

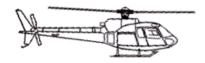
ATTACHMENT 3

AIRWORTHINESS GROUP CHAIRMAN'S FACTUAL REPORT

NTSB No. CEN15MA290

Airbus Helicopters Service Bulletin No. AS350-67.00.66 Revision 1 (31 Pages)



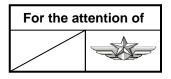


Civil version(s): B3

SERVICE BULLETIN

SUBJECT: ROTOR FLIGHT CONTROLS

Functional check procedure for the tail rotor load compensator



Revision No.	Date of issue
Revision 0	2015-08-26
Revision 1	2015-10-22

Summary:

To insert the appended pages into the Flight Manual, pending the Flight Manual update.

Reason for last Revision:

The purpose of Revision 1 is:

- to specify the procedure for inserting appendices into the Flight Manual,
- to make improvements in the appended pages.

Compliance:

Airbus Helicopters recommends compliance with this Service Bulletin.



1. PLANNING INFORMATION

1.A. EFFECTIVITY

1.A.1. Helicopters/installed equipment or parts

AS350 helicopters, version B3.

1.A.2. Non-installed equipment or parts

Not applicable.

1.B. ASSOCIATED REQUIREMENTS

Not applicable.

1.C. REASON

Following the issue of Safety Information Notice No. 2944-S-29, Airbus Helicopters asks you to insert the appended pages into the Flight Manual, pending the Flight Manual update.

Revision 1

The purpose of Revision 1 is:

- to specify the procedure for inserting appendices into the Flight Manual,
- to make improvements in the appended pages.

Revision 1 affects compliance with Revision 0.

1.D. DESCRIPTION

Compliance with this Service Bulletin consists in inserting the appended pages into the Flight Manual, pending the Flight Manual update.

1.E. COMPLIANCE

1.E.1. Compliance at H/C manufacturer level



1.E.2. Compliance in service

The work on the helicopter is to be performed by the operator.

Helicopters/installed equipment or parts:

Upon receipt of Revision 1 of this Service Bulletin, Airbus Helicopters recommends:

- removing the pages inserted as part of compliance with Revision 0 of this Service Bulletin from the Flight Manual.
- inserting the appended pages (according to configuration) into the Flight Manual.
 - . For AS350 B3 version helicopters equipped with an ARRIEL 2B engine, insert the pages given in Appendix 4.A.
 - . For AS350 B3 version helicopters equipped with an ARRIEL 2B1 engine, insert the pages given in Appendix 4. B.
 - . For AS350 B3e helicopters (or H125 Commercial name of AS350 B3 version helicopters equipped with an ARRIEL 2D engine), insert the pages given in Appendix 4.C. (EASA) or Appendix 4.D. (FAA/ANAC/IAC-AR).

The pages given in the Appendix are a partial update of some points of the existing pages (like a Rush Revision of the Flight Manual).

Each page given in the Appendix must be inserted (without deleting the Service Bulletin header/footer) in front of the page that contains the existing text to be modified or completed. No pages are removed from the Flight Manual.

The information which has been updated and integrated in the Appendices is detailed in:

- Figures 1 and 2, for AS350 B3 version helicopters equipped with an ARRIEL 2B engine,
- Figures 3 to 5, for AS350 B3 version helicopters equipped with an ARRIEL 2B1 engine,
- <u>Figures 6 and 7</u> (EASA) or <u>Figures 8 to 10</u> (FAA/ANAC/IAC-AR), for AS350 B3e version helicopters (or H125 Commercial name of AS350 B3 version helicopters equipped with an ARRIEL 2D engine.

Non-installed equipment or parts:



1.F. APPROVAL

Approval of modifications:

Not applicable.



Approval of this document:

The technical information contained in this Service Bulletin Revision 0 was approved on August 26, 2015 under the authority of EASA Design Organization Approval No. 21J.056 for helicopters of civil versions subject to an Airworthiness Certificate.

The technical information contained in this Service Bulletin Revision 1 was approved on October 22, 2015 under the authority of EASA Design Organization Approval No. 21J.056 for helicopters of civil versions subject to an Airworthiness Certificate.

