evidence the appropriate closed loop approval workflow process was used and is traceable in the Cirrus PDM database.