



Aviation Investigation Factual Report

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|--------------------------------|--------------------------------------|-------------------------|------------|
| Location: | Nuevo, California | Accident Number: | ANC23FA031 |
| Date & Time: | March 24, 2023, 12:17 Local | Registration: | N14Z |
| Aircraft: | BELL HELICOPTER TEXTRON 407 | Aircraft Damage: | Destroyed |
| Defining Event: | Low altitude operation/event | Injuries: | 2 Fatal |
| Flight Conducted Under: | Part 91: General aviation - Personal | | |

On March 24, 2023, at 1217 Pacific daylight time, a Bell 407 helicopter, N14Z, was destroyed when it was involved in an accident near Nuevo, California. The pilot and pilot-rated passenger were fatally injured. The helicopter was operated as a Title 14 *Code of Federal Regulations* Part 91 personal flight.

According to a family member, the helicopter was en route to Big Bear, California, as part of a weekend skiing trip. The pilot's cell phone was enabled with a crash notification feature, which triggered an automatic call to a family member. The family member then notified first responders, who located the wreckage near the peak of a rocky hilltop at 1243.

Review of ADS-B data revealed the helicopter departed French Valley Airport (F70), Murrieta/Temecula, California, at 1205 and flew in a northerly direction at an altitude of about 500 to 600 ft above the ground (agl). The last ADS-B reporting point was collocated with the accident site, about 18 miles north of F70.

Pilot Information

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|----------------------------------|--|--|------------------|
| Certificate: | Private | Age: | 61, Male |
| Airplane Rating(s): | None | Seat Occupied: | Right |
| Other Aircraft Rating(s): | Helicopter | Restraint Used: | |
| Instrument Rating(s): | None | Second Pilot Present: | Yes |
| Instructor Rating(s): | None | Toxicology Performed: | Yes |
| Medical Certification: | Class 3 With waivers/limitations | Last FAA Medical Exam: | October 1, 2022 |
| Occupational Pilot: | No | Last Flight Review or Equivalent: | November 1, 2022 |
| Flight Time: | 2190 hours (Total, all aircraft), 132 hours (Total, this make and model) | | |

Pilot-rated passenger Information

| | | | |
|----------------------------------|----------------------------------|--|---------------|
| Certificate: | Private | Age: | 62, Male |
| Airplane Rating(s): | Single-engine land | Seat Occupied: | Left |
| Other Aircraft Rating(s): | Helicopter | Restraint Used: | |
| Instrument Rating(s): | None | Second Pilot Present: | Yes |
| Instructor Rating(s): | None | Toxicology Performed: | Yes |
| Medical Certification: | Class 3 With waivers/limitations | Last FAA Medical Exam: | April 1, 2012 |
| Occupational Pilot: | No | Last Flight Review or Equivalent: | |
| Flight Time: | 1600 hours (Total, all aircraft) | | |

Aircraft and Owner/Operator Information

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|--------------------------------------|---|---------------------------------------|----------------|
| Aircraft Make: | BELL HELICOPTER TEXTRON | Registration: | N14Z |
| Model/Series: | 407 | Aircraft Category: | Helicopter |
| Year of Manufacture: | | Amateur Built: | |
| Airworthiness Certificate: | Normal | Serial Number: | 54704 |
| Landing Gear Type: | None; High skid | Seats: | 7 |
| Date/Type of Last Inspection: | November 1, 2022 Continuous airworthiness | Certified Max Gross Wt.: | 5250 lbs |
| Time Since Last Inspection: | | Engines: | 1 Turbo shaft |
| Airframe Total Time: | 1750.2 Hrs at time of accident | Engine Manufacturer: | Rolls-Royce |
| ELT: | C126 installed, activated, did not aid in locating accident | Engine Model/Series: | 250C47B/8 |
| Registered Owner: | On file | Rated Power: | 813 Horsepower |
| Operator: | On file | Operating Certificate(s) Held: | None |