1.G. MANPOWER

Not applicable.

1.H. WEIGHT AND BALANCE

Not applicable.

1.I. **POWER CONSUMPTION**

Not applicable.

1.J. SOFTWARE UPGRADES/UPDATES

Not applicable.

1.K. REFERENCES

Not applicable.

1.L. OTHER AFFECTED PUBLICATIONS



Airbus Helicopters will update the Flight Manual (PMV) in accordance with this Service Bulletin.

This document will be issued subsequently.

1.M. PART INTERCHANGEABILITY OR MIXABILITY



2. **EQUIPMENT OR PARTS INFORMATION**

2.A. EQUIPMENT OR PARTS: PRICE - AVAILABILITY - PROCUREMENT

Not applicable.

2.B. LOGISTIC INFORMATION

Not applicable.

2.C. EQUIPMENT OR PARTS REQUIRED PER HELICOPTER/COMPONENT

Not applicable.

2.D. EQUIPMENT OR PARTS TO BE RETURNED



3. **ACCOMPLISHMENT INSTRUCTIONS**

3.A. GENERAL

Not applicable.

3.B. WORK STEPS

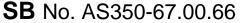
Not applicable.

3.C. COMPLIANCE CONFIRMATION

Compliance with this document:

Record compliance with this document, with the revision number, in the helicopter documents.

