Post-it* Fax Note	7671	Date 4/29 pages /
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Co/Dept. NTS	•	CO. CESCNA
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GENERAL INFORMATION

Year <u>197</u>	6 Model 421C	Serial Numbe	r <u>421C013</u>	38 Registra	ition Number	N3911C
Mishap Date	9/2/95	Mishap Time1	215 MST	Mishap Location	on Phoenix,	AZ
Cessna Inve	stigator Stephen T. W	ilson	Engine Ma	nufacturer <u>1</u>	eledyne Contin	ental
Registered Owner	May Industries, Inc. 3640 West Osborne Phoenix, AZ 85019		Operator	Same as owne	er.	

2. SUMMARY

The pilot, flying alone, was on short final approach to Runway 7R at the Deer Valley Airport in Phoenix, Arizona, when four air traffic tower controllers observed the aircraft turn left and descend into a commercial lot. Other witnesses observed the aircraft roll right and left before descending into the ground. The aircraft was destroyed and came to rest along the airport perimeter road. Fire erupted immediately. The pilot and one person on the ground were fatally injured. The temperature was 101° Fahrenheit, and density altitude was approximately 4,700 feet. The aircraft was almost entirely consumed by the post-mishap fire. Subsequent examinations of the airframe, engines, and propellers revealed no evidence of pre-mishap conditions that were attributed to the cause of the mishap. Symmetrical damage to both propellers, including chordwise scoring and substantial twisting and bending, was observed. Subsequent autopsy toxicology analysis of the pilot revealed "mild focal patchy bronchopneumonia" and "therapeutic amounts" of diphenhydramine, a chemical ingredient of common over-the-counter drugs. According to the NTSB, other drugs were detected by the Civil Aeromedical Institute in Oklahoma City, although their report was not complete at the time of this report. Alcohol was not detected.

LEGEND:

UNK = Unknown N/R = Not Reliable D = Destroyed

N/A = Not Applicable N/O = Not Obtainable

		3.	FLIGHT DA	ATA		
Type of Flight:	Local _	х		Cross Co	ountry	
	PIC/Solo _			Dual		
Purpose:	Pleasure _		FT	Bus	NDV	
	Instruction _	-UK	144	Air T	UI_I	
	Practice _	ווע		Other (Sp	pecify)	
Flight Plan Filed?	☐ Yes	⊠ No		Type of F	Right Plan	·
Departure Point	Deer Valley, A	VZ (DVT)		Date	9/2/95	Time
Destination	DVT			ETA		ETE
Routing	UNK			~ ~ ~	Altitude	
Weather Briefing?	Prior to takeo	ff? ∐ Yes ⊠ji	No □ Unkno	wn Me	thod of Briefing	
	After takeof?	☐ Yes ⊠ i	No 🗆 Unkno	wn		
Last Known Refuel	ling Location:	Westwind Avia	ation, DVT, 6/	19/95		
Amount of Fuel Ad	ded <u>103.6 ga</u>	lions	Type of Fuel	Added	100LL	
Amount of Fuel at	Takeoff UNK		Estimated A	mount of F	fuel at Occurrer	nce
Center of Gravity V	Vithin Limits	At Takeoff	⊠ Yes	□ No	Inches	
		At Occurrent	æ ⊠Yes	□ No	inches	
Gross Weight With	in Limit	At Takeoff	⊠ Yes	□ No	Pounds	
		At Occurrent	ce 🛭 Yes	□ No	Pounds	
Baggage Number	er/Description o	fitems None	observed.			
Weight		Arm				

4.	OC	CU	P۵	IN	8
-	~~			4	

Seat	Name	Address	Status	Age	Injury
1	Charles Gene May	Peoria, AZ 85345	PIC	59	Fatal
	pr	AFT PAU			
		MEI UUI			

