

Continental Motors

ENGINE FIELD INSPECTION REPORT

ENGINE MODEL	E225-8
ENGINE SERIAL NUMBER	35818-D-0-8-R
AIRCRAFT MAKE & MODEL	Beechcraft 35G Bonanza
AIRCRAFT SERIAL NUMBER	D-4863
AIRCRAFT REGISTRATION	N394CW
CMI FILE NUMBER	15-238

NAME	SIGNATURE	DATE
Mike Council		11-17-2015

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	GENERAL INFORMATION				
EX	AMINA	TION	ACCIDEN	IT DATA	
DATE	10-13	3-2015	NTSB ACCIDENT #	CEN16FA011	
FACILITY	Beeg	les Aircraft Services	NTSB INVESTIGATOR	Josh Lindberg	
ADDRESS			FAA INVESTIGATOR	Wesley R. Dollahite	
			ACCIDENT DATE	10-13-2015	
			ACCIDENT LOCATION	Eaton, Colorado	
		ENGINE	INFORMATION		
ENGINE POSI	ENGINE POSITION Single				
TOTAL	TIME	5288.19 according to	o NTSB		
TIME	SOH	479.5 according to N	NTSB		
TYPE & TIM	E SLI	100 Hour Inspection	5/17/2015		
BUILD [DATE	Not in record			
IN SERVICE I	DATE	Not in record			

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Significant logbook information:

Subject:	AIRFRAME	ENGINE	PROPELLER
Make:	Beechcraft	Continental	Hartzell
Model:	G-35	E225-8	HC-A2V20-4A1
Serial Number:	D-4863	35818-D-O-8-R	AK837
Times as of Date:	May 7, 2015	May 7, 2015	May 7, 2015
Times Verified By:	Logbook Entry	Logbook Entry	Logbook Entry
Tach Time	778.60	778.60	778.60
Total Time:	5091.00	5288.19	UNK
TSOH:	NA	778.60	70.50
OH Date:	NA	July 7, 1992	December 8, 2011
TSLightOH:	NA	479.50	NA
TSTOH:	NA	87.00	NA
TOH Date:	NA	January 25, 2011	NA
Last Inspection: Annual		100 Hr.	100 Hr.
Last Inspection Date:	May 7, 2015	May 17, 2015	May 17, 2015

ENGINE NOTES:	Major Overhaul	July 7, 1992	0.00
	#4 Cyl replaced	May 3, 1996	147.90
	R&R All Cyl.	June 20, 1997	270.50
	Engine Teardown	February 2, 1999	299.10
	Engine Teardown #1 & 3 Cyl	February 28, 2002 November 11,	454.30
	replaced	2004	556.10
	#6 Cyl replaced	March 3, 2006 November 15,	566.00
	R&R All Cyl.	2007	599.80
	Rehoned All Cyl.	June 19, 2009	626.50
	R&R All Cyl.	January 25, 2011	691.60
	#4 Cyl replaced	April 1, 2013	763.00

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Report Summary:

Search Code(s):

Examination of the engine was performed by the CMI Investigator under supervision of the NTSB Investigator. The inspection of this engine did not reveal any pre-impact anomalies which would have prevented Its ability to produce rated horsepower.

Disposition of engine following exam: The engine remains at Beegles Aircraft Services until released by the NTSB IIC to the insurance company.

	INSPECTION WITNESSES				
NAME	Mike Council	NAME	Paul Yoos		
ADDRESS		ADDRESS			
	Mobile, Alabama				
			Wichita, KS 67206-2556		
ORGANIZATION	Continental Motors, ASI	ORGANIZATION	Textron ASI		
PHONE		PHONE			
NAME	Wesley R. Dollahite	NAME	Brad Bigelow		
ADDRESS		ADDRESS			
ORGANIZATION	FAA, Safety Inspector	ORGANIZATION	NTSB, ASI		
PHONE		PHONE			

EXTERNAL INSPECTION OF ENGINE

The engine data tag remained attached to the crankcase. The data tag indicated that the engine was a model E225-8 engine, serial number 35818-D-0-8-R. No mechanical or impact damage was noted on the crankcase halves.

The combination oil tank/ cooler was impact damaged and was removed to gain access to the remaining engine components. The engine oil rod was removed, wiped and reinserted into the oil rod sleeve. The oil rod indicated approximately five quarts of dark, oily substance was contained in the oil sump.

Right and left exhaust components were impact damaged and crushing was noted to the mufflers. The induction system exhibited impact damage. The air box and induction filter separated and were impact damaged.

The engine driven fuel pump separated from the accessory case but remained attached by a fuel hose. The fuel pump housing was impact damaged and could not be rotated manually. The drive coupling and gear were separated and not recovered.

The pressure carburetor was intact and in place with damage noted to the attach pad. The carburetor throttle control and mixture arms remained attached and secured.

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The aftermarket propeller governor and "T" pad adapter separated from the accessory case due to impact forces. The accessory case housing exhibited impact damage.