3.D. OPERATING AND MAINTENANCE INSTRUCTIONS





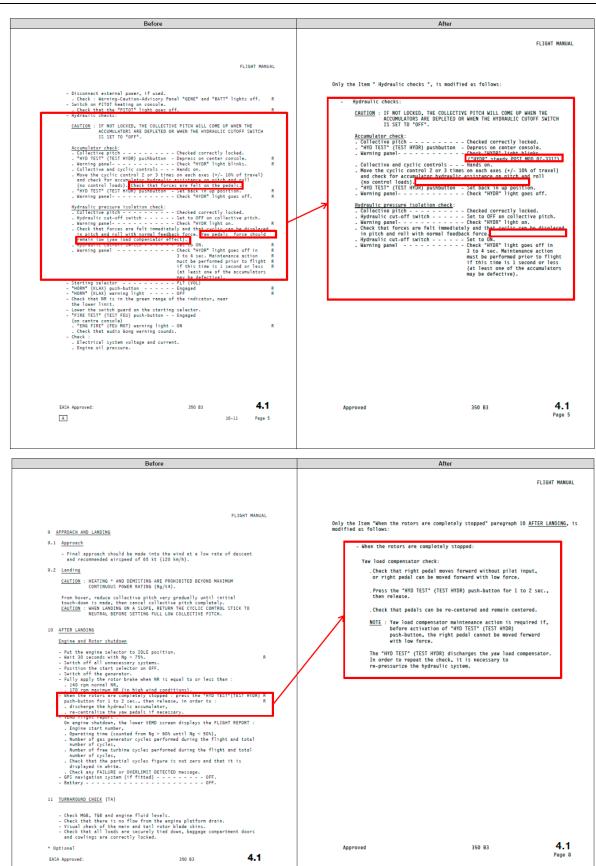


Figure 1 - ARRIEL 2B



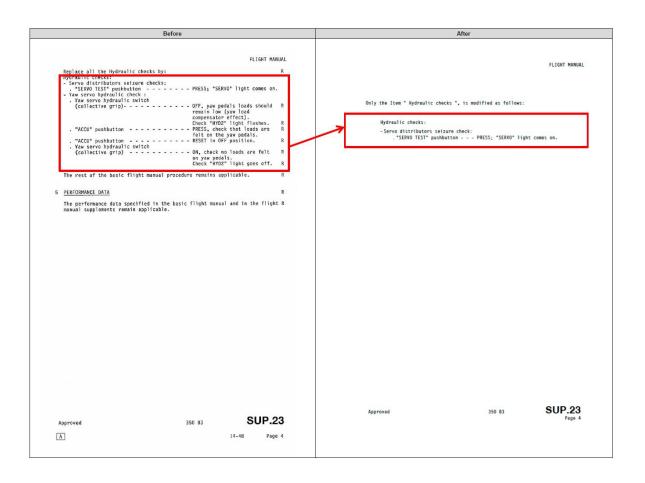


Figure 2 - ARRIEL 2B



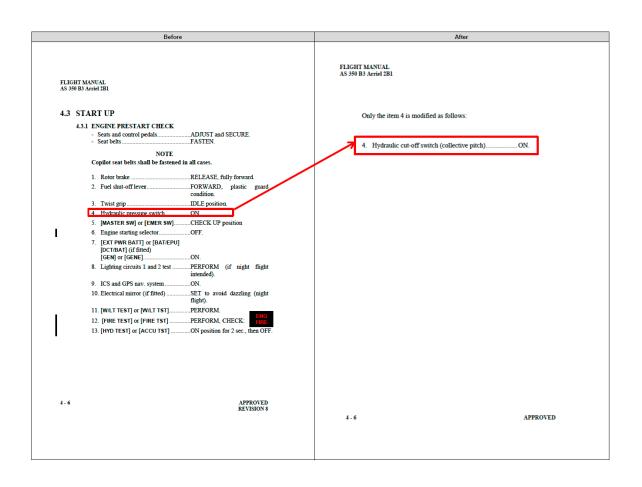
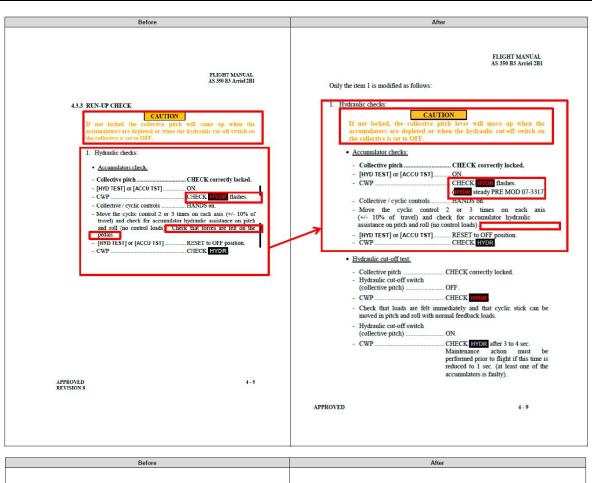


Figure 3 - ARRIEL 2B1







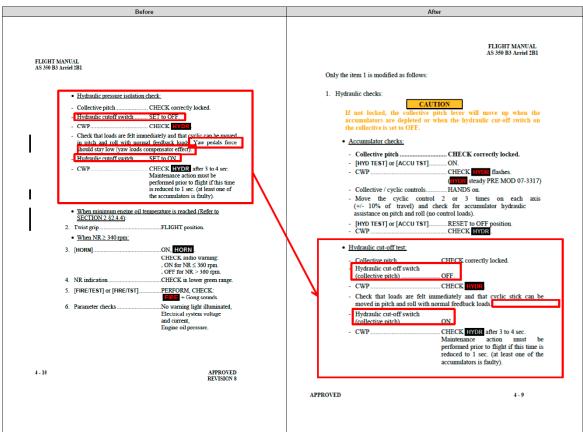
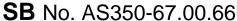
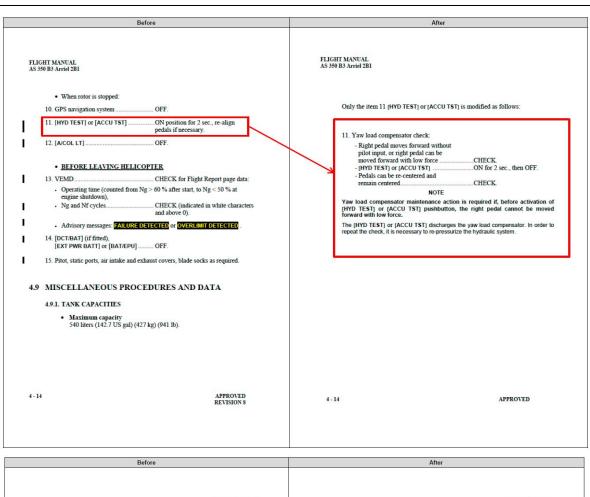