5. CREW HISTORY

Name	Charles Gene h	May Date	of Birth		_ 8	leat Occupied	1
Certificate No.	Date	lssued 1/26/		pe of Med		_ Date Issued	7/19/94
				mitations		e lenses for nea	ar and
☐ ATP		SE Land		stant visior			
☐ Commercial	☐ Rotorcraft	SE Sea		topsy Per			No
☑ Private	☐ Glider	ME Land	R	esuits?	See палта	itive.	
☐ Student		☐ ME Sea	_				
Fit Instructor	Other				Performed?		No
☐ Other	Type Ratings			esults?	See name	itive.	
(See Narrative for d	etails on 'Other' and	d 'Type Ratings')					
Flight Time							
Total Time	879.2 SE	112.5	ME		62.0	PIC	747.5
Actual Instrument	21.9 Dual	218	Day	8	52.4	Night	268
Simul. Instrument	24.5 This M	odel	Last 90	Days		Last Biennial	3/22/94
Source of Information	on 🛛 Lagbook	FAA	MITSI	3 🗆 0	Other		
Name		Date	e of Birth	1		Seat Occupied	
Certificate No.	Date	Issued		pe of Med	tical	Date Issued	
	. ,		Li	mitations			
T ATP	☐ Airplane	SE Land					
Commercial	Rotorcraft	SE Sea		utopsy Per	formed?	☐ Yes ☐	No
☐ Private	☐ Glider		R	esuits?			
☐ Student	☐ Instrument	📋 ME Sea	_				
☐ Fit Instructor	☐ Other				Performed?	Yes] No
☐ Other	☐ Type Ratings			esults?			
(See Narrative for o	letails on 'Other' an	d 'Type Ratings')				
Flight Time							
Total Time	SE		ME	_		PIC	
Actual Instrument	Dual		Day			Night	
Simul. Instrument	This M	lodel	Last 90	Davs		Last Biennial	
	on Logbook		□ NTS		Other	•	

6. MISHAP SITE								
☐ On Airpor	t ⊠ Off A	irport	Elevation	1,480 ft. MSL				
Latitude 3	33° 41' N	Longitu	de <u>112° 5' W</u>					
Terrain Feat	tures							
⊠ Levei	☐ Rolling	□ Hitty	☐ Mountainou	s 🗍 Wooded	☐ Plowed Field			
□ Сторе	☐ Brush	☐ Swamp	☐ Desert	☐ Water	☑ City Area			
☐ Other (sp	ecify)			 -				
Conditions	of T erra in							
⊠ Hard	⊠ Dry	☐ Soft	□ Wet	Snow Covered				
☑ Other (sp	ecify) Paved	commercial lot.						
Light Condi ☑ Day ☐ Night		DI	RAFT	COPY	, ——,			
			7. MISHAP DA	TA				
Obstacles S	itruck Before Pr	incipal impact						
☐ Wires	☐ Trees	D Brush	☐ Buildings	None Non				
Other (sp	ecify)							
Approximat	e Attitude at im	pact						
Pitch		_ Roll	 	Yaw				
Flight Path								

8. W	8. WRECKAGE DIAGRAM							
	Elevation: Latitude: Longitude: Terrain Profile:							

The wreckage was recovered prior to the arrival of the writer.

