

THE **adlog**TM AIRCRAFT
MAINTENANCE
RECORDKEEPING
SYSTEM

**AIRFRAME
MAINTENANCE
RECORDS**

AIRFRAME MAINTENANCE RECORDS

Log No. 1

Aircraft Registration No. 6770X

Aircraft Mfg. CESSNA Model 310F Serial No. 310F-0070

Engine Mfg. CONT. Model I0-470D Serial No. CS105176-8-D-R

Engine Mfg. CONT Model 10-470D Serial No. 79315-0-D

Propeller Mfg. HARTZELL Model HC-A2XF-2 Serial No. 1198

Hub Design No. _____ Hub Serial No. _____

Blade Design No. _____ Blade Serial No's. _____

Propeller Mfg. HARTZELL Model HC-A2XF-2 Serial No. Y428

Hub Design No. _____ Hub Serial No. _____

Blade Design No. _____ Blade Serial No's. _____

(All applicable information must be filled in)



Steve Sorrell Aviation, Inc.

Fort Pierce, FL 34946

Phone: [REDACTED]

Fax: [REDACTED]

Cell: [REDACTED]

LOG BOOK ENTRY

August 15, 2007

N6770X

Hobbs: 765.6

ACFTT: 5765.3

Replaced carpet Hafner Textile Pebble Beach Lot#382010905 8110-3 attached. Recovered headliner side panels and seats cream illusion and COBI #61610 burn seats and 8110.3 attached. Replaced NLG doors PN 0813300-13, 0813300-1&2, Hing assy 0813300-7, 0843538-1, 0813300-93, 0813300-92, 0813300-8. Bearings 13830, 13889, seals MS28782-32, S2418-1, 0841200-15, 0841200-25. Installed STC single piece windshield #5A391GL in accordance with drawing no. W-2070 Rev A See 337 form. Replaced cabin windows with Great Lakes FAA PMA parts. Replaced RH tire, LH and RH brake linings LH navigating light. Sanded and painted aircraft G-8003 Matterhorn white 44469xGold, 99xBlack, Flight controls removed, balance checked and reinstalled in accordance with Cessna Service Manual. Repacked LH and RH brake calipers. Replaced ELT battery, Inspected and tested in accordance with 91.207D. Expiration Feb/09. Replaced LH fuel boost pump with serviceable part.

[REDACTED] A&P [REDACTED]

August 115, 2007

N6770X

Hobbs: 765.6

ACFTT: 5765.3

I certify this aircraft has been inspected in accordance with an annual inspection found airworthy and approved for return to service all A.D.s complied with thru 8/07. [REDACTED] IA [REDACTED]

12-30-08

775.

I HAVE INSPECTED THIS AIRCRAFT AND FOUND IT TO BE SAFE FOR FERRY FLIGHT PER SPECIAL FLIGHT PERMIT DATED [REDACTED] REPORTED VAC PRESS. LINE CLEAR OF PILOT'S CONTROL COLUMN. CHECKED GEAR RETRACTION/EXTENSION - C/O GOOD. AD 72-14-08 R1 C/W BY INSP. PER AD.

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TOTAL—Carry forward to next page

DATE

TOTAL TIME IN SERVICE

TACH OR RECORDING METER TIME

DESCRIPTION OF WORK PERFORMED—
SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK

N6670X Cessna 310F S/N 310-0070 hobbs 780 TTA 5780

I certify that this aircraft has been inspected and is in condition for ferry flight from KFPR to X21 for maintance.

6-28-11

Michael Cam A&P [REDACTED]

N6770X

HOBBS 784.1

AIRFRAME

10
12-01-11

TT: 5784

Carried out Annual Inspection IAW FAR Part 43 App D. Opened up aircraft removed Interior. Jacked up aircraft replaced Ldg. Down Lock Springs and all tires , packed bearings , Lube Gear, Removed nose drag leg, wash ,striped, primed , painted & reinstalled. Cleaned corrosion out of nose wheel well & primed. Install nose tire fender. Preformed retraction test , checked LDG. Lights & horn. Removed replaced rudder trim actuator. Installed tail tie down. Installed RH. Overhauled Boost Pump by Aero Accessories in San Antonio Tx. see 8130-3. Installed new #2 manifold pressure hose to left gauge and 1 in Rt. eng compartment. Installed airspeed indicator from Ford Inst. Removed extending ELT, installed new AK450 in its place. Tested IAW 91.407 Due Dec. 2017 . Installed repair kit in compass. Had Transponder, and encoder Alt. certified by Volusia Aviation. Installed 2 new plates on top of engine nacelles. Preformed heater decay test. Installed new fuel drains in tip tanks & fuel placards on tip tanks. Installed serviceable artificial horizon, and filter to horizon and Dg. Checked AD's see report in Logs.Ground run, Opps. Check OK.