Figure 4 - ARRIEL 2B1







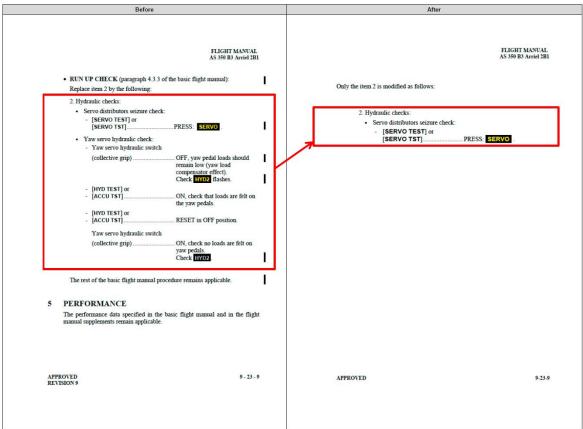


Figure 5 - ARRIEL 2B1





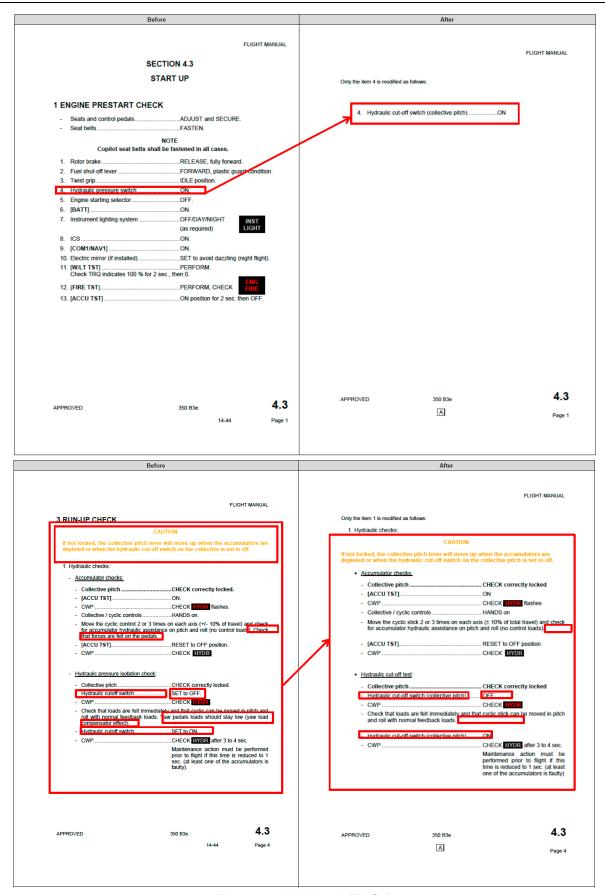
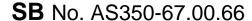


Figure 6 - ARRIEL 2D (EASA)





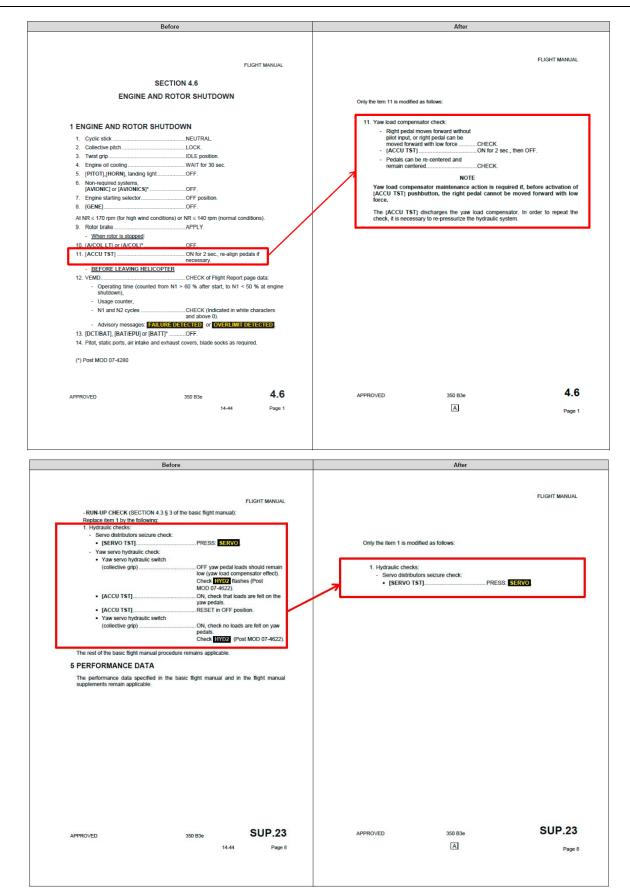


Figure 7 - ARRIEL 2D (EASA)