DRAFT COPY

9. WEATHER DATA								
Weather at Neare	est Reporting Poin	t						
Location Deer	/alley ATCT		· · · · · ·			Time _	1217 MS	T
Cloud Cover Cle	<u> </u>			Visibility	30	Tempera	ture .	101°F
Dew Point 51°F	Wind Direction	130	O" Wind	l Velocity	5 knots	Altimeter (Setting	inHg
Remarks Aircra	ift mishap							
Estimated Weath	er at Accident Site	•						
Cloud Cover				Visibility		Tempera	ture	•F
Dew Point°	F Wind Direction	n	Win	d Velocity		Altimeter	Setting	inHg
Mishap Site Weath	ner Data Source	Same a	s above.					
Remarks				001	MA			
	nn	ΛC			PY			
	Uh	H	. ARPL	NE HIST	ORY		-	
Year 1976 M	Model 421C S	Serial N	lumber 4	21C0138	F	Registration I	Viember	N3911C
Airframe Total Tim				otal Time at				100770
at Occurrence	UNK hours		100-hr insp		. Last	hours	Date	
Airframe Total Tim			_					
Periodic Inspection	UNK h	ours	Date	12/20/9	4		Tot	al Time Since
	Manufacturer	1	Model	Seria	Number	Total Time		v or Overhaul
#1 Engine	TCM		O-520-L	246013	-R			
#2 Engine	TCM		O-520-L	272089				
#1 Propeller	McCauley	90UM		763225				
#2 Propeller	McCauley	90ÚM	18-0	763221				
#4 b B _ SI _ J _	0-7-10		1 Din 4 - 4	1/2000	-	1 = =		
#1 Propeller Blade	Senai Numbers		Blade 1	K56995		Blade 2	K57064	
#2 Propeller Blade	Carial Number		Blade 3 Blade 1	K57442 K56991		Blade 4 Blade 2	VE7450	
#4 FTODERET DIAGE	OCHAI MUINEIS		Blade 3	K57444 m	iecino	Blade 4	K57458	· · · · · · · · · · · · · · · · · · ·
Source of Informat	ion 🛛 Logbo	ooks	⊠ Pa		Other (sp			
Flight Manual on B	loard? ⊠ Yes [_ No	☐ Unk	Logbooks	on Board?	⊠ Yes [] No [] Unk
Last Pitot/Static Sy	stem Check <u>UN</u>	K			•			

11. WRECKAGE DATA

Communication and Navigation	,	Engine instruments				
Qn	Off	Fn	eq		#1 Engine	#2 Engine
COM 1				Hourmeter		
COM 2				Tachometer - RPM	0	0
DME				Tachometer - Hours		
ADF 🗆				Manifold Pressure	28.5"	28.5°
NAV 1				Cylinder Head Temp	OSB	OSB
NAV 2				Oil Pressure	>0 <10	0
RNAV	_ [Oil Temperature	OSB	OSB
Loran				Fuel Pressure		
GPS				Exhaust Gas Temperature		
Autopilot				Turbine Inlet Temperature		
Transponder	On 🗆 Alt			Ammeter		
Flight Instruments				Voltmeter		
	Left	Ri	ght	Vacuum Pressure	Red	Red
Airspeed Indicator	0	D	0		V DSL	OSL
Altimeter	?,500		R	o team ater		
Altimeter Setting		U		achameter and a	×	
Directional Gyro	330°	13	30°	N2 Tachometer		
Heading Bug	185°			Environmental Controls		
Vertical Speed Indicator	+1,100		250		On	Off
Attitude Indicator (pitch)	10° Dn		t up	Cabin Heater		×
Attitude Indicator (roll)	65° L	5	·L	Air Conditioner		
Turn & Bank Indicator (Needle)	1/setct R			Cabin Vent		
Turn & Bank Indicator (Ball)	Left					
Turn Coordinator (Airplane)						
Turn Coordinator (Ball)				Pressurization Controls		
Magnetic Compass						
NAV1 OBS	170°			Cabin VSI		0
NAV2 OBS				Cabin Altitude	< 2	2,000
RNAV Bearing				Differential Pressure		0
RNAV Distance				Pressurization Safety Val		iurize on
Clock				Pressurization Dump Valv	re	
Fuel Management						
	Left		R	light		
Fuel Quantity (Main Tanks)	25 lbs			D		
Fuel Quantity (Aux Tanks)				·		
Fuel Quantity Gauge Selection						
Fuel Selector Handles	D			D		
Fuel Selector Valve	Off			Off		