Meteorological Information and Flight Plan

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|---|----------------------------------|---|------------------|
| Conditions at Accident Site: | Visual (VMC) | Condition of Light: | Day |
| Observation Facility, Elevation: | KHMT, 1514 ft msl | Distance from Accident Site: | 7 Nautical Miles |
| Observation Time: | 12:55 Local | Direction from Accident Site: | 150° |
| Lowest Cloud Condition: | Scattered / 3800 ft AGL | Visibility | 10 miles |
| Lowest Ceiling: | None | Visibility (RVR): | |
| Wind Speed/Gusts: | 4 knots / None | Turbulence Type Forecast/Actual: | None / None |
| Wind Direction: | 180° | Turbulence Severity Forecast/Actual: | N/A / N/A |
| Altimeter Setting: | 30.21 inches Hg | Temperature/Dew Point: | 13°C / 0°C |
| Precipitation and Obscuration: | No Obscuration; No Precipitation | | |
| Departure Point: | Murrietta, CA (F70) | Type of Flight Plan Filed: | None |
| Destination: | Big Bear, CA (L35) | Type of Clearance: | VFR |
| Departure Time: | 12:05 Local | Type of Airspace: | Class G |

Wreckage and Impact Information

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|----------------------------|---------|-----------------------------|---------------------------|
| Crew Injuries: | 1 Fatal | Aircraft Damage: | Destroyed |
| Passenger Injuries: | 1 Fatal | Aircraft Fire: | None |
| Ground Injuries: | | Aircraft Explosion: | None |
| Total Injuries: | 2 Fatal | Latitude, Longitude: | 33.789279,-117.09542(est) |

The helicopter came to rest about 330 ft downslope on a rocky, hilly terrain, at an elevation of 2,560 ft above mean sea level. All of the helicopter and engine components were located within the main wreckage site.

The debris field extended along the route of flight, starting at the top of a small hill and extending down the hillside. The beginning of the debris field consisted of very small fragments of fiberglass, consistent with main rotor blade material. These fragments were located on top of, and immediately adjacent to, a large rocky outcrop at the top of the hill. Ground scars about 60 ft past the initial impact point were consistent with multiple tail rotor blade strikes. No evidence of main rotor strikes were found after the initial impact point.

The airframe was extensively damaged, and the tailboom was separated. The main rotor hub remained attached to the transmission and the transmission deck was fractured from the

passenger compartment. Two main rotor blades exhibited extensive impact damage. The two other main rotor blades sustained less damage.

The cockpit and flight controls sustained extensive damage due to impact forces. The helicopter was equipped with a set of left side dual controls, which consisted of a cyclic, collective, and tail rotor anti torque pedals. Main rotor flight control continuity for both cyclic and collective systems was confirmed from the flight control input sticks to the main rotor hub assembly, with several overload fractures observed consistent with impact forces. No preimpact anomalies were observed in the main and tail rotor flight controls or hydraulics and all observed fractures were consistent with overload forces at impact. The hydraulic reservoir contained hydraulic fluid, and the screen was free of debris.

The three main rotor servos were all connected to the servo support, which remained attached to the surrounding fractured roof section. The servos were removed and functionally tested; no anomalies were noted.

The rotating (outer ring) swashplate rotated freely by hand. The swashplate also moved freely up and down with hand movement of the collective lever assembly. Drive shaft continuity was confirmed through a series of impact fractures from the engine gearbox to the tail rotor system. No preimpact anomalies were found in the fuel system; fuel remained in the bottom of the aft main fuel cell. Postaccident examination of the engine and airframe revealed no evidence of any preimpact mechanical malfunctions or failures that would have precluded normal operation.

Medical and Pathological Information

An autopsy of the pilot was performed by Riverside County Sheriff's Coroner in Perris, California. The autopsy report was reviewed by the NTSB Investigator-In-Charge. According to the autopsy report, the cause of death was multiple blunt force injuries, and the manner of death was accident.