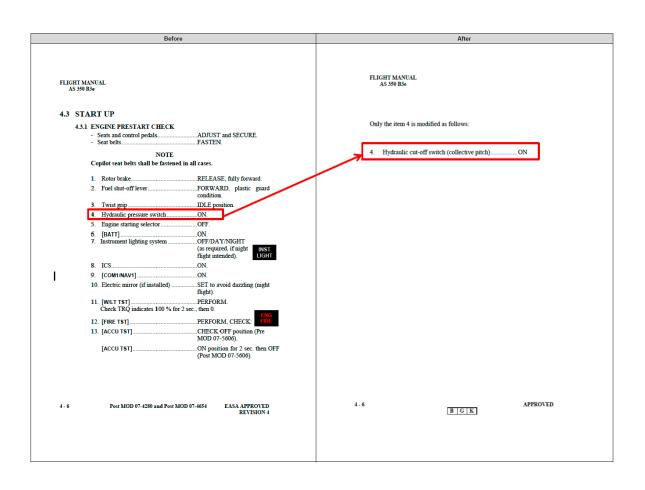
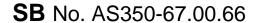


Figure 8 - ARRIEL 2D (FAA/ANAC/IAC-AR)





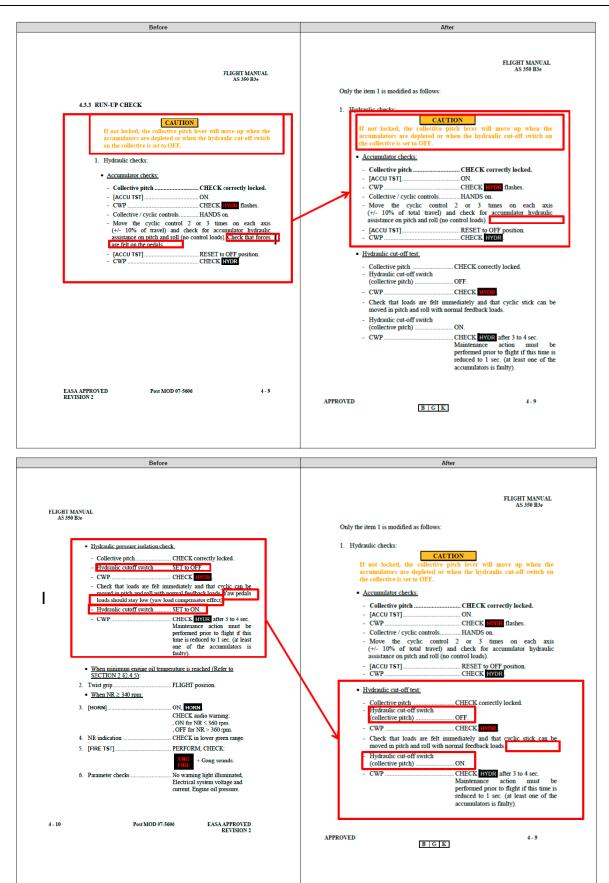
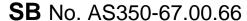


Figure 9 - ARRIEL 2D (FAA/ANAC/IAC-AR)