I Certify That This AIRFRAME Has Been Inspected
I.A.W. ANNUAL Inspection And Determined To Be
In AIRWORTHY Condition
TT: 5784 Date: 12-10-11
Michael Cam [REDACTED] IA

10/19/2012 Cessna N6770X Ser: 310F-0070 Hobbs: 804.2

Removed directional gyro for overhaul at Porter-Straight (see yellow tag dated 09-10-2012), installed two G-25 batteries, Adjusted vacuum regulator to get desired suction. Performed operational check of all affected components and found all components to be operating properly. All work was performed in reference with the manufactures maintenance manual. I certify that this engine is airworthy in regard to the work performed above.

Keith Thomas [REDACTED]
AP [REDACTED]

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DATE 04/03/2013 Cessna N6770X Ser:310-0070 Tach: 804.9 AFTT: 5804.2

Removed panels for inspection, serviced aircraft battery, and fuel strainer, serviced heater fuel filter and replaced seals, replaced induction air filter P/N BA-2305, replaced instrument air filter. Checked all flight control tensions and travels adjusted flap tensions, adjusted rudder tension, installed serviceable flap rod on left outboard flap inboard flap rod, patched right hand flap at inboard trailing edge. Inspected ELT in accordance with 14 CFR 91.207, ELT battery due March 2017, next inspection due April 2014. Cleaned left landing light switch due to corrosion, cleaned left nav light connection at ground, Packed all wheel bearings replaced nose wheel bearing and races, resealed right brake caliper, replaced right gear upper truss resealed strut and serviced with fluid, replaced shimmy dampner rod, resealed shimmy dampner and serviced with fluid, filed aluminum where step was rubbing on belly of aircraft, removed coaxial cable from underneath instrument panel that went no where, swung compass and filled out correction card this date, re-connected nav #2 indicator wiring, replaced all seat belts, tightened up all fittings on right fuel selector valve and cleaned off fuel stains, replaced missing buttons on the bottom of both left and right bottom cowlings, and replaced rivets on right hand engine lower cowling, patched a hole in bottom right engine lower cowling, replaced popped rivets in rudder, replaced missing screws at top of vertical fin, lubed all pulleys and cables, Reinstalled panels after inspection. Performed AD search and complied with the following ADS, AD96-12-22 on both left and right engine by verification of torque on the oil filter adapter nut and that no leakage was present at this time next due at AFTT 5904.2, AD72-14-08R1 by pressuring up fuel lines with fuel pump and inspected for leak, no leaks noted at this time, next due at AFTT:5864.2, AD76-07-12R1 by verification that ignition switch turns engine off when in the off position next due at AFTT: 5904.2. AD81-09-09 was previously complied with on date 12/28/1993 at heater hobbs: 2.8, and the heater decay test was performed 12/10/2011 and is next due at heater hobbs:202.8 or December 2013 whichever occurs first. All work was performed in reference with the manufactures maintenance manual. I certify that this annual inspection was performed in accordance with 14 CFR 43 appendix D and using manufacturers maintenance manual as reference and this AIRCRAFT was determined to be in an airworthy condition for return to service.

Keith Thomas [REDACTED]
[REDACTED] IA

DENTON AVIATION CONSULTING INC.

FAA CRS# Q6DR312Y N 6770X Date 5/9/14


Test Set	Encoder	Altimeter	Baro	Reading	Difference
-1000	-1000	1005	28.10	-1045	+5
0	0	+20	28.50	-720	+20
500	500	500	29.00	-220	0
1000	1000	1000	29.50	+250	0
1500	1500	1520	29.92	+640	0
			30.50	+1180	+10
			30.90	+1540	+20
			30.99	+1620	+0

S/W 714

MASTER READS	ROOM TEMP °C	LOW TEMP °C
40	40	
50	50	
60	61	
70	71	
80	81	
90	91	
100	101	
110	111	
120	121	
130	130	
140	140	
150	150	
160	160	
170	170	