Ignition and Electrical Sw	itches				Engine Control Positi	one (Fnaine)	<u></u>
Left Right						#1 Engine	#2 Engine
	On	Off	On	Off	Throttle		
Magnetos (#1 Engine)		X	<u> </u>	Х	Power Lever		
Magnetos (#2 Engine)	X		X		Emer Power Lever		
Ignition	-				Mixture Control		
Alternator/Generator	X		X		Fuel Control Unit		
Fuel Boost Pumps		Х		- x−	Propeller Governor		
·			<u> </u>		Cowi Fiaps		
					Carburetor Heat		
					Alternate Air		
					Inertial Separator		:
					monier osperawi	<u> </u>	
	C	n	0	ff	Engine Control Positi	ons (Cocknit)	
Master Switch	7	(#1 Engine	#2 Engine
Avionics Switch (#1)	 ,	(Throttle-	Full	Full
Avionics Switch (#2)	-		- #	7	Para lever		
Inverter Switch (#1)	AR			++1	mer hower Lever		
Inverter Switch (#2)	UT			Jt	Mixture Control	Full rich	Full rich
Pitot Heat	M	#1	* X		Fuel Condition Lever		,
Navigation Lights	<u> </u>		Х		Propeller Control	High RPM	Back 1/2"
Rotating Beacon (s)		·			Cowl Flaps		
Landing Lights			Х		Carburetor Heat		
Taxi Lights			Х		Alternate Air		
Strobe Lights	 ,	(Primer		
Instrument Lights					Inertial Separator		
Stail Heat			X		·		- ,
Icing Equipment	!	****** *			Landing Gear and Fis	p Positions	
	. 0	n	Of	Ť		Up	Down
Surface De-Ice			Х		Left Main		
Surface Anti-Ice					Right Main		
Windshield De-Ice					Nose Gear		
Windshield Anti-Ice			Х		Gear Selector	N/R	
Propeller Anti-Ice			X		Gear Actuator		
Propeller De-Ice							
Propeller Sync/Phase					Left Flap		
Autofeather				. "	Right Flap		
					Flap Indicator		30°
					Flap Actuator		
					Flap Selector		Btwn 0&15°
	<u> </u>						

Flight Controls			
ragnt Controls	Continuity Established?	Aircraft Damage Assessment	
Rudder	No No	Engine #1	Dostania
Elevator	No.	1 ·	Destroyed
Ailerons		Engine #2	Destroyed
1	No	Propeller #1	Destroyed
Flaps Rudder Tab	No	Propeller #2	Destroyed
1	No	Fuselage	Destroyed
Elevator Tab	No	Wing Center Section	Destroyed
Aileron Tab	No	Tailcone	Destroyed
	.	Left Wing	Destroyed
	Position?	Right Wing	Destroyed
Rudder Tab	Actuator: 2.7	Left Flap	Destroyed
Elevator Tab	Actuator: 1.7"	Right Flap	Destroyed
Aileron Tab		Left Aileron	Destroyed
Rudder Tab Indicator	D	Right Aileron	Destroyed
Elevator Tab Indicator	D	Left Horizontal Stabilizer	Destroyed
Aileron Tab Indicator	D	Right Horizontal Stabilizer	Destroyed
Remarks	····	Left Elevator	Destroyed
DDAFT	0001/	Right Elevator	Destroyed
TIRNE	THUV	Vertical Stabilizer	Destroyed
וואוע		Rudder	Destroyed
Miscellaneous information		Aileron Tab	Destroyed
		Rudder Tab	Destroyed
Dual Controls Installed?	Yes	Elevator Tab	Destroyed
Oxygen installed?	Yes	Left Main Gear	Destroyed
De-loe Boots Installed?	Yes	Right Main Gear	Destroyed
Certified into known icing?	Yes	Nose/Tail Wheel	Destroyed
Emergency Locator Transmitt	er information		
ELT installed	UNK	Î	
ELT Type		1	
ELT Serial Number		1	
ELT Battery Due Date			
ELT armed?		İ	
ELT activated?		ł	
LLI GOLITARON:	<u> </u>		