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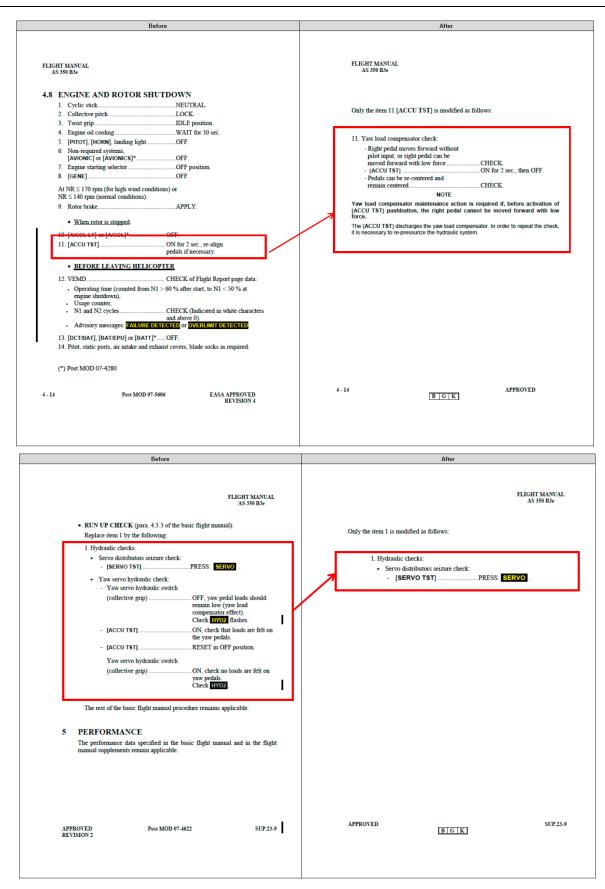


Figure 10 - ARRIEL 2D (FAA/ANAC/IAC-AR)



4. APPENDIX

4.A. FLIGHT MANUAL - ARRIEL 2B

FLIGHT MANUAL Only the Item " Hydraulic checks ", is modified as follows: Hydraulic checks: CAUTION : IF NOT LOCKED, THE COLLECTIVE PITCH WILL COME UP WHEN THE ACCUMULATORS ARE DEPLETED OR WHEN THE HYDRAULIC CUTOFF SWITCH IS SET TO "OFF". Accumulator check: . Collective pitch - - - - - - - - Checked correctly locked. "HYD TEST" (TEST HYDR) pushbutton - Depress on center console. . Warning panel- - - - - - - - Check "HYDR" light blinks ("HYDR" steady POST MOD 07-3317) . Collective and cyclic controls - - - Hands on. . Move the cyclic control 2 or 3 times on each axes (+/- 10% of travel) and check for accumulator hydraulic assistance on pitch and roll (no control loads).
. "HYD TEST" (TEST HYDR) pushbutton - Set back in up position.
. Warning panel- - - - - - - - Check "HYDR" light goes off. Hydraulic pressure isolation check: Collective pitch - - - - - - - - - Checked correctly locked.

Hydraulic cut-off switch - - - - - Set to OFF on collective pitch.

Warning panel- - - - - - - - Check "HYDR" light on. . Check that forces are felt immediately and that cyclic can be displaced in pitch and roll with normal feedback force. . Hydraulic cut-off switch - - - - - Set to ON. . Warning panel - - - - - - - Check "HYDR" light goes off in 3 to 4 sec. Maintenance action must be performed prior to flight if this time is 1 second or less (at least one of the accumulators may be defective). 4.1 Approved 350 B3 Page 5



4.A. FLIGHT MANUAL - ARRIEL 2B (continued)

FLIGHT MANUAL

Only the Item "When the rotors are completely stopped" paragraph 10 $\underline{\sf AFTER\ LANDING}$, is modified as follows:

- When the rotors are completely stopped:

Yaw load compensator check:

- .Check that right pedal moves forward without pilot input, or right pedal can be moved forward with low force.
- . Press the "HYD TEST" (TEST HYDR) push-button for 1 to 2 sec., then release.
- . Check that pedals can be re-centered and remain centered.

NOTE : Yaw load compensator maintenance action is required if, before activation of "HYD TEST" (TEST HYDR) push-button, the right pedal cannot be moved forward with low force.

The "HYD TEST" (TEST HYDR) discharges the yaw load compensator. In order to repeat the check, it is necessary to re-pressurize the hydraulic system.