MASTER READS	ROOM TEMP °C	LOW TEMP °C
180	180	
190	190	
200	200	
210	210	
220	220	
230	231	
240	242	
250	252	
260		
270		
280		
290		
300		

Name: Cessna
 Model: 310
 Serial #: 310-0070
 Tach: 804.2 Hobbs
 ATIS: 120-r
 Temp: 200
 Tested by: [REDACTED]

TESTED BY:  **KELLEY INSTRUMENTS, INC.** DATE 5-7-2014
 FAA. C.R.S. 4131 MAY
 OP2R065L WICHITA, KANSAS 67209-2838

DATE 04/03/2013 Cessna N6770X Ser:310-0070 Tach: 804.9 AFTT: 5804.2

Removed panels for inspection, serviced aircraft battery, and fuel strainer, serviced heater fuel filter and replaced seals, replaced induction air filter P/N BA-2305, replaced instrument air filter. Checked all flight control tensions and travels adjusted flap tensions, adjusted rudder tension, installed serviceable flap rod on left outboard flap inboard flap rod, patched right hand flap at inboard trailing edge. Inspected ELT in accordance with 14 CFR 91.207, ELT battery due March 2017, next inspection due April 2014. Cleaned left landing light switch due to corrosion, cleaned left nav light connection at ground, Packed all wheel bearings replaced nose wheel bearing and races, resealed right brake caliper, replaced right gear upper truss resealed strut and serviced with fluid, replaced shimmy dampner rod, resealed shimmy dampner and serviced with fluid, filed aluminum where step was rubbing on belly of aircraft, removed coaxial cable from underneath instrument panel that went no where, swung compass and filled out correction card this date, re-connected nav #2 indicator wiring, replaced all seat belts, tightened up all fittings on right fuel selector valve and cleaned off fuel stains, replaced missing buttons on the bottom of both left and right bottom cowlings, and replaced rivets on right hand engine lower cowling, patched a hole in bottom right engine lower cowling, replaced popped rivets in rudder, replaced missing screws at top of vertical fin, lubed all pulleys and cables, Reinstalled panels after inspection. Performed AD search and complied with the following ADS, AD96-12-22 on both left and right engine by verification of torque on the oil filter adapter nut and that no leakage was present at this time next due at AFTT 5904.2, AD72-14-08R1 by pressuring up fuel lines with fuel pump and inspected for leak, no leaks noted at this time, next due at AFTT:5864.2, AD76-07-12R1 by verification that ignition switch turns engine off when in the off position next due at AFTT: 5904.2. AD81-09-09 was previously complied with on date 12/28/1993 at heater hobbs: 2.8, and the heater decay test was performed 12/10/2011 and is next due at heater hobbs:202.8 or December 2013 whichever occurs first. All work was performed in reference with the manufactures maintenance manual. I certify that this annual inspection was performed in accordance with 14 CFR 43 appendix D and using manufacturers maintenance manual as reference and this AIRCRAFT was determined to be in an airworthy condition for return to service.

Keith Thomas

IA

DENTON AVIATION CONSULTING INC.

FAA CRS# Q6DR312Y

N

6770X

Date

5/9/14

Test Set	Encoder	Altimeter	Baro	Reading	Difference
-1000	-1000	1005	28.10	-1045	+5
0	0	+20	28.50	-720	+20
500	500	500	29.00	-220	0
1000	1000	1000	29.50	+250	0
1500	1500	1520	29.92	+640	0
2000	2000	2000	30.50	+1180	+10
3000	3000	3010	30.90	+1540	+20
4000	4000	4010	30.99	+1620	+0
6000	6000	6010	Name: <u>Cessna</u>		
8000	8000	8010	Model: <u>310</u>		
10000	10000	10020	Serial #: <u>310-0070</u>		
12000	12000	12020	Tachs <u>809.2 Hobbs</u>		
14000	14000	14020	ATIS: <u>124-r</u>		
16000	16000	16020	Temp: <u>200</u>		
18000	18000	18020	Tested by [Redacted]		
20000	20000	20000			

FORM ALTCORR 010609 Rev 3

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1. Approving Civil Aviation Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 133795
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4. Organization Name and Address: Kelley Instruments, Inc. [REDACTED] Wichita, KS 67209		5. Work Order/Contract/Invoice Number: 133795
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6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	AIRSPEED	S1471N2	1	714	OVERHAULED

12. Remarks:

Complete Tear Down, Cleaned, Calibrated, Assembled, Functional Tested & Certified

RANGE MARKED TO CUSTOMERS REQUEST

Final Tested and Certified Per Manual: S1471 DRAWING

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: [REDACTED]	14c. Approval/Certificate No.:
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): MICHAEL E. ESTES	14e. Date (dd/mmm/yyyy): MAY 07 2014

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.


Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

ALTIMETER S/N 8B537

STANDARD ALTITUDE IN FT.	ALTIMETER READING
-1000	-1000
0	0
500	500
1000	1000
1500	1500
2000	2000
3000	3000
4000	4010
6000	6010
8000	8020
10000	10025
12000	12040

STANDARD ALTITUDE IN FT.	ALTIMETER READING
14000	14060
16000	16070
18000	18080
20000	20070
22000	22060
25000	25050
30000	29970
35000	XXXXXXXXXX
40000	
45000	
50000	
51000	
53000	

 Porter-Strait Instrument Co., Inc.
 2708 N. Sheridan Rd • Tulsa, OK 74115
 918-838-8711 FAA Repair Station #EI2R795K

TESTED BY: 
 DATE: Mar 02 2014

1. Reporting Civil Aviation Authority/Country: FAA/United States	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 100061
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4. Organization Name and Address: PORTER-STRAIT INSTRUMENT CO., INC., [REDACTED] TULSA, OK 74115	5. Work Order/Contract/Invoice Number: 100061
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6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	ALTIMETER	5934PA-1	ONE	8B537	REPAIRED

12. Remarks:

RESET BAR, TESTED AND CERTIFIED
I/A/W FAR 43 APP E PAR B TO 30,000 FEET.

A completed description of work performed is on file at the above referenced organization under the work order and system tracking reference number indicated in blocks 3 and 5. Doc. No.: FAR 43 APP E (B)

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: [REDACTED]	14c. Approval/Certificate No.:
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): J. BUSH	14e. Date (dd/mmm/yyyy): 02/MAY/2014

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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			TOTAL brought forward from previous page
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N6670X c310f SN 0077 ACTT: 5807.6 Hobbs 808.3 May 09/2014
 Repaired bad ground on LT nav light, removed corrosion on interior overhead lights.
 Repositioned ELT inspected per far91.207Dbattery good to 03/2017.Replaced both
 ship batteries with G25 batteries. Repaired and lubricated LT ALT air cable,ops checked good.
 Installed RT aft nose landing gear door hinge bolt. lubed control hinges and pulleys. Gear swing
 in normal and emergency mode tested good .Repaired Altimeter 5934PA-1 sn 8B537 .
 Replaced airspeed Indicator with S1471N2 sn 714 . CW AD 2014-03-03
 by (g)2 (i) and (ii)A installing placards. I certify that this aircraft Has been inspected IAW an
 annual inspection per 14 CFR 43 appendix D and was determined to be in airworthy condition.
 ads current through this date see list.

[Redacted Signature]

Gaylord A. Syrstad A&P [Redacted] IA

DENTON AVIATION CONSULTING, INC.
 CRS# [Redacted]
 CERTIFIES THAT THE ALTIMETER
 MOD# 5934PA-1 S/N HTT 8B537
 PITOTSTATIC SYS HAS BEEN TESTED
 TO 20,000 FT PER FAR 91.411 &
 FAR 43 APP. E

DENTON AVIATION CONSULTING, INC.
 CRS# [Redacted]
 CERTIFIES THAT TRANSPONDER
 MOD# C300 S/N 19114
R354A
 HAS BEEN TESTED PER FAR 91.413 &
 FAR 43 APP. F

W2050914 [Redacted] TACH/HOBBS 809.2 05/09/14

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DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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1 of 1

MAKE: Cessna
MODEL: 310F
S/N: 310-0070
REG. NO: 6770X
WORK ORDER: 6644



Tennessee Aircraft Services, Inc.
[Redacted]
Jackson, TN 38301
Phone: 731-424-7227

DATE: 22 August, 2014
AVC TSN:
TACH:

Airframe Entries

- Installed serviceable horizontal stabilizer assembly using new hardware. Reconnected flight controls and rigged. Work performed in accordance with Cessna 310F Service Manual section 2 paragraphs 24, 27, 30, 31, 39, 40, 41, 44, 45, 54, 102, 108, 113, 117, 124, 127, 131, 135, 136, 137, 139, 142, and figures 6, 7, 8, 9, 19, & 20.
- Drilled a few drain holes in the aircraft belly.
- Installed a serviceable elevator trim tab horn p/n: 0831000-50.
- Rerouted & rigged rudder cable and set rudder cable system tensions in accordance with the Cessna 310F maintenance Manual Instructions.
- Rerouted elevator trim cables from rear cabin bulkhead to actuator. Set cable tensions, adjusted travel stops, saftied turn barrel & pulley guards in accordance with Cessna 310F Maintenance Manual Instructions.
- Work performed by Doug Kail, Robert Batterman, David Carter, Blake Sides, and Joe Norris.