The electric starting motor remained attached with a slight amount of impact damage to the rear casing. The engine driven generator remained attached and intact. The engine driven "wet" type vacuum pump remained attached and intact. The air/oil separator was impact damaged. Right and left magnetos remained intact and mounted to the accessory case.



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ENGINE TEARDOWN AND COMPONENT EXAMINATION

EXHAUST SYSTEM

Condition:

Right and left mufflers and right tailpipe were crushed by impact forces but remained attached. The left tailpipe separated. Muffler flame cones were intact.









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INDUCTION SYSTEM

Condition:

The induction system was impact damaged but remained attached. The front mounted air box and induction filter was impact damaged









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IGNITION SYSTEM

LEFT MAGNETO

Manufacturer: TCM

P/N: S6LN-21

S/N: E189201DR

Condition:

The magneto received impact damage. The bushing cup was cracked. The magneto was removed and manually rotated using a drill motor. Spark was

produced to each ignition lead.





RIGHT MAGNETO

Manufacturer: TCM

P/N: S6LN-???

S/N: Undetermined

Condition:

The magneto did not display damage. The magneto was removed and manually rotated using a drill motor. Spark was produced to each ignition lead. The data decal was damaged and the serial number was not legible.





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IGNITION HARNESS

Manufacturer: Undetermined

P/N: Undetermined

S/N: Undetermined

Condition:

The ignition harness remained intact with the exception of leads to cylinders one and six which were separated. (top sparkplugs)





SPARK PLUGS

Manufacturer: Champion

P/N: REM40E

Condition:

When compared to a Champion "Check A Plug" chart, sparkplugs from cylinder one was "worn out severe", while sparkplugs from cylinders 2, 3, 4, 5 and 6 exhibited "normal" wear signatures. Sparkplugs from cylinders 2, 4, and 6 were wet with oil. Note: the engine was positioned with cylinders 2, 4, and 6 positioned downward for nearly 24 hours.





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FUEL SYSTEM

FUEL PUMP

Manufacturer: Thompson Products, Inc.

P/N: M128983A

S/N: 444249W

Condition:

The engine driven fuel pump separated from the accessory case but remained attached by a fuel hose. The fuel pump housing was impact damaged and could not be rotated manually. The drive coupling and gear were separated and not recovered.





CARBURATOR

Manufacturer: Bendix

P/N: PS-5C

S/N:772805

Condition:

The pressure carburetor was intact and in place with damage noted to the mounting flange. The fuel screen was removed and a small quantity of contamination was noted. No fuel was released from the carburetor during the bench inspection. The carburetor throttle control and mixture arms remained attached and secured.





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LUBRICATION SYSTEM

OIL SCREEN Manufacturer: TCM P/N:A25131

Condition: A small amount of carbon was found on the screen surface.





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OIL TANK/ COOLER

Manufacturer: Undetermined

P/N: Undetermined

S/N: Undetermined

Condition:

The combination oil tank and cooler was impact damaged





CYLINDERS

Condition:

All six cylinders remained intact and in place with little or no impact damage. All six cylinders displayed orange paint on each mounting pad indicating that the internal cylinder barrels were "channel chrome" plated. All six cylinders achieved thumb compression when the engine was manually rotated. A lighted borescope inspection was performed and all exhaust and intake valves were noted to be intact. Cylinder rocker covers were removed and rocker arm movement was noted when the crankshaft was manually rotated.





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CRANKSHAFT ASSEMBLY

CRANKSHAFT Forging Number:
Undetermined S/N: Undetermined Heat code:
Undetermined

Condition: The crankshaft appeared to be undamaged. Continuity was established by

manually rotating the engine and observing timing gear, magneto and camshaft

rotation.

ACCESSORIES

STARTER Manufacturer: Delco RemyServiced by EMI Aviation
Parts P/N: Undetermined S/N: Undetermined

Condition: The starter motor was impact damaged.









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ALT/GEN #1 Manufacturer: Delco Remy P/N: 11018885 S/N: 5428

Condition: The engine driven generator remained attached and intact





VACUUM Manufacturer: Garwin

P/N: G45?

S/N:2260

Condition:

The engine driven "wet" type vacuum pump remained attached and intact. The air/oil separator was impact damaged.





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PROPELLER

PROPELLER GOVERNOR		Manufacturer: Undetermined P/N: Undetermined S/N: U		S/N: Undetermined
Condition: The aftermarket propeller governor and "T" pad adapter separated from the accessory case due to impact forces.			parated from the	
PROPELLE	ER	Manufacturer: Hartzell	M/N: HC-A2V20- 4A1/V8433	S/N:AK847
Blade 1 S	/N:	: 87993		
Blade 2 S	/N:	85703		

Condition:

Blade A remained attached to the propeller hub but was loose in the hub assembly. Blade A displayed heavy scratching and scoring on the cambered face and the blade displayed a smooth rearward bending signature beginning approximately 16 inches from the blade root.

Blade B separated from the hub and exhibited a smooth forward bending signature beginning approximately eight inches from the blade hub. There was a deep gouge on the trailing edge approximately five inches from the blade tip.









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