Approved 350 B3

4.1Page 8



4.A. FLIGHT MANUAL - ARRIEL 2B (continued)

	FLIGHT	MANUAL
Only the Item " Hydraulic checks ", is modified as follows:		
Hydraulic checks:		
-Servo distributors seizure check: . "SERVO TEST" pushbutton PRESS; "SERVO" light comes	on.	
Approved 350 B3	SUP	.23 Page 4



4.B. FLIGHT MANUAL - ARRIEL 2B1

FLIGHT MANUAL AS 350 B3 Arriel 2B1	
Only the item 4 is modified as follows:	
Hydraulic cut-off switch (collective pitch)ON.	
4 - 6 APPROVED	



4.B. FLIGHT MANUAL - ARRIEL 2B1 (continued)

FLIGHT MANUAL AS 350 B3 Arriel 2B1

Only the item 1 is modified as follows:

Hydraulic checks:

CAUTION

If not locked, the collective pitch lever will move up when the accumulators are depleted or when the hydraulic cut-off switch on the collective is set to OFF.

Accumulator checks:

- Collective pitch CHECK correctly locked.
- [HYD TEST] or [ACCU TST].....ON.
- CWP CHECK HYDR flashes.

(HYDR steady PRE MOD 07-3317)

- Move the cyclic control 2 or 3 times on each axis (+/- 10% of travel) and check for accumulator hydraulic assistance on pitch and roll (no control loads).
- [HYD TEST] or [ACCU TST]........... RESET to OFF position.
- CWPCHECK HYDR.

· Hydraulic cut-off test:

- Hydraulic cut-off switch (collective pitch)OFF.
- Check that loads are felt immediately and that cyclic stick can be moved in pitch and roll with normal feedback loads.
- Hydraulic cut-off switch (collective pitch)ON.
- CWP CHECK HYDR after 3 to 4 sec.

Maintenance action must be performed prior to flight if this time is reduced to 1 sec. (at least one of the accumulators is faulty).

APPROVED 4 - 9



4.B. FLIGHT MANUAL - ARRIEL 2B1 (continued)

FLIGHT MANUAL AS 350 B3 Arriel 2B1

Only the item 11 [HYD TEST] or [ACCU TST] is modified as follows:

- 11. Yaw load compensator check:
 - Right pedal moves forward without pilot input, or right pedal can be moved forward with low forceCHECK.
 - [HYD TEST] or [ACCU TST]ON for 2 sec., then OFF.
 - Pedals can be re-centered and

NOTE

Yaw load compensator maintenance action is required if, before activation of [HYD TEST] or [ACCU TST] pushbutton, the right pedal cannot be moved forward with low force.

The [HYD TEST] or [ACCU TST] discharges the yaw load compensator. In order to repeat the check, it is necessary to re-pressurize the hydraulic system.

4 - 14 APPROVED



4.B. FLIGHT MANUAL - ARRIEL 2B1 (continued)

FLIGHT MANUAL AS 350 B3 Arriel 2B1
Only the item 2 is modified as follows:
2. Hydraulic checks: • Servo distributors seizure check: - [SERVO TEST] or [SERVO TST]PRESS: SERVO .
APPROVED 9-23-9



4.C. FLIGHT MANUAL - ARRIEL 2D - EASA

			FLIGHT MANUAL
Only the ite	em 4 is modified as follows:		
4.	Hydraulic cut-off switch (o	collective pitch)ON	
APPROVE	ED	350 B3e	4.3
		A	Page 1



4.C. FLIGHT MANUAL - ARRIEL 2D - EASA (continued)

FLIGHT MANUAL Only the item 1 is modified as follows: 1. Hydraulic checks: CAUTION If not locked, the collective pitch lever will move up when the accumulators are depleted or when the hydraulic cut-off switch on the collective pitch is set to off. · Accumulator checks: - Collective pitch CHECK correctly locked - [ACCU TST]ON - Collective / cyclic controls HANDS on Move the cyclic stick 2 or 3 times on each axis (± 10% of total travel) and check for accumulator hydraulic assistance on pitch and roll (no control loads). · Hydraulic cut-off test: - Collective pitch CHECK correctly locked - Hydraulic cut-off switch (collective pitch) OFF - CWP CHECK HYDR - Check that loads are felt immediately and that cyclic stick can be moved in pitch and roll with normal feedback loads. - Hydraulic cut-off switch (collective pitch) ON - CWP CHECK HYDR after 3 to 4 sec. Maintenance action must be performed prior to flight if this time is reduced to 1 sec. (at least one of the accumulators is faulty) 4.3 APPROVED 350 B3e Α Page 4