The results of the pilot's postmortem toxicological testing indicated that he had used the sedating medication zolpidem. The precise timing of his last zolpidem use, and whether it was exerting significant impairing effects at the time of the crash, could not be determined from the zolpidem levels measured in cavity blood and urine. The pilot's toxological tests were also positive for zolpidem, chlorothiazide, hydrochlorothiazide and lisinopril, none of which are typically impairing.

The pilot-rated passenger's toxicological tests were positive for acetaminophen, amlodipine, atorvastatin, chlorothiazide, hydrochlorothiazide, sildenafil, desmethylsildenafil, and rivaroxaban, which are not typically impairing.

Additional Information

Two GoPro cameras were found within the debris field and were sent to the National Transportation Safety Board (NTSB) Vehicle Recorders Laboratory in Washington, DC for examination. Video recovered from one of the cameras ended before takeoff. The other video captured the entirety of the 9-minute flight from takeoff to impact from its mounted position on the helicopter's right skid. The video shows the helicopter depart and proceed toward the intended destination. On two occasions, the video depicted low-level maneuvering over elevated terrain. On the first low pass over a hilltop, about 6 minutes after takeoff, embedded GPS data indicated the helicopter traversed the crest of the hill at an altitude between 30 to 40 ft agl.

About 3 minutes later, the pilot flew over the next line of hills at an altitude of about 30 ft agl, then turned left and flew low-level along a ridgeline. As the pilot approached the crest of the ridgeline, he initiated a steep right turn in a bank of about 60°. See figure 1.



Figure 1. Video frame from on board Go-Pro camera seconds before impact, with main rotor blade seen in upper left of photo.

Moments later, as the helicopter passed over a rocky outcropping at the top of the hill, the main rotor blade contacted terrain and shards of main rotor blades can be seen on the video. See figure 2.



Figure 2. Video frame from on board Go-Pro camera at point of impact, with portions of fragmented main rotor blade seen in upper left of photo.

Administrative Information

Investigator In Charge (IIC): Ward, Mark

Additional Participating Persons: Jeffrey Newcomer; Federal Aviation Administration; Riverside, CA
Mark Stuntzner; Bell Flight Safety; Fort Worth, TX
Jack Johnson; Rolls-Royce Corporation; Indianapolis, IN

Report Date:

Last Revision Date:

Investigation Class: [Class 3](#)

Note:

Investigation Docket: <https://data.nts.gov/Docket?ProjectID=106951>

The National Transportation Safety Board (NTSB) is an independent federal agency charged by Congress with investigating every civil aviation accident in the United States and significant events in other modes of transportation—railroad, transit, highway, marine, pipeline, and commercial space. We determine the probable causes of the accidents and events we investigate, and issue safety recommendations aimed at preventing future occurrences. In addition, we conduct transportation safety research studies and offer information and other assistance to family members and survivors for each accident or event we investigate. We also serve as the appellate authority for enforcement actions involving aviation and mariner certificates issued by the Federal Aviation Administration (FAA) and US Coast Guard, and we adjudicate appeals of civil penalty actions taken by the FAA.

The NTSB does not assign fault or blame for an accident or incident; rather, as specified by NTSB regulation, “accident/incident investigations are fact-finding proceedings with no formal issues and no adverse parties ... and are not conducted for the purpose of determining the rights or liabilities of any person” (Title 49 *Code of Federal Regulations* section 831.4). Assignment of fault or legal liability is not relevant to the NTSB’s statutory mission to improve transportation safety by investigating accidents and incidents and issuing safety recommendations. In addition, statutory language prohibits the admission into evidence or use of any part of an NTSB report related to an accident in a civil action for damages resulting from a matter mentioned in the report (Title 49 *United States Code* section 1154(b)). A factual report that may be admissible under 49 *United States Code* section 1154(b) is available [here](#).