Seats				<u>i</u> _		
	Seat Feet Intact?	Seat Back intact?	Seat Base Intact?	Seat Rail Intact?		
Seat#	Yes No	Yes No	Yes No	Yes No		
1						

Lap Belts					. —	Shouk	Jer Harr	iesses					
•		Used?		Inta	ct?		Used?		Inta	ct?	1	installed	7
Seat#	Yes	No	Unk	Yes	No	Yes	No	Unk	Yes	No	Yes	No	Unk
1			Ø		X		X			X	X		
												ä	
	.		77	FA /		-				*******			
			- 1 2	U I	11	T	ne	9 89 1					

Aircraft and Occupant Configuration								
	Occuj	pied?	Seat	Orienta	ition?	Comments		
Seat #	Yes	No	Fwd	Aft	Side			
1	X		X					
			П					

<u>Note</u>: Further descriptions of the seats, lapbelts, and shoulder harnesses may be found in 12. Narrative under Seats/Restraint Systems/Cabin Environment

12. NARRATIVE

HISTORY OF ELIGHT

The pilot, owner of May Industries, was flying alone during day VFR conditions. According to NTSB investigator, George Petterson, the pilot's wife knew her husband was at the Deer Valley Airport in north Phoenix, Arizona, cleaning their company owned aircraft, but she did not know he was going to fly. The pilot contacted the Deer Valley Air Traffic Control Tower and reported 15 miles northwest of the airport inbound for landing. The pilot was instructed to enter a left base for Runway 7R. The tower controller observed the aircraft turn onto final approach approximately one mile out. After the aircraft was established on final, four air traffic controllers observed the aircraft turn left and descend into a commercial lot. Other witnesses observed the aircraft roll right and left before descending into the ground. The aircraft was destroyed and came to rest beneath utility lines along the airport perimeter road, North 19th Avenue. Fire erupted immediately. Temperature was 101° F. Density attitude was approximately 4,700 feet.

INJURIES TO PERSONS

32 year-old, Robert Glen Jones, Market Phoenix, AZ, Market Was closing a gate at the close of business at the commercial lot when the mishap occurred. Witnesses observed Mr. Jones being involved by the fire and rushed to his assistance. The Phoenix Police Department report indicated Mr. Jones received burns on 92% of his body. He was taken to Maricopa County Medical Center in critical condition. According to insurance adjuster, Bob McMullin, Mr. Jones succumbed to his injury the next morning when life support was removed. Mr. McMullin also advised that Mr. Jones was a superintendent for his father's business, C.S. Construction.

The pilot was fatally injured. The Maricopa County Medical Examiner's autopsy report reads, "Autopsy revealed extensive thermal burns and evidence of smoke inhalation with soot in the airway. The heart was enlarged and microscopic study of the heart showed old fibrosis. Microscopic sections of the lungs showed mild focal patchy bronchopneumonia as well as soot inhalation. Carbon monoxide level was 8%, which was too low for carbon monoxide toxicity." The report listed the cause of death as smoke inhalation and thermal burns. Maricopa County toxicology detected "therapeutic amounts" of diphenhydramine, a chemical ingredient of drugs such as Bufferin, Excedrin P.M., Benadryl, Tylenol PM, and Unisom. According to NTSB Investigator Petterson, other drugs were detected by the Civil Aeromedical Institute in Oklahoma City, although their report was not complete at the time of this report. Alcohol was not detected.

INVESTIGATION

The writer traveled to Phoenix on the evening the mishap occurred. Before his arrival the aircraft was recovered from the site and taken to Air Transport, a Phoenix aircraft salvage company, at the direction of NTSB Investigator Petterson. Per Mr. Petterson's instruction, the writer met FAA Inspector Larry Jones and AlliedSignal Engineer Mike Cummings at the Air Transport facility the following morning for a post-salvage examination of the wreckage. Murle Williams, of Air Transport, assisted. Following the examination at Air Salvage, the writer drove to the mishap site and took photographs.