4.C. FLIGHT MANUAL - ARRIEL 2D - EASA (continued)

FLIGHT MANUAL				
Only the tem 11 is modified as follows:				
11. Yaw load compensator check:				
- Right pedal moves forward without pilot input, or right pedal can be moved forward with low forceCHECK [ACCU TST]ON for 2 sec., then OFF.				
Pedals can be re-centered and remain centeredCHECK.				
NOTE				
NOTE Yaw load compensator maintenance action is required if, before activation of [ACCU TST] pushbutton, the right pedal cannot be moved forward with low force.				
The [ACCU TST] discharges the yaw load compensator. In order to repeat the check, it is necessary to re-pressurize the hydraulic system.				
APPROVED 350 B3e 4.6				
A Page 1				
Page 1				



4.C. FLIGHT MANUAL - ARRIEL 2D - EASA (continued)

			FLIGHT MANUAL
Only the ite	m 1 is modified as follows:		
- Ser	ulic checks: vo distributors seizure check: SERVO TST]	PRESS: SERVO.	
APPROVED	350 B3e		SUP.23
	A		Page 8



4.D. FLIGHT MANUAL - ARRIEL 2D - FAA/ANAC/IAC-AR

	FLIGHT MANUAL AS 350 B3e
	Only the item 4 is modified as follows:
	4. Hydraulic cut-off switch (collective pitch)ON
4 - 6	B G K



4.D. FLIGHT MANUAL - ARRIEL 2D - FAA/ANAC/IAC-AR (continued)

FLIGHT MANUAL AS 350 B3e

Only the item 1 is modified as follows:

1. Hydraulic checks:

CAUTION

If not locked, the collective pitch lever will move up when the accumulators are depleted or when the hydraulic cut-off switch on the collective is set to OFF.

- Accumulator checks:
 - Collective pitch CHECK correctly locked.
 - [ACCU TST].....ON.
 - CWP CHECK HYDR flashes.

 - Move the cyclic control 2 or 3 times on each axis (+/- 10% of total travel) and check for accumulator hydraulic assistance on pitch and roll (no control loads).
 - [ACCU TST] RESET to OFF position.
 - CWP CHECK HYDR
- Hydraulic cut-off test:

 - Hydraulic cut-off switch (collective pitch)OFF.
 - CWP.....CHECK HYDR.
 - Check that loads are felt immediately and that cyclic stick can be moved in pitch and roll with normal feedback loads.
 - Hydraulic cut-off switch (collective pitch)ON.
 - CWP CHECK HYDR after 3 to 4 sec.

Maintenance action must be performed prior to flight if this time is reduced to 1 sec. (at least one of the accumulators is faulty).

APPROVED 4 - 9

B G K



4.D. FLIGHT MANUAL - ARRIEL 2D - FAA/ANAC/IAC-AR (continued)

FLIGHT MANUAL AS 350 B3e Only the item 11 [ACCU TST] is modified as follows: 11. Yaw load compensator check: - Right pedal moves forward without pilot input, or right pedal can be moved forward with low forceCHECK. - [ACCU TST]ON for 2 sec., then OFF. Pedals can be re-centered and NOTE Yaw load compensator maintenance action is required if, before activation of [ACCU TST] pushbutton, the right pedal cannot be moved forward with low force. The [ACCU TST] discharges the yaw load compensator. In order to repeat the check, it is necessary to re-pressurize the hydraulic system. 4 - 14 APPROVED B G K



4.D. FLIGHT MANUAL - ARRIEL 2D - FAA/ANAC/IAC-AR (continued)

		FLIGHT 1 AS 35	
Only the item 1 is modified as f	collows:		
Hydraulic checks: Servo distributor [SERVO TS]	s seizure check: T]PRE	ess: <mark>Servo</mark> .	
APPROVED	B G K		SUP.23-9

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