DATE: 22 August, 2014

SIGNED: [Redacted]
Doug Kail, IA: [Redacted]

Work Order: 6644

Printed by EBis 3 (datcomedia.com)

9 JUN 15 5855.2 855.9 Washed aircraft, lube pulleys and cables and linkages IAW maintenance manual, neutralized, and removed corrosion on bottom of left wing, sealed surface with Alidinc. Gear swing in normal and emergency mode test good. Inspected and tested ELT IAW 14CFR 91.207 batteries due 3/17. I certify this airframe has been inspected IAW 14CFR 43 appendix D and the manufacturer's maintenance manual for 100 hr. inspection and has been found in a condition safe for continued operation. [Redacted]

Nicholas A. Frankt
A+P [Redacted]

10 JUN 15 5855.2 855.9 Verified aircraft was inspected IAW 14CFR 43 appendix D and manufacturer's maintenance manual. Verified AD compliance is current through this date. I certify that this aircraft has been inspected IAW [Redacted] and was found to be in airworthy condition.

A+P [Redacted] IA

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DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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30 SEP 15	5891.9	892.6	Removed G-25 battery installed new G-25 SN G02863688 [REDACTED] A+P [REDACTED] Wick Franke
15 NOV 15	5896.4	907.1	Removed Rt. engine propeller cable and conduit replaced with PN 9910268-12. Removed left eng air box repaired by welding reinstalled. Replaced broken prop conduit on left side. All repairs and adjustments made in accordance with Cessna maintenance manual and A/C 43-13-1B [REDACTED] A+P [REDACTED]
8 FEB 16	5933.3	937.0	Washed aircraft, lubed pulleys and cables and linkage systems IAW Manufacturers Maintenance Manual. Gear swing in normal and emergency mode test good. Inspected and tested ELT IAW 14 CFR 91.207, batt due 3/17. Replaced seal in ^{left} fuel tank drain. Inspecting this airframe has been inspected IAW 14 CFR 43 appendix D and the Manufacturers Maintenance Manual 100 hour inspection criteria and has been found to be in a condition safe for continued flight. [REDACTED] A+P [REDACTED]
27 MAR	5962.3	966.0	Installed left main landing gear tire part # A133M4, 650-10 6ply Air Hawk with new tube. All work done I.A.W. Cessna maintenance manual. [REDACTED] A+P [REDACTED]
1 MAY 16	6035.3	1039.0	Removed Left wingtip main fuel tank and replaced with tank from N6799X ser 310F0099 as removed AD 87-21-02R1 complied with while installed
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DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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			on N6799K log entry 5-6-1993 "filler casters installed" per ATP 431231864711 [REDACTED] ATP [REDACTED]
1 MAY 16	6035.3	1039.0	Washed airplane lubricated pulleys, cables and linkages in accordance with Cessna Maintenance Manual. Tested landing gear in normal and emergency mode both test OK. Inspected and tested ELT in accordance with 14CFR 91.207 next due at 100 hour/annual inspection both due 3/17 I certify this aircraft has been inspected IAW 100 hr 14CFR 43 appendix D and the Manufacturer's Maintenance Manual and has been found to be in a safe condition for continued flight. ATP [REDACTED]
MAY 1, 2016	6035.3	1039.0	Verified aircraft was inspected IAW a 100 hour inspection per 14CFR appendix D, and the manufacturer's maintenance manual. C/W AD 76-07-12 ignition switch function check. C/W AD 16-07-24 Replacement of elevator trim push-pull hardware IAW MEB-27-02 dated February 29, 2016. Verified AD compliance through this date (See listing) I certify that this aircraft was inspected IAW an Annual Inspection and was determined to be in Airworthy Condition. [REDACTED] ATP [REDACTED] IA

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