Damage to Aircraft: The entire aircraft was thermally destroyed, although the cockpit area remained generally intact for the purpose of examination. Switch positions on the electrical console were examined after debris was carefully removed. The annunciator panel and landing gear position lights were removed for NTSB examination. Cabin structure aft of the wing area was not present except for the cabin door and adjacent door structure. The outboard section of the left wing stub was not present. The inboard section of the wing stub was swept aft and detached at the forward attach

point. The right wing stub remained attached to the fuselage. Attached to the right wing stub were the following: a portion of the right engine nacelle, the air conditioner hydraulic pump, the landing gear hydraulic pump, the right turbo system, the a fuel pump, a portion of the right induction system, and the right fuel metering valve. A lower outboard section of the left wing including the wing tip was present. The left alleron was present, and exhibited comparatively less thermal damage. Other major sections of the wings were not found. The tailcone was not present. Only the most inboard sections of the empennage remained.

Landing Gear and Flaps: All three landing gear were detached from the aircraft structure. Both lower main landing gear struts were detached from the upper trunions. The left main wheel assembly and lower strut were not thermally damaged. The left main gear door was present. The nose wheel assembly appeared to have burned from the strut. The nose strut drag link remained attached to the upper trunion and was bent, forming an acute angle. Neither flap was present, nor the adjacent wing structure. The flap actuator was present but detached from airframe structure. Neither flap chain was engaged in the actuator cogs.

Left Engline and Propeller: The left engine exhibited the mall lamage and locating firoughout. The accessories remained attached, except the propeller governor was detached from the left forward side. Both magnetos were thermally destroyed. All ignition leads appeared in place except one left lead was cut. The outboard side of the oil cooler was melted away. The exhaust risers on the right side appeared intact. The left exhaust risers were crushed upward. The induction system was burned away. The propeller hub remained attached to the engine with two propeller blades (K57442 and K56995) in place, but loose. The pitch change links were detached from the blades. The outboard half span of blade K57442 was melted away. It exhibited diagonal scoring and no leading edge damage in the inboard section. It was bent aft in the root section and forward in the most outboard section. Blade number K56995 was curled aft about 180° in the outboard section. It exhibited some leading edge damage and chordwise scoring which was greatest within 4 Inches of the tip. Blade number K57064 was detached from the hub and exhibited very similar deformation to two of the right propeller blades, except it was sooted and burned in appearance. See description of the right propeller, below.

Right Engine and Propeller: Like the left engine, the right engine exhibited thermal damage and sooting throughout. The alternator, starter, vacuum pump, propeller governor, and magnetos remained attached. The hydraulic pumps were detached. As previously reported, the hydraulic pumps remained attached to the right wing stub. The ignition leads appeared in place. The induction manifold was in place. The right induction tubes were melted away. The left tubes were intact, but detached from the manifold. The right side exhaust risers were crushed upward. The left side exhaust risers were intact. The propeller hub was broken from the engine, except the aft wall of the hub remained bolted to the crankshaft flange. Two of the right propeller blades (K57458 and K57444) were present and detached from the hub. The third right blade was missing. As mentioned above, the two right blades and the detached left blade were very similar in appearance. They were each curled aft and twisted (toward low pitch). Each blade exhibited heavy chordwise scoring in the outboard third span on the cambered surface. Blade K57458 exhibited no fire damage. Blade K57444 was sooted.

Seats and Seat Belts: The left front seat remained attached to the cabin floor. It was partially burned. The lap belt was unclasped. The male fitting was evenly scoted. The female lap belt fitting left a burn shadow on the seat cushion. See photograph 3-AIDN-1. The shoulder harness was not attached to the lap belt fitting.

