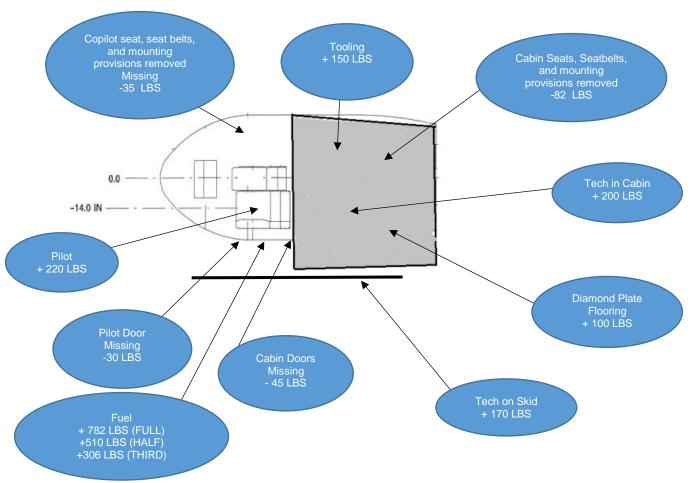
N602BP WEIGHT AND BALANCE ESTIMATION

Figure 1 – Aircraft Configuration at Time of Accident $\,^{1}$ $\,^{2}$ $\,^{3}$



Flight Manual – 600N (600RFM-1)

MD-600N Flight Manual 2-1 Flight Restrictions 4

Flight with doors removed is approved under the following conditions.

- Approved doors-off configurations:
- Both pilot's doors removed. OR
- Either pilot door removed. OR
- All cabin doors removed. OR
- All cabin doors, plus left pilot door removed. OR
- Right side aft cabin door removed. OR
- Right side mid and aft cabin doors removed. OR
- Left side aft cabin door removed.

¹ This weight and balance is an estimation based on the configuration of the aircraft at the time of the accident, a review of manufacturer technical data, and research and analysis of N602BP airworthiness records.

² No load manifest depicting the current configuration of the aircraft or adjustments for changes was found in the aircraft records nor was a load manifest produced by the operator following the accident depicting the removal of doors, seats, or installation of the cabin floor.

³ Pilot and Technician weights include physical weight plus winter attire and safety harnesses

⁴ As depicted in Section 2-1 of the MD600-N FAA approved flight manual, operations with the left side cockpit and left side cabin doors removed or copilot seat assembly removed are not authorized. CFR 91.107(a)(3) requires all passengers to be seated in an approved seat and properly secured with a seatbelt during aircraft movement

MD-600 Flight Manual Section 2-4 Weight Limitations

For internal gross weight operations above 3850 LB, refer to Section XI.

Maximum gross weight: 3850 LB.

Minimum flying gross weight: Refer to Figure 2-3

Cargo deck capacity: 1350 lbs. (not to exceed 115 lbs. per square foot).

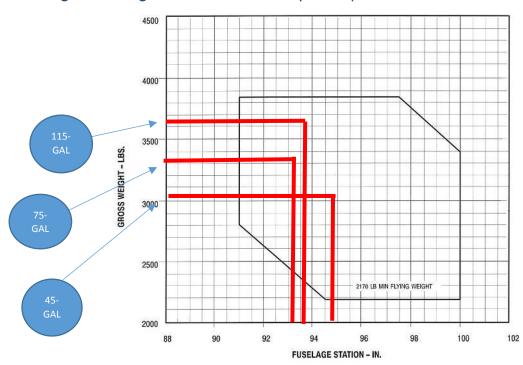
CG Limits: Ensure helicopter CG and weight are within approved limits throughout flight.

Lateral CG limits ±5.0 IN

Figure 2 – Estimated Weight and Balance Table ⁵ ⁶

					BAS	IC WEI	GHT AN	ID BAL	ANCE RE	CORD						
AIRCRAFT MO	SERIAL NUMBER RN0025										REGISTRATION NUMBER					
											N602BP					
DESCRIPTION		-	ADDED	DDED (+)			REMOVE (-)					RUNNING TOTAL				
DESC	WT	LONG	LAT	LONG	LAT	WT	LONG	LAT	LONG	LAT	WT	LONG	LAT	LONG	LAT	
		ARM	ARM	мом	мом		ARM	ARM	MOM	мом		ARM	ARM	мом	мом	
Actual Weight											2196.2	105.3	0.0	230575	-26	
LH Cockpit Door						30	43.0	-20	1290	-60						
LH Cabin Door						30	78.5	-20	2355	-60						
LH Cabin Door						15	96.9	-20	1453.5	-300						
Cabin Flooring	100	74	0.0	7400.0	0.0											
Coplt Seat Assy						35.0	43.0	13.5	1505.0	472.5						
Cabin Seat Assy						30.0	110.2	0.0	3306.0	0.0						
Cabin Seat Assy						52.0	66.5	0.0	3458.0	0.0						
GNS-430	4.1	12.0	-1.1	49.0	-5.0											
GTX-327	4.1	12.0	-1.1	49.0	-5.0											
FM Radio	2.8	12.0	-1.1	33.0	-3.0											
Cargo Hook	7.0	84.8	-3.1	594.0	-22.0											
ELT	10.0	12.0	-1.1	120.0	-11.0											
SUBTOTAL	128.0	64.41	4	8245.0	-46.0	192.0	43.21	2.22	8296.0	426.5	2132.2	108.12	023	230524	-498	
Pilot	220	43.0	-14	9460	-3080											
Technician	200	72.5	-16	8600	-3200											
Technician	170	94.0	-30	15980	-5100											
Tooling	150	107.0	0.0	16050	0.0											
Fuel- 115 GAL	782	82.1	0.0	67199	0.0											
TOTAI											3654.2	93.79	-3.11	342741.5	-11373.8	
Fuel – 75 GAL	510	81.3	0.0	41463.0	0.0						3382.2	93.73	-3.36	317005.5	-11373.8	
Fuel -45 GAL	306	83.2	0.0	25459.2	0.0						3180.2	94.65	-3.58	301001.7	-11373.8	

Figure 3 – Longitudinal Center of Gravity Envelope



⁵ Last physical weight and balance performed on N602BP was February 28, 2013. These numbers are annotated as the Actual Weight as depicted in Figure 2, Estimated Weight and Balance Table. Some major alterations were found to have been embodied into the aircraft following the physical weigh and were added to the basic weight adjustment.

⁶ Weight and balance estimation appears to depict N602BP to have been operating with the longitudinal and lateral CG range at the time of the accident.