SUBSEQUENT EXAMINATIONS

On October 5, 1995, both engines were disassembled and examined at the Teledyne Continental Motors (TCM) Analytical facility in Mobile, Alabama, under the supervision of NTSB investigator Tom Wilcox. TCM's inspection reports are included as Attachment 8 to this report. It should be noted that the engines were mismarked as received. Throughout the TCM reports, the left engine as-written

should be right, and the right engine as-written should be left. In summary, no evidence of discrepancy was noted with either engine that was attributed to the cause of the mishap.

At the direction of NTSB Investigator Petterson, the turbocharger system was examined on October 17, 1995, at the AltiedSignal Aerospace Equipment Facility in Torrance, California. AltiedSignal Engineer Mike Cummings reported, "no pre-existing were found that would prevent normal operation." According to Mr. Cummings, rub marks were present in both turbochargers which evidenced rotation at the moment of the mishap.

On October 24, 1995, the writer met NTSB Investigator Tom Wilcox; FAA Inspectors Bruce Smith and John Sills; Allied Signal Investigator Mike Cummings; and McCauley Engineers Tom Knopp and Wayne Edmonds for a propeller examination at Air Transport in Phoenix. By examining deformation to the butt-end of the propeller blades, Mr. Knopp determined that the right propeller was in an operating position, but he was not able to determine the blade angle. According to Mr. Knopp, the left propeller was in a low pitch operating position indicated by damage to the butt-end of blade K56995. A copy of Mr. Knopp's report to NTSB Investigator Petterson is included as Attachment 7 of this report. In the conclusion section of the report, Mr. Knopp stated, "...All propeller damage was of the type associated with impact forces, with gross deflections, was of sudden failure type, and there were no indications of any type of fatigue failure. ...Blade bending and overall propeller damage was of the type and extent indicative of power at impact, with blade bending damage similar for each propeller indicating symmetrical power."

According to NTSB Investigator Petterson, no information was obtained from the NTSB examination of annunciator panel and landing gear position lights.

CREW INFORMATION

The pilot's logbook indicated he completed a biennial flight review on March 22, 1994, which included 1.5 hours of flight. In February, 1995, the pilot logged three flights with instructor Harry Oliphant. Investigator Petterson talked with Mr. Oliphant who reported the pilot was "behind the aircraft." In addition to seven flights logged in 1995, the pilot's wife advised Mr. Petterson of one flight to Bullhead City on June 10, 1995. Flight times listed on Page 3 of this report were taken from the pilot's logbook. It was noted that columns of total day flight plus total night flight did not equal total "duration of flight." According to Mr. Petterson the pilot began flying on November 10, 1981, and started his multi-engine flying on August 9, 1982. Copies of the last page of the pilot's logbook and a copy of his medical certificate application were provided by Mr. Petterson. They are included as Attachment 3 of this report.

AIRCRAFT INFORMATION

A copy of invoices pertaining to the aircraft were provided by Mr. Petterson, and are included as Attachment 4 of this report. The last annual inspection was performed by Alberto Marin, an employee of Sawyer Aviation, on his own time. According to Mr. Petterson, Mr. Marin worked "out of his car." Mr. Marin billed May Industries for 69 hours labor which included the following in addition to the inspection: both oil and air filters were changed, the left main wheel was replaced, the right and nose gear uplock actuators were resealed, the left landing light motor and gears were replaced, and the fuel inlet valves in both wings were replaced. A copy of Mr. Marin's written statement is included in Attachment 4 of this report. The pilot logged 17.3 hours in addition to the one known unlogged trip to Bulihead City since the date of the annual inspection.

The following notes were taken from the burned remains of the aircraft logbooks during the propeller examination on October 24, 1995:

- Tag Knisley Welding Inc., Repair station number NJ3R712L, 3450 Swetzer Rd., Loomis, CA, 95650, 916-652-5891.
- 1986 logbook entry, Time 539.3.